Roxi

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Abstract

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- 1.2 Team Members

2 Overall Design

3 Mechanical Design

3.1 Structure Overview

This years drive base was guided by the following main goals 123

coofing performance

educed overall mass

oad accommodation

to hold a new laptop

l ride characteristics

3.2 Drive System

Layout & Suspension

Motors & Encoder Modification

3.3 Waterproofing

3.4 Electronics Accommodation

Laptop

Camera

LIDAR

Interface Panel

Power Electronics

4 Electronics Design

The electronics for Roxi can be broken into three major categories: Senors, Power, and computers.

4.1 Sensors

Roxi uses vision and LIDAR as its primary sensors used for the navigation challenge and also has GPS and wheel encoders to allow for waypoint navigation.

4.1.1 Vision

The vision system consists of a AVT Guppy F-036C camera connected via an IEEE 1394a link to the main computer. This camera is capable of 752 x 480 resolution at 64 fps. This camera is polled at approximately 10 Hz to send a new frame to the vision algorithm. The camera is placed at the top of the mast, facing forwards and down to allow the lines and obstacle in front of the robot to be sensed.

4.1.2 Wheel Encoder

Each wheel is connected to a quadrature wheel encoder, allowing wheel rate to be measured. This allows the velocity of the robot to be measured. The encoder is a US Digital E3-200-375-I-H-M-B, with 200 counts per revolution and an index channel. This allows for wheel rates up to XX speed to be sensed. The quadrature lines drive interrupts on a microcontroller, which then feeds the state of the lines to a state machine which increments or decrements a wheel counter. Wheel angular velocity is measured by differencing the number of counts over a 5 ms period, allowing a bandwidth of XX Hz and XX m / s. The microcontrollers are capable of sending both rate and count information to the laptop, allowing for speed control and odometry operations.

4.1.3 LIDAR

Two front and rear mounted Sick NAV300 LIDAR are used as object and ramp detectors. The LIDAR have a 270° FOV and a 10 meter range. The front facing LIDAR is used as an object finder, while the back facing LIDAR is used as a safety feature allowing the robot to sense if an object / person is moved behind it after the robot has moved through an area.

4.1.4 GPS

A GPS is used to provide world position to the robot, allowing obstacles to be placed in world space and allowing waypoints to be followed. A Garmin "GPS 18-5Hz" gps is mounted to the

mast to allow a clear view of the sky. This GPS is accurate to ¡15 m / ¡3 m (GPS / WAAS) and has a time to first fix of 45 seconds. The GPS updates 5 times per second.

4.2 Power

4.2.1 Main Power

Main power for the robot comes from two sealed lead acid gell-cell batteries. These batteries are connected in series to produce a nominal 24 VDC supply for the motors and other systems. This provides approximately $XX \ W \cdot hr$ of energy, XX hours of runtime of the motors, and XX hours of runtime of the electronics and motors.

The batteries are connected to a power distribution board, which allows the connection to each motor to be fused with XX Amps, allowing power to be cut in the event of a motor stall to prevent damage to the H-bridge. Power is also provided to several DC-DC boost converters, which output 5 VDC, 9 VDC, and 19.5 VDC for other electronics on the robot.

4.2.2 H Bridge

Each motor is connected to an OSMC H-bridge. This board is used to allow a low power signal from the microcontrollers to generate a high power PWM input to the motors. Each OSMC is capible of switching up to 50 VDC at 160A cont / 300A peak, allowing significant margin above our standard operating power of around 24 VDC / 20 A.

4.2.3 Component Power

Other systems are provided power through the use of DC-DC converters to produce voltages at 5 V, 9 V, and 19.5 V. This allows for the usb tethered microcontrollers, the sensors, and the main computer to be powered off of the main lead acid batteries. This greatly simplifies charging the robot, as only one battery system needs to be maintained.

4.3 Computers

4.3.1 Main Computer

Nearly all computation is performed on a single laptop containing a quadcore Intel Core i7 cpu, cuda enabled NVIDIA 285M gpu, and 6 GB of RAM. This computer is responsible for all vision, lidar, and GPS data processing and all path planning and control algorithms. It also forms the core of the sensor interconnects, providing the firewire and USB bus the camera, GPS, and microcon-

trollers use. This laptop replaces the main computer used in previous years, and was replaced with support from Northrup Grumman.

4.3.2 MCU

Microcontrollers are used on Roxi as data acquisition boards to collect data from the wheel encoders, and as motor control boards to generate PWM signals to drive the H-bridges. There are 6 ATmega328p based Arduino Duemilanove boards on the robot, 4 interfacing with the wheel encoders, and 2 to drive the motors.

5 Software Design

- 5.1 Architecture
- 5.1.1
- 5.1.2
- 5.2 Algorithms

6 Performance

References

- [1] *Space Network User's Guide*, pages 35–257, 317–327, 365–389. Number 450-SNUG. NASA GSFC, 9th edition, August 2007.
- [2] Advanced Scientific Concepts Inc. *DragonEye 3D Flash LIDAR Space Camera*. http://advancedscientificconcepts.com/products/dragoneye.html.
- [3] Allied Vision Technologies. Big Family.
- [4] Allied Vision Technologies. Prosilica GE4900 Datasheet, v2.0.1 en edition.
- [5] G. Richard Curry. Radar Systems Performance Modeling. Artech House Publishers, 2004.
- [6] Zhaoxu Dong. *Mechanical Behavior of Silica Nanoparticle Impregnated Kevlar Fabrics*. PhD thesis, Purdue University, 2008.
- [7] Du Pont. Technical Guide Kevlar Aramid Fiber.
- [8] European Space Agency. New Rdar Satellite Technique Sheds Light on Ocean Current Dynamics. http://www.esa.int/esaEO/SEMZRQEMKBF_economy_0.html.
- [9] European Space Agency. Sentinel-1. http://www.esa.int/esaLP/SEMBRS4KXMF_LPgmes_0.html.
- [10] Navid S. Fatemi, Howard E. Pollard, Hong Q. Hou, and Paul R. Sharps. Solar Array Trades Between Very High-Efficiency Multi-Junction and Si Space Solar Cells. pages 2–3. Emcore Photovoltaics, September 2003.
- [11] Christophe Geuzaine and Jean-Francois Remacle. Gmash: A three dimensional finite element mesh generator with built-in pre- and post- processing facilities. *International Journal for Numerical Methods in Engineering*, 79(11):1309–1331, 2009.
- [12] Honeywell. M50 Controll Moment Gyroscope, 7th edition, January 2006.
- [13] Jorgen Jensen and George Townsend, editors. *Orbital Flight Handbook, Part 1 Basic Techniques and Data*, volume 1 of *Space Flight Handbooks*, pages V–52. Martin Company Space Systems Division, 1967.
- [14] Bassem Mahafza. *Radar Systems Analysis and Design Using MATLAB*. Chapman and Hall, 2005.
- [15] Malin Space Science Systems. ECAM-C50.

- [16] Malin Space Science Systems. ECAM-DVR4.
- [17] Malin Space Science Systems. *ECAM-IR1*.
- [18] Malin Space Science Systems. ECAM Optics.
- [19] et al Mikhalaylovskiy, Yuriy. Impact. http://impact.sourceforge.net/.
- [20] PRC Laser Corp. FH Series High Power Lasers, 2005.
- [21] Ramsey Electronics. The 'LOGI' Log Periodic Antenna.
- [22] Mark A. Richards. Fundamentals of Radar Signal Processing. McGraw-Hill, 2005.
- [23] Wolfgang O. Schall. Orbital derbis removal by laser radiation. *Acta Astronuatica*, 24:343–351, 1991.
- [24] Merrill Skolnik. Introduction to Radar Systems. McGraw-Hill, 2002.
- [25] SpaceX. Falcon 1 Launch Vehicle Payload User's Guide, 7th edition, May 2008.
- [26] SpaceX. Falcon 9 Launch Vehicle Payload User's Guide, 1st edition, 2009.
- [27] John F. Stocky and Christopher M. Stevens. Guidelines for preparing project risk management plans. Technical report, NASA, 2005.
- [28] J.D. Weinberg, R. Craig, P. Earhart, I. Gravseth, and K.L. Miller. Flash lidar systems for hazard detection, surface navigation and autonomous rendezvous and docking. page 1. Ball Aerospace & Technologies Corp., 2007.
- [29] James R. Wertz and Wiley J. Larson, editors. *Space Mission Analysis and Design*, pages 301–497, 894–897. Microcosm Press and Springer, 3rd edition, 2008.