# Mathematics

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# April 3, 2023

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- 1 Calculus
- 1.1 Differentiation

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- 3 Multivariable Calculus
- 4 Linear Algebra
- 5 Differential Equations
- 6 Partial Differential Equations

# 7 Vector Calculus

- 7.1 Operators
- 7.1.1 Grad
- 7.1.2 Div
- 7.1.3 Curl
- 7.1.4 Laplacian
- 7.2 Integral Theorems
- 7.2.1 Divergence Theorem
- 7.2.2 Stokes's Theorem

# 8 Fluid Mechanics

### 8.1 Kinematics

#### 8.1.1 Coordinates

**Lagrangian**  $\underline{x}(\underline{a}, t)$ : The motion of individual particles is studied; the position  $\underline{x}$  of a particle at time t is related to its position at a reference point in time  $\underline{a}$  (typically at t = 0).

**Eulerian**  $(\underline{x}, t)$ : The 'flow field' is considered as a whole and the state of a fluid is described in terms of the values at a fixed location  $\underline{x}$  and at a fixed time

#### 8.1.2 Velocity

In Cartesian coordinates the velocity of a fluid particle at position  $\underline{x}(x,y,z)$  is given by:

$$\underline{u}(x,y,z) = u(x,y,z)\underline{\hat{i}} + v(x,y,z)\underline{\hat{j}} + w(x,y,z)\underline{\hat{k}}$$

### 8.1.3 Stagnation Points

Stagnation points occur when the velocity vector  $\underline{u}$  is equal to  $\underline{0}$ 

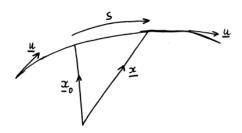
$$u = 0$$

$$v = 0$$

$$w = 0$$

#### 8.1.4 Streamlines

A streamline is a curve C drawn at one point in time such that the fluid velocity vector  $\underline{u}$  is tangent to C at every point along C.



$$\frac{d\underline{x}}{ds} = \underline{u}$$

$$\frac{dx}{ds} = u, \frac{dy}{ds} = v, \frac{dz}{ds} = w$$

$$\boxed{\frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w}(=ds)}$$

#### 8.1.5 Particle Paths

Particle path is obtained by solving the initial value problem:

$$\frac{d\underline{x}}{dt} = \underline{u}(\underline{x}, t) , \underline{x} = x_0 \text{ at } t = 0$$

$$\frac{dx}{dt} = u , x(0) = x_0$$

$$\frac{dy}{dt} = v, y(0) = y_0$$

$$\frac{dz}{dt} = w, z(0) = z_0$$

### 8.1.6 Steady Flow

**Steady Flow**: The flow velocity vector  $\underline{u}$  is independent of time t **Unsteady Flow**:  $\underline{u}$  depends on t; the pattern of streamlines changes with t

#### 8.1.7 Convective Derivative

The convective derivative tells us how a property changes as it moves with a flow.

General

$$\boxed{\frac{D*}{Dt} = \frac{\partial *}{\partial t} + (\underline{u} \cdot \nabla)* = \frac{\partial *}{\partial t} + u\frac{\partial *}{\partial x} + v\frac{\partial *}{\partial y} + w\frac{\partial *}{\partial z}}$$

Scalar

$$\frac{D\rho}{Dt} = \frac{\partial\rho}{\partial t} + (\underline{u}\cdot\nabla)\rho = \frac{\partial\rho}{\partial t} + u\frac{\partial\rho}{\partial x} + v\frac{\partial\rho}{\partial y} + w\frac{\partial\rho}{\partial z}$$

Vector

$$\frac{D\underline{\mathbf{u}}}{Dt} = \frac{\partial\underline{\mathbf{u}}}{\partial t} + (\underline{u} \cdot \nabla)\underline{\mathbf{u}} = \frac{\partial\underline{\mathbf{u}}}{\partial t} + u\frac{\partial\underline{\mathbf{u}}}{\partial x} + v\frac{\partial\underline{\mathbf{u}}}{\partial y} + w\frac{\partial\underline{\mathbf{u}}}{\partial z}$$

## 8.1.8 Vorticity

Vorticity  $\underline{\omega}$  is a measure of the local rotation of fluid particles in flow.

$$\underline{\omega} = \nabla \times \underline{u}$$

$$\nabla \times \underline{u} = \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ u & v & w \end{vmatrix}$$

Irrotational Flow:

$$\underline{\omega} = \underline{0}$$

## 8.1.9 Incompressible Flow

$$\nabla \cdot \underline{u} = 0$$

When this is true the convective derivative of the fluid density is zero:

$$\frac{D\rho}{Dt} = 0$$

### 8.1.10 Velocity Potential

For an irrotational flow the velocity can be described as the gradient of a scalar field known as the *Velocity Potential*.

$$\nabla \times u = 0$$

The curl of the gradient of a scalar field is zero:

$$\nabla \times (\nabla \phi) = 0$$
$$u = \nabla \phi$$

If the flow is also incompressible

$$\nabla \cdot \underline{u} = 0$$
$$\nabla \cdot (\nabla \phi) = 0$$

Therefore the velocity potential of an irrotational, incompressible flow satisfies Laplace's Equation:

$$\nabla^2 \phi = 0$$

### 8.1.11 Equipotential Surfaces

Lines/surfaces of constant  $\phi$  are equipotentials.

The velocity potential can be considered a surface:

$$\phi(x, y, z) = c$$

Let  $\underline{a}$  be tangent to the surface, the derivative of  $\phi$  in the direction of  $\underline{a}$ :

$$\underline{a} \cdot \nabla \phi = 0$$

Because the derivative of a constant c is zero:  $\nabla \phi$  is normal to the surface.

$$\hat{\underline{n}} = \frac{\nabla \phi}{|\nabla \phi|}$$

#### 8.1.12 The Stream Function

Considering incompressible two dimensional flow:

$$\nabla \cdot \underline{u} = 0$$
$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

We can introduce the stream function  $\psi(x,y,t)$  such that:

$$\boxed{u = \frac{\partial \psi}{\partial y}, v = -\frac{\partial \psi}{\partial x}}$$

This satisfies the previous equations:

$$\frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial^2 \psi}{\partial y \partial x} = 0$$

Vorticity of an incompressible two dimensional flow:

$$\underline{\omega} = \nabla \times \underline{u}$$

$$\underline{\omega} = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$$

$$\underline{\omega} = -\frac{\partial^2 \psi}{\partial^2 x} - \frac{\partial^2 \psi}{\partial^2 y}$$

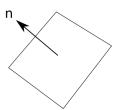
$$\underline{\omega} = -\nabla^2 \psi$$

## 8.2 Pressure in a Fluid

### 8.2.1 Pressure and Force

Let the normal vector  $\underline{n}$  be pointing into the fluid.

$$\partial \underline{F} = -\partial F \underline{\underline{n}}$$



 $\partial F$  is the force exerted over a small area  $\partial S$ 

$$p = \lim_{\partial S \to 0} \left( \frac{\partial F}{\partial S} \right)$$
$$\underline{F} = -\int \int_{S} p\underline{n} dS$$

#### **Equations of State** 8.2.2

Ideal Gas Law

$$pV = nRT$$

$$\frac{pV}{T} = nR$$

$$V \propto 1/\rho$$

$$\boxed{\frac{p}{\rho T} = constant}$$

#### **Isothermal Gas**

For a gas at constant temperature T

Boyle's Law:

$$p \propto \rho$$

$$p\rho^{-1} = p_0 \rho_0^{-1}$$

### Adiabatic Gas

Quick fluctuations in pressure (such as in sound waves) are described by the adiabatic law.

$$p\rho^{-\gamma} = c$$

Where  $\gamma$  is the *Polytropic Index* 

#### 8.2.3Hydrostatics

The force due to the pressure of the fluid

$$F = -\int \int_{\Omega} pndS$$

 $\underline{F} = -\int \int_S p\underline{n}dS$  Applying the divergence theorem for a scalar:  $\underline{F} = -\int \int \int_V \nabla p dV$ 

$$F = -\int \int \int_{V} \nabla p dV$$

Calculating the weight of the fluid region:

$$\underline{W} = \int \int \int_{V} \rho g dV$$

Since the fluid is stationary, the total force on the fluid is zero

$$\begin{split} \frac{W}{V} + \underline{F} &= 0\\ \int \int \int_{V} \rho \underline{g} dV - \int \int \int_{V} \nabla p dV &= 0\\ \int \int \int_{V} (\rho \underline{g} - \nabla p) dV &= 0\\ \text{Therefore:}\\ \boxed{\rho \underline{g} = \nabla p} \end{split}$$

$$g = -g\underline{\hat{k}}$$

$$\frac{\partial \overline{p}}{\partial z} = -\rho g$$
Integrating gives:
$$p = p_a - \rho gz$$

### 8.2.4 Buoyancy

Buoyancy is the force exerted on a submerged body due to the pressure of the surrounding static fluid.

$$\begin{split} \underline{B} &= -\int \int_S p\underline{n}dS \\ \text{Applying the divergence theorem for a scalar:} \\ \underline{B} &= -\int \int \int_V \nabla p dV \\ \nabla p &= \rho \underline{g}, \ \underline{g} = -g \underline{\hat{k}} \\ \underline{B} &= \underline{\hat{k}}g \int \int \int_V \rho dV \\ \underline{B} &= \rho V g \underline{\hat{k}} = m_w g \underline{\hat{k}} \end{split}$$

**Archimedes' Principle** – the buoyancy force on a body is equal in magnitude to the weight of fluid that is displaced by the body

- 8.3 Flow Dynamics
- 8.4 Tow-dimensional Flow
- 8.5 Vorticity Dynamics
- 8.6 Free Surface Waves

# 9 Numerical Methods