# Mathematics

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### 7 Vector Calculus

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### 8 Fluid Mechanics

#### 8.1 Kinematics

#### 8.1.1 Coordinates

**Lagrangian**  $\underline{x}(\underline{a}, t)$ : The motion of individual particles is studied; the position  $\underline{x}$  of a particle at time t is related to its position at a reference point in time  $\underline{a}$  (typically at t = 0).

**Eulerian**  $(\underline{x}, t)$ : The 'flow field' is considered as a whole and the state of a fluid is described in terms of the values at a fixed location  $\underline{x}$  and at a fixed time

#### 8.1.2 Velocity

In Cartesian coordinates the velocity of a fluid particle at position  $\underline{x}(x,y,z)$  is given by:

$$\underline{u}(x,y,z) = u(x,y,z)\underline{\hat{i}} + v(x,y,z)\underline{\hat{j}} + w(x,y,z)\underline{\hat{k}}$$

#### 8.1.3 Stagnation Points

Stagnation points occur when the velocity vector  $\underline{u}$  is equal to  $\underline{0}$ 

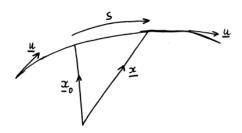
$$u = 0$$

$$v = 0$$

$$w = 0$$

#### 8.1.4 Streamlines

A streamline is a curve C drawn at one point in time such that the fluid velocity vector  $\underline{u}$  is tangent to C at every point along C.



$$\frac{d\underline{x}}{ds} = \underline{u}$$

$$\frac{dx}{ds} = u, \frac{dy}{ds} = v, \frac{dz}{ds} = w$$

$$\boxed{\frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w}(=ds)}$$

#### 8.1.5 Particle Paths

Particle path is obtained by solving the initial value problem:

$$\frac{d\underline{x}}{dt} = \underline{u}(\underline{x}, t) , \underline{x} = x_0 \text{ at } t = 0$$

$$\frac{dx}{dt} = u , x(0) = x_0$$

$$\frac{dy}{dt} = v, y(0) = y_0$$

$$\frac{dz}{dt} = w, z(0) = z_0$$

#### 8.1.6 Steady Flow

**Steady Flow**: The flow velocity vector  $\underline{u}$  is independent of time t **Unsteady Flow**:  $\underline{u}$  depends on t; the pattern of streamlines changes with t

#### 8.1.7 Convective Derivative

The convective derivative tells us how a property changes as it moves with a flow.

General

$$\boxed{\frac{D*}{Dt} = \frac{\partial *}{\partial t} + (\underline{u} \cdot \nabla)* = \frac{\partial *}{\partial t} + u\frac{\partial *}{\partial x} + v\frac{\partial *}{\partial y} + w\frac{\partial *}{\partial z}}$$

Scalar

$$\frac{D\rho}{Dt} = \frac{\partial\rho}{\partial t} + (\underline{u}\cdot\nabla)\rho = \frac{\partial\rho}{\partial t} + u\frac{\partial\rho}{\partial x} + v\frac{\partial\rho}{\partial y} + w\frac{\partial\rho}{\partial z}$$

Vector

$$\frac{D\underline{\mathbf{u}}}{Dt} = \frac{\partial\underline{\mathbf{u}}}{\partial t} + (\underline{u} \cdot \nabla)\underline{\mathbf{u}} = \frac{\partial\underline{\mathbf{u}}}{\partial t} + u\frac{\partial\underline{\mathbf{u}}}{\partial x} + v\frac{\partial\underline{\mathbf{u}}}{\partial y} + w\frac{\partial\underline{\mathbf{u}}}{\partial z}$$

### 8.1.8 Vorticity

Vorticity  $\underline{\omega}$  is a measure of the local rotation of fluid particles in flow.

$$\underline{\omega} = \nabla \times \underline{u}$$

$$\nabla \times \underline{u} = \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ u & v & w \end{vmatrix}$$

Irrotational Flow:

$$\underline{\omega} = \underline{0}$$

#### 8.1.9 Incompressible Flow

$$\nabla \cdot \underline{u} = 0$$

When this is true the convective derivative of the fluid density is zero:

$$\frac{D\rho}{Dt} = 0$$

#### 8.1.10 Velocity Potential

For an irrotational flow the velocity can be described as the gradient of a scalar field known as the Velocity Potential.

$$\nabla \times \underline{u} = 0$$

The curl of the gradient of a scalar field is zero:

$$\nabla \times (\nabla \phi) = 0$$
$$u = \nabla \phi$$

If the flow is also incompressible

$$\nabla \cdot \underline{u} = 0$$
$$\nabla \cdot (\nabla \phi) = 0$$

Therefore the velocity potential of an irrotational, incompressible flow satisfies

Laplace's Equation:

$$\nabla^2 \phi = 0$$

#### 8.1.11 Equipotential Surfaces

Lines/surfaces of constant  $\phi$  are equipotentials. The velocity potential can be considered a surface:

$$\phi(x, y, z) = c$$

Let  $\underline{a}$  be tangent to the surface, the derivative of  $\phi$  in the direction of  $\underline{a}$ :

$$\underline{a} \cdot \nabla \phi = 0$$

Because the derivative of a constant c is zero:  $\nabla \phi$  is normal to the surface.

$$\hat{\underline{n}} = \frac{\nabla \phi}{|\nabla \phi|}$$

#### 8.1.12 The Stream Function

Considering incompressible two dimensional flow:

$$\nabla \cdot \underline{u} = 0$$
$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

We can introduce the stream function  $\psi(x, y, t)$  such that:

$$\boxed{u = \frac{\partial \psi}{\partial y}, v = -\frac{\partial \psi}{\partial x}}$$

This satisfies the previous equations:

$$\frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial^2 \psi}{\partial y \partial x} = 0$$

Vorticity of an incompressible two dimensional flow:

$$\underline{\omega} = \nabla \times \underline{u}$$

$$\underline{\omega} = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$$

$$\underline{\omega} = -\frac{\partial^2 \psi}{\partial^2 x} - \frac{\partial^2 \psi}{\partial^2 y}$$

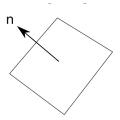
$$\underline{\omega} = -\nabla^2 \psi$$

#### 8.2 Pressure in a Fluid

#### 8.2.1 Pressure and Force

Let the normal vector  $\underline{n}$  be pointing into the fluid.

$$\partial \underline{F} = -\partial F \underline{n}$$



 $\partial F$  is the force exerted over a small area  $\partial S$ 

$$p = \lim_{\partial S \to 0} \left( \frac{\partial F}{\partial S} \right)$$
$$\underline{F} = -\int \int_{S} p\underline{n} dS$$

#### 8.2.2 Equations of State

Ideal Gas Law

$$pV = nRT$$
 
$$\frac{pV}{T} = nR$$
 
$$V \propto 1/\rho$$
 
$$\frac{p}{\rho T} = constant$$

#### **Isothermal Gas**

For a gas at constant temperature T

Boyle's Law:

$$p \propto \rho$$

$$p\rho^{-1} = p_0 \rho_0^{-1}$$

#### Adiabatic Gas

Quick fluctuations in pressure (such as in sound waves) are described by the adiabatic law.

$$p\rho^{-\gamma} = c$$

Where  $\gamma$  is the *Polytropic Index* 

#### 8.2.3 Hydrostatics

The force due to the pressure of the fluid  $\underline{F} = -\int \int_S p\underline{n}dS$  Applying the divergence theorem for a scalar:  $\underline{F} = -\int \int \int_V \nabla p dV$ 

Calculating the weight of the fluid region:

$$\underline{W} = \int \int \int_{V} \rho g dV$$

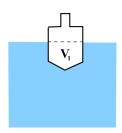
Since the fluid is stationary, the total force on the fluid is zero

#### 8.2.4 Buoyancy

Buoyancy is the force exerted on a submerged body due to the pressure of the surrounding static fluid.

$$\begin{split} \underline{B} &= -\int \int_S p\underline{n}dS \\ \text{Applying the divergence theorem for a scalar:} \\ \underline{B} &= -\int \int \int_V \nabla p dV \\ \nabla p &= \rho \underline{g}, \ \underline{g} = -g \underline{\hat{k}} \\ \underline{B} &= \underline{\hat{k}}g \int \int \int_V \rho dV \\ \underline{B} &= \rho V g \underline{\hat{k}} = m_w g \underline{\hat{k}} \end{split}$$

**Archimedes' Principle** – the buoyancy force on a body is equal in magnitude to the weight of fluid that is displaced by the body.



$$\begin{split} V_1 &= \text{Volume od displaced water,} \\ \rho_w &= \text{Density of water,} \\ m_B &= \text{Mass of the body} \\ \text{At equilibrium:} \\ \underline{B} + \underline{W} &= 0 \\ \rho_w V_1 g \hat{\underline{k}} - m_B g \hat{\underline{k}} = \underline{0} \\ \hline m_B &= \rho_w V_1 \end{split}$$

Center of Mass (Gravity): A theoretical point in the body where the body's total mass (weight) is thought to be concentrated.

Center of Buoyancy: The centre of mass of the displaced water

A floating body is stable if its centre of gravity is below its centre of buoyancy.

#### 8.2.5 Oscillation of a Floating Body

When a floating body is submerged below its equilibrium depth d it begins to oscillate.

Consider a cylindrical body with mass m, circular area A, equilibrium depth d:

$$mz''(t) = \rho_w g(d - z(t))A - mg$$
$$= -\rho_w gz(t)A$$
$$= mg\frac{z(t)}{d}$$

Therefore you can form the differential equation:

$$z''(t) = \frac{g}{d}z(t)$$

$$z(0) = -z_0$$
The solution:
$$z(t) = -z_0 \cos(\sqrt{\frac{g}{d}}t)$$

#### 8.3 Flow Dynamics

#### 8.3.1 Reynolds Transport Theorem

$$\frac{d}{dt} \int_{\Omega(t)} F(\underline{x}, t) dV = \lim_{\Delta t \to 0} \left[ \int_{\Omega(t + \Delta t)} F(\underline{x}, t + \Delta t) dV - \int_{\Omega(t)} F(\underline{x}, t) dV \right]$$

Using the Taylor series expansion:

$$\frac{d}{dt} \int_{\Omega(t)} F(\underline{x}, t) dV = \lim_{\Delta t \to 0} \left[ \int_{\Omega(t + \Delta t)} F(\underline{x}, t) + \frac{\partial F}{\partial t}(\underline{x}, t) \Delta t + O(\Delta t)^2 dV - \int_{\Omega(t)} F(\underline{x}, t) dV \right]$$

Let  $\Delta\Omega$  be the difference between the regions  $\Omega(t)$  and  $\Omega(t + \Delta t)$ :

$$\frac{d}{dt} \int_{\Omega(t)} F(\underline{x}, t) dV = \lim_{\Delta t \to 0} \left[ \int_{\Delta \Omega} F(\underline{x}, t) dV + \int_{\Omega(t + \Delta t)} \frac{\partial F}{\partial t}(\underline{x}, t) \Delta t + O(\Delta t)^2 dV \right]$$

Letting  $U_n$  be the normal velocity to the boundary  $\partial\Omega$ 

$$\frac{d}{dt} \int_{\Omega(t)} F(\underline{x}, t) dV = \lim_{\Delta t \to 0} \left[ \int_{\partial \Omega} F(\underline{x}, t) U_n(\underline{x}, t) \Delta t dS \right] + \lim_{\Delta t \to \infty} \left[ \int_{\Omega(t + \Delta t)} \frac{\partial F}{\partial t}(\underline{x}, t) \Delta t dV \right]$$

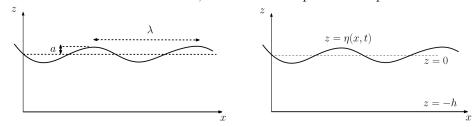
$$\frac{d}{dt} \int_{\Omega(t)} F(\underline{x}, t) dV = \int_{\partial \Omega} F(\underline{x}, t) U_n(\underline{x}, t) dS + \int_{\Omega(t)} \frac{\partial F}{\partial t}(\underline{x}, t) dV$$

**Leibniz's Integral Rule**: The rule can be derived from considering a one dimensional case of *Reynold's Transport Theorem*.

$$\frac{d}{dt} \int_{a(t)}^{b(t)} F(\underline{x}, t) dx = \int_{a(t)}^{b(t)} \frac{\partial F}{\partial t} (\underline{x}, t) dx + F(b(t), t) \frac{db}{dt} - F(a(t), t) \frac{da}{dt}$$

- 8.3.2 Conservation of Mass
- 8.4 Tow-dimensional Flow
- 8.5 Vorticity Dynamics
- 8.6 Free Surface Waves
- 8.6.1 Periodic Gravity Waves

The free surface of the liquid is described by  $z = \eta(x,t)$ , z = 0 is the undisturbed surface level, z = -h is the depth of the liquid.



 $\eta$  is assumed to take the form:

$$\eta(x,t) = a\sin(kx - \omega t)$$

k is the wave number, a is the amplitude,  $\omega$  is the angular frequency,  $\lambda$  is the wavelength.

$$k = \frac{2\pi}{\lambda}$$

We also assume the liquid is incompressible and irrotational

$$\nabla \times \underline{u} = 0 \text{ and } \nabla \cdot \underline{u} = 0$$
Therefore
$$\underline{u} = \nabla \phi \text{ and } \nabla^2 \phi = 0$$

#### 8.6.2 Boundary Conditions

No Normal Flow along the bottom at z = 0

$$\begin{array}{c}
\underline{\underline{u}} \cdot \underline{h} \\
= \underline{\underline{u}} \cdot \underline{\hat{k}} \\
= \nabla \phi \cdot \underline{\hat{k}} \\
= 0 \\
\text{Hence:} \\
\hline
\frac{\partial \phi}{\partial z} = 0 \text{ at } z = 0
\end{array}$$

Dynamic Boundary Condition at  $z = \eta(x, t)$ 

Fluid pressure at the free surface = Atmospheric pressure  $p_0$ 

Bernoulli's Equation relates 
$$p_0$$
 and  $\phi$ :
$$p_0 + \rho(\frac{\partial \phi}{\partial t} + \frac{1}{2}\underline{u}^2 + gz) = f(t)$$

$$\frac{\partial \phi}{\partial t} + \frac{1}{2}\underline{u}^2 + gz = \frac{f(t) - p_0}{\rho} = 0$$

$$\boxed{\frac{\partial \phi}{\partial t} + \frac{1}{2}|\nabla \phi|^2 + g\eta(x, t) = 0 \text{ at } z = \eta(x, t)}$$

### Kinematic Boundary Condition at $z = \eta(x, t)$

Fluid particles can not leave the free surface thus the surface must move with the fluid.

## 9 Numerical Methods