

Q. 주어진 궤도 정보로부터 주어진 날 자정부터 다음날 자정까지 하루 동안의 위치를 계산하여 Plot 수행

- 1. 주어진 정보를 loading 하는데 GPS, QZSS, BDS 중 선택
- 2. 관찰하고자 하는 시간을 설정
- 3. 관측자의 현재 위치(위도, 경도, 고도) 입력
- 4. Elevation angle의 cutoff 설정

```
%% Loading Information
         load('nav.mat')
         %% INPUT (Satetllite)
         disp("1)GPS 2)OZSS 3)BDS"):
         C = input("Choose from GPS, QZSS, or BDS which satellite information you want to receive ");
10
11
12
13
             a = nav.GPS.a*0.001; % Semi-major axis(km)
14
             e = nav.GPS.e; % Ecceinticity
             i = nav.GPS.i; % Inclination(rad)
15
             RAAN = nav.GPS.OMEGA; % RAAN(rad)
16
17
             w = nav.GPS.omega; % Argument of perigee(rad)
             M0 = nav.GPS.M0; % Mean anomaly at toc(rad)
18
             toc = nav.GPS.toc; %toc
19
20
         elseif C == 2
21
             a = nav.OZSS.a*0.001: % Semi-major axis(km)
                                                                                    %% INPUT (Time)
             e = nav.QZSS.e; % Ecceinticity
22
                                                                                    disp('Enter the date to observe')
23
             i = nav.OZSS.i; % Inclination(rad)
                                                                                    t = input("([YYYY,MM,DD,hh,mm,ss])format:");
                                                                          40
24
             RAAN = nav.QZSS.OMEGA; % RAAN(rad)
                                                                          41
                                                                                    Initial time = datetime(t):
25
             w = nav.OZSS.omega: % Argument of perigee(rad)
                                                                          42
                                                                                    Final time = datetime(t) + hours(24); % Observation time = 24hr
             M0 = nav.QZSS.M0; % Mean anomaly at toc(rad)
26
                                                                          43
                                                                                    % Time set
27
             toc = nav.QZSS.toc; %toc
                                                                          44
                                                                                    t toc = datetime(toc);
28
         elseif C == 3
                                                                          45
                                                                                    t t = linspace(Initial time, Final time, 1440);
29
             a = nav.BDS.a*0.001; % Semi-major axis(km)
                                                                          46
30
             e = nav.BDS.e; % Ecceinticity
                                                                          47
                                                                                    %% INPUT (Condition)
31
             i = nav.BDS.i; % Inclination(rad)
                                                                          48
                                                                                    ground lat = input('Enter lattitude of observation site (degree): ');
32
             RAAN = nav.BDS.OMEGA; % RAAN(rad)
                                                                          49
                                                                                    ground lon = input('Enter longitude of observation site (degree): ');
33
             w = nav.BDS.omega; % Argument of perigee(rad)
             M0 = nav.BDS.M0; % Mean anomaly at toc(rad)
                                                                          50
                                                                                    height = input('Enter height of observation site (km): ');
34
35
                                                                          51
                                                                                    el mask = input('Enter elevation mask (degree): ');
             toc = nav.BDS.toc; %toc
36
                                                                          52
37
```

- 5. 받은 입력값을 통해 궤도 계산 진행
- ① Orbital Elemets → PQW(Perifocal frame)
- \bigcirc PQW \rightarrow ECI
- ③ ECI → ECEF → ECEF frame에서 geoplot을 위한 위성의 위도 경도 계산
- ④ ECEF → ENU → ENU frame에서 skyplot을 위한 Az,El 계산

Cf) 경과하는 시간에 따라 값을 받아야하므로 반복문 사용

```
M = Mean(t toc, t t(k), a, M0);
61
              E = M2E(M, e);
62
             v = E2v(E, e);
63
64
              %% POW
              R POW = solveRangeInPerifocalFrame(a, e, v);
65
              V POW = solveVelocityInPerifocalFrame(a, e, v);
66
67
              %% POW -> ECI
68
69
             DCM P E = POW2ECI(w, i, RAAN);
70
              R_{ECI} = DCM_P_E * R_PQW;
71
             V ECI = DCM P E * V PQW;
72
73
              %% ECI -> ECEF
74
             DCM_E_E = ECI2ECEF(t_t(k));
              R ECEF = DCM E E * R ECI;
75
76
             V ECEF = DCM E E * V ECI;
77
78
              %% ECEF -> Geodetic
79
              wgs84 = wgs84Ellipsoid('kilometer');
80
              [lat, lon, h] = ecef2geodetic(wgs84, R_ECEF(1), R_ECEF(2), R_ECEF(3), "degrees");
81
              lat_geoplot = [lat_geoplot, lat];
82
              lon geoplot = [lon geoplot, lon];
83
              %% ECEF -> ENU
84
85
              [R_ENU(1), R_ENU(2), R_ENU(3)] = ecef2enu(R_ECEF(1), R_ECEF(2), R_ECEF(3), ground_lat, ground_lon, height, wgs84);
              R ENU = [R ENU(1); R ENU(2); R ENU(3)];
86
87
             % Azimuth angle, Elevation angle
88
              az = azimuth1(R ENU);
89
90
              el = elevation1(R_ENU, el_mask);
91
              az skyplot = [az skyplot, az];
92
              el_skyplot = [el_skyplot, el];
93
         end
```

6. Geoplot, Skyplot, Ground track

관측자 위도: 37(deg), 경도: 127(deg) 고도: 500(km)

Elevation Cutoff: 10(deg)

관측시간: 86400(s) = 24시간 관측날짜: 2023.06.22.00.00

GPS = 1, QZSS = 2, BDS = 3

```
1) GPS 2) QZSS 3) BDS
Choose from GPS, QZSS, or BDS which satellite information you want to receive 1
Enter the date to observe
([YYYY,MM,DD,hh,mm,ss]) format: [2023,06,22,00,00,00]
Enter lattitude of observation site (degree): 37
Enter longitude of observation site (degree): 127
Enter height of observation site (km): 500
Enter elevation mask (degree): 10
```

```
94
95
           %% Geoplot / Skyplot
96
          geoplot(lat geoplot, lon geoplot,'*');
97
           figure;
          skyplot(az skyplot, el skyplot);
98
99
100
           %% Simulation(GroundTrack)
101
           sampleTime = 60;
          sc = satelliteScenario(Initial time,Final time,sampleTime);
102
103
          simul = satellite(sc,a*1000,e,i,RAAN,w,v);
104
           show(simul)
105
           groundTrack(simul,LeadTime=3600)
          play(sc,PlaybackSpeedMultiplier=100)
106
107
```



