

Orbits are ellipses

Ellipse defined by foci

Amount of flattening: eccentricity

The longest axis: major axis

shortest axis: minor axis

1: Each planet's orbit about the sun is an ellipse

The sun's center is always located at one focus of the orbital ellipse

2: The imaginary line joining a planet and the Sun sweeps equal areas of space

during equal time intervals as the planet orbits.

Perihelion Aphelion

3: The squares of the orbital periods of the planets are directly proportional to the cubes of the semi-major axes of their orbits.

Kepler's Third Law

$$T^2 \propto R^3$$

$$T^2 = KR^3$$

T: Orbital period

R: Orbital radius

K is the Kepler's Constant which is the same for all satellites of a central body

Ex: K is the same for all planets in the Solar System because they all orbit the sun → central mass.

Example 1:

Mercury orbits the Sun in 88 days.

Find orbital radius

Earth: $T_E = 1 \text{ year}$

$R_E = 1 \text{ AU}$

$$K = \frac{T^2}{R^3} = \frac{1 \text{ yr}^2}{1 \text{ AU}^3}$$

Mercury: $T_m = 88 \text{ days} = 0.241 \text{ yr}$

$$K = \frac{1 \text{ yr}^2}{1 \text{ AU}^3}$$

$$R_m = \sqrt[3]{\frac{T_m^2}{K}} = \sqrt[3]{\frac{0.241^2}{1}} \rightarrow \sqrt[3]{\frac{1 \text{ yr}^2}{1 \text{ AU}^3}}$$

$$= 0.387 \text{ AU}$$

Ex 2:

Saturn has orbital radius of 9 AU.

Find one trip around the sun

$T_s?$ $R_s = 9 \text{ AU}$

$$T = \sqrt{K R_s^3} = \sqrt{1 (9)^3}$$

$$= 27 \text{ yr} \times \frac{365 \text{ d}}{\text{yr}} = 9855 \text{ days}$$

Newton proves Kepler's Laws

What factors affect the force of gravity?

Planetary Motion

$$F_g \propto \frac{mM}{R^2}$$

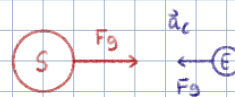
M = central mass

m = smaller mass

R = orbital radius

G = universal gravitational constant

$$6.67 \times 10^{-11} \frac{\text{Nm}^2}{\text{kg}^2}$$



$$\vec{F}_{\text{net}} = \vec{F}_g$$

$$m_E \vec{a}_c = \frac{G m_E M_S}{R^2}$$

$$\frac{4\pi^2 R}{T^2} = \frac{G M_S}{R^2}$$

$$T^2 = \frac{4\pi^2 R^3}{G M_S}$$

K
Kepler's
→ constant

$g = 9.81 \text{ (earth)}$

$$F_g = \frac{Gm}{R^2} m$$

$$F_g \propto \frac{GmM}{R^2}$$

$$G = \frac{F_g R^2}{mM}$$

$$= \frac{(\text{N})(\text{m}^2)}{(\text{kg})^2}$$

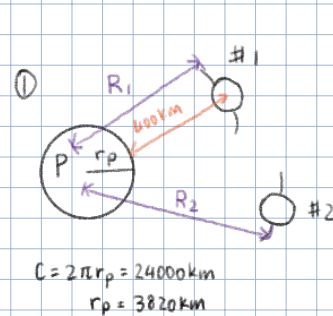
Kepler's Law Example 3

A planet has a circumference of 24000km. A moon is orbiting at an altitude of 400km and completes 2 revolutions in 3 hours. A 2nd moon orbits 2 times in a week.

Find:

a) The closest the two moons can get

b) The density of the planet



#1

$$T_1 = \frac{3 \text{ hr}}{2 \text{ rev}} = 1.5 \text{ hr/rev}$$

$$R_1 = r_p + \text{Alt} = 3820 + 400$$

$$= 4220 \text{ km}$$

$$K = \frac{T_1^2}{R_1^3} = \frac{1.5^2}{4220^3}$$

$$= 2.99 \times 10^{-11} \left(\frac{\text{hr}^2}{\text{km}^3} \right)$$

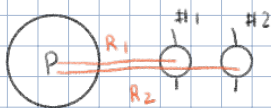
#2

$$T_2 = \frac{7 \text{ days}}{2 \text{ rev}} \times \frac{24 \text{ h}}{1 \text{ day}}$$

$$= 84 \text{ hr/rev}$$

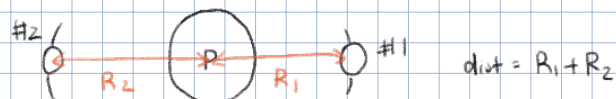
$$R_2 = \sqrt[3]{\frac{T_2^2}{K}} = \sqrt[3]{\frac{(84)^2}{2.99 \times 10^{-11}}}$$

$$= 61769 \text{ km}$$

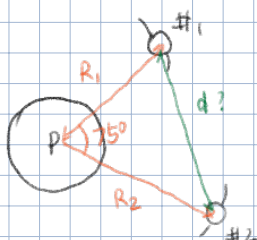


Closest distance: $R_2 - R_1 = 61769 - 4220$
 $= 57549 \text{ km}$

Variation to a) #1: Furthest distance

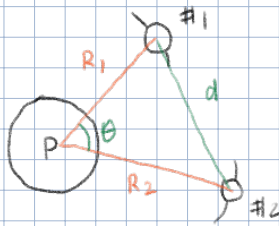


Variation to a) #2: if 75° ,



$$d = \sqrt{R_1^2 + R_2^2 - 2R_1R_2 \cos \theta}$$

Variation to a) #3: if they are 59000 km apart, θ ?



$$\cos \theta = \frac{d^2 - R_1^2 - R_2^2}{-2 R_1 R_2}$$

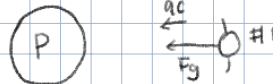
b) Density of planet

$$D = \frac{M_p}{V_p}$$

$$= \frac{1.53 \times 10^{24}}{2.33 \times 10^{20}} = 6567 \frac{\text{kg}}{\text{m}^3}$$

$$r_p = 3820 \text{ km}$$

$$\begin{aligned} V_p &= \frac{4}{3} \pi r_p^3 \\ &= \frac{4}{3} (\pi) (3820)^3 \\ &= 2.33 \times 10^{11} \text{ km}^3 \\ &= 2.33 \times 10^{20} \text{ m}^3 \end{aligned}$$



$$\vec{F}_{\text{net}} = \vec{F}_g$$

$$m_1 \vec{a}_c = \frac{G m_1 M_p}{R_1^2}$$

Show Work

$$\frac{4\pi^2 R}{T_1^2} = \frac{G M_p}{R_1^2}$$

$$M_p = \frac{4\pi^2 R_1^3}{G T_1^2}$$

$$M_p = \frac{4\pi^2 (4220000)^3}{6.67 \times 10^{-11} (5400)^2}$$

$$M_p = 1.53 \times 10^{24} \text{ kg}$$

$$R_1 = 4220000 \text{ m}$$

$$T_1 = 1.5 \text{ h} \times \frac{3600}{\text{h}}$$

$$= 5400 \text{ s}$$