Progress Report for simulating the collision of UAVs with commercial airplanes

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We seek to solve dynamic equilibrium equations. The formulation from Liu and Quek assumes among other things:

- Linear Elastic Deformation grows proportionally with external force.
- 2 Isotropic Material property is direction independent.

We distinguish between four kinds of objects: beams, trusses, plates and shells.

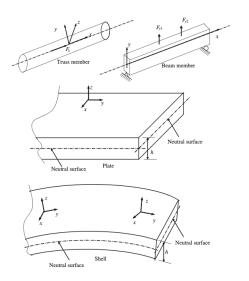


Figure: The four kinds of objects. Liu & Quek, 2013

Taking a queue from Liu and Quek, we start off by defining DEE for an idealized infinitesimal, linearly elastic, isotropic material.

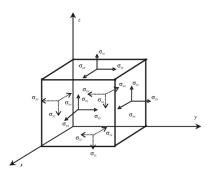


Figure: Forces on an idealized cube. Liu & Quek, 2013

Observe that because of equilibrium $\sigma_{xy} = \sigma_{yx}$; $\sigma_{xz} = \sigma_{zx}$; $\sigma_{yz} = \sigma_{yz}$ Which give sus the stress tensors: $\sigma^T = \{\sigma_{xx}, \sigma_{yy}, \sigma_{zz}, \sigma_{yz}, \sigma_{xz}, \sigma_{xy}\}$

We also get six strain components $\epsilon^T = \{\epsilon_{xx}, \epsilon_{yy}, \epsilon_{zz}, \gamma_{yz}, \gamma_{xz}, \gamma_{xy}\}$ Where:

$$\begin{split} \epsilon_{xx} &= \frac{\partial u}{\partial x}; \quad \epsilon_{yy} = \frac{\partial v}{\partial y}; \quad \epsilon_{zz} = \frac{\partial w}{\partial z}; \\ \gamma_{xy} &= 2\epsilon_{xy} = \frac{\partial u}{\partial_y} + \frac{\partial v}{\partial x}; \\ \gamma_{xy} &= 2\epsilon_{xz} = \frac{\partial u}{\partial_z} + \frac{\partial w}{\partial x}; \\ \gamma_{yz} &= 2\epsilon_{yz} = \frac{\partial v}{\partial_z} + \frac{\partial w}{\partial y}; \end{split}$$

Provided
$$\mathbf{U} = \begin{pmatrix} u \\ v \\ w \end{pmatrix}$$
 the displacement matrix we get the system $\epsilon = \mathbf{L}\mathbf{U}$

Where the differential operator is:
$$\mathbf{L} = \begin{pmatrix} \frac{\partial}{\partial x} & 0 & 0\\ 0 & \frac{\partial}{\partial y} & 0\\ 0 & 0 & \frac{\partial}{\partial z} & 0\\ 0 & \frac{\partial}{\partial z} & \frac{\partial}{\partial y} \\ \frac{\partial}{\partial z} & 0 & \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} & 0 \end{pmatrix}$$

For an external force in the y direction the equilibrium forces are:

$$(\sigma_{xx} + d\sigma_{xx})dydz - \sigma_{xx}dydz + (\sigma_{yx} + d\sigma_{yx})dxdz - \sigma_{yx}dxdz + (\sigma_{zx} + d\sigma_{zx})dxdy - \sigma_{zx}dxdy + f_xdxdydz = \rho \ddot{u}dxdydz$$
(1)

But since:

$$d\sigma_{xx} = \frac{\partial \sigma_{xx}}{\partial x} dx, d\sigma_{yx} = \frac{\partial \sigma_{yx}}{\partial y} dy, d\sigma_{zx} = \frac{\partial \sigma_{zx}}{\partial z} dz$$
 (2)

Substituting (2) into (1) gives us:

$$\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{yx}}{\partial y} + \frac{\partial \sigma_{zx}}{\partial z} + f_{x} = \rho \ddot{u}$$
 (3)

Using analogous equations for different directions, y and z, we get:

$$\frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_{yy}}{\partial y} + \frac{\partial \sigma_{zy}}{\partial z} + f_y = \rho \ddot{v}$$
$$\frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yz}}{\partial y} + \frac{\partial \sigma_{zz}}{\partial z} + f_z = \rho \ddot{w}$$

Yielding the linear system:

$$L^{T}\sigma + f_{b} = \rho \ddot{U} \tag{4}$$

And since Hooke's Law states $\sigma = c\epsilon$ we can further simplify to:

$$L^T c L U + f_b = \rho \ddot{U}$$

General structure of a deal.ii step.

- include deal.ii headers
- ② Create class StepN with public methods StepN(), run() and private methods make_grid(),setup_system(),assemble_system(),solve(), output_results(), and attributes often Triangulation<dim>, FE_Q <dim>, DoFHandler<dim>
- Oefine the methods outside, sometimes all encaspulated in a template for dimension independent programming.

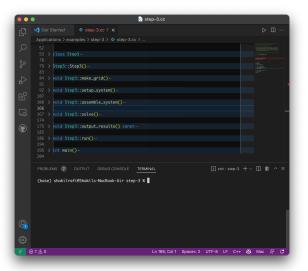


Figure: VSCode of Step-3 with comments stripped, code folding, minor refactoring $_{11/22}$

How to solve it



Figure: Grid refinement on a 2-simplex

Results of solving Laplace Equation on Different Domains:

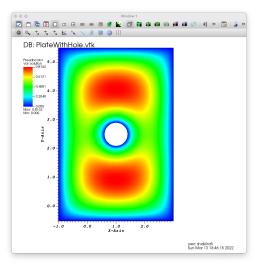


Figure: Plate with one hole

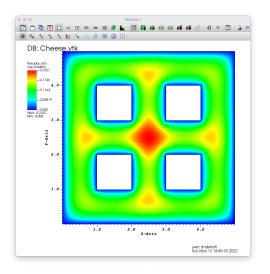
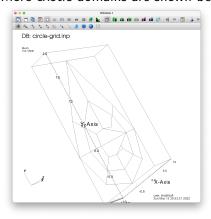
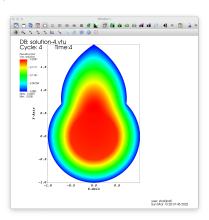


Figure: Cheese domain

More exotic domains are shown below:





How to solve it

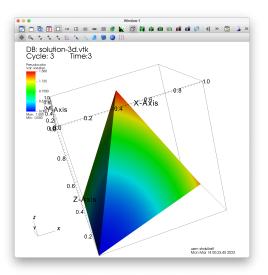


Figure: The Laplace problem solved on a tetrahedral domain

How to solve it

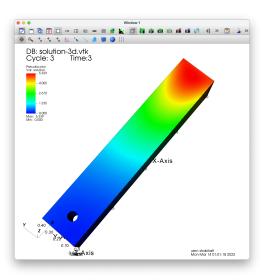


Figure: Same but on a channel with holes

Step-8

Step-8 presents the elastic equations, an extension of Laplace equations with vector-valued solutions:

$$-\operatorname{div}(C\nabla u) = f \tag{5}$$

Step-8

The x-displacement on the teardrop domain is:

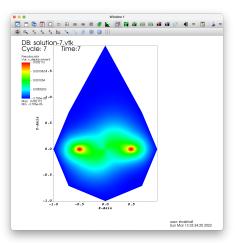


Figure: x-displacement on the solution to the elastic equations on the teardrop domain

Future Timeline

Our next step is to master step-44. Step-44 deals exclusively with a certain variation of the dynamic structure equations for a plate.

Mesh data for airplanes from UIUC Airfoil data is not in the format that deal.ii can understand. In particular converting from the Lednicer format in a text file proves challenging and prohibitively time consuming.

Future Timeline

We hope to master Step-44 by the end of spring break by which time we will have usable mesh data. Our aim is to solve the dynamic equilibrium equations on the domains provided by the mesh data.

Bibliography

- Liu G. R. & S. S. Quek, *The Finite Element Method: A Practical Course*, 2nd Edition. Butterworth-Heinemann, 2013.
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