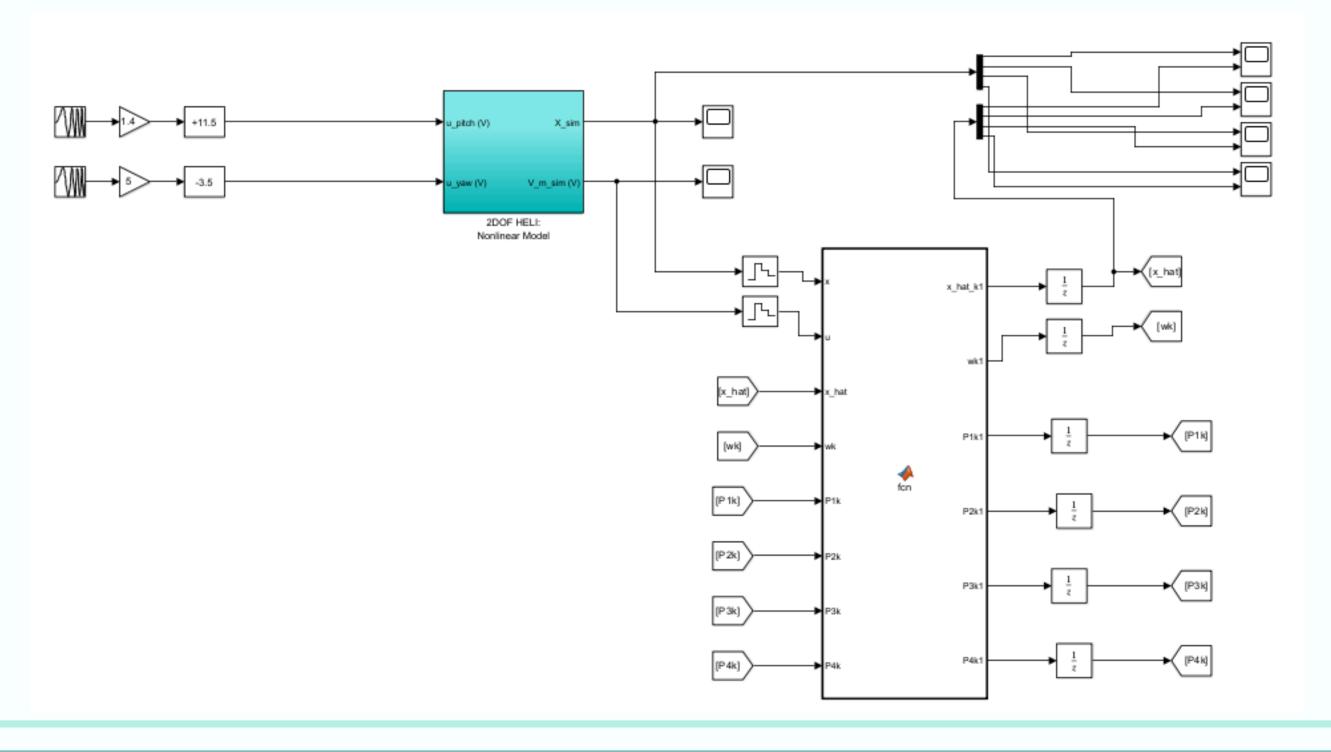
HELICOPTERO_2DOF_EKF

Implementar un identificador neuronal para el modelo del helicóptero Quanzer

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MODELO



$$(J_{eq,p} + m_{heli}l_{cm}^2)\ddot{\theta} = K_{pp}V_{m,p} + K_{py}V_{m,y} - B_p\dot{\theta} - m_{heli}gl_{cm}$$

$$(J_{eq,y} + m_{heli}l_{cm}^2)\ddot{\psi} = K_{yy}V_{m,y} + K_{yp}V_{m,p} - B_y\dot{\psi} + 2m_{heli}l_{cm}^2\theta\dot{\psi}\dot{\theta}.$$
 cross-coupling

PLANTA $X=[heta,\ \psi,\ \dot{ heta},\ \dot{\psi}]$

$$X=[heta,\ \psi,\ \dot{ heta},\ \dot{\psi}]$$

Entradas principales

u_pitch (V): Voltaje, responsable del movimiento de pitch. (inclinación vertical) u_yaw (V): Voltaje,genera el movimiento de yaw. (rotación horizontal)

> Pitch: movimiento vertical del brazo del helicóptero Inercia del eje (Jp) Torque principal generado por el motor de pitch (Kpp)

Torque cruzado inducido por el motor de yaw (Kyp). Par gravitacional, que depende del ángulo de inclinación Amortiguamiento viscoso; fricción (Bp). Sus salidas son el ángulo de pitch (θ) y su velocidad angular $(\dot{\theta})$.

Yaw: rotación horizontal del helicóptero Inercia del eje (Jy) Torque del motor lateral (Kyy) Efecto cruzado del motor de pitch (Kpy). Amortiguamiento; fricción (By) Sus salidas son el ángulo de yaw (ψ) y su velocidad angular ($\dot{\psi}$).

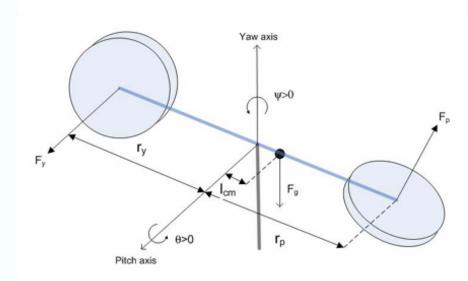


Figure 12: Dynamics of 2 DOF Helicopter.

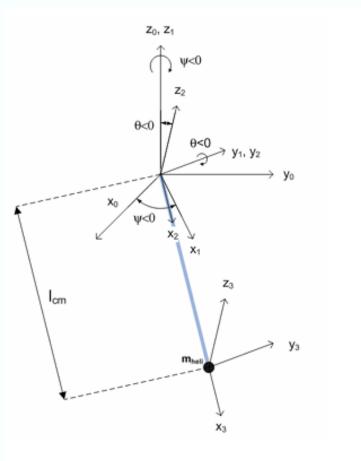


Figure 13: Kinematics of 2 DOF Helicopter.

MODELO NEURONAL

Propuesta 1

%% Neural states

%% H matrix coeff

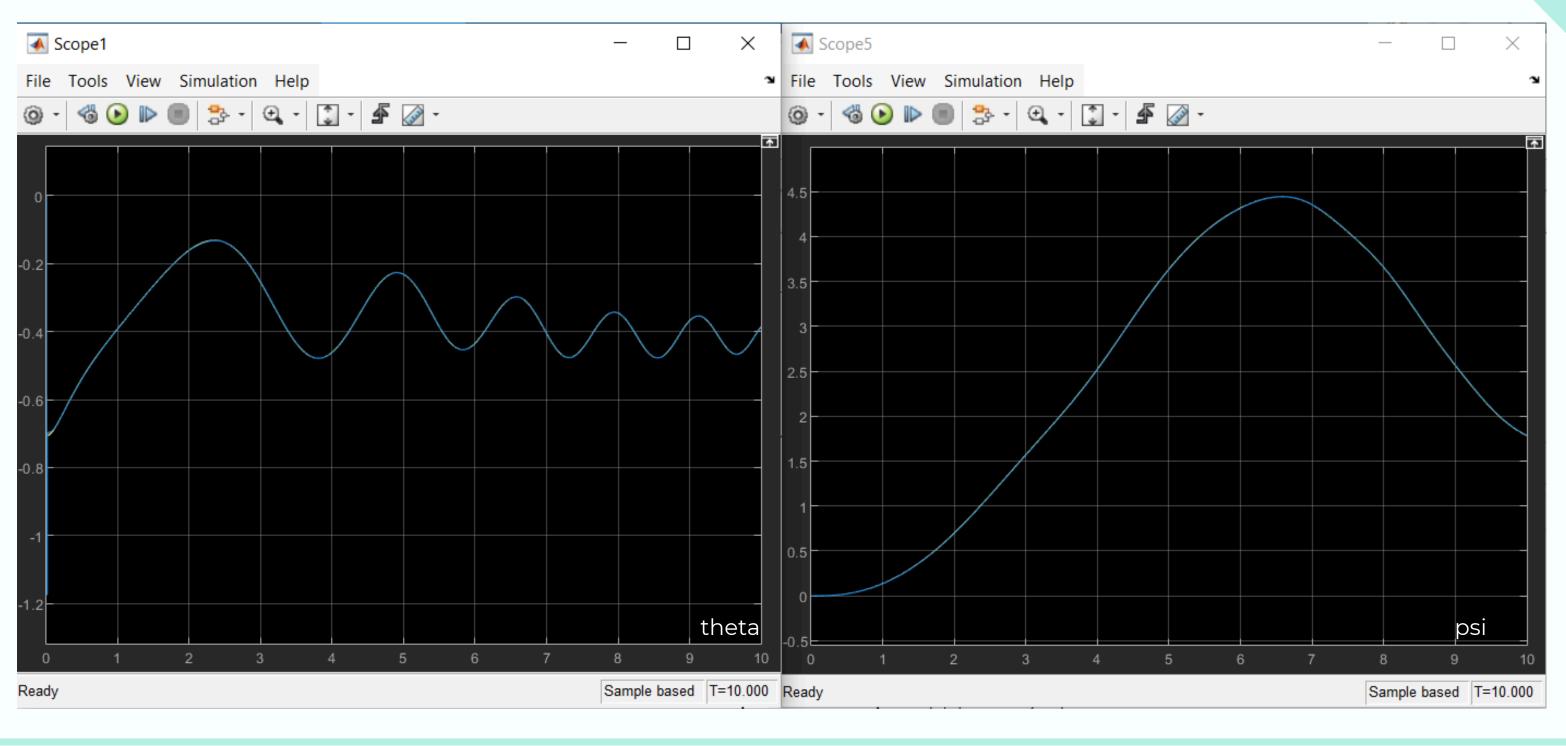
```
H1=[tanh(x(3)) tanh(x(1))]';

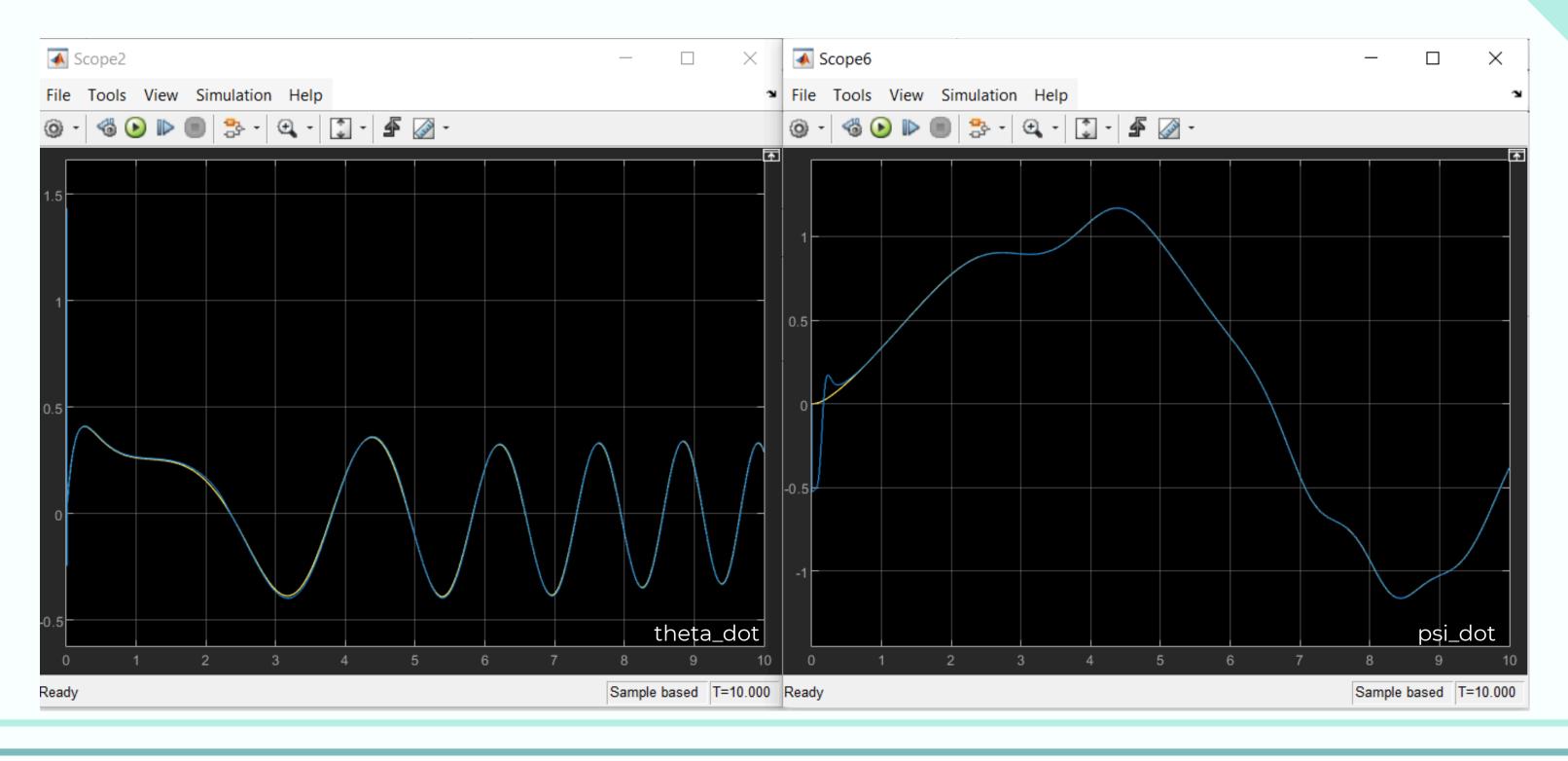
H2=[tanh(x(4)) tanh(x(2))]';

H3=[tanh(x(3)) tanh(x(1))]';

H4=[tanh(x(4)) tanh(x(2))]';
```

```
%error
error=x-x hat;
% First filter
K1=(P1k*H1)/(R1+H1'*P1k*H1);
w1k1=[wk(1);wk(2)]+K1*error(1);
P1k1=P1k-K1*H1'*P1k+Q1;
% Second filter
K2=P2k*H2/(R2+H2'*P2k*H2);
w2k1=[wk(3);wk(4)]+K2*error(2);
P2k1=P2k-K2*H2'*P2k+Q2;
% Third filter
K3=(P3k*H3)/(R3+H3'*P3k*H3);
w3k1=[wk(5);wk(6)]+K3*error(3);
P3k1=P3k-K3*H3'*P3k+Q3;
% Fourth filter
K4=(P4k*H4)/(R4+H4'*P4k*H4);
w4k1=[wk(7);wk(8)]+K4*error(4);
P4k1=P4k-K4*H4'*P4k+Q4;
wk1=[w1k1;w2k1;w3k1;w4k1];
```





MODELO NEURONAL

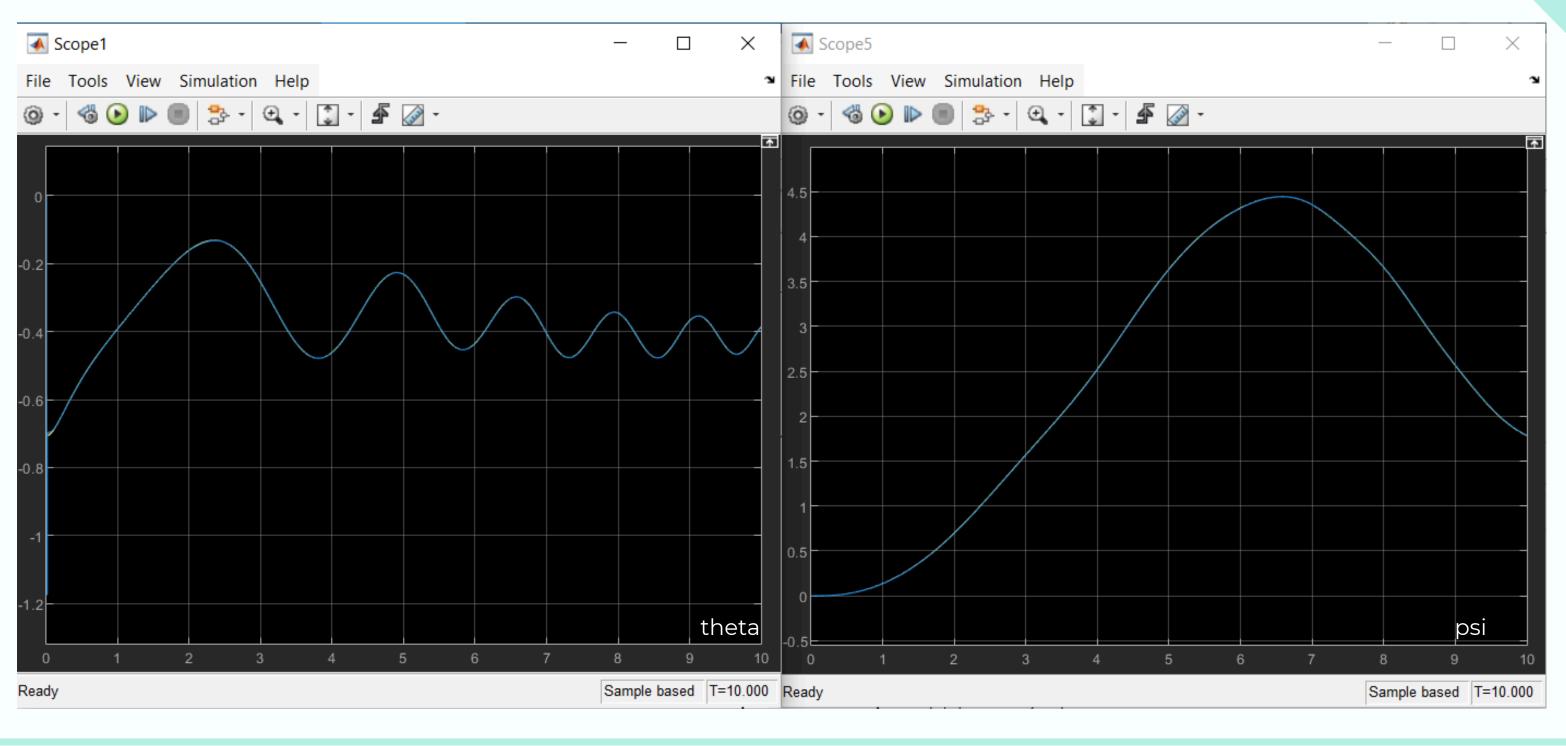
Propuesta 2

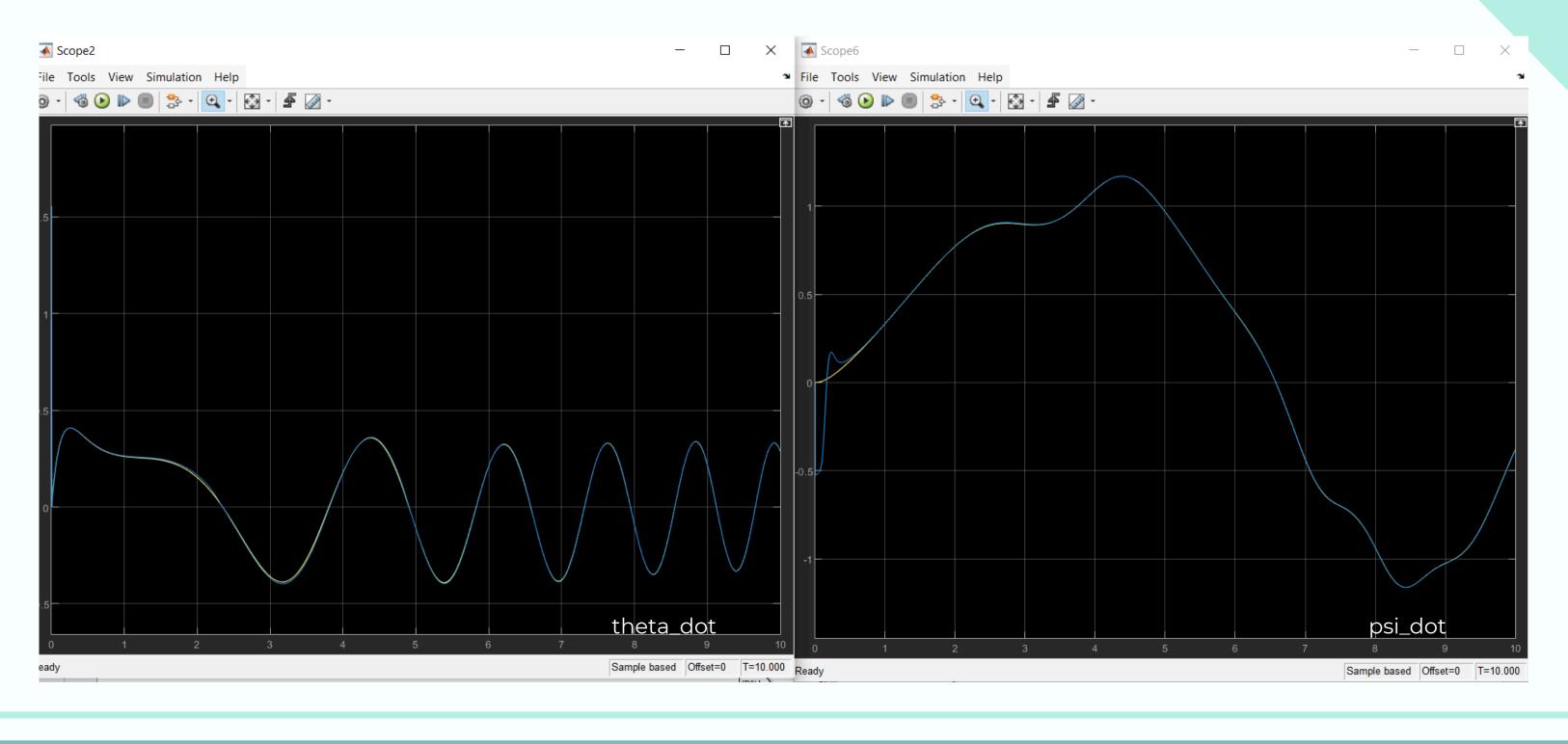
%% Neural states x_hat_k_1 = wk(1)*tanh(x(3)) + wk(2)*tanh(x(1)); x_hat_k_2 = wk(3)*tanh(x(4)) + wk(4)*tanh(x(2)); x_hat_k_3 = wk(5)*tanh(x(3)) + wk(6)*tanh(x(1))*cos(x(1))^2 + w_1*u(1); x_hat_k_4 = wk(7)*tanh(x(4)) + wk(8)*tanh(x(2))*cos(x(2)) + w_1*u(2); x_hat_k_1 = [x_hat_k_1; x_hat_k_2; x_hat_k_3; x_hat_k_4];

%% H matrix coeff

```
 \begin{aligned} &\text{H1=[tanh(x(3)) tanh(x(1))]';} \\ &\text{H2=[tanh(x(4)) tanh(x(2))]';} \\ &\text{H3=[tanh(x(3)) tanh(x(1))*cos(x(1))^2]';} \\ &\text{H4=[tanh(x(4)) tanh(x(2))*cos(x(2))]';} \end{aligned}
```

```
%error
error=x-x hat;
% First filter
K1=(P1k*H1)/(R1+H1'*P1k*H1);
w1k1=[wk(1);wk(2)]+K1*error(1);
P1k1=P1k-K1*H1'*P1k+Q1;
% Second filter
K2=P2k*H2/(R2+H2'*P2k*H2);
w2k1=[wk(3);wk(4)]+K2*error(2);
P2k1=P2k-K2*H2'*P2k+Q2;
% Third filter
K3=(P3k*H3)/(R3+H3'*P3k*H3);
w3k1=[wk(5);wk(6)]+K3*error(3);
P3k1=P3k-K3*H3'*P3k+Q3;
% Fourth filter
K4=(P4k*H4)/(R4+H4'*P4k*H4);
w4k1=[wk(7);wk(8)]+K4*error(4);
P4k1=P4k-K4*H4'*P4k+04;
wk1=[w1k1;w2k1;w3k1;w4k1];
```





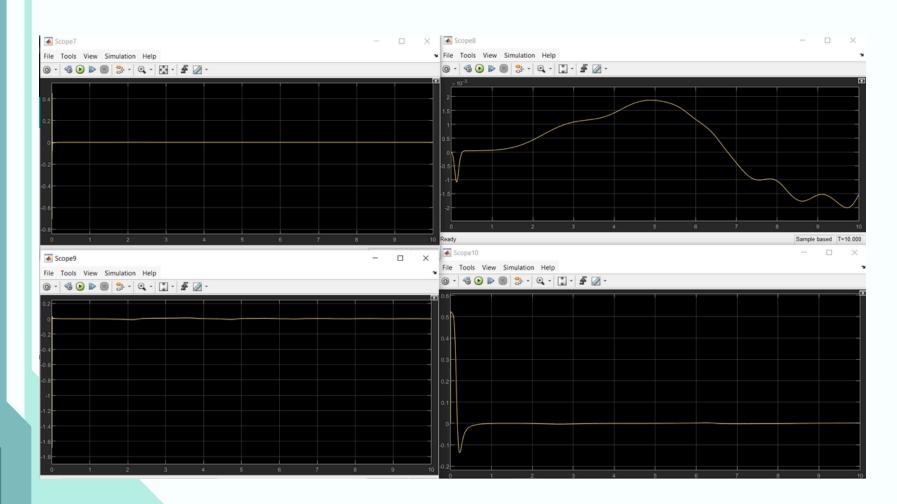
CONCLUSIONES

1.441e-05 z^2 - 1.441e-05 z _____ z^3 - 2.989 z^2 + 2.978 z - 0.9892

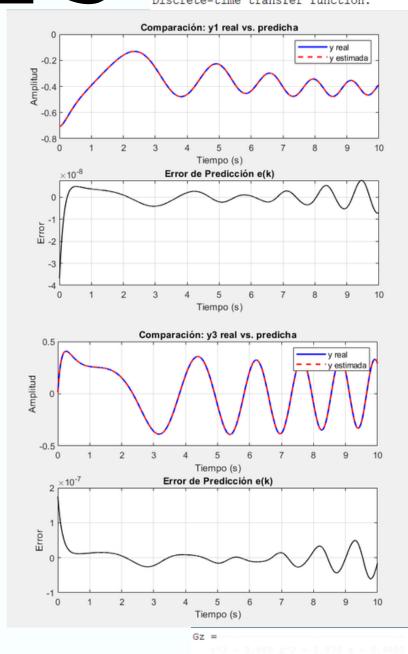
Sample time: 0.001 seconds Discrete-time transfer function.

9.204e-05 z^2 - 9.194e-05 z ----z^3 - 2.458 z^2 + 1.916 z - 0.4578

Sample time: 0.001 seconds Discrete-time transfer function.

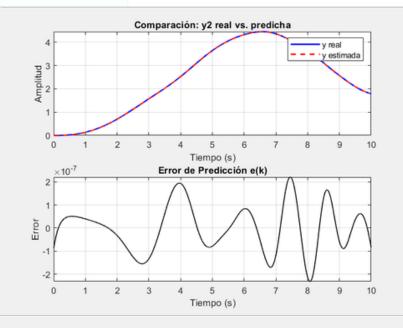


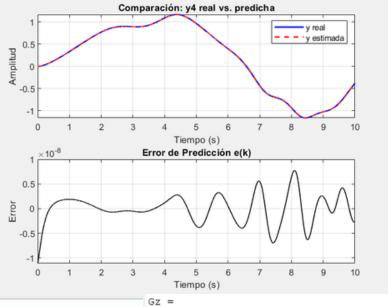




5.057e-07 z^2 - 5.058e-07 z z^3 - 2.999 z^2 + 2.998 z - 0.9991

Sample time: 0.001 seconds Discrete-time transfer function.





-8.781e-07 z^2 + 8.78e-07 z _____ z^3 - 3.003 z^2 + 3.005 z - 1.003 Sample time: 0.001 seconds Discrete-time transfer function.