

普朗特边界层

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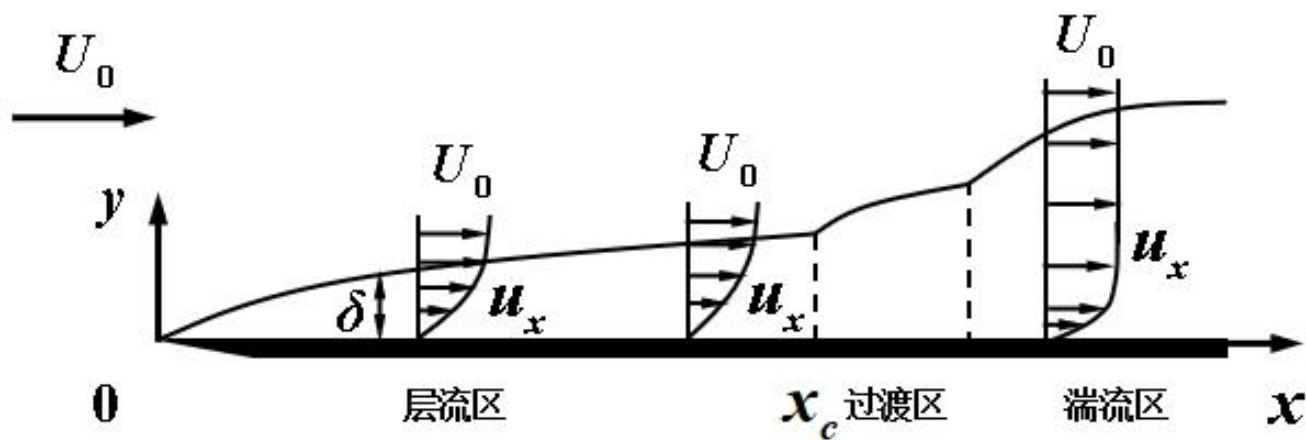
第六讲. 普朗特边界层

- 1. 边界层**
- 2. 普朗特边界层方程**
- 3. 边界层动量积分方程**
- 4. 边界层计算**

1. 边界层

1904年，普朗特提出“边界层”概念

高雷诺数下，流体区域以 $99\%U_0$ 作为边界，
内层区为粘性流体 u_x ，外层区为理想流体 U_0 。



边界层特点

①. 慢：边界层内 $u_x < U_0$ ，壁面 $u_x = 0$ 。

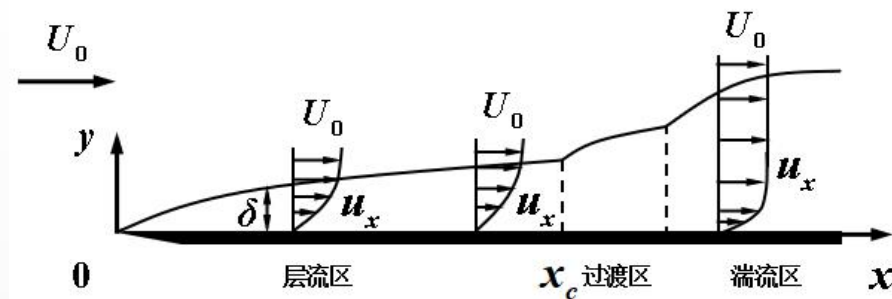
②. 薄： $\delta \ll x$ 。 ③. 陡： $\frac{du_x}{dy}$ 很大。

④. 增： $x \uparrow, \delta \uparrow$ 。 ⑤. 旋：微团有旋。

⑥. 惯、粘同量级：惯性力与粘性力在边界层内量级相当。

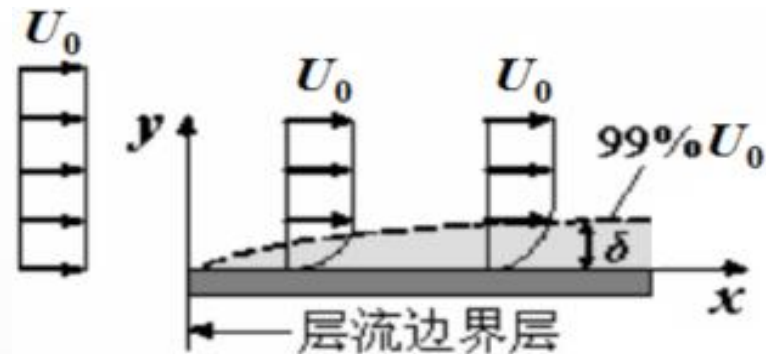
⑦. 截面等压力：无压差流动。 ⑧. 流型会转变： $x > x_c$ 时，层流→湍流。

⑨. 逆压，失速会分离（绕曲面流动时的表现）。



2. 普朗特边界层方程

物理分析



定常: $\frac{\partial u_x}{\partial t} = \frac{\partial u_y}{\partial t} = 0$

二维流动: $\begin{cases} u_x \neq 0 \\ u_y \neq 0 \\ u_z = 0 \end{cases} \quad \begin{cases} \frac{\partial u_x}{\partial x} \neq 0 \\ \frac{\partial u_x}{\partial y} \neq 0 \\ \frac{\partial u_x}{\partial z} = 0 \end{cases} \quad \begin{cases} \frac{\partial u_y}{\partial x} \neq 0 \\ \frac{\partial u_y}{\partial y} \neq 0 \\ \frac{\partial u_y}{\partial z} = 0 \end{cases} \quad \begin{cases} \frac{\partial^2 u_x}{\partial x^2} \neq 0 \\ \frac{\partial^2 u_x}{\partial y^2} \neq 0 \\ \frac{\partial^2 u_x}{\partial z^2} = 0 \end{cases} \quad \begin{cases} \frac{\partial^2 u_y}{\partial x^2} \neq 0 \\ \frac{\partial^2 u_y}{\partial y^2} \neq 0 \\ \frac{\partial^2 u_y}{\partial z^2} = 0 \end{cases}$

无压差流动: $\frac{\partial p}{\partial x} = \frac{\partial p}{\partial y} = 0$

忽略重力: $X = Y = 0$

问题探讨 边界层内流动是二维流动吗？

$$\rho \left(\frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} + u_z \frac{\partial u_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \rho X + \mu \left(\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} + \frac{\partial^2 u_x}{\partial z^2} \right)$$

奈维-斯托克斯方程简化可得：

$$u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = \nu \left(\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right)$$

同理，y 方向可得：

$$u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} = \nu \left(\frac{\partial^2 u_y}{\partial x^2} + \frac{\partial^2 u_y}{\partial y^2} \right)$$

连续性方程：

$$\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0$$

量级比较

边界层厚度薄, $\delta \ll x$ 。

量级表示: $\begin{cases} x \sim O(L) \\ y \sim O(\delta) \end{cases}$

连续性方程: $\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0$

$$u_x \sim O(U_0) \quad \frac{\partial u_x}{\partial x} \sim O\left(\frac{U_0}{L}\right)$$

$$\frac{\partial u_y}{\partial y} \sim O\left(\frac{U_0}{L}\right) \quad u_y \sim O\left(\frac{\delta U_0}{L}\right)$$

量级表示

$$\begin{cases} u_x \sim O(U_0) \\ u_y \sim O\left(\frac{\delta U_0}{L}\right) \end{cases}$$

$$\begin{cases} \frac{\partial u_x}{\partial x} \sim O\left(\frac{U_0}{L}\right) \\ \frac{\partial u_x}{\partial y} \sim O\left(\frac{U_0}{\delta}\right) \\ \frac{\partial u_y}{\partial x} \sim O\left(\frac{\delta U_0}{L^2}\right) \\ \frac{\partial u_y}{\partial y} \sim O\left(\frac{U_0}{L}\right) \end{cases}$$

$$\begin{cases} \frac{\partial^2 u_x}{\partial x^2} \sim O\left(\frac{U_0}{L^2}\right) \\ \frac{\partial^2 u_x}{\partial y^2} \sim O\left(\frac{U_0}{\delta^2}\right) \\ \frac{\partial^2 u_y}{\partial x^2} \sim O\left(\frac{\delta U_0}{L^3}\right) \\ \frac{\partial^2 u_y}{\partial y^2} \sim O\left(\frac{U_0}{\delta L}\right) \end{cases}$$

x 方向

$$u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = \nu \left(\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right)$$

量级表示

$$\frac{U_0^2}{L} \quad \frac{U_0^2}{L} \quad \frac{U_0}{L^2} \ll \frac{U_0}{\delta^2}$$

根据边界层内**惯性力与粘性力量级相当**，则有：

$$u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = \nu \frac{\partial^2 u_x}{\partial y^2}$$

y 方向

$$u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} = \nu \left(\frac{\partial^2 u_y}{\partial x^2} + \frac{\partial^2 u_y}{\partial y^2} \right)$$

量级表示

$$\frac{\delta U_0^2}{L^2} \quad \frac{\delta U_0^2}{L^2} \quad \frac{\delta U_0}{L^3} \ll \frac{U_0}{\delta L}$$

惯性力项

粘性力项

**x 方向
与
 y 方向
比较**

$$\frac{\frac{U_0^2}{L}}{\frac{\delta U_0^2}{L^2}} = \frac{L}{\delta} \gg 1$$

$$\frac{\frac{U_0}{\delta^2}}{\frac{U_0}{\delta L}} = \frac{L}{\delta} \gg 1$$

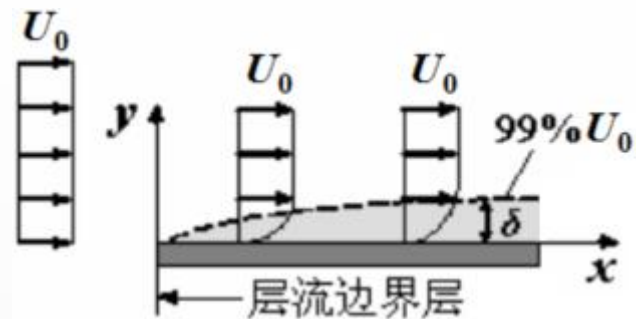
结论：忽略 y 方向的流动

量级比较简化可得
普朗特边界层方程：

$$\begin{cases} u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = \nu \frac{\partial^2 u_x}{\partial y^2} \\ \frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0 \end{cases}$$

边界条件：

$$\begin{cases} y = 0, u_x = u_y = 0 \\ y \rightarrow \infty, u_x = U_0 \end{cases}$$



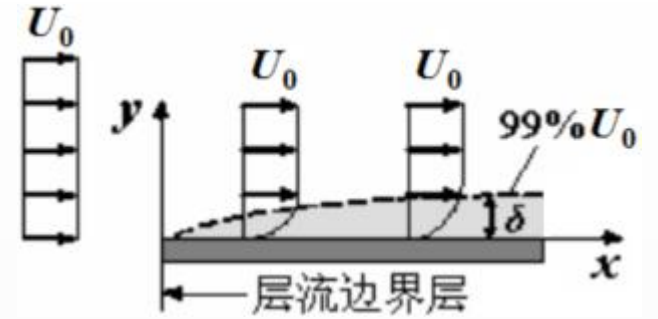
引入流函数的目的是将速度变量 u_x, u_y 用一个变量 ψ 代替, 从而使方程的求解得以简化。

$$\begin{cases} u_x = \frac{\partial \psi}{\partial y} \\ u_y = -\frac{\partial \psi}{\partial x} \end{cases} \quad \text{可得:} \quad \frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = \nu \frac{\partial^3 \psi}{\partial y^3}$$

$$\text{边界条件:} \quad \begin{cases} y = 0, \frac{\partial \psi}{\partial y} = \frac{\partial \psi}{\partial x} = 0 \\ y \rightarrow \infty, \frac{\partial \psi}{\partial y} = U_0 \end{cases}$$

相似变换

令速度分布为: $\frac{u_x}{U_0} = \varphi\left(\frac{y}{\delta}\right)$



根据边界层内**惯性力与粘性力量级相当**，则有：

$$\frac{U_0^2}{L} \sim \nu \frac{U_0}{\delta^2} \Rightarrow \delta \sim \sqrt{\frac{\nu L}{U_0}} \Rightarrow \delta \sim \sqrt{\frac{\nu x}{U_0}}$$

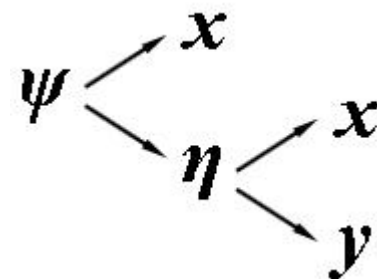
令 $\eta = \frac{y}{\delta}$ 则有: $\eta = y \sqrt{\frac{U_0}{\nu x}}$ $\frac{u_x}{U_0} = \varphi(\eta) = \varphi\left(y \sqrt{\frac{U_0}{\nu x}}\right)$

流函数
$$\begin{cases} u_x = \frac{\partial \psi}{\partial y} \\ u_y = -\frac{\partial \psi}{\partial x} \end{cases}$$

$$\psi = \int_0^y u_x dy = \int_0^y U_0 \varphi \left(y \sqrt{\frac{U_0}{\nu x}} \right) dy = \sqrt{\nu x U_0} \int_0^\eta \varphi(\eta) d\eta$$

令 $\int_0^\eta \varphi(\eta) d\eta = f(\eta)$

则有 $\psi = \sqrt{\nu x U_0} f(\eta)$

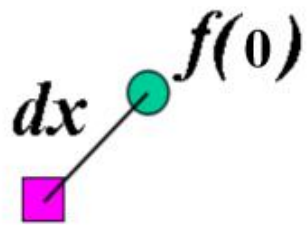


$$\left\{ \begin{aligned} \frac{\partial \psi}{\partial y} &= \frac{\partial \psi}{\partial \eta} \frac{\partial \eta}{\partial y} = U_0 f'(\eta) \\ \frac{\partial \psi}{\partial x} &= f(\eta) \frac{\partial \sqrt{\nu x U_0}}{\partial x} + \sqrt{\nu x U_0} f'(\eta) \frac{\partial \eta}{\partial x} = \frac{1}{2} \sqrt{\frac{\nu U_0}{x}} [f(\eta) - \eta f'(\eta)] \\ \frac{\partial^2 \psi}{\partial x \partial y} &= U_0 \frac{\partial f'(\eta)}{\partial \eta} \frac{\partial \eta}{\partial x} = U_0 f''(\eta) \left(-\frac{1}{2} \frac{\eta}{x} \right) = -\frac{1}{2} \frac{U_0}{x} \eta f''(\eta) \\ \frac{\partial^2 \psi}{\partial y^2} &= U_0 \frac{\partial f'(\eta)}{\partial \eta} \frac{\partial \eta}{\partial y} = U_0 f''(\eta) \sqrt{\frac{U_0}{\nu x}} = U_0 \sqrt{\frac{U_0}{\nu x}} f''(\eta) \\ \frac{\partial^3 \psi}{\partial y^3} &= U_0 \sqrt{\frac{U_0}{\nu x}} \frac{\partial f''(\eta)}{\partial \eta} \frac{\partial \eta}{\partial y} = U_0 \sqrt{\frac{U_0}{\nu x}} f'''(\eta) \sqrt{\frac{U_0}{\nu x}} = \frac{U_0^2}{\nu x} f'''(\eta) \end{aligned} \right.$$

代入整理得： $2f'''(\eta) + f(\eta)f''(\eta) = 0$

边界条件： $\begin{cases} \eta = 0, f(\eta) = f'(\eta) = 0 \\ \eta \rightarrow \infty, f'(\eta) = 1 \end{cases}$

麦克劳林级数


$$f(x) = f(0) + f'(0)x + \frac{f''(0)}{2!}x^2 + \dots$$

布拉修斯求解: $2f'''(\eta) + f(\eta)f''(\eta) = 0$

$$f(\eta) = f(0) + f'(0)\eta + \frac{f''(0)}{2!}\eta^2 + \dots + \frac{f^n(0)}{n!}\eta^n + \dots$$

$$\frac{u_x}{U_0} = f'(\eta) = f'(0) + f''(0)\eta + \frac{f'''(0)}{2!}\eta^2 + \dots + \frac{f^n(0)}{(n-1)!}\eta^{n-1} + \dots$$

在 $\eta=0$ 附近展开, 解得速度分布: $\eta = 0, f(\eta) = f'(\eta) = 0$

$$\frac{u_x}{U_0} = f'(\eta) = f''(0)\eta - \frac{f''(0)^2}{2} \frac{\eta^4}{4!} + \frac{11f''(0)^3}{4} \frac{\eta^7}{7!} - \frac{375f''(0)^4}{8} \frac{\eta^{10}}{10!} + \dots$$

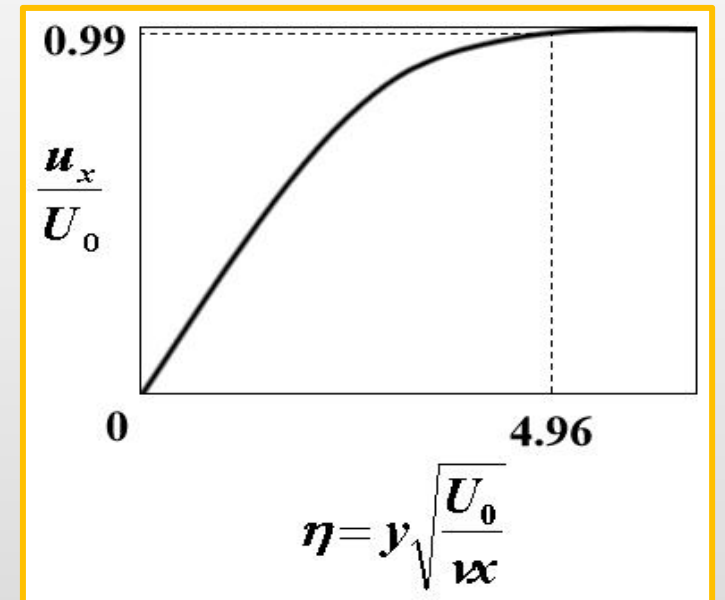
由 $\eta \rightarrow \infty, f'(\eta) = 1$ 推得 $f''(0) = 0.332$

$$\frac{u_x}{U_0} = 0.332\eta - 2.2963 \times 10^{-3} \eta^4 + 1.9967 \times 10^{-5} \eta^7 - 1.5694 \times 10^{-7} \eta^{10} + \dots$$

当 $\frac{u_x}{U_0} = 0.99$ 时, $\eta = 4.96$ 则有:

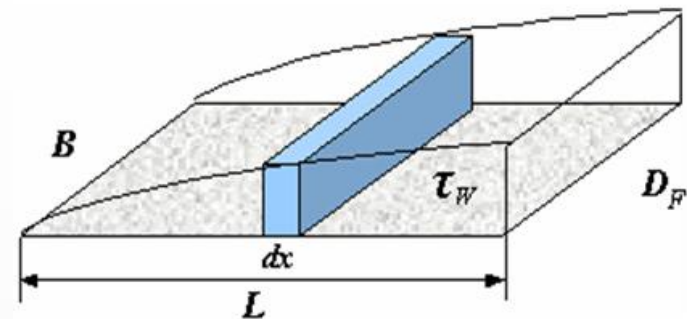
$$\eta = y \sqrt{\frac{U_0}{\nu x}} \Rightarrow y = \eta \sqrt{\frac{\nu x}{U_0}} \Rightarrow \delta = 4.96 \sqrt{\frac{\nu x}{U_0}}$$

$$\frac{\delta}{x} = \frac{4.96}{\sqrt{Re_x}}$$



壁面剪切应力

$$\tau_w = \mu \left. \frac{\partial u_x}{\partial y} \right|_{y=0} = \mu U_0 \sqrt{\frac{U_0}{\nu x}} f''(0) = 0.332 \mu U_0 \sqrt{\frac{U_0}{\nu x}}$$



则壁面摩擦阻力 D_F :

$$D_F = \int_0^L \tau_w B dx = 0.332 \mu U_0 \sqrt{\frac{U_0}{\nu}} \int_0^L \frac{1}{\sqrt{x}} dx = 0.664 B U_0 \sqrt{\mu \rho L U_0}$$

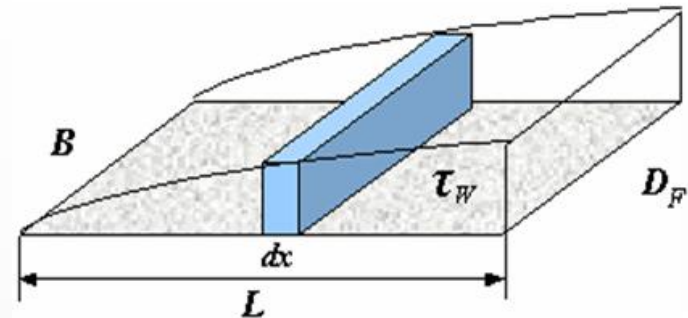
$$D_F = \frac{1.328}{\sqrt{Re_L}} B L \frac{1}{2} \rho U_0^2$$

定义摩擦阻力系数 C_D :

$$D_F = \frac{1.328}{\sqrt{Re_L}} BL \frac{1}{2} \rho U_0^2 = C_D A \frac{1}{2} \rho U_0^2$$

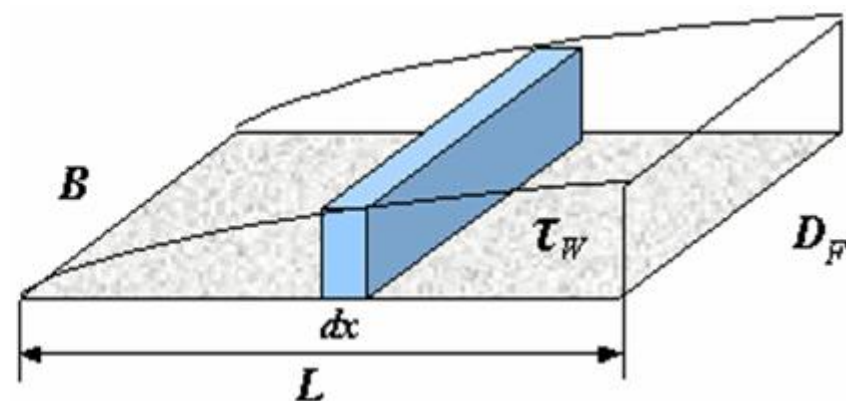
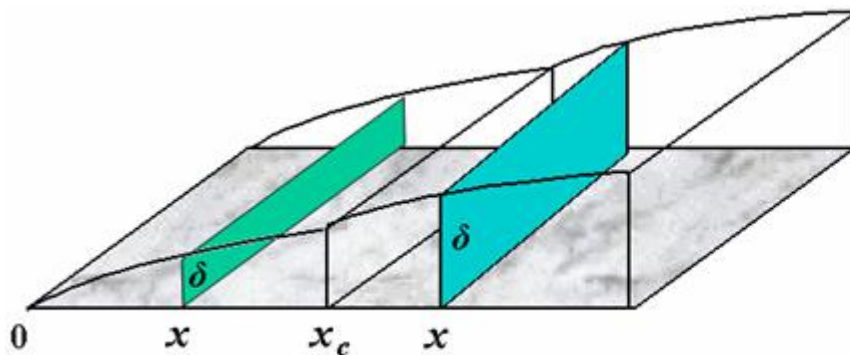
$$C_D = \frac{1.328}{\sqrt{Re_L}}$$

$$D_F = C_D A \frac{1}{2} \rho U_0^2$$



问题探讨 上述阻力公式只适用层流？

3. 边界层动量积分方程



临界雷诺数 $Re_{xc} = 5 \times 10^5$

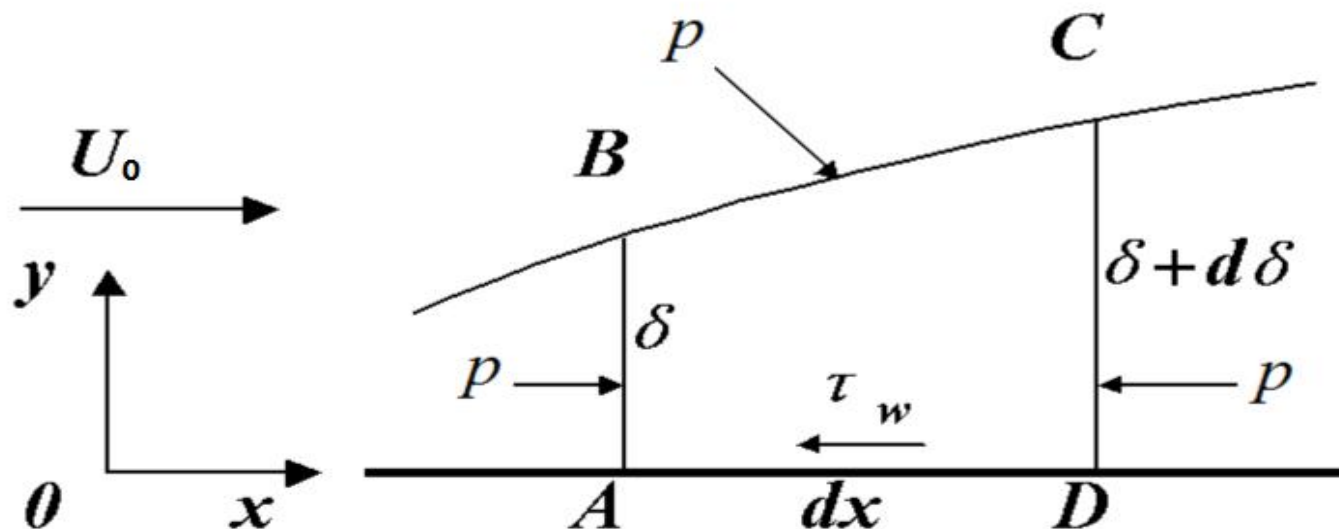
层流 $\frac{\delta}{x} = \frac{4.64}{\sqrt{Re_x}}$

$$D_F = \frac{1.292}{\sqrt{Re_L}} \cdot BL \cdot \frac{1}{2} \rho U_0^2$$

湍流 $\frac{\delta}{x} = \frac{0.376}{\sqrt[5]{Re_x}}$

$$D_F = \frac{0.072}{\sqrt[5]{Re_L}} \cdot BL \cdot \frac{1}{2} \rho U_0^2$$

冯·卡门 边界层动量积分方程



选取控制体 $ABCD$, 单位宽度

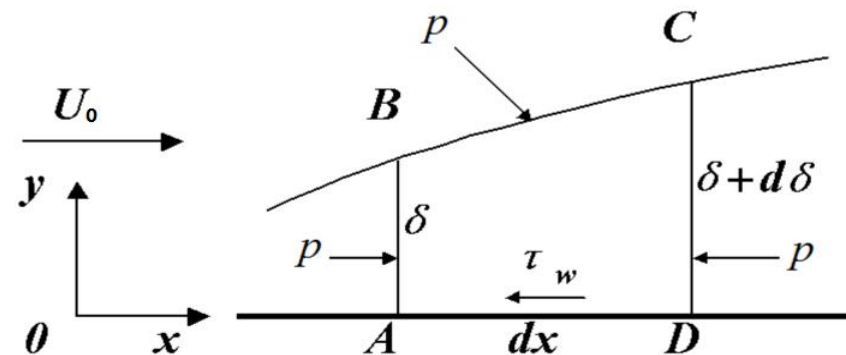
流动为无压差流动

$$\frac{\partial p}{\partial x} = 0$$

动量守恒:

$$\frac{\partial(m\bar{u}_x)}{\partial t} = (w\bar{u})_{1x} - (w\bar{u})_{2x} + \Sigma \vec{F}_x$$

对定常流体: $\Sigma \vec{F}_x = (w\bar{u})_{2x} - (w\bar{u})_{1x}$



合力: $\Sigma \vec{F}_x = p\delta + pd\delta - p(\delta + d\delta) - \tau_w dx$

$$\Sigma \vec{F}_x = -\tau_w dx$$

动量变化率:	进 AB 面	微元	AB 面
	质量流率	$\rho u_x dy$	$\int_0^\delta \rho u_x dy$
	动量速率	$\rho u_x^2 dy$	$\int_0^\delta \rho u_x^2 dy$

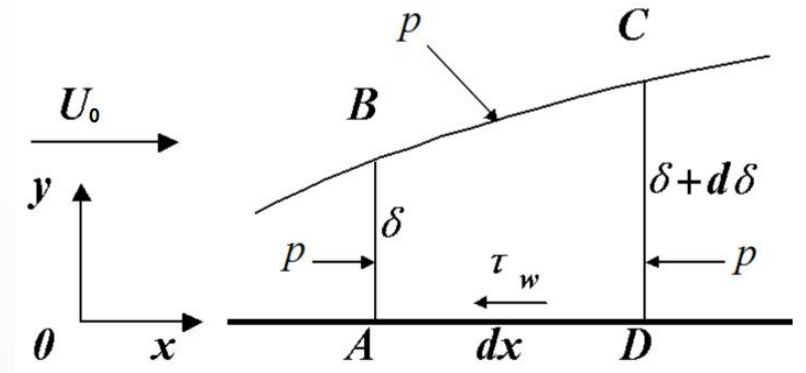
出 CD 面:

质量流率

$$\int_0^{\delta} \rho u_x dy + \frac{\partial}{\partial x} \left(\int_0^{\delta} \rho u_x dy \right) dx$$

动量速率

$$\int_0^{\delta} \rho u_x^2 dy + \frac{\partial}{\partial x} \left(\int_0^{\delta} \rho u_x^2 dy \right) dx$$



进 BC 面:

质量流率

$$\frac{\partial}{\partial x} \left(\int_0^{\delta} \rho u_x dy \right) dx$$

动量速率

$$U_0 \frac{\partial}{\partial x} \left(\int_0^{\delta} \rho u_x dy \right) dx$$

动量变化率:

$$(w\bar{u})_{2x} - (w\bar{u})_{1x} = \frac{\partial}{\partial x} \left(\int_0^\delta \rho u_x^2 dy \right) dx - U_0 \frac{\partial}{\partial x} \left(\int_0^\delta \rho u_x dy \right) dx$$

$$(w\bar{u})_{2x} - (w\bar{u})_{1x} = \rho dx \frac{\partial}{\partial x} \int_0^\delta (u_x - U_0) u_x dy$$

根据动量守恒:

$$\rho \frac{\partial}{\partial x} \int_0^\delta (U_0 - u_x) u_x dy = \tau_w$$

边界层动量积分方程

层流边界层，设速度分布：

$$\frac{u_x}{U_0} = a + b\left(\frac{y}{\delta}\right) + c\left(\frac{y}{\delta}\right)^2 + d\left(\frac{y}{\delta}\right)^3$$

边界条件：

$$\begin{cases} y = 0, & u_x = 0; & y = 0, & \frac{\partial^2 u_x}{\partial y^2} = 0 \\ y = \delta, & u_x = U_0; & y = \delta, & \frac{\partial u_x}{\partial y} = 0 \end{cases}$$

速度分布：

$$\frac{u_x}{U_0} = \frac{3}{2}\left(\frac{y}{\delta}\right) - \frac{1}{2}\left(\frac{y}{\delta}\right)^3$$

壁面切应力：

$$\tau_w = \mu \left. \frac{\partial u_x}{\partial y} \right|_{y=0} = \mu \frac{3}{2} \frac{U_0}{\delta}$$

将 u_x , τ_w 代入动量积分方程求得：

$$\delta = \frac{4.64x}{\sqrt{Re_x}} + C$$

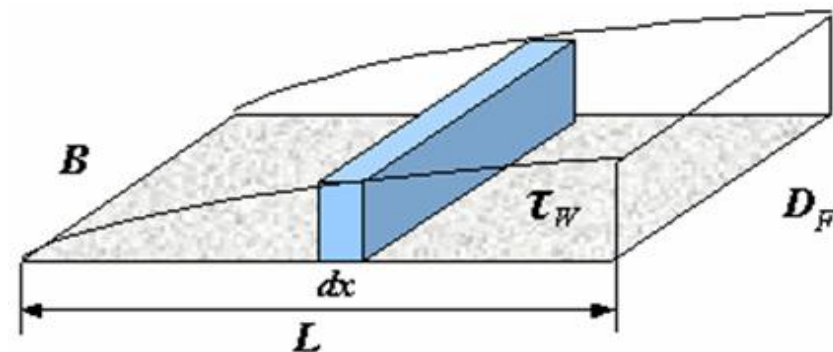
$\because x=0, \delta=0 ; \therefore C=0$ 。

$$\delta = \frac{4.64x}{\sqrt{Re_x}}$$

此式只适用 $x < x_c$

壁面切应力:

$$\tau_w = \mu \frac{3}{2} \frac{U_0}{\delta} = 0.323 \rho U^2 Re_x^{-\frac{1}{2}}$$



对 $B \times L$ 壁面总阻力:

$$D_F = \int_0^L \tau_w dx \cdot B = 0.646 B \sqrt{\mu \rho L U_0^3}$$

阻力系数:

$$C_D = \frac{\frac{D_F}{A}}{\frac{1}{2} \rho U_0^2}$$

层流:

$$C_D = \frac{1.292}{\sqrt{Re_L}}$$

湍流边界层，设速度分布：

$$\frac{u_x}{U_0} = \left(\frac{y}{\delta} \right)^{\frac{1}{7}}$$

代入边界层动量积分方程

$$\rho \frac{\partial}{\partial x} \int_0^{\delta} (U_0 - u_x) u_x dy = \tau_w$$

得：

$$\tau_w = \frac{7}{72} \rho U_0^2 \frac{d\delta}{dx}$$

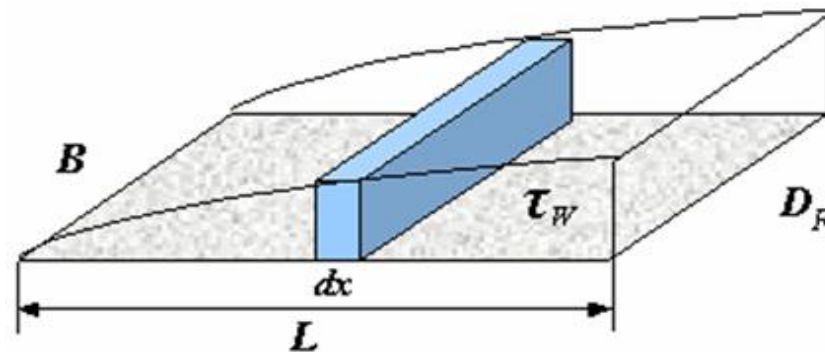
湍流的壁面剪切应力：

$$\tau_w = 0.023 \rho U_0^{\frac{7}{4}} \left(\frac{\nu}{\delta} \right)^{\frac{1}{4}}$$

问题探讨 上述湍流 τ_w 公式如何求得？

$$\frac{7}{72} \rho U_0^2 \frac{d\delta}{dx} = 0.023 \rho U_0^{\frac{7}{4}} \left(\frac{\nu}{\delta} \right)^{\frac{1}{4}}$$

$$\int_0^\delta \delta^{\frac{1}{4}} d\delta = \int_0^x 0.236 \left(\frac{\nu}{U_0} \right)^{\frac{1}{4}} dx$$



湍流

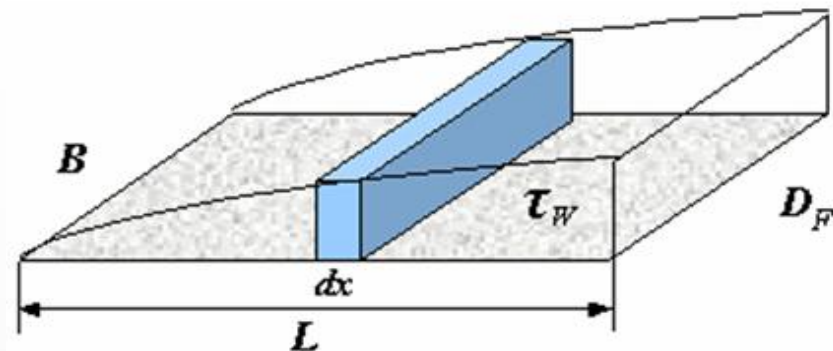
注意 $x = 0$, $\delta = 0$ 。说明假定一开始就是湍流。

$$\frac{\delta}{x} = \frac{0.376}{\sqrt[5]{Re_x}}$$

此式只适用 $x > x_c$

对 $B \times L$ 壁面总阻力:

假定一开始就是湍流

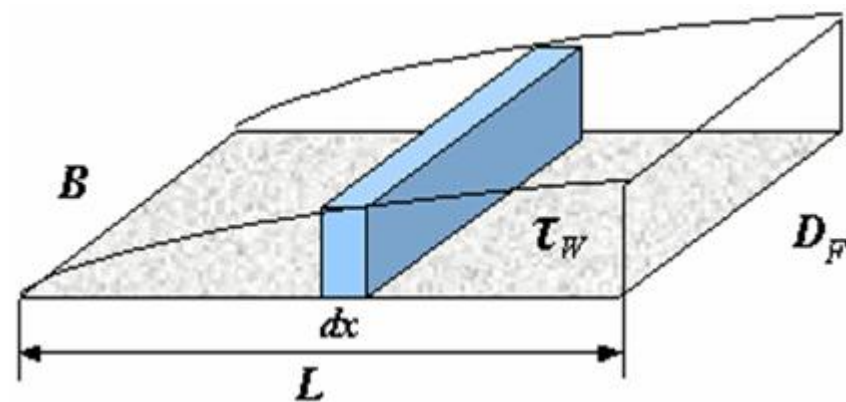
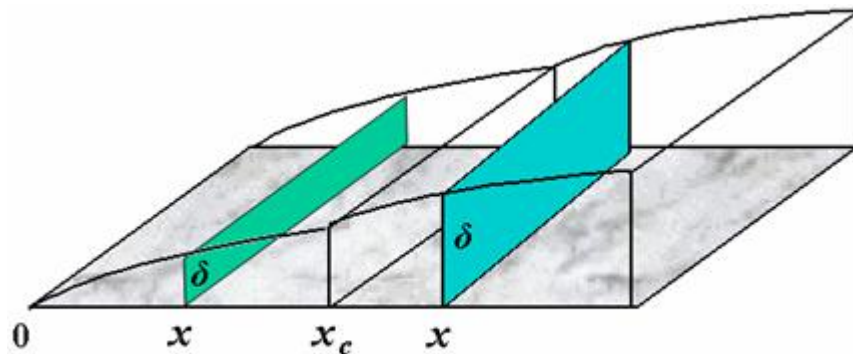


$$D_F = \int_0^L \tau_w dx \cdot B = \frac{0.073}{\sqrt[5]{Re_L}} BL \frac{1}{2} \rho U_0^2$$

湍流边界层阻力系数:

$$C_D = \frac{0.073}{\sqrt[5]{Re_L}}$$

4. 边界层计算



临界雷诺数 $Re_{xc}=5\times 10^5$

层流 $\frac{\delta}{x} = \frac{4.64}{\sqrt{Re_x}}$

$$D_F = \frac{1.292}{\sqrt{Re_L}} \cdot BL \cdot \frac{1}{2} \rho U_0^2$$

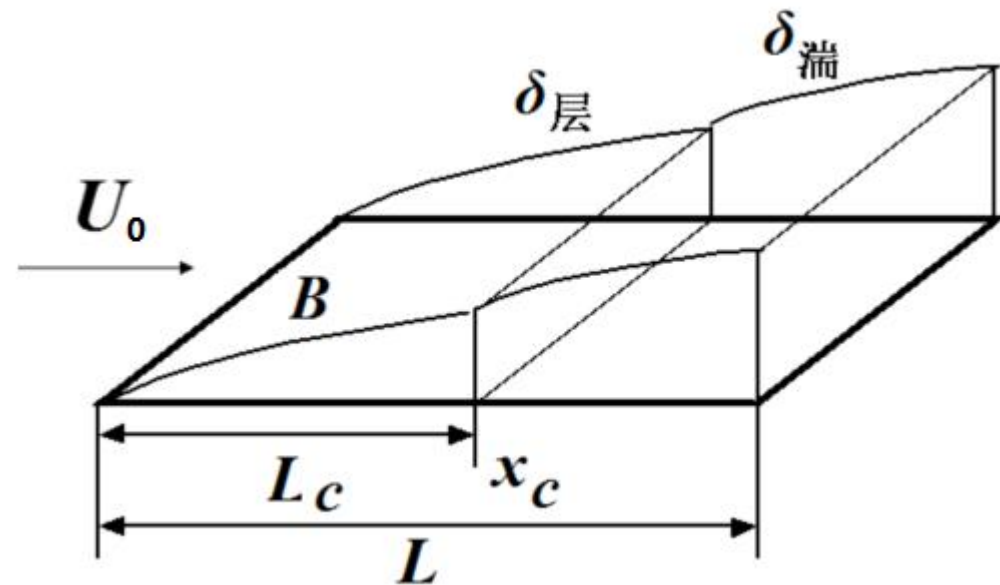
湍流 $\frac{\delta}{x} = \frac{0.376}{\sqrt[5]{Re_x}}$

$$D_F = \frac{0.072}{\sqrt[5]{Re_L}} \cdot BL \cdot \frac{1}{2} \rho U_0^2$$

当边界层达到湍流时

当 $L > x_c$ 时, 求 D_F

层流段 $L_c = x_c$: $D_{F_{\text{层}}, L_c}$



全部以湍流计算: $D_{F_{\text{湍}}, L}$

误算作湍流的 $D_{F_{\text{湍}}, L_c}$

$$\therefore D_F = D_{F_{\text{层}}, L_c} + D_{F_{\text{湍}}, L} - D_{F_{\text{湍}}, L_c}$$

平板阻力计算

已知：流体 $\nu=10^{-6}\text{m}^2/\text{s}$, $U_0=2.4\text{m/s}$, $Re_{xc}=5\times 10^5$, $\rho=1000\text{ kg/m}^3$ 。

求：①. x_c ②. $\delta_{x=3\text{ m}}$ ③. $D_F, L=2\text{ m}, B=1\text{ m}$

解：①. $Re_{xc} = \frac{x_c U_0}{\nu}$ $x_c = \frac{\nu Re_{xc}}{U_0} = 0.208\text{m}$

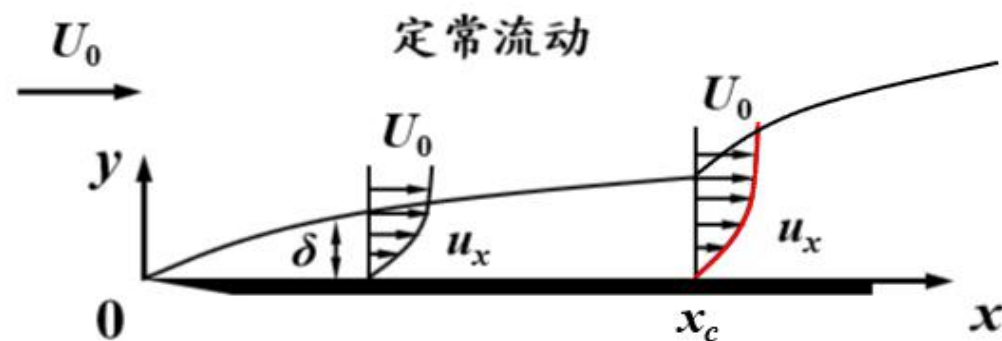
②. $x=3\text{m} > x_c$ 湍流 $\delta = \frac{0.376x}{\sqrt[5]{Re_x}} = 0.048\text{m}$

③. $L=2\text{m} > L_c$ 湍流 $\therefore D_F = D_{F\text{层}, Lc} + D_{F\text{湍}, L} - D_{F\text{湍}, Lc} = 17.08\text{N}$

课后思考

1.还记得静止流体中的平板启动吗？

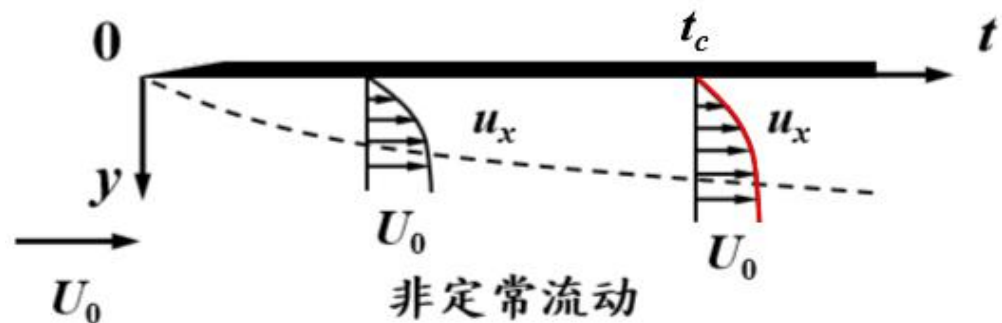
平板边界层流动



临界距离

空间场

静止流体中的平板启动



临界时间

时间场