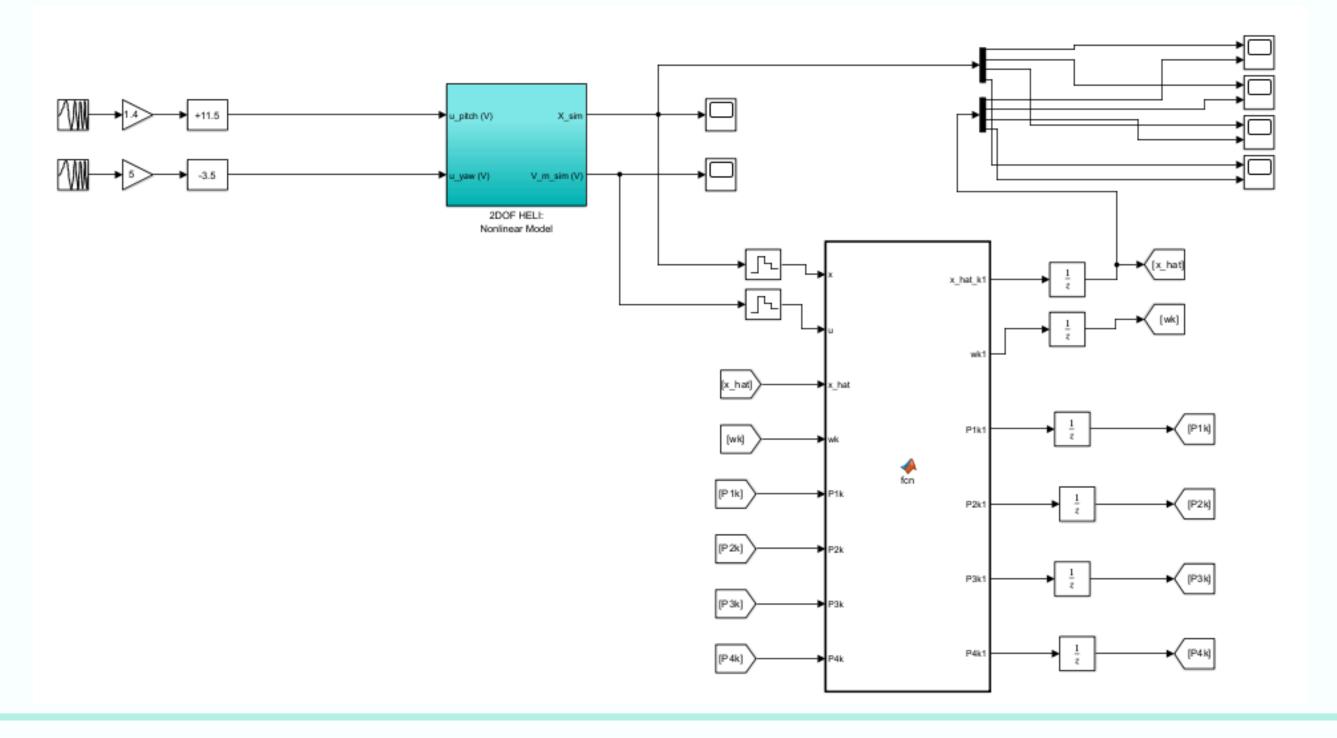
# HELICOPTERO\_2DOF\_EKF

Implementar un identificador neuronal para el modelo del helicóptero Quanzer

> Zhivago Ramos 0246042 Juan Pablo Larios 0244215 CONTROL DIFUSO



## MODELO



$$(J_{eq,p} + m_{heli}l_{cm}^2)\ddot{\theta} = K_{pp}V_{m,p} + K_{py}V_{m,y} - B_p\dot{\theta} - m_{heli}gl_{cm}$$
 
$$(J_{eq,y} + m_{heli}l_{cm}^2)\ddot{\psi} = K_{yy}V_{m,y} + K_{yp}V_{m,p} - B_y\dot{\psi} + 2m_{heli}l_{cm}^2\theta\dot{\psi}\dot{\theta}.$$
 cross-coupling

## PLANTA $X=[ heta,\ \psi,\ \dot{ heta},\ \dot{\psi}]$

$$X = [\theta, \ \psi, \ \dot{\theta}, \ \dot{\psi}]$$

Entradas principales

u\_pitch (V): Voltaje, responsable del movimiento de pitch. (inclinación vertical) u\_yaw (V): Voltaje,genera el movimiento de yaw. (rotación horizontal)

> Pitch: movimiento vertical del brazo del helicóptero Inercia del eje (Jp)

Torque principal generado por el motor de pitch (Kpp) Torque cruzado inducido por el motor de yaw (Kyp). Par gravitacional, que depende del ángulo de inclinación Amortiguamiento viscoso; fricción (Bp). Sus salidas son el ángulo de pitch  $(\theta)$  y su velocidad angular  $(\dot{\theta})$ .

Yaw: rotación horizontal del helicóptero Inercia del eje (Jy) Torque del motor lateral (Kyy) Efecto cruzado del motor de pitch (Kpy). Amortiguamiento; fricción (By) Sus salidas son el ángulo de yaw ( $\psi$ ) y su velocidad angular ( $\dot{\psi}$ ).

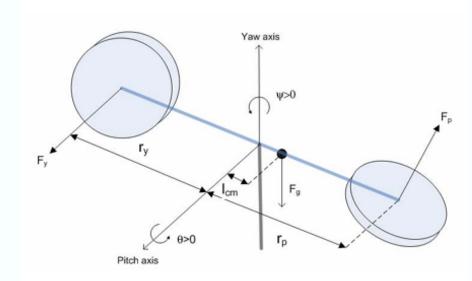


Figure 12: Dynamics of 2 DOF Helicopter.

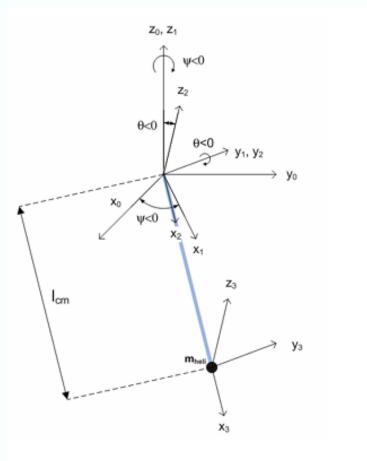


Figure 13: Kinematics of 2 DOF Helicopter.

#### MODELO NEURONAL

#### **Propuesta 1**

#### **%% Neural states**

#### %% H matrix coeff

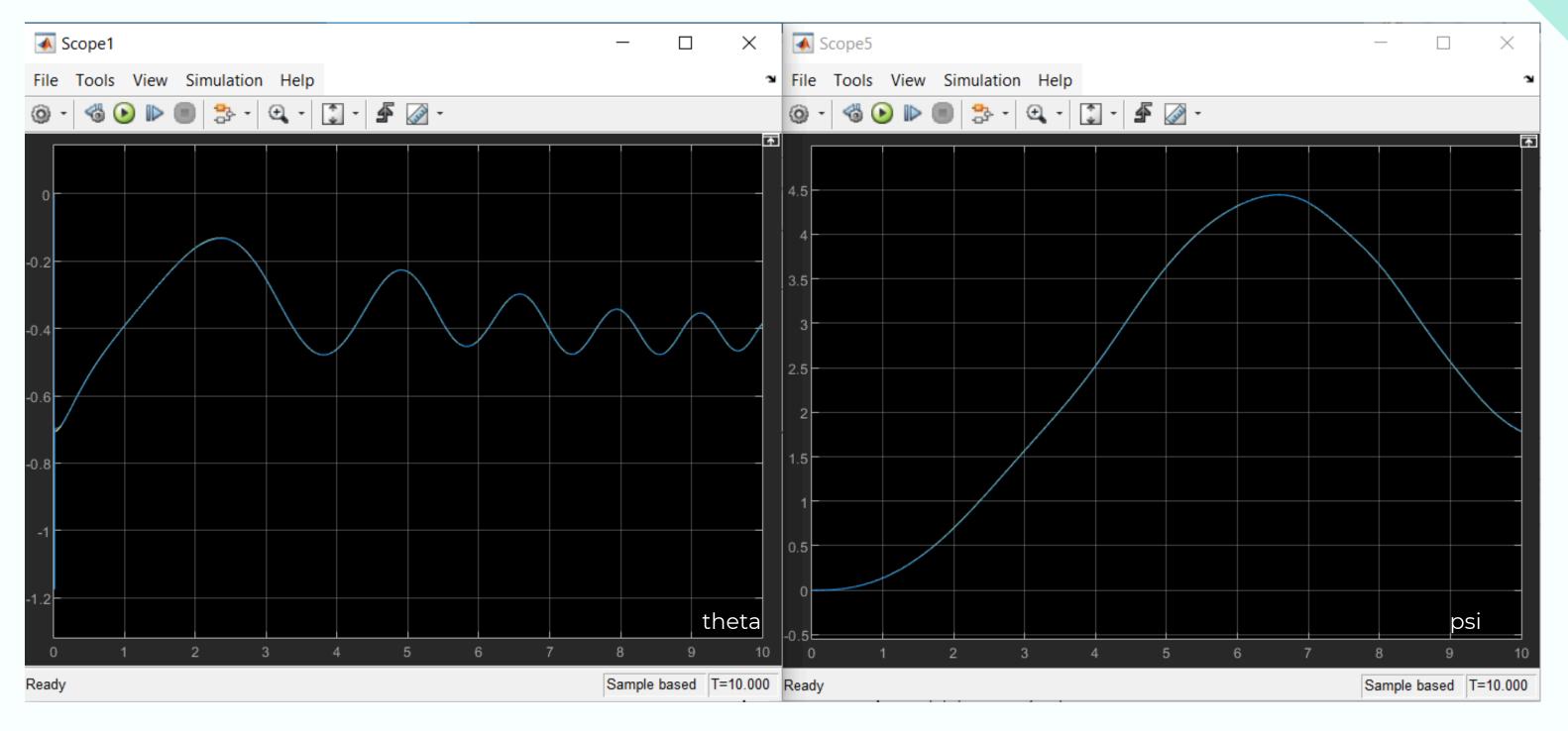
```
H1=[tanh(x(3)) tanh(x(1))]';

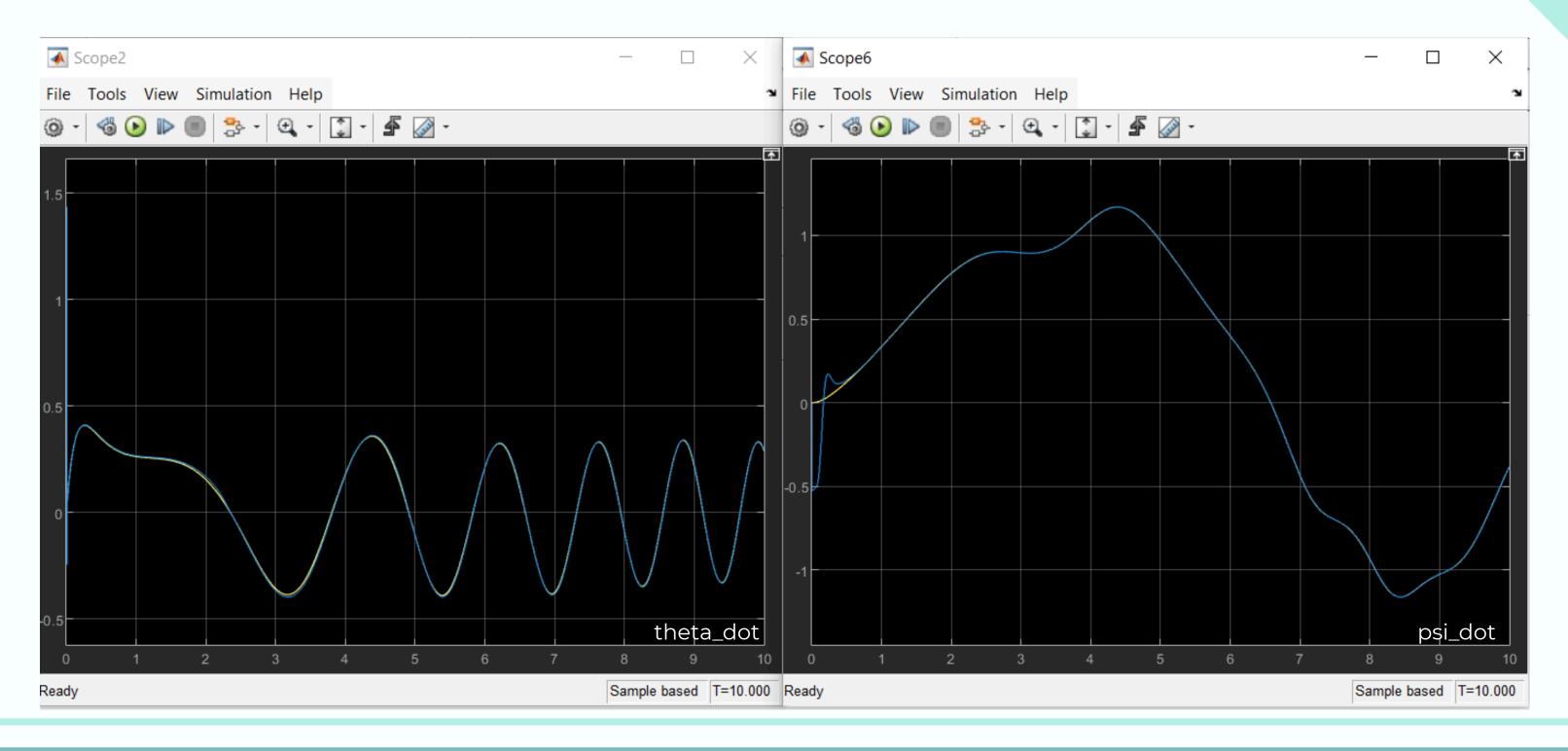
H2=[tanh(x(4)) tanh(x(2))]';

H3=[tanh(x(3)) tanh(x(1))]';

H4=[tanh(x(4)) tanh(x(2))]';
```

```
%error
error=x-x hat;
% First filter
K1=(P1k*H1)/(R1+H1'*P1k*H1);
w1k1=[wk(1);wk(2)]+K1*error(1);
P1k1=P1k-K1*H1'*P1k+Q1;
% Second filter
K2=P2k*H2/(R2+H2'*P2k*H2);
w2k1=[wk(3);wk(4)]+K2*error(2);
P2k1=P2k-K2*H2'*P2k+Q2;
% Third filter
K3=(P3k*H3)/(R3+H3'*P3k*H3);
w3k1=[wk(5);wk(6)]+K3*error(3);
P3k1=P3k-K3*H3'*P3k+Q3;
% Fourth filter
K4=(P4k*H4)/(R4+H4'*P4k*H4);
w4k1=[wk(7);wk(8)]+K4*error(4);
P4k1=P4k-K4*H4'*P4k+Q4;
wk1=[w1k1;w2k1;w3k1;w4k1];
```





### MODELO NEURONAL

#### **Propuesta 2**

# %% Neural states x\_hat\_k\_1 = wk(1)\*tanh(x(3)) + wk(2)\*tanh(x(1)); x\_hat\_k\_2 = wk(3)\*tanh(x(4)) + wk(4)\*tanh(x(2)); x\_hat\_k\_3 = wk(5)\*tanh(x(3)) + wk(6)\*tanh(x(1))\*cos(x(1))^2 + w\_1\*u(1); x\_hat\_k\_4 = wk(7)\*tanh(x(4)) + wk(8)\*tanh(x(2))\*cos(x(2)) + w\_1\*u(2); x\_hat\_k\_1 = [x\_hat\_k\_1; x\_hat\_k\_2; x\_hat\_k\_3; x\_hat\_k\_4];

#### %% H matrix coeff

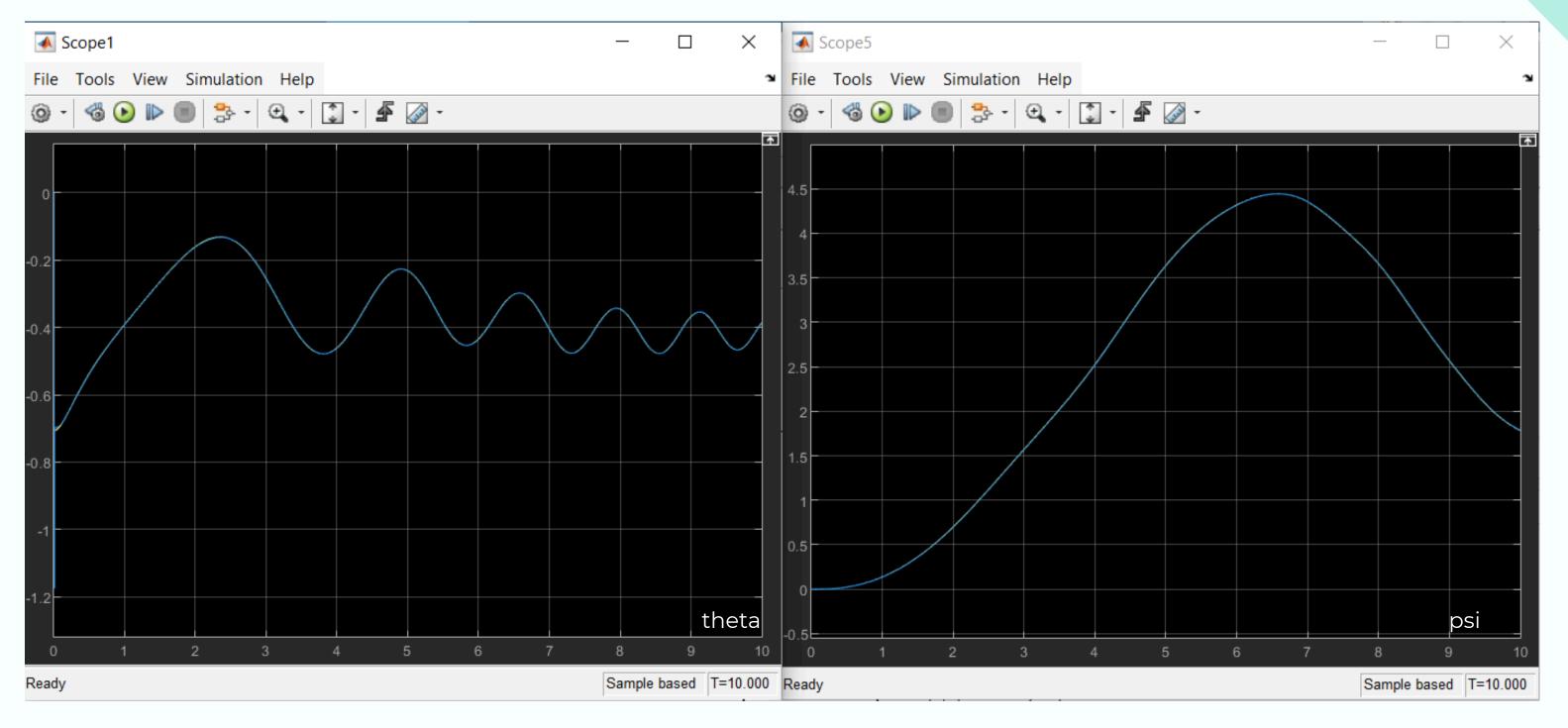
```
H1=[tanh(x(3)) tanh(x(1))]';

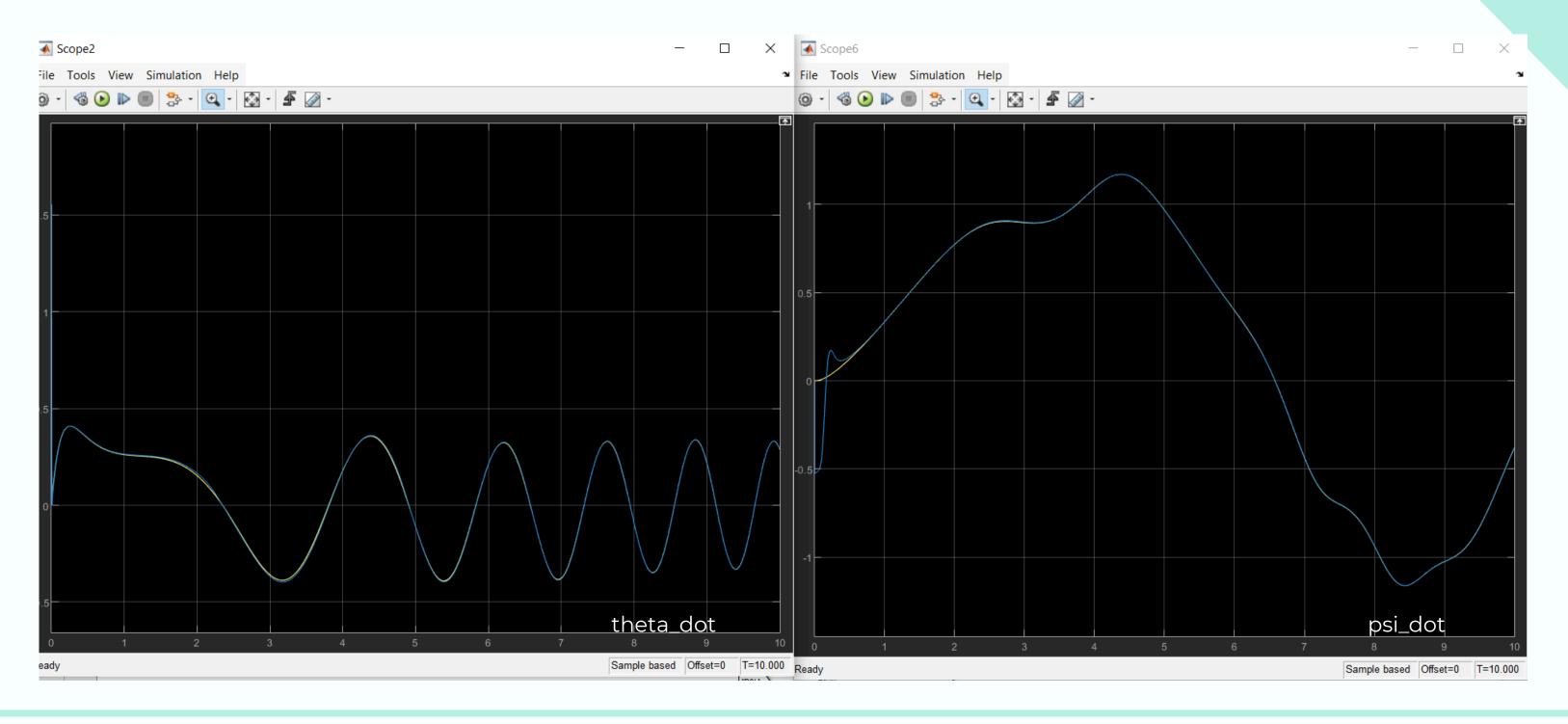
H2=[tanh(x(4)) tanh(x(2))]';

H3=[tanh(x(3)) tanh(x(1))*cos(x(1))^2]';

H4=[tanh(x(4)) tanh(x(2))*cos(x(2))]';
```

```
%error
error=x-x hat;
% First filter
K1=(P1k*H1)/(R1+H1'*P1k*H1);
w1k1=[wk(1);wk(2)]+K1*error(1);
P1k1=P1k-K1*H1'*P1k+Q1;
% Second filter
K2=P2k*H2/(R2+H2'*P2k*H2);
w2k1=[wk(3);wk(4)]+K2*error(2);
P2k1=P2k-K2*H2'*P2k+Q2;
% Third filter
K3=(P3k*H3)/(R3+H3'*P3k*H3);
w3k1=[wk(5);wk(6)]+K3*error(3);
P3k1=P3k-K3*H3'*P3k+Q3;
% Fourth filter
K4=(P4k*H4)/(R4+H4'*P4k*H4);
w4k1=[wk(7);wk(8)]+K4*error(4);
P4k1=P4k-K4*H4'*P4k+04;
wk1=[w1k1;w2k1;w3k1;w4k1];
```





## CONCLUSIONES

G2 =

1.441e-05 z^2 - 1.441e-05 z

z^3 - 2.989 z^2 + 2.978 z - 0.9892

Sample time: 0.001 seconds Discrete-time transfer function.

Comparación: y1 real vs. predicha

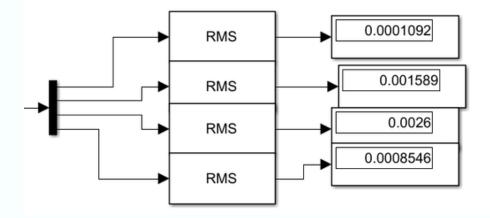
\_

9.204e-05 z^2 - 9.194e-05 z

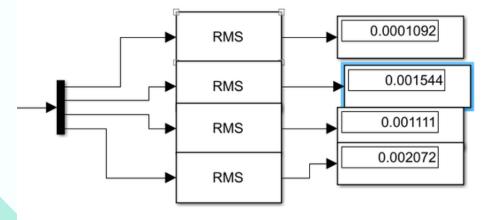
z^3 - 2.458 z^2 + 1.916 z - 0.4578

Sample time: 0.001 seconds Discrete-time transfer function.

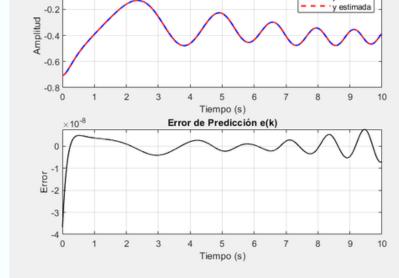
#### **RHONN 1**

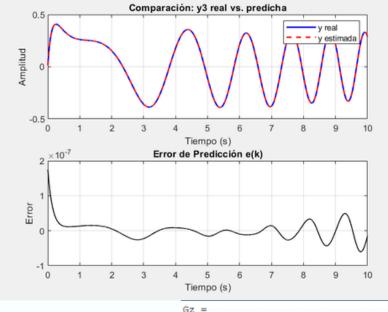


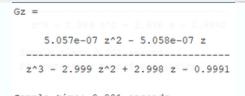
#### **RHONN 2**



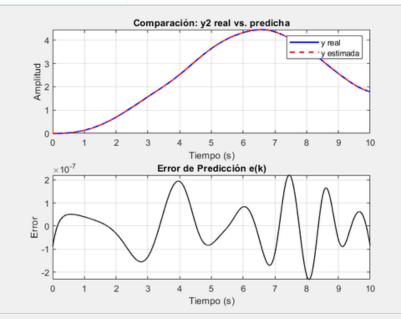
**EKF** 

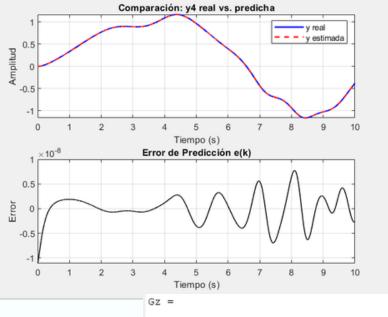






Sample time: 0.001 seconds Discrete-time transfer function.





Discrete-time transfer function.