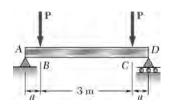
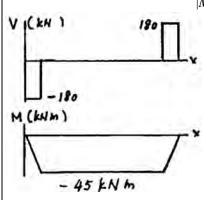
CHAPTER 8



A W250×101 rolled-steel beam supports a load **P** as shown. Knowing that P=180 kN, a=0.25 m, and $\sigma_{\rm all}=126$ MPa, determine (a) the maximum value of the normal stress σ_m in the beam, (b) the maximum value of the principal stress $\sigma_{\rm max}$ at the junction of the flange and web, (c) whether the specified shape is acceptable as far as these two stresses are concerned.

SOLUTION



$$|V|_{\text{max}} = 180 \text{ kN}$$

 $|M|_{\text{max}} = (180)(0.25) = 45 \text{ kNm}$

For W250×101 rolled steel section,

$$\begin{aligned} d &= 264 \text{ mm} & b_f &= 257 \text{ mm} & t_f &= 19.6 \text{ mm} \\ t_w &= 11.9 \text{ mm} & I_x &= 164 \times 10^6 \text{ mm}^4 & S_x &= 1240 \times 10^3 \text{ mm}^3 \\ c &= \frac{1}{2} d = 132 \text{ mm} & y_b &= c - t_f &= 112.4 \text{ mm} \end{aligned}$$

(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{45 \times 10^3}{1240 \times 10^{-6}}$$
 $\sigma_m = 36.3 \text{ MPa}$

$$\sigma_b = \frac{y_b}{c} \sigma_m = \left(\frac{112.4}{132}\right) (36.3) = 30.9 \text{ MPa}$$

$$A_f = b_f t_f = 5037 \text{ mm}^2$$

$$\overline{y}_f = \frac{1}{2} (c + y_b) = 122.2 \text{ mm}$$

$$Q_b = A_f \overline{y}_f = 615521 \text{ mm}^3$$

$$\tau_{xy} = \frac{|V|_{\text{max}} Q_b}{I_x t_w} = \frac{(180 \times 10^3)(615521 \times 10^{-9})}{(164 \times 10^{-6})(0.0119)} = 56.77 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_{xy}^2} = 58.83 \text{ MPa}$$

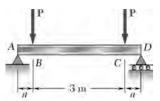
(b)
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 74.29 \text{ MPa}$$

$$\sigma_{\text{max}} = 74.3 \text{ MPa} \blacktriangleleft$$

(c) Since
$$\sigma_{\text{max}} < \sigma_{\text{all}} (= 126 \text{ MPa})$$
,

W250 × 101 is acceptable.
$$\blacktriangleleft$$

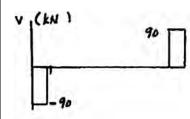




Solve Prob. 8.1, assuming that P = 90 kN, a = 0.5 m.

PROBLEM 8.1 A W250×101 rolled-steel beam supports a load **P** as shown. Knowing that P=180 kN, a=0.25 m, and $\sigma_{\rm all}=126$ MPa, determine (a) the maximum value of the normal stress σ_m in the beam, (b) the maximum value of the principal stress $\sigma_{\rm max}$ at the junction of the flange and web, (c) whether the specified shape is acceptable as far as these two stresses are concerned.

SOLUTION



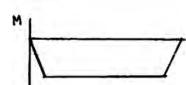
$$|V|_{\text{max}} = 90 \text{ kN}$$

$$|M|_{\text{max}} = (90)(0.5) = 45 \text{ kNm}$$

For $W250 \times 101$ rolled steel section,

$$d = 264 \text{ mm}$$
 $b_f = 257 \text{ mm}$ $t_f = 19.6 \text{ mm}$
 $t_w = 11.9 \text{ mm}$ $I_x = 164 \times 10^6 \text{ mm}^4$ $S_x = 1240 \times 10^3 \text{ mm}^3$

$$c = \frac{1}{2}d = 132 \text{ mm}$$
 $y_b = c - t_f = 112.4 \text{ mm}$



(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{45 \times 10^3}{1240 \times 10^{-6}}$$
 $\sigma_m = 36.3 \text{ MPa}$

$$\sigma_b = \frac{y_b}{c} \sigma_m = \left(\frac{112.4}{132}\right) (36.3) = 30.9 \text{ MPa.}$$

$$A_f = b_f t_f = 5037 \text{ mm}^2$$

$$\overline{y}_f = \frac{1}{2}(c + y_b) = 122.2 \text{ mm}$$

$$Q_b = A_f \overline{y}_f = 615521 \text{ mm}^3$$

$$\tau_{xy} = \frac{|V|_{\text{max}} Q_b}{I_x t_w} = \frac{(45 \times 10^3)(615521 \times 10^{-9})}{(164 \times 10^{-6})(0.119)} = 4.12 \text{ MPa}$$

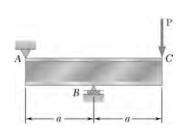
$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_{xy}^2} = 15.99 \text{ MPa}$$

(b)
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 31.44 \text{ MPa}$$

$$\sigma_{\rm max} = 31.4 \; {\rm MPa} \; \blacktriangleleft$$

(c) Since
$$\sigma_{\text{max}} < \sigma_{\text{all}} (= 126 \text{ MPa}),$$

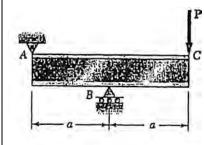
W250 × 101 is <u>acceptable</u>. \blacktriangleleft



(a)

An overhanging W920×449 rolled-steel beam supports a load **P** as shown. Knowing that P = 700 kN, a = 2.5 m, and $\sigma_{all} = 100$ MPa, determine (a) the maximum value of the normal stress σ_m in the beam, (b) the maximum value of the principal stress $\sigma_{ ext{max}}$ at the junction of the flange and web, (c) whether the specified shape is acceptable as far as these two stresses are concerned.

SOLUTION



$$|V|_{\text{max}} = 700 \text{ kN} = 700 \times 10^3 \text{ N}$$

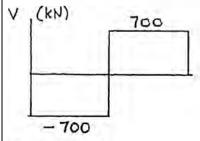
 $|M|_{\text{max}} = (700 \times 10^3)(2.5) = 1.75 \times 10^6 \text{ N} \cdot \text{m}$

For W920×449 rolled steel beam,

$$d = 947 \text{ mm}, \qquad b_f = 424 \text{ mm}, \qquad t_f = 42.7 \text{ mm},$$

$$t_w = 24.0 \text{ mm}, \qquad I_x = 8780 \times 10^6 \text{ mm}^4, \quad S_x = 18,500 \times 10^3 \text{ mm}^3$$

$$c = \frac{1}{2}d = 473.5 \text{ mm}, \qquad y_b = c - t_f = 430.8 \text{ mm}$$



$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{1.75 \times 10^6}{18500 \times 10^{-6}}$$

$$\sigma_m = 94.595 \text{ MPa}$$
 $\sigma_m = 94.6 \text{ MPa}$

$$A_f = b_f t_t = 18.1048 \times 10^3 \text{ mm}^2$$

 $\overline{y}_f = \frac{1}{2} (c + y_b) = 452.15 \text{ mm}$

 $\sigma_b = \frac{y_b}{2} \sigma_m = \frac{430.8}{473.5} (94.595) = 86.064 \text{ MPa}$

$$\overline{y}_f = \frac{1}{2}(c + y_b) = 452.15 \text{ mm}$$

$$Q_b = A_f \overline{y}_f = 8186.1 \times 10^3 \text{ mm}^3 = 8186.1 \times 10^{-6} \text{ m}^3$$

$$\tau_{xy} = \frac{|V|_{\text{max}} Q_b}{I_x t_w} = \frac{(700 \times 10^3)(8186.1 \times 10^{-6})}{(8780 \times 10^{-6})(24.0 \times 10^{-3})} = 27.194 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_{xy}^2} = \sqrt{\left(\frac{86.064}{2}\right)^2 + 27.194^2} = 50.904 \text{ MPa}$$

$$(b) \qquad \sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 93.9 \text{ MPa}$$

$$\sigma_{\text{max}} = 93.9 \text{ MPa} \blacktriangleleft$$

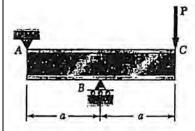
Since 94.6 MPa $< \sigma_{all} (= 100 \text{ MPa}),$ (c)

W920×449 is acceptable. ◀

Solve Prob. 8.3, assuming that P = 850 kN and a = 2.0 m.

PROBLEM 8.3 An overhanging W920×449 rolled-steel beam supports a load **P** as shown. Knowing that P = 700 kN, a = 2.5 m, and $\sigma_{all} = 100$ MPa, determine (a) the maximum value of the normal stress σ_m in the beam, (b) the maximum value of the principal stress σ_{max} at the junction of the flange and web, (c) whether the specified shape is acceptable as far as these two stresses are concerned.

SOLUTION



$$|V|_{\text{max}} = 850 \text{ kN} = 850 \times 10^3 \text{ N}$$

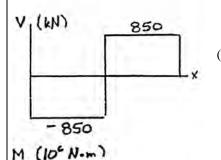
 $|M|_{\text{max}} = (850 \times 10^3)(2.0) = 1.70 \times 10^6 \text{ N} \cdot \text{m}$

For W920×449 rolled steel section,

$$d = 947 \text{ mm}, \qquad b_f = 424 \text{ mm}, \qquad t_f = 42.7 \text{ mm},$$

$$t_w = 24.0 \text{ mm}, \qquad I_x = 8780 \times 10^6 \text{ mm}^4, \quad S_x = 18500 \times 10^3 \text{ mm}^3$$

$$c = \frac{1}{2}d = 473.5 \text{ mm} \qquad y_b = c - t_f = 430.8 \text{ mm}$$



(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{1.70 \times 10^6}{18,500 \times 10^{-6}}$$
 $\sigma_m = 91.892 \text{ MPa}$ $\sigma_m = 91.9 \text{ MPa}$

$$A_f = b_f t_f = 18.1048 \times 10^3 \text{ mm}^2$$

 $\overline{v}_f = \frac{1}{2} (c_f + v_f) = 452.15 \text{ mm}$

$$\overline{y}_f = \frac{1}{2}(c + y_b) = 452.15 \text{ mm}$$

$$Q_b = A_f \overline{y}_f = 8186.1 \times 10^3 \,\text{mm}^3 = 8186.1 \times 10^{-6} \,\text{m}^3$$

 $\sigma_b = \frac{y_b}{c} \sigma_m = \frac{430.8}{473.5} (91.892) = 83.605 \text{ MPa}$

$$\tau_{xy} = \frac{|V|_{\text{max}} Q_b}{I_x t_w} = \frac{(850 \times 10^3)(8186.1 \times 10^{-6})}{(8780 \times 10^{-6})(24.0 \times 10^{-3})} = 33.021 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_{xy}^2} = \sqrt{\left(\frac{83.605}{2}\right)^2 + 33.021^2} = 53.271 \text{ MPa}$$

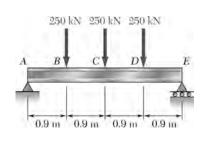
(b)
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 95.1 \text{ MPa}$$

$$\sigma_{\rm max} = 95.1 \, \mathrm{MPa} \, \blacktriangleleft$$

(c) Since 95.1 MPa $< \sigma_{all} (= 100 \text{ MPa}),$

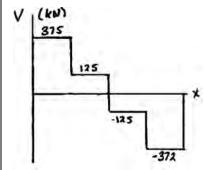
-1.70

W920×449 is acceptable. ◀



(a) Knowing that $\sigma_{\rm all} = 160$ MPa and $\tau_{\rm all} = 100$ MPa, select the most economical metric wide-flange shape that should be used to support the loading shown. (b) Determine the values to be expected for σ_m , τ_m , and the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web of the selected beam.

SOLUTION



Reactions:
$$R_A = 375 \text{ kN}$$
, $R_E = 375 \text{ kN}$

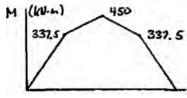
$$|V|_{\text{max}} = 375 \text{ kN}$$

$$|M|_{\text{max}} = 450 \text{ kN} \cdot \text{m}$$

|V| at point C: 125 kN

$$S_{\text{min}} = \frac{M_{\text{max}}}{\sigma_{\text{all}}} = \frac{450 \times 10^3}{160 \times 10^6} = 2.8125 \times 10^{-3} \,\text{m}^3$$

= 2812.5×10³ mm³



	Shape	$S_x(10^3\mathrm{mm}^3)$
	W840×176	5890
	W760×147	4410
\rightarrow	W690×125	3510
	W610×155	4220
	W530×150	3720
	W460×158	3340
	W360×216	3800

(a) Use W690×125. ◀

d = 678 mm $t_f = 16.30 \text{ mm}$ $t_w = 11.7 \text{ mm}$

(b)
$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{450 \times 10^3}{3510 \times 10^{-6}} = 128.2 \times 10^6 \text{ Pa}$$

$$\sigma_m = 128.2 \text{ MPa} \blacktriangleleft$$

$$\tau_m = \frac{|V|_{\text{max}}}{A_w} = \frac{|V|_{\text{max}}}{dt_w} = \frac{375 \times 10^3}{(678 \times 10^{-3})(11.7 \times 10^{-3})}$$

$$\tau_m = 47.3 \text{ MPa} \blacktriangleleft$$

PROBLEM 8.5 (Continued)

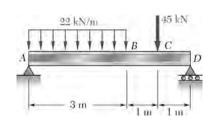
At point C,
$$\tau_{w} = \frac{V}{A_{w}} = \frac{125 \times 10^{3}}{(678 \times 10^{-3})(11.7 \times 10^{-3})} = 15.76 \times 10^{6} \,\text{Pa} = 15.76 \,\text{MPa}$$

$$c = \frac{1}{2}d = \frac{678}{2} = 339 \,\text{mm} \qquad y_{b} = c - t_{f} = 339 - 16.30 = 322.7 \,\text{mm}$$

$$\sigma_{b} = \frac{y_{b}}{c}\sigma_{m} = \left(\frac{322.7}{339}\right)(128.2) = 122.0 \,\text{MPa}$$

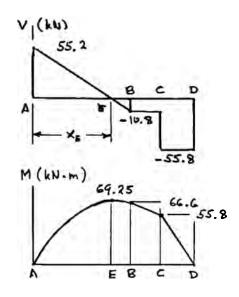
$$R = \sqrt{\left(\frac{\sigma_{b}}{2}\right)^{2} + \tau_{w}^{2}} = \sqrt{(61.0)^{2} + (15.76)^{2}} = 63.0 \,\text{MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_{b}}{2} + R = 61.0 + 63.0 \qquad \sigma_{\text{max}} = 124.0 \,\text{MPa}$$



(a) Knowing that $\sigma_{\rm all} = 160$ MPa and $\tau_{\rm all} = 100$ MPa, select the most economical metric wide-flange shape that should be used to support the loading shown. (b) Determine the values to be expected for σ_m , τ_m , and the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web of the selected beam.

SOLUTION



+)
$$\Sigma M_D = 0$$
:
-5 $R_A + (3.5)(3)(22) + (1)(45) = 0$
 $R_A = 55.2 \text{ kN}$ | $R_D = 55.8 \text{ kN}$ |

Draw shear and bending moment diagrams.

Locate point where V = 0 and M_E .

$$55.2 - 22x_E = 0 x_E = 2.5091 \text{ m.}$$

$$M_E = \frac{1}{2}(55.2)(2.5091) = 69.25 \text{ kN} \cdot \text{m}$$

$$S_{\text{min}} = \frac{|M_{\text{max}}|}{\sigma_{\text{all}}} = \frac{69.25 \times 10^3}{160 \times 10^6} = 432.8 \times 10^{-6} \text{ m}^3$$

$$= 432.8 \times 10^3 \text{ mm}^3$$

	Shape	$S(10^3 \text{mm}^3)$
	W410×38.8	637
\rightarrow	W360×32.9	474
	W310×38.7	549
	W 250×44.8	535
	W 200×46.1	448
	1	

(a) Use W 360×32.9

For
$$W360 \times 32.9$$
 $S = 474 \times 10^3 \text{ mm}^3 = 474 \times 10^{-6} \text{ m}^3$

$$A_{\text{web}} = dt_w = (349)(5.8) = 2024.2 \text{ mm}^2 = 2024.2 \times 10^{-6} \text{ m}^2$$

At point E:
$$\sigma_m = \frac{M_E}{S} = \frac{69.25 \times 10^3}{474 \times 10^{-6}}$$

 $\sigma_m = 146.1 \text{ MPa} \blacktriangleleft$

PROBLEM 8.6 (Continued)

At point C:
$$\sigma_m = \frac{M_c}{S} = \frac{55.8 \times 10^3}{474 \times 10^{-6}}$$
 $\sigma_m = 117.7 \text{ MPa}$

$$\tau_m = \frac{|V|}{A_{\text{web}}} = \frac{55.8 \times 10^3}{2024.2 \times 10^{-6}}$$
 $\tau_m = 27.6 \text{ MPa} \blacktriangleleft$

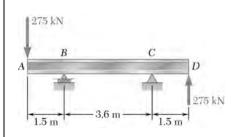
$$c = \frac{1}{2}d = \frac{1}{2}(349) = 174.5 \text{ mm} \quad y_b = c - t_f = 166 \text{ mm}$$

$$\sigma_b = \frac{y_b}{c}\sigma_m = \left(\frac{166}{174.5}\right)(117.7) = 112.0 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_m^2} = 62.4 \text{ MPa}$$

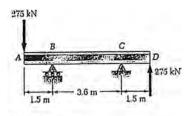
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R$$

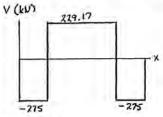
$$\sigma_{\text{max}} = 118.4 \text{ MPa} \blacktriangleleft$$

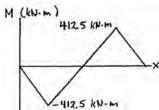


(a) Knowing that $\sigma_{\rm all} = 160$ MPa and $\tau_{\rm all} = 100$ MPa, select the most economical metric wide-flange shape that should be used to support the loading shown. (b) Determine the values to be expected for σ_m , τ_m , and the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web of the selected beam.

SOLUTION







$R_B = 504.17 \text{ kN} \uparrow \qquad R_C = 504.17 \downarrow$
$ V _{\text{max}} = 275 \text{ kN} M _{\text{max}} = 412.5 \text{ kN} \cdot \text{m}$
$S_{\text{min}} = \frac{ M _{\text{max}}}{\sigma_{\text{all}}} = \frac{412.5 \times 10^2}{160 \times 10^6} = 2578 \times 10^{-6} \text{ m}^3$
$= 2578 \times 10^3 \text{ mm}$

	Shape	$S_x(10^3 \mathrm{mm}^3)$
	W760×147	4410
\rightarrow	W690×125	3490
	W530×150	3720
	W460×158	3340
	W360×216	3800

(a) Use W690 \times 125.

$$d = 678 \text{ mm}$$
 $t_f = 16.3 \text{ mm}$ $t_w = 11.7 \text{ mm}$

(b)
$$\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{412.5 \times 10^3}{3490 \times 10^{-6}} = 118.195 \times 10^6 \text{ Pa}$$
 $\sigma_m = 118.2 \text{ MPa} \blacktriangleleft$

$$\tau_m = \frac{|V|_{\text{max}}}{A_{\text{or}}} = \frac{|V|_{\text{max}}}{dt_{\text{or}}} = \frac{275 \times 10^3}{(678 \times 10^{-3})(11.7 \times 10^{-3})} = 34.667 \times 10^6 \text{ Pa}$$

$$\tau_m = 34.7 \text{ MPa} \blacktriangleleft$$

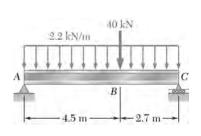
$$c = \frac{1}{2}d = \frac{67.8}{2} = 339 \text{ mm}, \quad t_f = 16.3 \text{ mm}, \quad y_b = c - t_f = 339 - 16.3 = 322.7 \text{ mm}$$

$$\sigma_b = \frac{y_b}{c}\sigma_m = \left(\frac{322.7}{339}\right)(118.195) = 112.512 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_m^2} = \sqrt{(56.256)^2 + (34.667)^2} = 66.080 \text{ MPa}$$

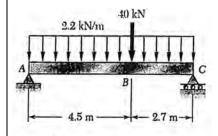
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 56.256 + 66.080$$

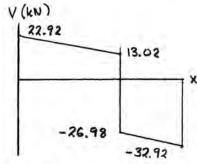
 $\sigma_{\rm max} = 122.3 \, \mathrm{MPa} \, \blacktriangleleft$

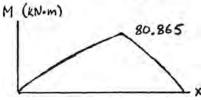


(a) Knowing that $\sigma_{\rm all} = 160$ MPa and $\tau_{\rm all} = 100$ MPa, select the most economical metric wide-flange shape that should be used to support the loading shown. (b) Determine the values to be expected for σ_m , τ_m , and the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web of the selected beam.

SOLUTION







$$+\sum M_C = 0$$
:
- 7.2 R_A + (2.2)(7.2)(3.6) + (40)(2.7) = 0

$$R_A = 22.92 \text{ kN}$$

$$V_A = R_A = 22.92 \text{ kN}$$

$$V_B^- = 22.92 - (2.2)(4.5) = 13.02 \text{ kN}$$

$$V_R^+ = 13.02 - 40 = -26.98 \text{ kN}$$

$$V_C = -26.98 - (2.2)(2.7) = -32.92 \text{ kN}$$

$$M_A = 0$$

$$M_B = 0 + \frac{1}{2}(22.92 + 13.02)(4.5) = 80.865 \text{ kN} \cdot \text{m}$$

$$M_C = 0$$

$$S_{\text{min}} = \frac{|M|_{\text{max}}}{\sigma_{\text{all}}} = \frac{80.865 \times 10^3}{165 \times 10^6} = 490 \times 10^{-6} \text{ m}^3$$

$$= 490 \times 10^3 \text{ mm}^3$$

Shape	$S(10^3 \text{ mm}^3)$
W360 × 39	578
$W310 \times 38.7$	547 ←
$W250 \times 44.8$	531
$W200 \times 52$	511

(a) Use W310
$$\times$$
 38.7.

$$d = 310 \text{ mm}$$
 $t_f = 9.65 \text{ mm}$

$$t_w = 5.84 \text{ mm}$$

PROBLEM 8.8 (Continued)

(b)
$$\sigma_m = \frac{M_B}{S} = \frac{80.865 \times 10^3}{547 \times 10^{-6}} = 147.834 \times 10^6 \text{ Pa}$$

 $\sigma_m = 147.8 \text{ MPa}$

$$\tau_m = \frac{|V|_{\text{max}}}{dt_w} = \frac{32.92 \times 10^3}{(310 \times 10^{-3})(5.84 \times 10^{-3})} = 18.1838 \times 10^6 \text{ Pa}$$

 $\tau_m = 18.18 \, \text{MPa} \, \blacktriangleleft$

$$c = \frac{1}{2}d = 155 \text{ mm}$$
 $y_b = c - t_f = 155 - 9.65 = 145.35 \text{ mm}$

$$\sigma_b = \frac{y_b}{c}\sigma_m = \left(\frac{145.35}{155}\right)(147.834) = 138.630 \text{ MPa}$$

At point B,

$$\tau_w = \frac{V}{dt_w} = \frac{(26.98 \times 10^3)}{(310 \times 10^{-3})(5.84 \times 10^{-3})} = 14.9028 \text{ MPa}$$

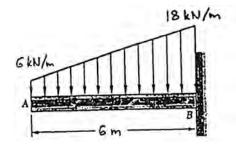
$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_w^2} = \sqrt{(69.315)^2 + (14.9028)^2} = 70.899 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 69.315 + 70.899$$

 $\sigma_{\rm max} = 140.2 \, \mathrm{MPa} \, \blacktriangleleft$

The following problem refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\rm all}$. For the selected design (use the loading of Prob. 5.73 and selected W530 × 66 shape), determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web.

SOLUTION

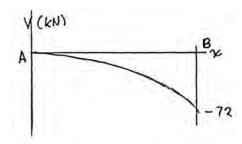


From Prob. 5.73, $\sigma_{\text{all}} = 160 \text{ MPa}$

$$|M|_{\text{max}} = 180 \text{ kN} \cdot \text{m}$$
 at section B.
 $|V| = 72 \text{ kN}$ at section B.

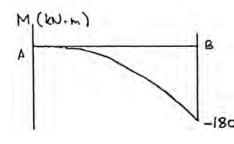
For W530×66 rolled steel section,

$$d = 526 \text{ mm},$$
 $b_f = 165 \text{ mm},$ $t_f = 11.4 \text{ mm},$ $t_w = 8.89 \text{ mm},$ $I = 351 \times 10^6 \text{ mm}^4,$ $S = 1340 \times 10^3 \text{ mm}^3$ $c = \frac{1}{2}d = 263 \text{ mm}$



(a) $|M|_{\text{max}} = 180 \times 10^3 \,\text{N} \cdot \text{m}$ $S = 1340 \times 10^{-6} \,\text{m}^3$ $\sigma_m = \frac{|M|_{\text{max}}}{S} = 134.328 \times 10^6, \qquad \sigma_m = 134.3 \,\text{MPa} \blacktriangleleft$

(b)
$$y_b = c - t_f = 251.6 \text{ mm}$$
 $\overline{y} = \frac{1}{2}(c + y_b) = 257.3 \text{ mm}$
 $A_f = b_f t_f = 1881 \text{ mm}^2$



At section B, $V = 72 \times 10^3 \text{ N}$

$$\tau_b = \frac{VQ}{It} = \frac{VA_f \overline{y}}{It} = \frac{(72 \times 10^3)(1881 \times 10^{-6})(257.3 \times 10^{-3})}{(351 \times 10^{-6})(8.89 \times 10^{-3})}$$
$$= 11.1674 \times 10^6 = 11.1674 \text{ MPa}$$

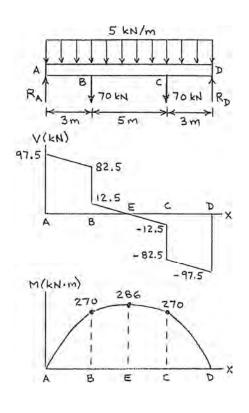
-180
$$\sigma_b = \frac{y_b}{c} \sigma_m = \frac{257.3}{263} (134.328) = 131.417 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = 66.651 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = \frac{131.417}{2} + 66.651$$
 $\sigma_{\text{max}} = 132.4 \text{ MPa}$

The following problem refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\text{all}}$. For the selected design (use the loading of Prob. 5.74 and selected W530 × 92 shape), determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress σ_{max} at the junction of a flange and the web.

SOLUTION



Reactions:
$$R_A = 97.5 \text{ kN} \uparrow R_D = 97.5 \text{ kN} \uparrow$$

$$|V|_{\text{max}} = 97.5 \text{ kN}$$

$$|M|_{\text{max}} = 286 \text{ kN} \cdot \text{m}$$

For W530 \times 92 rolled steel section,

$$d = 533 \text{ mm},$$
 $b_f = 209 \text{ mm},$ $t_f = 15.6 \text{ mm},$

$$t_w = 10.2 \text{ mm}, \qquad c = \frac{1}{2}d = 266.5 \text{ mm}$$

$$I = 554 \times 10^6 \text{ mm}^4$$
 $S = 2080 \times 10^3 \text{ mm}^3$

(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{286 \times 10^3}{2080 \times 10^{-6}} = 137.5 \times 10^6 \text{ Pa}$$

$$\sigma_m = 137.5 \, \mathrm{MPa} \, \blacktriangleleft$$

$$y_b = c - t_f = 250.9 \text{ mm}$$

$$A_f = b_f t_f = 3260.4 \text{ mm}^2$$

$$\overline{y} = \frac{1}{2}(c + y_b) = 258.7 \text{ mm}$$

$$Q = A_f \overline{y} = 843.47 \times 10^3 \,\text{mm}^3$$

At midspan:
$$V = 0$$
 $\tau_b = 0$

$$\sigma_b = \frac{y_b}{c} \sigma_m = \frac{250.9}{266.5} (137.5) = 129.5 \text{ MPa}$$

$$\sigma_{\rm max} = 129.5 \, \mathrm{MPa} \, \blacktriangleleft$$

PROBLEM 8.10 (Continued)

At sections *B* and *C*:

$$\sigma_{m} = \frac{M}{S} = \frac{270 \times 10^{3}}{2080 \times 10^{-6}} = 129.808 \text{ MPa}$$

$$\sigma_{b} = \frac{y_{b}}{c} \sigma_{m} = \frac{250.9}{266.5} (129.808) = 122.209 \text{ MPa}$$

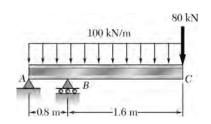
$$\tau_{b} = \frac{VQ}{It} = \frac{VA_{f}\overline{y}}{It_{w}} = \frac{(82.5 \times 10^{3})(3260.4 \times 10^{-6})(258.7 \times 10^{-3})}{(554 \times 10^{-6})(10.2 \times 10^{-3})}$$

$$= 12.3143 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_{b}}{2}\right)^{2} + \tau_{b}^{2}} = 62.333 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_{b}}{2} + R$$

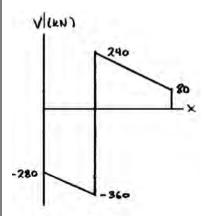
$$\sigma_{\text{max}} = 123.4 \text{ MPa} \blacktriangleleft$$

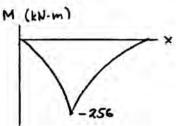


Each of the following problems refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\text{all}}$. For the selected design, determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress σ_{max} at the junction of a flange and the web.

PROBLEM 8.12 Loading of Prob. 5.75 and selected S20×66 shape.

SOLUTION





From Problem 5.75 $\sigma_{\text{all}} = 160 \text{ MPa}$ $|M|_{\text{max}} = 256 \text{ kN} \cdot \text{m} \text{ at point } B$ |V| = 360 kN at B

For S20×66 rolled steel section

$$d = 508 \text{ mm},$$
 $b_f = 159 \text{ mm},$ $t_f = 20.2 \text{ mm}$ $t_w = 12.8 \text{ mm},$ $I_x = 495 \times 10^6 \text{ mm}^4,$ $S_z = 1950 \times 10^3 \text{ mm}^3$ $c = \frac{1}{2}d = 254 \text{ mm}$

(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S_x} = \frac{256 \times 10^3}{1950 \times 10^{-6}} = 131.3 \text{ MPa}$$

 $y_b = c - t_f = 233.8$

$$\sigma_b = \frac{y_b}{\sigma_m} \sigma_m = 120.9 \text{ MPa}$$

$$\sigma_b = \frac{y_b}{c} \sigma_m = 120.9 \text{ MPa}$$
 $\frac{\sigma_b}{2} = 60.45 \text{ MPa}$

$$A_f = b_f t_f = 3212 \text{ mm}^2$$

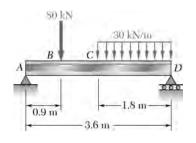
$$\overline{y} = \frac{1}{2}(c + y_b) = 243.9 \text{ mm}$$

$$Q = A_f \overline{y} = 783.4 \times 10^3 \text{ mm}^3$$

$$\tau_b = \frac{VQ}{It_w} = \frac{(360 \times 10^3)(783.4 \times 10^{-6})}{(495 \times 10^{-6})(12.8 \times 10^{-3})} = 44.5 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = \sqrt{60.45^2 + 44.5^2} = 75.06 \text{ MPa}$$

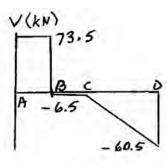
(b)
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 60.45 + 75.06 = 135.5 \text{ MPa}$$

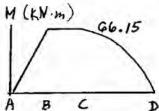


Each of the following problems refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\text{all}}$. For the selected design, determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress σ_{max} at the junction of a flange and the web.

PROBLEM 8.11 Loading of Prob. 5.78 and selected S310×47.3 shape.

SOLUTION





From Problem 5.78 $\sigma_{all} = 160 \text{ MPa}$

$$|M|_{\text{max}} = 66.15 \text{ kN} \cdot \text{m}$$
 at section B.

$$|V| = 73.5 \text{ kN at } B$$

For S310×47.3 rolled steel section

$$d = 305 \text{ mm},$$
 $b_f = 127 \text{ mm},$ $t_p = 13.8 \text{ mm}$

$$t_w = 8.9 \text{ mm}, \qquad I = 90.6 \times 10^6 \text{ mm}^4, \qquad S = 593 \times 10^3 \text{ mm}^3$$

$$c = \frac{1}{2}d = 152.5 \text{ mm}$$

(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{66.15 \times 10^3}{593 \times 10^{-6}}$$

$$\sigma_m$$
 = 111.6 MPa \blacktriangleleft

$$y_b = c - t_f = 138.7 \text{ mm}$$

$$\sigma_b = \frac{y_b}{c} \sigma_m = 101.5 \text{ MPa}$$

$$A_f = b_f t_f = 1752.6 \text{ mm}^2$$

$$\overline{y} = \frac{1}{2}(c + y_b) = 145.6 \text{ mm}$$

$$Q = A_f \overline{y} = 255.18 \times 10^3 \text{ mm}^3 = 255.18 \times 10^{-6} \text{ m}^3$$

$$\tau_b = \frac{VQ}{It_w} = \frac{(73.5 \times 10^3)(255.18 \times 10^{-6})}{(90.6 \times 10^{-6})(8.9 \times 10^{-3})} = 23.26 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = \sqrt{50.75^2 + 23.26^2} = 55.83 \text{ MPa}$$

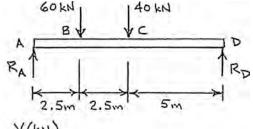
(b)
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = 106.6 \text{ MPa}$$

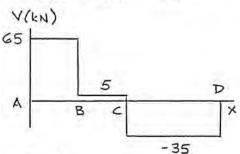
 $\sigma_{\rm max}$ = 106.6 MPa \blacktriangleleft

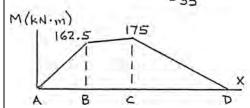
SOLUTION

The following problem refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\rm all}$. For the selected design (use the loading of Prob. 5.75 and selected S460 × 81.4 shape), determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web.









At section C:
$$\sigma_b = \frac{y_b}{c} \sigma_m = \frac{210.9}{228.5} (119.863) = 110.631 \text{ MPa}$$

$$\tau_b = \frac{VQ}{It} = \frac{VA_f \overline{y}}{It_w} = \frac{(35 \times 10^3)(2675.2 \times 10^{-6})(219.7 \times 10^{-3})}{(333 \times 10^{-6})(11.7 \times 10^{-3})} = 5.2799 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = 55.567 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R$$

Reactions:
$$R_A = 65 \text{ kN} \uparrow R_D = 35 \text{ kN} \uparrow$$

$$|V|_{\text{max}} = 65 \text{ kN}$$
$$|M|_{\text{max}} = 175 \text{ kN} \cdot \text{m}$$

For \$460×81.4 rolled steel section,

$$d = 457 \text{ mm},$$
 $b_f = 152 \text{ mm},$ $t_f = 17.6 \text{ mm},$ $t_w = 11.7 \text{ mm},$ $c = \frac{1}{2}d = 228.5 \text{ mm}$ $I = 333 \times 10^6 \text{ mm}^4$ $S = 1460 \times 10^3 \text{ mm}^3$

(a)
$$\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{175 \times 10^3}{1460 \times 10^{-6}} = 119.863 \times 10^6 \text{ Pa}$$

$$\sigma_m = 119.9 \,\mathrm{MPa}$$

(b)
$$y_b = c - t_f = 210.9 \text{ mm}$$

 $A_f = b_f t_f = 2675.2 \text{ mm}^2$
 $\overline{y} = \frac{1}{2}(c + y_b) = 219.7 \text{ mm}$
 $Q = A_f \overline{y} = 587.74 \times 10^3 \text{ mm}^3$

 $\sigma_{\rm max} = 110.9 \, \mathrm{MPa} \, \blacktriangleleft$

PROBLEM 8.13 (Continued)

At section *B*:
$$\sigma_m = \frac{M}{S} = \frac{162.5 \times 10^3}{1460 \times 10^{-6}} = 111.301 \text{ MPa}$$

$$\sigma_b = \frac{y_b}{c} \sigma_m = \frac{210.9}{228.5} (111.301) = 102.728 \text{ MPa}$$

$$\tau_b = \frac{VQ}{It} = \frac{VA_f \overline{y}}{It_w} = \frac{(65 \times 10^3)(2675.2 \times 10^{-6})(219.7 \times 10^{-3})}{(333 \times 10^{-6})(11.7 \times 10^{-3})} = 9.8055 \text{ MPa}$$

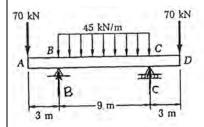
$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = 52.292 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R$$

$$\sigma_{\text{max}} = 103.7 \text{ MPa} \blacktriangleleft$$

The following problem refers to a rolled-steel shape selected in a problem of Chap. 5 to support a given loading at a minimal cost while satisfying the requirement $\sigma_m \le \sigma_{\text{all}}$. For the selected design (use the loading of Prob. 5.76 and selected S510 \times 98.2 shape), determine (a) the actual value of σ_m in the beam, (b) the maximum value of the principal stress σ_{max} at the junction of a flange and the web.

SOLUTION



From Prob. 5.76,

$$\sigma_{\rm all} = 160 \, \mathrm{MPa}$$

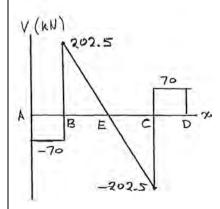
Bending moments:

 $-210 \text{ kN} \cdot \text{m}$ at sections B and C.

 $245.625 \text{ kN} \cdot \text{m}$ at midspan E.

 $|V|_{\text{max}} = 202.5 \text{ kN}$ at sections B and C.

For $S510 \times 98.2$ rolled steel section,



$$d = 508 \text{ mm},$$
 $b_f = 159 \text{ mm},$ $t_f = 20.2 \text{ mm},$ $t_w = 12.8 \text{ mm},$ $I = 495 \times 10^6 \text{ mm}^4, S = 1950 \times 10^3 \text{ mm}^3$ $c = \frac{1}{2}d = 254 \text{ mm}$

$$c = \frac{1}{2}d = 254 \text{ mm}$$

(a)
$$|M|_{\text{max}} = 245.625 \times 10^3 \text{ N} \cdot \text{m}$$
 $S = 1960 \times 10^{-6} \text{ m}^3$

$$\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{245.625 \times 10^3}{1950 \times 10^{-6}} = 125.962 \times 10^6 \text{ Pa}$$
 $\sigma_m = 126.0 \text{ MPa}$

(b)
$$y_b = c - t_f = 233.8 \text{ mm}, \quad A_f = b_f t_f = 3.212 \times 10^3 \text{ mm}^2, \quad \overline{y} = \frac{1}{2}(c + y_b) = 243.9 \text{ mm}$$

V = 0 $\tau_b = 0$ At midspan:

$$\sigma_b = \frac{y_b}{c}\sigma_m = \frac{233.8}{254}(125.962) = 115.9 \text{ MPa}$$

 $\sigma_{\text{max}} = 115.9 \text{ MPa}$

At sections *B* and *C*:

$$\sigma_m = \frac{M}{S} = \frac{210 \times 10^3}{1950 \times 10^{-6}} = 107.69 \text{ MPa}$$

$$\sigma_b = \frac{y_b}{c} \sigma_m = \frac{233.8}{254} (107.69) = 99.126 \text{ MPa}$$

$$\tau_b = \frac{VQ}{It} = \frac{VA_f \overline{y}}{It_w} = \frac{(202.5 \times 10^3)(3.212 \times 10^{-3})(243.9 \times 10^{-3})}{(495 \times 10^{-6})(12.8 \times 10^{-3})}$$

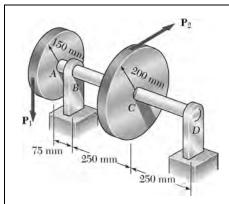
$$= 25.04 \text{ MPa}$$

PROBLEM 8.14 (Continued)

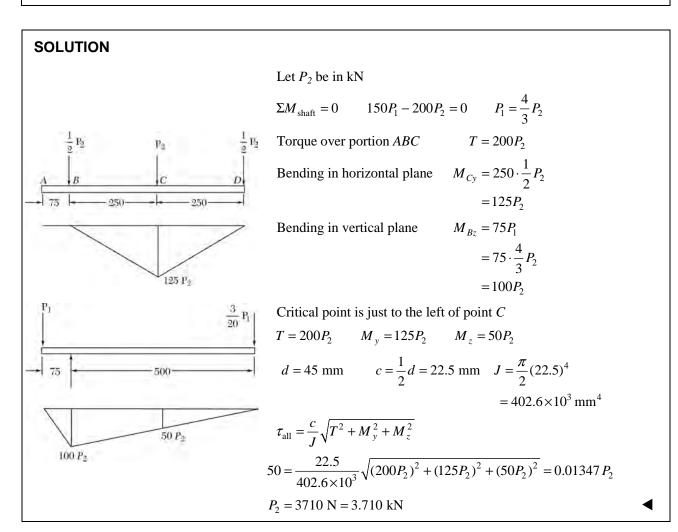
$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_b^2} = 55.53 \text{ MPa}$$

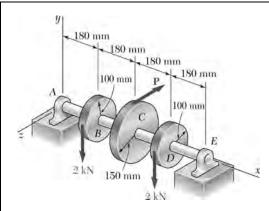
$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R$$

$$\sigma_{\rm max} = 105.1 \, \mathrm{MPa} \, \blacktriangleleft$$



The vertical force $\mathbf{P_1}$ and the horizontal force $\mathbf{P_2}$ are applied as shown to disks welded to the solid shaft AD. Knowing that the diameter of the shaft is 45 mm and that $\tau_{\rm all} = 50$ MPa, determine the largest permissible magnitude of the force $\mathbf{P_2}$.





The two 2-kN forces are vertical and the force **P** is parallel to the z axis. Knowing that $\tau_{\text{all}} = 55 \text{ MPa}$, determine the smallest permissible diameter of the solid shaft AE.

SOLUTION

$$\Sigma M_x = 0$$
: (2)(100) – 150P + (2)(100) = 0
P = 2.67 kN

Torques:

AB:
$$T = 0$$

BC:
$$T = -(2)(0.1) = -0.2 \text{ kNm}$$

CD:
$$T = (2)(0.1) = 0.2 \text{ kNm}$$

$$DE: T=0$$

Critical sections are either side of disk C.

$$T = 0.2 \text{ kNm}$$

$$M_{z} = 0.36 \text{ kNm}$$

$$M_{v} = 0.48 \text{ kNm}$$

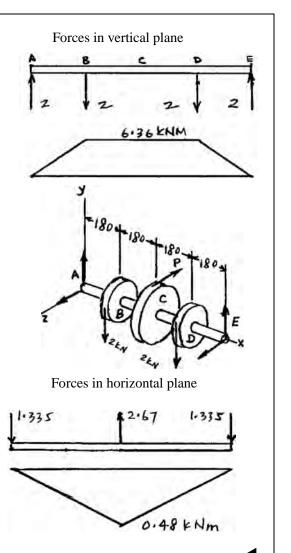
$$\tau_{\text{all}} = \frac{c}{J} \sqrt{M_y^2 + M_z^2 + T^2}$$

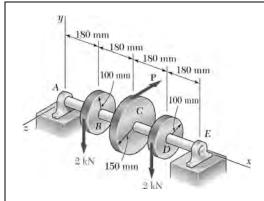
$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\sqrt{M_y^2 + M_z^2 + T^2}}{\tau_{\text{all}}}$$

$$= \frac{\sqrt{0.48^2 + 0.36^2 + 0.2^2}}{55 \times 10^3}$$

$$= 11.5 \times 10^{-6} \text{ m}^3$$

$$c = 0.0194 \text{ m}$$
 $d = 2c = 38.8 \text{ mm}$





For the gear-and-shaft system and loading of Prob. 8.19, determine the smallest permissible diameter of shaft AE, knowing that the shaft is hollow and has an inner diameter that is $\frac{2}{3}$ the outer diameter.

PROBLEM 8.19 The two 2-kN forces are vertical and the force **P** is parallel to the z axis. Knowing that $\tau_{\text{all}} = 55$ MPa, determine the smallest permissible diameter of the solid shaft AE.

SOLUTION

$$\Sigma M_x = 0$$
: $(2)(100) - 150P + (2)(100) = 0$
 $P = 2.67 \text{ kN}$

Torques:

$$AB$$
: $T = 0$

BC:
$$T = -(2)(0.1) = -0.2 \text{ kNm}$$

CD:
$$T = (2)(0.1) = 0.2 \text{ kNm}$$

$$DE$$
: $T = 0$

Critical sections are either side of disk C.

$$T = 0.2 \text{ kNm}$$

$$M_z = 0.36 \text{ kNm}$$

$$M_y = 0.48 \text{ kNm}$$

$$\tau_{\text{all}} = \frac{c}{J} \sqrt{M_y^2 + M_z^2 + T^2}$$

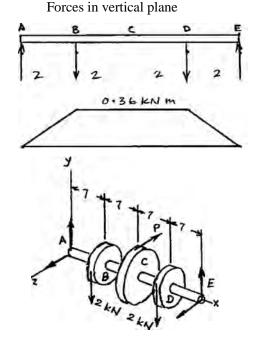
$$\frac{J}{c} = \frac{\pi}{2c} \left[c_o^4 - c_i^4 \right] = \frac{\pi}{2c} \left[c^4 - \left(\frac{2}{3} c \right)^4 \right]$$

$$=\frac{\pi}{2c}\frac{65c^4}{81} = \frac{65\pi}{162}c^3$$

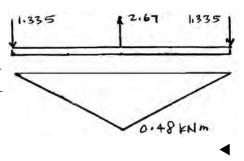
$$\frac{65\pi}{162}c^3 = \frac{\sqrt{M_y^2 + M_z^2 + T^2}}{\tau_{\text{all}}} = \frac{\sqrt{(0.48)^2 + (0.36)^2 + (0.2)^2}}{55 \times 10^3}$$

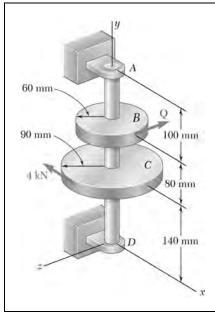
$$=11.5 \times 10^{-6} \,\mathrm{m}^3$$

$$c = 0.0209 \text{ m}$$
 $d = 2c = 41.8 \text{ mm}$



Forces in horizontal plane:





The 4-kN force is parallel to the x axis, and the force \mathbf{Q} is parallel to the z axis. The shaft AD is hollow. Knowing that the inner diameter is half the outer diameter and that $\tau_{\rm all} = 60$ MPa, determine the smallest permissible outer diameter of the shaft.

SOLUTION

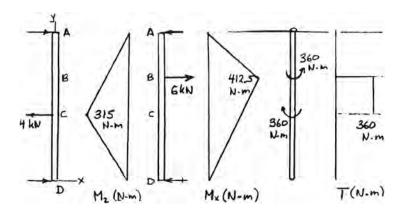
$$\Sigma M_y = 0$$
: $60 \times 10^{-3} \,\mathrm{Q} - (90 \times 10^{-3})(4 \times 10^3) = 0$
 $\mathrm{Q} = 6 \times 10^3 \,\mathrm{N} = 6 \,\mathrm{kN}$

Bending moment and torque diagrams.

In xy plane: $(M_z)_{\text{max}} = 315 \text{ N} \cdot \text{m}$ at C.

In yz plane: $(M_x)_{\text{max}} = 412.5 \text{ N} \cdot \text{m}$ at B.

About z-axis: $T_{\text{max}} = 360 \text{ N} \cdot \text{m}$ between B and C.



PROBLEM 8.18 (Continued)

At *B*:

$$M_z = \left(\frac{100}{180}\right)(315) = 175 \text{ N} \cdot \text{m}$$

$$\sqrt{M_x^2 + M_z^2 + T^2} = \sqrt{175^2 + 412.5^2 + 360^2} = 574.79 \text{ N} \cdot \text{m}$$

At *C*:

$$M_x = \left(\frac{140}{220}\right)(412.5) = 262.5 \text{ N} \cdot \text{m}$$

$$\sqrt{M_x^2 + M_z^2 + T^2} = \sqrt{315^2 + 262.5^2 + 360^2} = 545.65 \text{ N} \cdot \text{m}$$

 $d_o = 2c_o$

Largest value is 574.79 N·m.

$$\tau_{\text{max}} = \frac{\max \sqrt{M_x^2 + M_z^2 + T^2} c}{J}$$

$$\frac{J}{c} = \frac{\max \sqrt{M_x^2 + M_z^2 + T^2}}{\tau_{\text{max}}} = \frac{574.79}{60 \times 10^6}$$

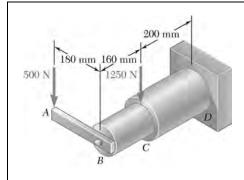
$$= 9.5798 \times 10^{-6} \,\text{m}^3 = 9.5798 \times 10^3 \,\text{mm}^3$$

$$\frac{J}{c} = \frac{\frac{\pi}{2} \left(c_o^4 - c_i^4\right)}{c_o} = \frac{\pi}{2} c_o^3 \left(1 - \frac{c_i^4}{c_o^4}\right) = \frac{\pi}{2} c_o^3 \left[1 - \left(\frac{1}{2}\right)^4\right]$$

$$= 1.47262 c_o^3$$

$$1.47262 c_o^3 = 9.5798 \times 10^3 \quad c_o = 18.67 \,\text{mm}$$

 $d_o = 37.3 \text{ mm}$



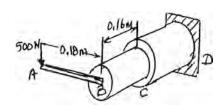
Neglecting the effect of fillets and of stress concentrations, determine the smallest permissible diameters of the solid rods BC and CD. Use $\tau_{\rm all} = 60$ MPa.

SOLUTION

$$au_{\text{all}} = 60 \times 10^6 \,\text{Pa}$$

$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}}$$

$$c^3 = \frac{2}{\pi} \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} \qquad d = 2c$$



Bending moments and torques.

Just to the left of *C*:

$$M = (500)(0.16) = 80 \text{ N} \cdot \text{m}$$

$$T = (500)(0.18) = 90 \text{ N} \cdot \text{m}$$

$$\sqrt{M^2 + T^2} = 120.416 \text{ N} \cdot \text{m}$$

Just to the left of *D*:

$$T = 90 \text{ N} \cdot \text{m}$$

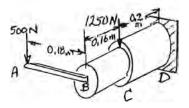
$$M = (500)(0.36) + (1250)(0.2) = 430 \text{ N} \cdot \text{m}$$

$$\sqrt{M^2 + T^2} = 439.32 \text{ N} \cdot \text{m}$$

Smallest permissible diameter d_{BC} .

$$c^3 = \frac{(2)(120.416)}{\pi(60 \times 10^6)} = 1.27765 \times 10^{-6} \text{m}^3$$

$$c = 0.01085 \text{ m} = 10.85 \text{ mm}$$



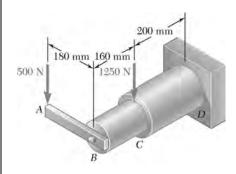
$$d_{RC} = 21.7 \text{ mm}$$

Smallest permissible diameter d_{CD} .

$$c^3 = \frac{(2)(439.32)}{\pi (60 \times 10^6)} = 4.6613 \times 10^{-6} \,\mathrm{m}^3$$

$$c = 0.01670 \text{ m} = 16.7 \text{ mm}$$

$$d_{CD} = 33.4 \text{ mm}$$



Knowing that rods BC and CD are of diameter 24 mm and 36 mm, respectively, determine the maximum shearing stress in each rod. Neglect the effect of fillets and of stress concentrations.

SOLUTION

Over *BC*:
$$c = \frac{1}{2}d = 12 \text{ mm} = 0.012 \text{ m}$$

Over *CD*:
$$c = \frac{1}{2}d = 18 \text{ mm} = 0.018 \text{ m}$$
$$\tau = \frac{\sqrt{M^2 + T^2}c}{L} = \frac{2}{\pi} \frac{\sqrt{M^2 + T^2}}{c^3}$$



Just to the left of *C*:

$$M = (500)(0.16) = 80 \text{ N} \cdot \text{m}$$

 $T = (500)(0.18) = 90 \text{ N} \cdot \text{m}$
 $\sqrt{M^2 + T^2} = 120.416 \text{ N} \cdot \text{m}$

Just to the left of *D*:

$$T = 90 \text{ N} \cdot \text{m}$$

$$M = (500)(0.36) + (1250)(0.2)$$

$$= 430 \text{ N} \cdot \text{m}$$

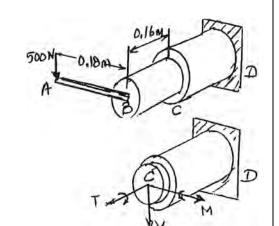
$$\sqrt{M^2 + T^2} = 439.32 \text{ N} \cdot \text{m}$$

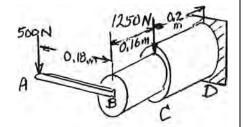
Maximum shearing stress in portion BC.

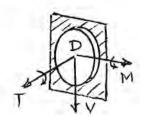
$$\tau_{\text{max}} = \frac{(2)(120.416)}{\pi (0.012)^3} = 44.36 \times 10^6 \text{ Pa}$$

Maximum shearing stress in portion CD.

$$\tau_{\text{max}} = \frac{(2)(439.32)}{\pi (0.018)^3} = 47.96 \times 10^6 \,\text{Pa}$$

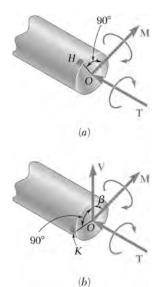






$$\tau_{\rm max} = 44.4 \ {\rm MPa} \ \blacktriangleleft$$

$$\tau_{\rm max} = 48.0 \; \mathrm{MPa} \; \blacktriangleleft$$



It was stated in Sec. 8.3 that the shearing stresses produced in a shaft by the transverse loads are usually much smaller than those produced by the torques. In the preceding problems, their effect was ignored and it was assumed that the maximum shearing stress in a given section occurred at point H (Fig. P8.21a) and was equal to the expression obtained in Eq. (8.5), namely,

$$\tau_H = \frac{c}{I} \sqrt{M^2 + T^2}$$

Show that the maximum shearing stress at point K (Fig. P8.21b), where the effect of the shear V is greatest, can be expressed as

$$\tau_K = \frac{c}{J} \sqrt{(M \cos \beta)^2 + \left(\frac{2}{3}cV + T\right)^2}$$

where β is the angle between the vectors **V** and **M**. It is clear that the effect of the shear V cannot be ignored when $\tau_K \ge \tau_H$. (*Hint:* Only the component of **M** along **V** contributes to the shearing stress at K.)

SOLUTION

Shearing stress at point K.

Due to V: For a semicircle,

$$Q = \frac{2}{3}c^3$$

For a circle cut across its diameter,

t = d = 2c

For a circular section,

$$I = \frac{1}{2}J$$

$$\tau_{xy} = \frac{VQ}{It} = \frac{(V)(\frac{2}{3}c^3)}{(\frac{1}{2}J)(2c)} = \frac{2}{3}\frac{Vc^2}{J}$$

Due to T:

$$\tau_{xy} = \frac{Tc}{J}$$

Since these shearing stresses have the same orientation,

$$\tau_{xy} = \frac{c}{J} \left(\frac{2}{3} V c + T \right)$$

PROBLEM 8.21 (Continued)

$$\sigma_{x} = \frac{Mu}{I} = \frac{2Mu}{J}$$

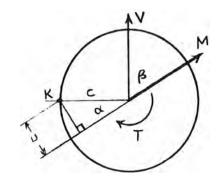
where u is the distance between point K and the neutral axis,

$$u = c \sin \alpha = c \sin \left(\frac{\pi}{2} - \beta\right) = c \cos \beta$$

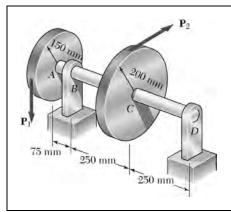
$$\sigma_x = \frac{2Mc\cos\beta}{J}$$

Using Mohr's circle,

$$\tau_K = R = \sqrt{\left(\frac{\sigma_x}{2}\right)^2 + \tau_{xy}^2}$$
$$= \frac{c}{J} \sqrt{(M\cos\beta)^2 + \left(\frac{2}{3}Vc + T\right)^2}$$



Cross section



Assuming that the magnitudes of the forces applied to disks A and C of Prob. 8.19 are, respectively, $P_1 = 4800 \text{ N}$ and $P_2 = 3600 \text{ N}$, and using the expressions given in Prob. 8.21, determine the values τ_H and τ_K in a section (a) just to the left of B, (b) just to the left of C.

SOLUTION

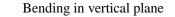
From Prob. 8.15, shaft diameter = 45 mm

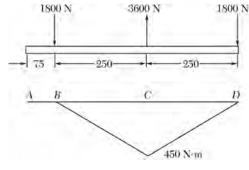
$$c = \frac{1}{2}d = 22.5 \text{ mm}$$
 $J = \frac{\pi}{2}c^4 = 402.6 \times 10^3 \text{ mm}^4$

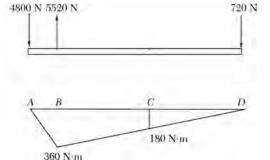
Torque over portion *ABC*:

$$T = (150)(4800) = (200)(3600) = 720000 \text{ N} \cdot \text{mm} = 720 \text{ N} \cdot \text{m}$$

Bending in horizontal plane







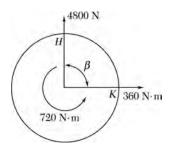
(a) Just to the left of point B:

$$V = 4800 \text{ N}$$

 $\beta = 90^{\circ}$

$$M = 360 \text{ N} \cdot \text{m}$$

 $T = 720 \text{ N} \cdot \text{m}$

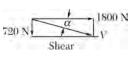


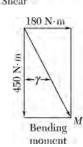
$$\tau_H = \frac{c}{J} \sqrt{M^2 + T^2} = \frac{0.0225}{402.6 \times 10^{-9}} \sqrt{(360)^2 + (720)^2}$$
$$= 44.98 \text{ MPa}$$

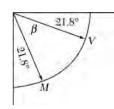
$$\tau_K = \frac{c}{J} \sqrt{(M \cos \beta)^2 + \left(\frac{2}{3}V_C + T\right)^2} = \frac{c}{J} \left[\frac{2}{3}V_C + T\right]$$

$$= \frac{0.0225}{402.6 \times 10^{-9}} \left[\left(\frac{2}{3} \right) (4800)(0.0225) + 720 \right] = 44.26 \text{ MPa}$$

PROBLEM 8.22 (Continued)







(b) Just to the left of point C

$$V = \sqrt{(720)^2 + (1800)^2} = 1938.6 \text{ N}$$

$$\alpha = \tan \frac{720}{1800} = 21.80^{\circ}$$

$$M = \sqrt{(180)^2 + (450)^2} = 484.66 \text{ N} \cdot \text{m}$$

$$\gamma = \tan^{-1} \frac{180}{450} = 21.80^{\circ}$$

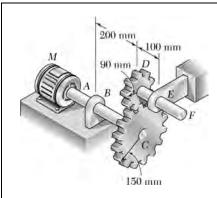
$$\beta = 90^{\circ} - 21.8^{\circ} - 21.8^{\circ} = 46.4^{\circ}$$

$$\tau_H = \frac{0.0225}{402.6 \times 10^{-9}} \sqrt{(720)^2 + (484.66)^2} = 48.50 \text{ MPa}$$

$$\frac{2}{3}V_C + T = \left(\frac{2}{3}\right)(1938.6)(0.0225) + 720 = 749 \text{ N} \cdot \text{m}$$

$$M\cos\beta = 484.66\cos 46.4^{\circ} = 334.23 \text{ N} \cdot \text{m}$$

$$\tau_k = \frac{0.0225}{402.6 \times 10^{-9}} \sqrt{(334.23)^2 + (749)^2} = 45.83 \text{ MPa}$$



The solid shafts ABC and DEF and the gears shown are used to transmit 15 kW from the motor M to a machine tool connected to shaft DEF. Knowing that the motor rotates at 240 rpm and that $\tau_{\rm all} = 50$ MPa, determine the smallest permissible diameter of (a) shaft ABC, (b) shaft DEF.

SOLUTION

$$240 \text{ rpm} = \frac{240}{60} = 4 \text{ Hz}$$

(a) Shaft ABC:
$$T = \frac{P}{2\pi f} = \frac{30}{(2\pi)(4)} = 1.194 \text{ kNm}$$

Gear C:
$$F_{CD} = \frac{T}{r_C} = \frac{1.194}{0.15} = 7.96 \text{ kN}$$

Bending moment at *B*:
$$M_B = (0.2)(7.96) = 1.592 \text{ kN}$$

$$\tau_{\text{all}} = \frac{c}{J} \sqrt{M^2 + T^2}$$

$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} = \frac{\sqrt{(1.194)^2 + (1.592)^2}}{50 \times 10^3} = 39.8 \times 10^{-6} \text{ m}^3$$

c 2
$$\tau_{\text{all}}$$
 50×10³
c = 0.0294 m $d = 2c = 58.8 \text{ mm} \blacktriangleleft$

(b) Shaft *DEF*:
$$T = r_D F_{CD} = (0.09)(7.96) = 0.7164 \text{ kNm}$$

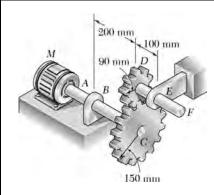
Bending moment at *E*:
$$M_E = (0.1)(7.96) = 0.796 \text{ kNm}$$

$$\tau_{\text{all}} = \frac{c}{J} \sqrt{M^2 + T^2}$$

$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} = \frac{\sqrt{(0.796)^2 + (0.7164)^2}}{50 \times 10^3} = 21.418 \times 10^{-6} \,\text{m}^3$$

$$c = 0.0239 \,\text{m}$$

$$d = 2c = 47.8 \,\text{mm}$$



Solve Prob. 8.23, assuming that the motor rotates at 360 rpm.

PROBLEM 8.23 The solid shafts ABC and DEF and the gears shown are used to transmit 15 kW from the motor M to a machine tool connected to shaft DEF. Knowing that the motor rotates at 240 rpm and that $\tau_{\text{all}} = 50 \text{ MPa}$, determine the smallest permissible diameter of (a) shaft ABC, (b) shaft DEF.

SOLUTION

$$360 \text{ rpm} = \frac{360}{60} = 6 \text{ Hz}$$

 $T = \frac{P}{2\pi f} = \frac{30}{(2\pi)(6)} = 0.796 \text{ kNm}$ Shaft ABC: (a)

 $F_{CD} = \frac{T}{r_C} = \frac{0.796}{0.15} = 5.31 \text{ kN}$ Gear C:

 $M_B = (0.2)(5.31) = 1.062 \text{ kNm}$ Bending moment at *B*:

 $\tau_{\rm all} = \frac{c}{I} \sqrt{M^2 + T^2}$

 $\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{oll}}} = \frac{\sqrt{1.062^2 + 0.796^2}}{50 \times 10^3} = 26.544 \times 10^{-6} \text{ m}^3$

c = 0.0257 m

d = 2c = 51.4 mm

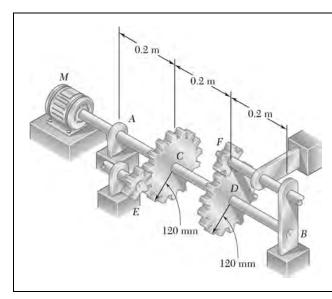
 $T = r_D F_{CD} = (0.09)(5.31) = 0.478 \text{ kNm}$ (*b*) Shaft DEF:

Bending moment at *E*: $M_E = (0.1)(5.31) = 0.531 \text{ kNm}$

> $\tau_{\rm all} = \frac{c}{I} \sqrt{M^2 + T^2}$ $\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{cut}}} = \frac{\sqrt{0.531^2 + 0.478^2}}{50 \times 10^3} = 14.29 \times 10^{-6}$

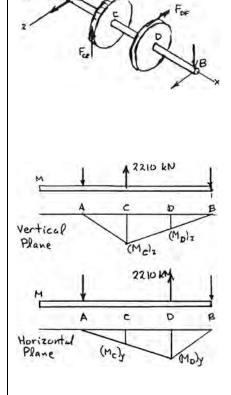
c = 0.0209 m

d = 2c = 41.8 mm



The solid shaft AB rotates at 360 rpm and transmits 20 kW from the motor M to machine tools connected to gears E and F. Knowing that $\tau_{\rm all} = 45$ MPa and assuming that 10 kW is taken off at each gear, determine the smallest permissible diameter of shaft AB.

SOLUTION



$$f = 360 \text{ rpm} = \frac{360}{60} = 6 \text{ Hz}$$

$$T_M = \frac{P_M}{2\pi f} = \frac{20 \times 10^3}{2\pi (6)} = 530.5 \text{ N} \cdot \text{m}$$

$$T_C = \frac{P_C}{2\pi f} = \frac{10 \times 10^3}{2\pi (6)} = 265.26 \text{ N} \cdot \text{m}$$

$$T_D = \frac{P_D}{2\pi f} = \frac{10 \times 10^3}{2\pi (6)} = 265.26 \text{ N} \cdot \text{m}$$

$$F_{CE} = \frac{T_C}{r_C} = \frac{265.26}{120 \times 10^{-3}} = 2.210 \times 10^3 \text{ N}$$

$$F_{DF} = \frac{T_D}{r_D} = \frac{265.26}{120 \times 10^{-3}} = 2.210 \times 10^3 \text{ N}$$

$$(M_C)_z = \frac{(0.2)(0.4)(2.210 \times 10^3)}{0.6} = 294.7 \text{ N} \cdot \text{m}$$

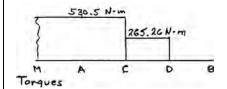
$$(M_D)_z = \frac{1}{2}(M_B)_z = 147.37 \text{ N} \cdot \text{m}$$

$$(M_C)_y = \frac{(0.4)(0.2)(2.210 \times 10^3)}{0.6} = 294.7 \text{ N} \cdot \text{m}$$

$$(M_D)_y = \frac{1}{2}(M_C)_y = 147.37 \text{ N} \cdot \text{m}$$

PROBLEM 8.25 (Continued)

Torques in shaft:



$$T_{MAC} = 530.5 \text{ N} \cdot \text{m}$$
$$T_{CD} = 265.26 \text{ N} \cdot \text{m}$$
$$T_{DB} = 0$$

Just to the left of gear *C*:

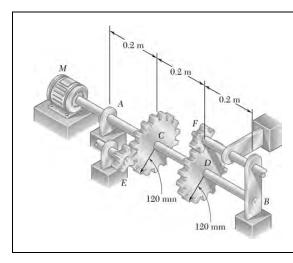
$$\max \sqrt{M_y^2 + M_z^2 + T^2} = 624.5 \text{ N} \cdot \text{m}$$

$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\max \sqrt{M_y^2 + M_z^2 + T^2}}{\tau_{\text{max}}}$$

$$= \frac{624.5}{45 \times 10^6} = 13.878 \times 10^{-6} \text{ m}^3$$

$$c = 20.67 \times 10^{-3} \text{ m} = 20.67 \text{ mm}$$

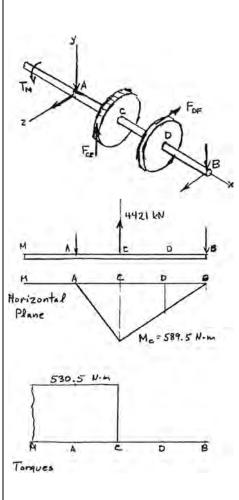
$$d = 2c \qquad d = 41.3 \text{ mm} \blacktriangleleft$$



Solve Prob. 8.25, assuming that the entire 20 kW is taken off at gear *E*.

PROBLEM 8.25 The solid shaft AB rotates at 360 rpm and transmits 20 kW from the motor M to machine tools connected to gears E and F. Knowing that $\tau_{\rm all} = 45$ MPa and assuming that 10 kW is taken off at each gear, determine the smallest permissible diameter of shaft AB.

SOLUTION



$$f = 360 \text{ rpm} = \frac{360}{60} = 6 \text{ Hz}$$

$$T_M = \frac{P_M}{2\pi f} = \frac{20 \times 10^3}{2\pi (6)} = 530.5 \text{ N} \cdot \text{m}$$

$$T_C = \frac{P_C}{2\pi f} = \frac{20 \times 10^3}{2\pi (6)} = 530.5 \text{ N} \cdot \text{m}$$

$$T_D = \frac{P_E}{2\pi f} = 0$$

$$F_{CE} = \frac{T_C}{r_C} = \frac{530.5}{120 \times 10^{-3}} = 4.421 \times 10^3 \text{ N}$$

$$F_{DF} = \frac{T_D}{r_D} = 0$$

Since $F_{DF} = 0$, there is no bending in the horizontal plane.

$$M_C = \frac{(0.2)(0.4)(4.421 \times 10^3)}{0.6} = 589.5 \text{ N} \cdot \text{m}$$

Torques in shaft. $T_{MAC} = 530.5 \text{ N} \cdot \text{m}$

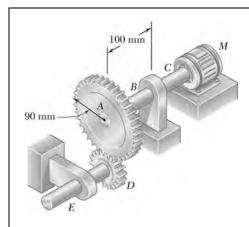
$$T_{CDE} = 0$$

Just to the left of gear C:

$$\max \sqrt{M_y^2 + M_z^2 + T^2} = \sqrt{0 + 589.5^2 + 530.5^2}$$
$$= 793.1 \text{ N} \cdot \text{m}$$

$$\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\max\sqrt{M_y^2 + M_z^2 + T^2}}{\tau_{\text{max}}} = \frac{793.1}{45 \times 10^6} = 17.624 \times 10^{-6} \text{ m}^3$$

$$c = 22.39 \times 10^{-3} = 22.39 \text{ mm}$$
 $d = 2c$ $d = 44.8 \text{ mm}$



The solid shaft ABC and the gears shown are used to transmit 10 kW from the motor M to a machine tool connected to gear D. Knowing that the motor rotates at 240 rpm and that $\tau_{\text{all}} = 60 \text{ MPa}$, determine the smallest permissible diameter of shaft ABC.

SOLUTION

$$f = \frac{240 \text{ rpm}}{60 \text{ sec/min}} = 4 \text{ Hz}$$

$$T = \frac{P}{2\pi f} = \frac{10 \times 10^3}{(2\pi)(4)} = 397.89 \text{ N} \cdot \text{m}$$

Gear A:

$$Fr_A - T = 0$$

$$F = \frac{T}{r_A} = \frac{397.89}{90 \times 10^{-3}} = 4421 \text{ N}$$

Bending moment at *B*:

$$M_B = L_{AB}F = (100 \times 10^{-3})(4421) = 442.1 \text{ N} \cdot \text{m}$$

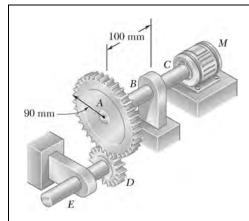
$$\tau_{\text{all}} = \frac{c}{J}\sqrt{M^2 + T^2}$$

$$\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}}$$

$$c^3 = \frac{2}{\pi}\frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} = \frac{(2)\sqrt{442.1^2 + 397.89^2}}{\pi(60 \times 10^6)} = 6.3108 \times 10^{-6} \text{m}^3$$

 $c = 18.479 \times 10^{-3} \,\mathrm{m}$ $d = 2c = 37.0 \times 10^{-3} \,\mathrm{m}$

d = 37.0 mm



Assuming that shaft *ABC* of Prob. 8.27 is hollow and has an outer diameter of 50 mm, determine the largest permissible inner diameter of the shaft.

PROBLEM 8.27 The solid shaft ABC and the gears shown are used to transmit 10 kW from the motor M to a machine tool connected to gear D. Knowing that the motor rotates at 240 rpm and that $\tau_{\rm all} = 60$ MPa, determine the smallest permissible diameter of shaft ABC.

SOLUTION

$$f = \frac{240 \text{ rpm}}{60 \text{ sec/min}} = 4 \text{ Hz}$$

$$T = \frac{P}{2\pi f} = \frac{10 \times 10^3}{(2\pi)(4)} = 397.89 \text{ N} \cdot \text{m}$$

Gear A:

$$Fr_A - T = 0$$

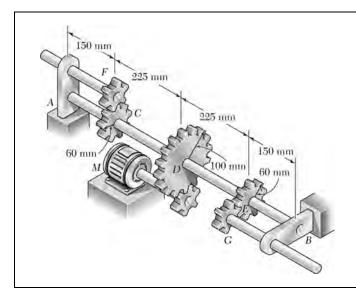
$$F = \frac{T}{r_A} = \frac{397.89}{90 \times 10^{-3}} = 4421 \text{ N}$$



Bending moment at *B*:

$$\begin{split} M_B &= L_{AB}F = (100 \times 10^{-3})(4421) = 442.1 \text{ N} \cdot \text{m} \\ \tau_{\text{all}} &= \frac{c_o}{J} \sqrt{M^2 + T^2} \qquad c_o = \frac{1}{2} d_o = 25 \times 10^{-3} \text{ m} \\ \frac{J}{c_o} &= \frac{\pi}{2} \frac{(c_o^4 - c_i^4)}{c_o} = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} \\ c_i^4 &= c_o^4 - \frac{2c_o \sqrt{M^2 + T^2}}{\pi \tau_{\text{all}}} \\ &= (25 \times 10^{-3})^4 - \frac{(2)(25 \times 10^{-3})\sqrt{442.1^2 + 397.89^2}}{\pi (60 \times 10^6)} \\ &= 390.625 \times 10^{-9} - 157.772 \times 10^{-9} = 232.85 \times 10^{-9} \\ c_i &= 21.967 \times 10^{-3} \text{ m} \qquad d_i = 2c_i = 43.93 \times 10^{-3} \text{ m} \end{split}$$

d = 43.9 mm



The solid shaft AB rotates at 450 rpm and transmits 20 kW from the motor M to machine tools connected to gears F and G. Knowing that $\tau_{\rm all} = 55$ MPa and assuming that 8 kW is taken off at gear F and 12 kW is taken off at gear G, determine the smallest permissible diameter of shaft AB.

SOLUTION

$$f = \frac{450}{60} = 7.5 \text{ Hz}$$

Torque applied at *D*:

$$T_D = \frac{P}{2\pi f} = \frac{20 \times 10^3}{2\pi (7.5)} = 424.41 \text{ N} \cdot \text{m}$$

Torques on gears *C* and *E*:

$$T_C = \frac{8}{20}T_D = 169.76 \text{ N} \cdot \text{m}$$

$$T_E = \frac{12}{20} T_D = 254.65 \text{ N} \cdot \text{m}$$

Forces on gears

$$F_D = \frac{T_D}{r_D} = \frac{424.41}{100 \times 10^{-3}} = 4244 \text{ N}$$

$$F_C = \frac{T_C}{r_C} = \frac{169.76}{60 \times 10^{-3}} = 2829 \text{ N}$$

$$F_E = \frac{T_E}{r_E} = \frac{254.65}{60 \times 10^{-3}} = 4244 \text{ N}$$

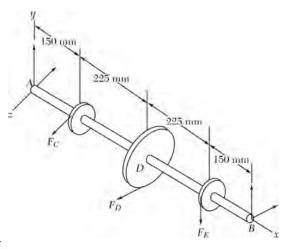
Torques in various parts

$$AC: T=0$$

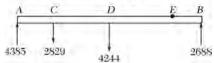
CD:
$$T = 169.76 \text{ N} \cdot \text{m}$$

$$DE: T = 254.65 \text{ N} \cdot \text{m}$$

EB:
$$T = 0$$



Forces in horizontal plane:

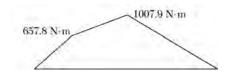


PROBLEM 8.29 (Continued)

Critical point lies just the right of *D*:

$$T = 254.65 \text{ N} \cdot \text{m}$$

 $M_y = 1007.9 \text{ N} \cdot \text{m}$
 $M_z = 318.3 \text{ N} \cdot \text{m}$



$$\left(\sqrt{M_y^2 + M_z^2 + T^2}\right)_{\text{max}} = 1087.2 \text{ N} \cdot \text{m}$$

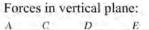
$$\tau_{\text{all}} = \frac{c}{J} \left(\sqrt{M_y^2 + M_z^2 + T^2} \right)_{\text{max}}$$

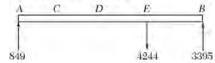
$$\frac{J}{c} = \frac{\pi}{2} c^3 = \frac{\left(\sqrt{M_y^2 + M_z^2 + T^2} \right)_{\text{max}}}{\tau_{\text{all}}} = \frac{1087.2}{55 \times 10^6}$$

$$= 19.767 \times 10^{-3} \text{ m}$$

$$d = 2c = 46.5 \times 10^{-3} \text{ m} = 46.5 \text{ m}$$

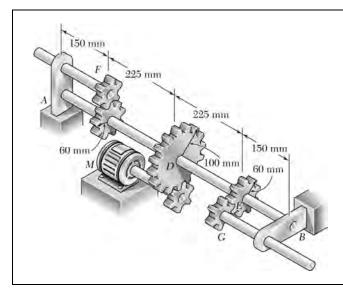
 $c = 23.26 \times 10^{-3} \text{ m}$







•



Solve Prob. 8.29, assuming that the entire 20 kW is taken off at gear G.

PROBLEM 8.29 The solid shaft AB rotates at 450 rpm and transmits 20 kW from the motor M to machine tools connected to gears F and G. Knowing that $\tau_{\rm all}$ = 55 MPa and assuming 8 kW is taken off at gear F and 12 kW is taken off at gear G, determine the smallest permissible diameter of shaft AB.

SOLUTION

$$f = \frac{450}{60} = 7.5 \text{ Hz}$$

Torque applied at *D*:

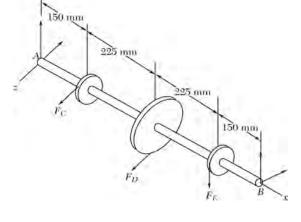
$$T_D = \frac{P}{2\pi f} = \frac{20 \times 10^3}{(2\pi)(7.5)} = 424.41 \text{ N} \cdot \text{m}$$

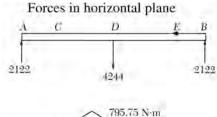
Torque on gears
$$E$$
:
$$T_E = T_D = 424.41 \text{ N} \cdot \text{m}$$

Forces on gears *D* and *E*:

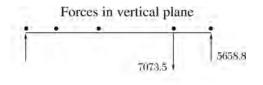
$$F_D = \frac{T_D}{r_D} = \frac{424.41}{100 \times 10^{-3}} = 4244 \text{ N}$$

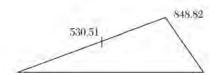
$$F_E = \frac{T_E}{r_D} = \frac{424.41}{60 \times 10^{-3}} = 7073.5 \text{ N}$$







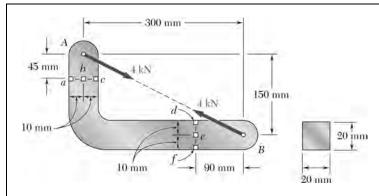




PROBLEM 8.30 (Continued)

Bending moments

$$\begin{split} M_D &= \sqrt{530.51^2 + 795.75^2} = 956.4 \text{ N} \cdot \text{m} \\ M_E &= \sqrt{848.82^2 + 318.3^2} = 906.5 \text{ N} \cdot \text{m} \\ \left(\sqrt{M^2 + T^2}\right)_{\text{max}} &= \sqrt{956.4^2 + 424.41^2} = 1046.3 \text{ N} \cdot \text{m} \\ \tau_{\text{all}} &= \frac{c}{J} \left(\sqrt{M^2 + T^2}\right)_{\text{max}} \\ \frac{J}{c} &= \frac{\pi}{2} c^3 = \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} = \frac{1046.3}{55 \times 10^6} = 19.024 \times 10^{-6} \text{ m}^3 \\ c &= 22.96 \times 10^{-3} \text{ m} \quad d = 2c = 45.9 \times 10^{-3} \text{ m} = 45.9 \text{ mm} \end{split}$$



Two 4-kN forces are applied to an L-shaped machine element AB as shown. Determine the normal and shearing stresses at (a) point a, (b) point b, (c) point c.

4×103 sinB

SOLUTION

Let β be the slope angle of line AB.

$$\tan \beta = \frac{150}{300}$$
 $\beta = 26.565^{\circ}$

Draw free body sketch of the portion of the machine element lying above section abc.

$$P = -(4 \times 10^{3}) \sin \beta$$

$$= -1.78885 \times 10^{3} \text{ N}$$

$$V = (4 \times 10^{3}) \cos \beta = 3.5777 \times 10^{3} \text{ N}$$

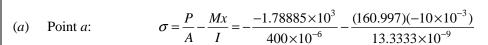
$$M = (45 \times 10^{-3})(4 \times 10^{3}) \cos \beta = 160.997 \text{ N} \cdot \text{m}$$

Section properties:

$$A = (20)(20) = 400 \text{ mm}^2 = 400 \times 10^{-6} \text{ m}^2$$

$$I = \frac{1}{12}(20)(20)^3 = 13.3333 \times 10^3 \text{ mm}^4$$

$$= 13.3333 \times 10^{-9} \text{ m}^4$$



= 116.3 MPa ◀

 $\tau = 0$

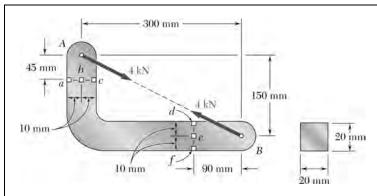
(b) Point b:
$$\sigma = \frac{P}{A} = -\frac{1.78885 \times 10^3}{400 \times 10^{-6}}$$
 = -4.47 MPa

$$Q = (20)(10)(5) = 1000 \text{ mm}^3 = 10^{-6} \text{ m}^3$$

$$\tau = \frac{VQ}{It} = \frac{(3.5777)(10^{-6})}{(13.3333 \times 10^{-6})(20 \times 10^{-3})} = 13.42 \text{ MPa} \blacktriangleleft$$

(c) Point c:
$$\sigma = \frac{P}{A} - \frac{Mx}{I} = -\frac{1.78885 \times 10^3}{400 \times 10^{-6}} - \frac{(160.997)(10 \times 10^{-3})}{13.3333 \times 10^{-9}} = -125.2 \text{ MPa}$$

 $\tau = 0$



Two 4-kN forces are applied to an L-shaped machine element AB as shown. Determine the normal and shearing stresses at (a) point d, (b) point e, (c) point f.

SOLUTION

Let β be the slope angle of line AB.

$$\tan \beta = \frac{150}{300}$$
 $\beta = 26.565^{\circ}$

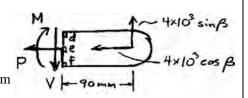
Draw free body sketch of the portion of the machine element lying to the right of section def.

$$P = -(4 \times 10^{3}) \cos \beta$$

$$= -3.5777 \times 10^{3} \text{ N}$$

$$V = (4 \times 10^{3}) \sin \beta = 1.78885 \times 10^{3} \text{ N}$$

$$M = (90 \times 10^{-3})(4 \times 10^{-3}) \sin \beta = 160.997 \text{ N} \cdot \text{m}$$



Section properties:

$$A = (20)(20) = 400 \text{ mm}^2 = 400 \times 10^{-6} \text{ m}^3$$
$$I = \frac{1}{12}(20)(20)^3 = 13.3333 \times 10^3 \text{ mm}^4$$
$$= 13.3333 \times 10^{-9} \text{ m}^4$$

(a) Point d:
$$\sigma = \frac{P}{A} - \frac{My}{I} = \frac{-3.5777 \times 10^3}{400 \times 10^{-6}} - \frac{(160.997)(10 \times 10^{-3})}{13.3333 \times 10^{-9}}$$

$$\tau = 0$$

(b) Point e:
$$\sigma = \frac{P}{A} = \frac{-3.5777 \times 10^3}{400 \times 10^{-6}}$$
 = -8.94 MPa

$$Q = (20)(10)(5) = 1000 \text{ mm}^3 = 10^{-6} \text{ m}^3$$

$$\tau = \frac{VQ}{It} = \frac{(1.78885 \times 10^3)(10^{-6})}{(13.3333 \times 10^{-9})(20 \times 10^{-3})} = 6.71 \,\text{MPa} \blacktriangleleft$$

(c) Point f:
$$\sigma = \frac{P}{A} - \frac{My}{I} = \frac{-3.5777 \times 10^3}{400 \times 10^{-6}} - \frac{(160.997)(-10 \times 10^{-3})}{13.3333 \times 10^{-9}} = 111.8 \text{ MPa} \blacktriangleleft$$

 $\tau = 0$

0.75 m 0.75 m A D 100 mm b 150 mm b 200 mm E B 0.3 m 0.6 m

PROBLEM 8.33

The cantilever beam AB has a rectangular cross section of 150×200 mm. Knowing that the tension in cable BD is 10.4 kN and neglecting the weight of the beam, determine the normal and shearing stresses at the three points indicated.

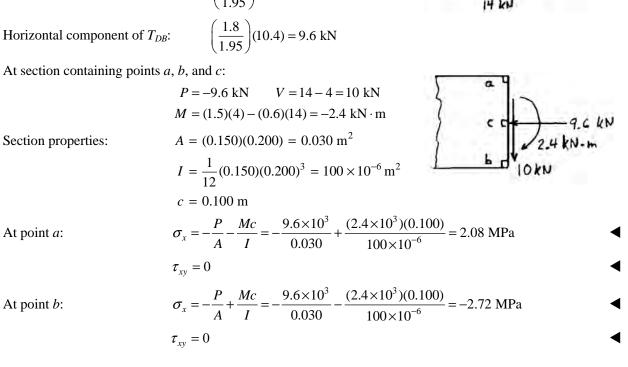
SOLUTION

$$\overline{DB} = \sqrt{0.75^2 + 1.8^2}$$

= 1.95 m

Vertical component of T_{DB} :

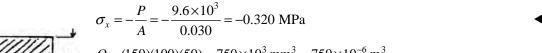
$$\left(\frac{0.75}{1.95}\right)(10.4) = 4 \text{ kN}$$



PROBLEM 8.33 (Continued)



- ISO -



$$Q = (150)(100)(50) = 750 \times 10^3 \text{ mm}^3 = 750 \times 10^{-6} \text{ m}^3$$

$$VO = (10 \times 10^3)(750 \times 10^{-6})$$

$$\tau_{xy} = -\frac{VQ}{It} = -\frac{(10 \times 10^3)(750 \times 10^{-6})}{(100 \times 10^{-6})(0.150)} = -0.500 \text{ MPa}$$

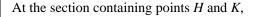
A 60 mm 9 kN 60 mm 12 mm 40 mm

PROBLEM 8.34

Member AB has a uniform rectangular cross section of 10×24 mm. For the loading shown, determine the normal and shearing stress at (a) point H, (b) point K.

SOLUTION

+)
$$\Sigma M_B = 0$$
: (9)(60 sin 30°) - 120 $R_A = 0$
 $R_A = 2.25 \text{ kN}$
+ $\Sigma F_x = 0$: 2.25 cos 30° - $B_x = 0$
 $B_x = 1.9486 \text{ kN} \leftarrow$
+ $\Sigma F_y = 0$: 2.25 sin 30° - 9 + $B_y = 0$
 $B_y = 7.875 \text{ kN} \uparrow$



$$P = 7.875\cos 30^{\circ} + 1.9486\sin 30^{\circ} = 7.794 \text{ kN}$$

$$V = 7.875\sin 30^{\circ} - 1.9486\cos 30^{\circ} = 2.25 \text{ kN}$$

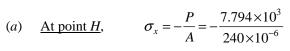
$$M = (7.875 \times 10^{3})(40 \times 10^{-3}\sin 30^{\circ})$$

$$- (1.9486 \times 10^{3})(40 \times 10^{-3}\cos 30^{\circ})$$

$$= 90 \text{ N} \cdot \text{m}$$

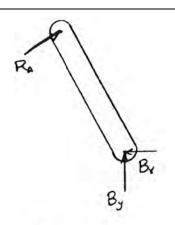
$$A = 10 \times 24 = 240 \text{ mm}^{2} = 240 \times 10^{-6} \text{ m}^{2}$$

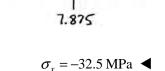
$$I = \frac{1}{12}(10)(24)^{3} = 11.52 \times 10^{3} \text{ mm}^{4} = 11.52 \times 10^{-9} \text{ m}^{4}$$



$$\tau_{xy} = \frac{3}{2} \frac{V}{A} = \frac{3}{2} \frac{2.25 \times 10^3}{240 \times 10^{-6}}$$

(b) At point
$$K$$
, $\sigma_x = -\frac{P}{A} - \frac{Mc}{I} = -\frac{7.794 \times 10^3}{240 \times 10^{-6}} - \frac{(90)(12 \times 10^{-3})}{11.52 \times 10^{-9}}$

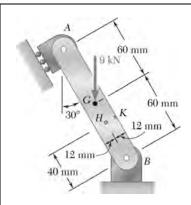




$$\tau_{xy} = 14.06 \, \text{MPa}$$

$$\sigma_x = -126.2 \,\mathrm{MPa}$$

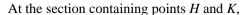
$$\tau_{xy} = 0$$



Member AB has a uniform rectangular cross section of 10×24 mm. For the loading shown, determine the normal and shearing stress at (a) point H, (b) point K.

SOLUTION

+)
$$\Sigma M_B = 0$$
: $(120\cos 30^\circ)R_A - (60\sin 30^\circ)(9) = 0$
 $R_A = 2.598 \text{ kN}$
+) $\Sigma F_y = 0$: $B_y - 9 = 0$ $B_y = 9 \text{ kN}$ \uparrow
+> $\Sigma F_x = 0$: $2.598 - B_x = 0$ $B_x = 2.598 \text{ kN} \leftarrow$



$$P = 9\cos 30^{\circ} + 2.598\sin 30^{\circ} = 9.093 \text{ kN}$$

$$V = 9\sin 30^{\circ} - 2.598\cos 30^{\circ} = 2.25 \text{ kN}$$

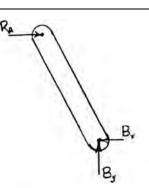
$$M = (9\times10^{3})(40\times10^{-3}\sin 30^{\circ})$$

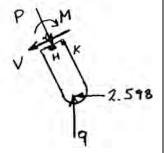
$$-(2.598\times10^{3})(40\times10^{-3}\cos 30^{\circ})$$

$$= 90 \text{ N} \cdot \text{m}$$

$$A = 10\times240 = 240 \text{ mm}^{2} = 240\times10^{-6} \text{ m}^{2}$$

$$I = \frac{1}{12}(10)(24)^{3} = 11.52\times10^{3} \text{ mm}^{4} = 11.52\times10^{-9} \text{ m}^{4}$$





(a) At point H:

$$\sigma_x = -\frac{P}{A} = -\frac{9.093 \times 10^3}{240 \times 10^{-6}}$$
$$\tau_{xy} = \frac{3}{2} \frac{V}{A} = \frac{3}{2} \frac{2.25 \times 10^3}{240 \times 10^{-6}}$$

$$\sigma = -37.9 \text{ MPa}$$

$$\tau = 14.06 \text{ MPa}$$

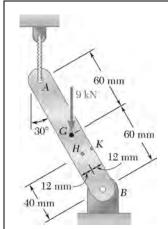
(b) At point K:

$$\sigma_x = -\frac{P}{A} - \frac{Mc}{I}$$

$$= -\frac{9.093 \times 10^3}{240 \times 10^{-6}} - \frac{(90)(12 \times 10^{-3})}{11.52 \times 10^{-9}}$$

$$\sigma = -131.6 \,\mathrm{MPa}$$

$$\tau = 0$$



Member AB has a uniform rectangular cross section of 10×24 mm. For the loading shown, determine the normal and shearing stress at (a) point H, (b) point K.

SOLUTION

At the section containing points H and K,

$$P = 4.5 \cos 30^{\circ} = 3.897 \text{ kN}$$

$$V = 4.5 \sin 30^{\circ} = 2.25 \text{ kN}$$

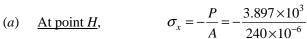
$$M = (4.5 \times 10^{3})(40 \times 10^{-3} \sin 30^{\circ})$$

$$= 90 \text{ N} \cdot \text{m}$$

$$A = 10 \times 24 = 240 \text{ mm}^{2} = 240 \times 10^{-6} \text{ m}^{2}$$

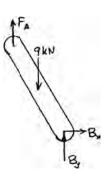
$$I = \frac{1}{12}(10)(24)^{3} = 11.52 \times 10^{3} \text{ mm}^{4}$$

$$= 11.52 \times 10^{-9} \text{ m}^{4}$$



$$\tau_{xy} = \frac{3}{2} \frac{V}{A} = \frac{3}{2} \frac{2.25 \times 10^3}{240 \times 10^{-6}}$$

(b) At point K,
$$\sigma_x = -\frac{P}{A} - \frac{Mc}{I} = -\frac{3.897 \times 10^3}{240 \times 10^{-6}} - \frac{(90)(12 \times 10^{-3})}{11.52 \times 10^{-9}}$$





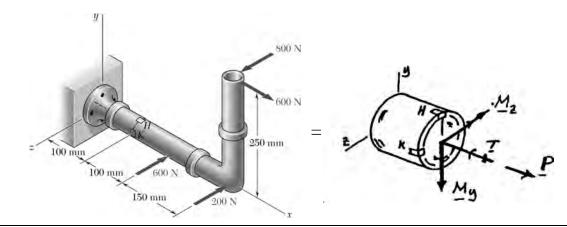
$$\sigma = -16.24 \text{ MPa}$$

$$\tau = 14.06 \, \text{MPa}$$

$$\sigma = 110.0 \, \text{MPa}$$

$$\tau = 0$$

Several forces are applied to the pipe assembly shown. Knowing that the pipe has inner and outer diameters equal to 40 mm and 48 mm, respectively, determine the normal and shearing stresses at (a) point H, (b) point K.



SOLUTION

Section properties:



$$P = 600 \text{ N}$$

$$T = (800 \text{ N})(0.25 \text{ m}) = 200 \text{ Nm}$$

$$M_z = (600 \text{ N})(0.25 \text{ m}) = 150 \text{ Nm}$$

$$M_y = (800 \text{ N} - 200 \text{ N})(0.25 \text{ m}) - (600)(0.1 \text{ m}) = 90 \text{ Nm}$$

$$V_z = 800 - 600 - 200 = 0$$

$$V_{v} = 0$$

$$A = \pi(24^2 - 22^2) = 289 \text{ mm}^2$$

$$I = \frac{\pi}{4} (24^4 - 22^4) = 76592 \text{ mm}^4$$

$$J = 2I = 153184 \text{ mm}^4$$

(a) Point H:

$$\sigma_H = \frac{P}{A} + \frac{M_z c}{I} = \frac{600}{289 \times 10^{-6}} + \frac{(150)(0.024)}{76592 \times 10^{-12}} = 49.078 \text{ MPa}$$

$$\sigma_H = 49.1 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\tau_N = \frac{Tc}{J} = \frac{(200)(0.024)}{153184 \times 10^{-12}} = 31.33 \text{ MPa}$$

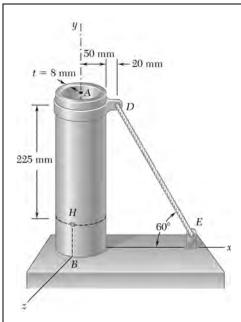
$$\tau_N = 31.3 \text{ MPa} \blacktriangleleft$$

$$\sigma_K = \frac{P}{A} + \frac{M_y c}{I} = \frac{600}{289 \times 10^{-6}} - \frac{(90)(0.024)}{76592 \times 10^{-12}} = -26.125 \text{ MPa}$$

$$\sigma_{\kappa} = -26.1 \, \text{MPa} \, \blacktriangleleft$$

$$\tau_K = \frac{Tc}{I} = \text{ same as for } \tau_N$$

$$\tau_K$$
 = 31.03 MPa ◀



The steel pipe AB has a 100-mm outer diameter and an 8-mm wall thickness. Knowing that the tension in the cable is 40 kN, determine the normal and shearing stresses at point H.

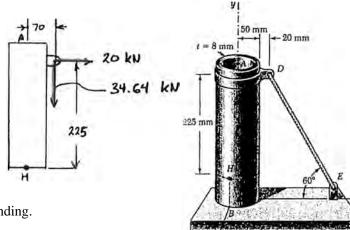
SOLUTION

Vertical force:

 $40\cos 30^{\circ} = 34.64 \text{ kN}$

Horizontal force:

 $40\sin 30^\circ = 20 \text{ kN}$



 $\sigma = -14.98 \, \mathrm{MPa} \, \blacktriangleleft$

Point *H* lies on neutral axis of bending.

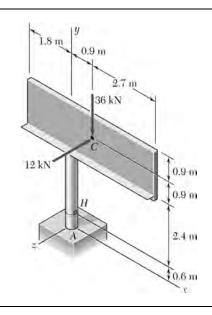
Section properties:

$$d_o = 100 \text{ mm}, \qquad c_o = \frac{1}{2}d_o = 50 \text{ mm}, \qquad c_i = c_o - t = 42 \text{ mm},$$

$$A = \pi(c_o^2 - c_i^2) = 2.312 \times 10^3 \text{ mm}^2 = 2.312 \times 10^{-3} \text{ m}^2$$

$$\sigma = -\frac{P}{A} = -\frac{34.64 \times 10^3}{2.312 \times 10^{-6}}$$

For thin pipe,
$$\tau = 2\frac{V}{A} = \frac{(2)(20 \times 10^3)}{2.314 \times 10^{-3}}$$
 $\tau = 17.29 \text{ MPa}$



The billboard shown weighs 36 kN and is supported by a structural tube that has a 375 mm outer diameter and a 12 mm wall thickness. At a time when the resultant of the wind pressure is 12 kN located at the center *C* of the billboard, determine the normal and shearing stresses at point *H*.

SOLUTION

At section containing point H

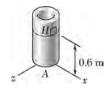
$$P = 36 \text{ kN (compression)}$$

$$T = (12)(0.9) = 10.8 \text{ kN} \cdot \text{m}$$

$$M_r = -(-3.3)(12) = -39.6 \text{ kN} \cdot \text{m}$$

$$M_{z} = -(0.9)(36) = -32.4 \text{ kN} \cdot \text{m}$$

$$V = 12 \text{ kN}$$



Section properties

$$d_o = 375 \text{ mm}$$
 $c_o = \frac{1}{2}d_o = 187.5 \text{ mm}$ $c_i = c_o - t = 175.5 \text{ mm}$

$$A = \pi \left(c_o^2 - c_i^2\right) = 13684.8 \text{ mm}^2$$

$$I = \frac{\pi}{4} \left(c_o^4 - c_i^4 \right) = 225.65 \times 10^6 \text{ mm}^4$$

$$J = 2I = 451.30 \times 10^6 \,\mathrm{mm}^4$$

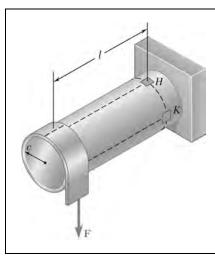
$$Q = \frac{2}{3} \left(c_o^3 - c_i^3 \right) = 790.9 \times 10^3 \text{ mm}^3$$

$$\sigma = -\frac{P}{A} - \frac{Mc}{I} = -\frac{36 \times 10^3}{13684.8 \times 10^{-6}} - \frac{(32.4 \times 10^3)(0.1875)}{225.65 \times 10^{-6}}$$

$$= -2.63 - 26.92 = -29.55$$
 MPa

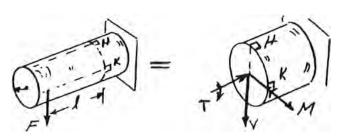
$$\tau = \frac{Tc}{J} + \frac{VQ}{It} = \frac{(10.8 \times 10^3)(0.1875)}{451.30 \times 10^{-6}} + \frac{(12 \times 10^3)(790.9 \times 10^{-6})}{(225.65 \times 10^{-6})(0.024)}$$

$$= 4.48 + 1.75 = 6.23$$
 MPa



A thin strap is wrapped around a solid rod of radius c = 20 mm as shown. Knowing that l = 100 mm and F = 5 kN, determine the normal and shearing stresses at (a) point H, (b) point K.

SOLUTION



At the section containing points H and K,

$$T = Fc$$
 $M = Fl$ $V = F$

$$J = \frac{\pi}{2}c^4 \qquad I = \frac{\pi}{4}c^4$$

Point *H*:

$$\sigma = \frac{Mc}{I} = \frac{Flc}{\frac{\pi}{4}c^4} \qquad \sigma = \frac{4Fl}{\pi c^2}$$

$$\tau = \frac{Tc}{J} = \frac{Fc^2}{\frac{\pi}{4}c^4} \qquad \tau = \frac{2F}{\pi c^2}$$

Point *K*: Point *K* lies on the neutral axis.

$$\sigma = 0$$

Due to torque:

$$\tau = \frac{Tc}{J} = \frac{2F}{\pi c^2}$$

Due to shear:

For a semicircle,

$$Q = \frac{2}{3}c^3$$
, $t = d = 2c$

$$\tau = \frac{VQ}{It} = \frac{F\frac{2}{3}c^3}{\frac{\pi}{4}c^4(2c)} = \frac{4F}{3\pi c^2}$$

Combined.

$$\tau = \frac{2F}{\pi c^2} + \frac{4F}{3\pi c^2}$$
 $\tau = \frac{10F}{3\pi c^2}$

PROBLEM 8.40 (Continued)

Data:
$$F = 5 \text{ kN} = 5 \times 10^3 \text{ N}, \quad l = 100 \text{ mm} = 0.100 \text{ m}$$

 $c = 20 \text{ mm} = 0.020 \text{ m}$

(a) Point H:
$$\sigma = \frac{(4)(5 \times 10^3)(0.100)}{\pi (0.020)^3}$$

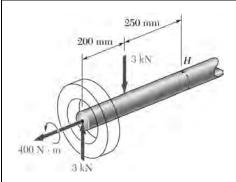
% (0.020)

 σ = 79.6 MPa

 $\tau = \frac{(2)(5 \times 10^3)}{\pi (0.020)^2}$ $\tau = 7.96 \text{ MPa} \blacktriangleleft$

(b) Point \underline{K} : $\sigma = 0$

 $\tau = \frac{(10)(5 \times 10^3)}{3\pi (0.020)^2}$ $\tau = 13.26 \text{ MPa}$



The axle of a small truck is acted upon by the forces and couple shown. Knowing that the diameter of the axle is 36 mm, determine the normal and shearing stress at point H located on the top of the axle.

SOLUTION

The bending moment causing normal stress at point H is

$$M = (0.2)(3) = 0.6 \text{ kNm}$$

$$C = \frac{1}{2}d = 18 \text{ mm}$$

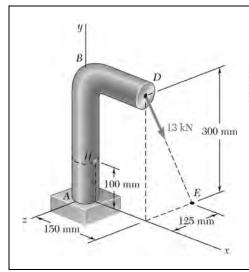
$$I = \frac{\pi}{4}C^4 = 82448 \text{ mm}^4$$
, $J = 2I = 164896 \text{ mm}^4$

Normal stress at *H*:

$$\sigma_H = -\frac{Mc}{I} = -\frac{(0.6 \times 10^6)(18)}{82448} = -131 \text{ MPa}$$

At the section containing point H, V = 0, T = 400 Nm

$$\tau_H = \frac{Tc}{J} = \frac{(400000)(18)}{164896} = 43.7 \text{ MPa}$$



A 13-kN force is applied as shown to the 60-mm-diameter cast-iron post ABD. At point H, determine (a) the principal stresses and principal planes, (b) the maximum shearing stress.

SOLUTION

$$DE = \sqrt{125^2 + 300^2} = 325 \text{ mm}$$

At point D,

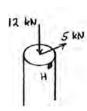
$$F_x = 0$$

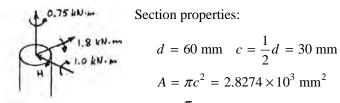
$$F_y = -\left(\frac{300}{325}\right)(13) = -12 \text{ kN}$$

$$F_z = -\left(\frac{125}{300}\right)(13) = -5 \text{ kN}$$

Moment of equivalent force-couple system at C, the centroid of the section containing point H:

$$\vec{M} = \begin{vmatrix} \vec{i} & \vec{j} & \vec{k} \\ 0.150 & 0.200 & 0 \\ 0 & -12 & -5 \end{vmatrix} = -1.00\vec{i} + 0.75\vec{j} - 1.8\vec{k} \text{ kN} \cdot \text{m}$$





Section properties:

$$d = 60 \text{ mm}$$
 $c = \frac{1}{2}d = 30 \text{ mm}$

$$A = \pi c^2 = 2.8274 \times 10^3 \text{ mm}^2$$

$$I = \frac{\pi}{4}c^4 = 636.17 \times 10^3 \text{ mm}^4$$

$$J = 2I = 1.2723 \times 10^6 \text{ mm}^4$$

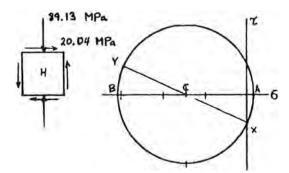
For a semicircle,

$$Q = \frac{2}{3}c^3 = 18.00 \times 10^3 \text{ mm}^3$$

PROBLEM 8.42 (Continued)

At point
$$H$$
, $\sigma_H = -\frac{P}{A} - \frac{Mc}{I} = -\frac{12 \times 10^3}{2.8274 \times 10^{-3}} - \frac{(1.8 \times 10^3)(30 \times 10^{-3})}{636.17 \times 10^{-9}} = -89.13 \text{ MPa}$

$$\tau_H = \frac{Tc}{J} + \frac{VQ}{It} = \frac{(0.75 \times 10^3)(30 \times 10^{-3})}{1.2723 \times 10^{-6}} + \frac{(5 \times 10^3)(18.00 \times 10^{-6})}{(636.17 \times 10^{-9})(60 \times 10^{-3})} = 20.04 \text{ MPa}$$



(a)
$$\sigma_{\text{ave}} = \frac{\sigma_H}{2} = -44.565 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_H}{2}\right)^2 + \tau_H^2} = 48.863 \text{ MPa}$$

$$\sigma_a = \sigma_{\text{ave}} + R$$
 $\sigma_a = 4.3 \,\text{MPa}$

$$\sigma_a = \sigma_{\text{ave}} + R$$
 $\sigma_a = 4.3 \text{ MPa}$ \bullet

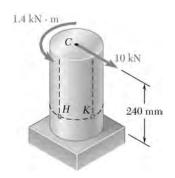
$$\sigma_b = \sigma_{\text{ave}} - R$$
 $\sigma_b = -93.4 \text{ MPa}$

$$\tan 2\theta_p = \frac{2\tau_H}{|\sigma_H|} = 0.4497$$

$$\theta_a = 12.1^\circ, \quad \theta_b = 102.1^\circ \blacktriangleleft$$

$$\tau_{\rm max} = R = 48.9 \, \mathrm{MPa} \, \blacktriangleleft$$

(*b*)



A 10-kN force and a 1.4-kN · m couple are applied at the top of the 65-mm diameter brass post shown. Determine the principal stresses and maximum shearing stress at (a) point H, (b) point K.

SOLUTION

At the section containing points H and K,

$$V = 10 \text{ kN} = 10 \times 10^{3} \text{ N}$$

$$M = (10 \times 10^{3})(240 \times 10^{-3}) = 2.4 \times 10^{3} \text{ N} \cdot \text{m}$$

$$T = 1.4 \times 10^{3} \text{ N} \cdot \text{m}$$

$$c = \frac{1}{2}d = 32.5 \text{ mm} = 0.0325 \text{ m}$$

$$J = \frac{\pi}{2}c^{4} = 1.75248 \times 10^{-6} \text{ m}^{4}$$

$$I = \frac{1}{2}J = 0.87624 \times 10^{-6} \text{ m}^{4}$$

$$Q = \frac{2}{3}c^{3} = \frac{2}{3}(0.0325)^{3} = 22.885 \times 10^{-6} \text{ m}^{3}$$

For a semicircle,

(a) Stresses at point *H*.

H lies on the neutral axis:
$$\sigma = 0$$

Due to torque:
$$\tau = \frac{Tc}{J} = \frac{(1.4 \times 10^3)(0.0325)}{1.75248 \times 10^{-6}}$$
$$= 25.963 \text{ MPa}$$

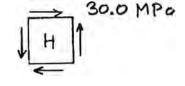
 $\tau = \frac{VQ}{It} = \frac{(10 \times 10^3)(22.885 \times 10^{-6})}{(0.87624 \times 10^{-6})(0.065)}$ Due to shear: = 4.018 MPa

Total at *H*: $\tau = 30.0 \text{ MPa}$

$$\sigma_{\text{ave}} = 0$$
, $R = 30.0 \text{ MPa}$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R$$

$$\sigma_{\text{min}} = \sigma_{\text{ave}} - R$$



 $\sigma_{\min} = -30.0 \text{ MPa}$

 $\sigma_{\rm max} = 30.0 \, \mathrm{MPa}$

 $\tau_{\rm max} = 30.0 \, \mathrm{MPa} \, \blacktriangleleft$

 $\tau_{\text{max}} = R$

PROBLEM 8.43 (Continued)

(b) Stresses at point K.

Due to shear: $\tau = 0$

Due to torque: $\tau = \frac{Tc}{I} = 25.963 \text{ MPa}$

Due to bending:
$$\sigma = -\frac{Mc}{I} = -\frac{(2.4 \times 10^3)(0.0325)}{(0.87624 \times 10^{-6})}$$

= -89.016 MPa

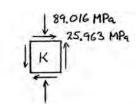
$$\sigma_{\text{ave}} = \frac{-89.016}{2} = -44.508 \text{ MPa}$$

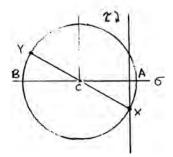
$$R = \sqrt{\left(\frac{89.016}{2}\right)^2 + (25.963)^2}$$
$$= 51.527 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R$$

$$\sigma_{\min} = \sigma_{\text{ave}} - R$$

$$\tau_{\text{max}} = R$$





$$\sigma_{\rm max} = 7.02 \; {\rm MPa} \; \blacktriangleleft$$

$$\sigma_{\min} = -96.0 \text{ MPa}$$

$$\tau_{\rm max} = 51.5 \; \mathrm{MPa} \; \blacktriangleleft$$

25 kN 25 kN 200 mm

PROBLEM 8.44

Three forces are applied to a 100-mm diameter plate that is attached to the solid 45-mm diameter shaft AB. At point H, determine (a) the principal stresses and principal plates, (b) the maximum shearing stress.

SOLUTION

At the section containing points H,

$$P = 50 \text{ kN}$$
 (compression)

$$V = 10 \text{ kN}$$

$$T = (10)(50) = 500 \text{ kNmm}$$

$$M = (10)(200) = 2000 \text{ kNmm}$$

$$d = 45 \text{ mm } C = \frac{1}{2}d = 22.5 \text{ mm}$$

$$A = \pi C^2 = 1590 \text{ mm}^2$$

$$I = \frac{\pi}{4}C^4 = 0.2013 \times 10^6 \text{ mm}^4$$

$$J = 2I = 0.4026 \times 10^6 \text{ mm}^4$$

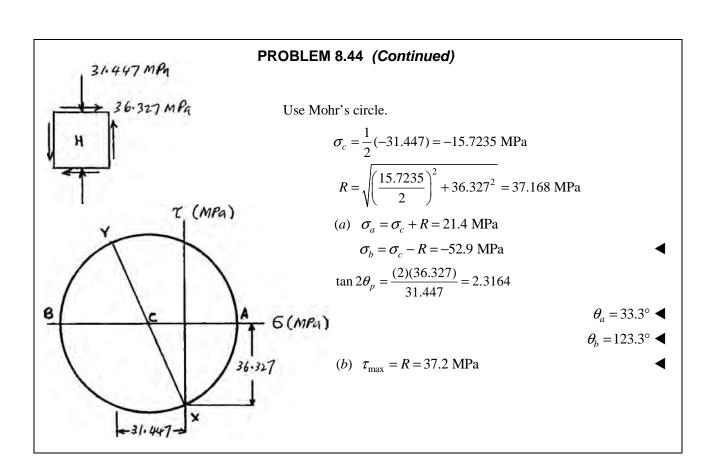
For a semicircle,

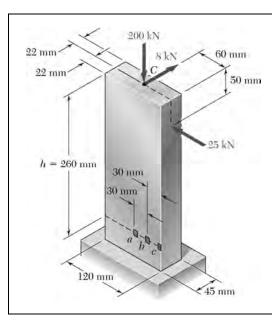
$$Q = \frac{2}{3}c^3 = 7.594 \times 10^3 \,\mathrm{mm}^3$$

Point *H* lies on neutral axis of bending.

$$\sigma_H = \frac{P}{A} = -\frac{50000}{1590} = -31.447 \text{ MPa}$$

$$\tau_H = \frac{Tc}{J} + \frac{VQ}{It} = \frac{(5 \times 10^5)(22.5)}{0.4026 \times 10^6} + \frac{(10 \times 10^3)(7.594 \times 10^3)}{(0.2013 \times 10^6)(45)} = 36.327 \text{ MPa}$$



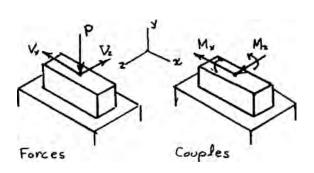


Three forces are applied to the bar shown. Determine the normal and shearing stresses at (a) point a, (b) point b, (c) point c.

SOLUTION

Calculate forces and couples at section containing points a, b, and c.

$$h = 260 \text{ mm}$$



$$P = 200 \text{ N}, \quad V_x = 25 \text{ kN}, \quad V_z = 8 \text{ kN}$$

$$M_z = (260 - 50)(25) = 5250 \text{ kN} \cdot \text{mm}$$

$$M_x = (260)(8) = 2080 \text{ kN} \cdot \text{mm}$$

Section properties.

$$A = (120)(44) = 5280 \text{ mm}^2$$

$$I_x = \frac{1}{12}(120)(44)^3 = 851840$$

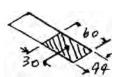
$$I_z = \frac{1}{12} (44)(120)^3 = 6.336 \times 10^6 \text{ mm}^4$$

Stresses.

$$\sigma = -\frac{P}{A} + \frac{M_z x}{I_z} + \frac{M_x z}{I_x} \qquad \tau = \frac{V_x Q}{I_z t}$$

PROBLEM 8.45 (Continued)

(a) Point a:

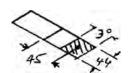


$$x = 0$$
, $z = 22 \text{ mm}$, $Q = (44)(60) \left(\frac{60}{2}\right) = 79200 \text{ mm}^3$

$$\sigma = -\frac{200 \times 10^3}{5280} + 0 + \frac{(2.08 \times 10^6)(22)}{0.85184 \times 10^6} = 15.8 \text{ MPa}$$

$$\tau = \frac{(25 \times 10^3)(79200)}{(6.336 \times 10^6)(44)} = 7.1 \text{ MPa}$$

(*b*) Point *b*:



$$x = 30 \text{ mm}, \quad z = 22 \text{ mm}, \quad Q = (44)(30)(45) = 59400 \text{ mm}^3$$

$$\sigma = -\frac{200 \times 10^3}{5280} + \frac{(5.25 \times 10^6)(30)}{6.336 \times 10^6} + \frac{(208 \times 10^6)(22)}{0.85184 \times 10^6} = 40.7 \text{ MPa}$$

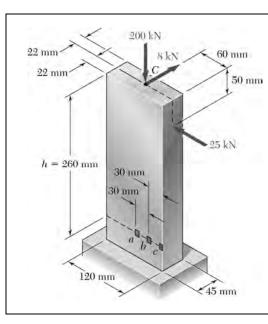
$$\tau = \frac{(25 \times 10^3)(59400)}{[(6.336 \times 10^6)(44)]} = 5.3 \text{ MPa}$$

(*c*) Point *c*:

$$x = 60 \text{ mm}, \quad z = 22 \text{ mm}, \quad Q = 0$$

$$\sigma = -\frac{200 \times 10^3}{5280} + \frac{(5.25 \times 10^6)(60)}{6.336 \times 10^6} + \frac{(2.08 \times 10^6)(22)}{0.85184 \times 10^6} = 65.6 \text{ MPa}$$

 $\tau = 0$

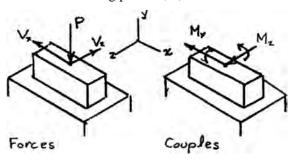


Solve Prob. 8.45, assuming that h = 300 mm

PROBLEM 8.45 Three forces are applied to the bar shown. Determine the normal and shearing stresses at (a) point a, (b) point b, (c) point c.

SOLUTION

Calculate forces and couples at section containing points a, b, and c. h = 300 mm



$$P = 200 \text{ kN}, \quad V_x = 25 \text{ kN}, \quad V_z = 8 \text{ kN}$$

$$M_z = (300 - 50)(25) = 6250 \text{ kN} \cdot \text{mm}$$

$$M_x = (300)(8) = 2400 \text{ kN} \cdot \text{mm}$$

Section properties.
$$A = (120)(44) = 5280 \text{ mm}^2$$

$$I_x = \frac{1}{12}(120)(44)^3 = 851840 \text{ mm}^4$$

$$I_z = \frac{1}{12} (44)(120)^3 = 6.336 \times 10^6 \text{ mm}^4$$

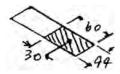
$$\sigma = -\frac{P}{A} + \frac{M_z x}{I_z} + \frac{M_x z}{I_x}$$

$$\tau = \frac{V_x Q}{I_z t}$$

Stresses.

PROBLEM 8.46 (Continued)

(a) Point a:

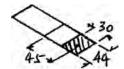


$$x = 0$$
, $z = 22 \text{ mm}$, $Q = (44)(60)(30) = 79200 \text{ mm}^3$

$$\sigma = -\frac{200 \times 10^3}{5280} + 0 + \frac{(2.4 \times 10^6)(22)}{0.85184 \times 10^6} = 24.1 \text{ MPa}$$

$$\tau = \frac{(25 \times 10^3)(79200)}{(6.336 \times 10^6)(44)} = 7.1 \text{ MPa}$$

(*b*) Point *b*:



$$x = 30 \text{ mm}, \quad z = 22 \text{ mm}, \quad Q = (30)(44)(45) = 59400 \text{ mm}^3$$

$$\sigma = -\frac{200 \times 10^3}{5280} + \frac{(6.25 \times 10^6)(30)}{10.336 \times 10^6} + \frac{(2.4 \times 10^6)(22)}{0.85184 \times 10^6} = 53.7 \text{ MPa}$$

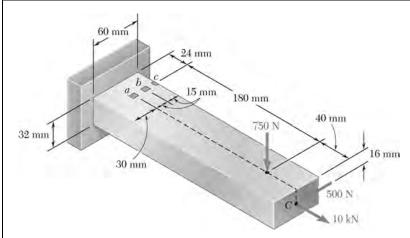
$$\tau = \frac{(25 \times 10^3)(59400)}{(6.336 \times 10^6)(44)} = 5.3 \text{ MPa}$$

(*c*) Point *c*:

$$x = 60 \text{ mm}, \quad z = 22 \text{ mm}, \quad Q = 0$$

$$\sigma = -\frac{200 \times 10^3}{5280} + \frac{(6.25 \times 10^6)(60)}{6.336 \times 10^6} + \frac{(2.4 \times 10^6)(22)}{0.85184 \times 10^6} = 83.3 \text{ MPa}$$

 $\tau = 0$



Three forces are applied to the bar shown. Determine the normal and shearing stresses at (a) point a, (b) point b, (c) point c.

SOLUTION

$$A = (60)(32) = 1920 \text{ mm}^2$$

$$= 1920 \times 10^{-6} \text{ m}^2$$

$$I_z = \frac{1}{12} (60)(32)^3 = 163.84 \times 10^3 \text{ mm}^4$$

$$= 163.84 \times 10^{-9} \text{ m}^4$$

$$I_y = \frac{1}{12} (32)(60)^3$$

$$= 579 \times 10^3 \text{ mm}^4$$

$$= 576 \times 10^{-9} \text{ m}^4$$

At the section containing points a, b, and c,

$$P = 10 \text{ kN}$$

$$V_y = 750 \text{ N},$$

$$V_z = 500 \text{ N}$$

$$M_z = (180 \times 10^{-3})(750)$$

$$= 135 \text{ N} \cdot \text{m}$$

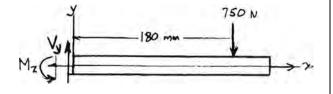
$$M_y = (220 \times 10^{-3})(500)$$

$$= 110 \text{ N} \cdot \text{m}$$

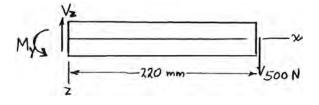
$$T = 0$$

$$\sigma = \frac{P}{A} + \frac{M_z y}{I_z} - \frac{M_y z}{I_y}$$

$$\tau = \frac{V_z Q}{I_y t}$$



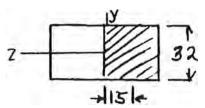
Side View



Top View

PROBLEM 8.47 (Continued)

(a) Point a: $y = 16 \text{ mm}, z = 0, Q = A\overline{z} = (32)(30)(15) = 14.4 \times 10^3 \text{ mm}^3$

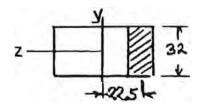


 $\sigma = \frac{10 \times 10^3}{1920 \times 10^{-6}} + \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} - 0$

 σ = 18.39 MPa

 $\tau = \frac{(500)(14.4 \times 10^{-6})}{(576 \times 10^{-9})(32 \times 10^{-3})}$

- $\tau = 0.391 \,\mathrm{MPa}$
- (b) Point b: $y = 16 \text{ mm}, z = -15 \text{ mm}, Q = A\overline{z} = (32)(15)(22.5) = 10.8 \times 10^3 \text{ mm}^3$

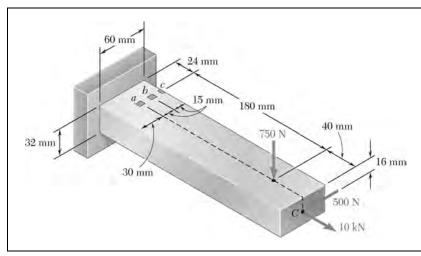


- $\sigma = \frac{10 \times 10^{3}}{1920 \times 10^{-6}} + \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} \frac{(110)(-15 \times 10^{-3})}{576 \times 10^{-9}}$
- $\sigma = 21.3 \,\mathrm{MPa}$

 $\tau = \frac{(500)(10.8 \times 10^{-6})}{(576 \times 10^{-9})(32 \times 10^{-3})}$

 $\tau = 0.293 \text{ MPa}$

- (c) Point c: y = 16 mm, z = -30 mm, Q = 0
 - $\sigma = \frac{10 \times 10^3}{1920 \times 10^{-6}} + \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} \frac{(110)(-30 \times 10^{-6})}{576 \times 10^{-9}}$
- σ = 24.1 MPa
 - $\tau = 0$



Solve Prob. 8.47, assuming that the 750-N force is directed vertically upward.

PROBLEM 8.47 Three forces are applied to the bar shown. Determine the normal and shearing stresses at (a) point a, (b) point b, (c) point c.

SOLUTION

$$A = (60)(32) = 1920 \text{ mm}^2$$

$$= 1920 \times 10^{-6} \text{ m}^2$$

$$I_z = \frac{1}{12} (60)(32)^3$$

$$= 163.84 \times 10^3 \text{ mm}^4$$

$$= 163.84 \times 10^{-9} \text{ m}^4$$

$$I_y = \frac{1}{12} (32)(60)^3$$

$$= 576 \times 10^3 \text{ mm}^4$$

$$= 576 \times 10^{-9} \text{ m}^4$$

At the section containing points a, b, and c,

$$P = 10 \text{ kN} \qquad T = 0$$

$$V_y = 750 \text{ N}$$

$$V_z = 500 \text{ N}$$

$$M_z = (180 \times 10^{-3})(750)$$

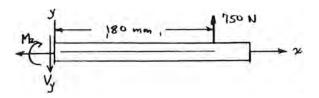
$$= 135 \text{ N} \cdot \text{m}$$

$$M_y = (220 \times 10^{-3})(500)$$

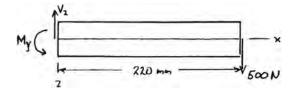
$$= 110 \text{ N} \cdot \text{m}$$

$$\sigma = \frac{P}{A} + \frac{M_z y}{I_z} - \frac{M_y z}{I_y}$$

$$\tau = \frac{V_z Q}{I_y t}$$



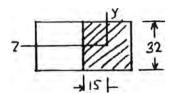
Side View



Top View

PROBLEM 8.48 (Continued)

 $y = 16 \text{ mm}, \quad z = 0, \quad Q = A\overline{z} = (32)(30)(15) = 14.4 \times 10^3 \text{ mm}^3$ Point *a*: (a)

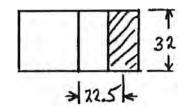


 $\sigma = \frac{10 \times 10^3}{1920 \times 10^{-6}} - \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} = 0$

 $\sigma = -7.98 \, \text{MPa}$

 $\tau = \frac{(500)(14.4 \times 10^{-6})}{(163.84 \times 10^{-9})(32 \times 10^{-3})}$

- $\tau = 0.391 \, \text{MPa}$
- $y = 16 \text{ mm}, \quad z = -15 \text{ mm}, \quad Q = A\overline{z} = (32)(15)(22.5) = 10.8 \times 10^3 \text{ mm}^3$ Point *b*: (b)

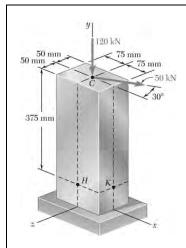


- $\sigma = \frac{10 \times 10^{3}}{1920 \times 10^{-6}} \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} \frac{(110)(-15 \times 10^{-3})}{576 \times 10^{-9}} \qquad \sigma = -5.11 \text{ MPa}$

 $\tau = \frac{(500)(10.8 \times 10^{-6})}{(163.84 \times 10^{-9})(32 \times 10^{-9})}$

 $\tau = 0.293 \, \text{MPa}$

- $y = 16 \text{ mm}, \quad z = -30 \text{ mm}, \quad Q = 0$ Point *c*:
 - $\sigma = \frac{10 \times 10^3}{1920 \times 10^{-6}} \frac{(135)(16 \times 10^{-3})}{163.84 \times 10^{-9}} \frac{(110)(-30 \times 10^{-3})}{576 \times 10^{-9}}$
- $\sigma = -2.25 \, \text{MPa}$
 - $\tau = 0$



For the post and loading shown, determine the principal stresses, principal planes, and maximum shearing stress at point H.

SOLUTION

Components of force at point *C*:

$$F_x = 50 \cos 30^\circ = 43.301 \text{ kN}$$

$$F_z = -50 \sin 30^\circ = -25 \text{ kN}$$

$$F_y = -120 \text{ kN}$$

Section forces and couples at the section containing points *H* and *K*:

$$P = 120 \text{ kN}$$
 (compression)

$$V_x = 43.301 \text{ kN}$$

$$V_{7} = -25 \text{ kN}$$

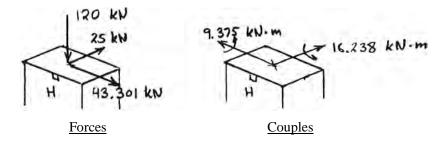
$$M_x = -(25)(0.375) = -9.375 \text{ kN} \cdot \text{m}$$

$$M_{v} = 0$$

$$M_z = -(43.301)(0.375) = -16.238 \text{ kN} \cdot \text{m}$$

$$A = (100)(150) = 15 \times 10^3 \text{ mm}^2 = 15 \times 10^{-3} \text{ m}^2$$

$$I_x = \frac{1}{12}(150)(100)^3 = 12.5 \times 10^6 \,\text{mm}^4 = 12.5 \times 10^{-6} \,\text{m}^4$$



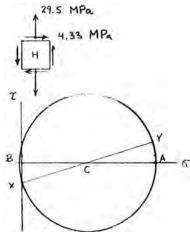
PROBLEM 8.49 (Continued)

Stresses at point *H*:

$$\sigma_H = -\frac{P}{A} - \frac{M_x z}{I_x} = -\frac{(120 \times 10^3)}{15 \times 10^{-3}} - \frac{(-9.375 \times 10^3)(50 \times 10^{-3})}{12.5 \times 10^{-6}} = 29.5 \text{ MPa}$$

$$\tau_H = \frac{3}{2} \frac{V_x}{A} = \frac{3}{2} \frac{43.301 \times 10^3}{15 \times 10^{-3}} = 4.33 \text{ MPa}$$

Use Mohr's circle.



$$\sigma_{\text{ave}} = \frac{1}{2}\sigma_{H} = 14.75 \text{ MPa}$$

$$R = \sqrt{\left(\frac{29.5}{2}\right)^{2} + 4.33^{2}} = 15.37 \text{ MPa}$$

$$\sigma_{a} = \sigma_{\text{ave}} + R$$

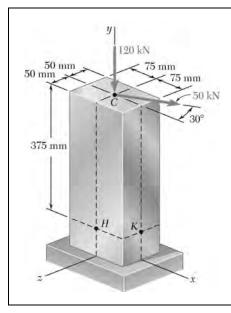
$$\sigma_{b} = \sigma_{\text{ave}} - R$$

$$\tan 2\theta_{p} = \frac{2\tau_{H}}{-\sigma_{H}} = -0.2936$$

$$\sigma_{a} = 30.1 \text{ MPa} \blacktriangleleft$$

$$\theta_a = -8.2^{\circ}$$
 $\theta_b = 81.8^{\circ}$

$$\tau_{\rm max} = R$$
 $\tau_{\rm max} = 15.37 \; {\rm MPa} \; \blacktriangleleft$



For the post and loading shown, determine the principal stresses, principal planes, and maximum shearing stress at point K.

SOLUTION

Components of force at point *C*:

$$F_x = 50\cos 30^\circ = 43.301 \,\mathrm{kN}$$

$$F_z = -50 \sin 30^\circ = -25 \text{ kN}$$
 $F_y = -120 \text{ kN}$

Section forces and couples at the section containing points *H* and *K*:

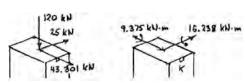
$$P = 120 \text{ kN (compression)}$$

$$V_x = 43.301 \,\text{kN}, \qquad V_z = -25 \,\text{kN}$$

$$M_x = -(25)(0.375) = -9.375 \text{ kN} \cdot \text{m}$$

$$M_{v} = 0$$

$$M_z = -(43.301)(0.375) = -16.238 \text{ kN} \cdot \text{m}$$



$$A = (100)(150) = 15 \times 10^{3} \text{ mm}^{2}$$

$$= 15 \times 10^{-3} \text{ m}^{2}$$

$$I_{z} = \frac{1}{12}(100)(150)^{3} = 28.125 \times 10^{6} \text{ mm}^{4}$$

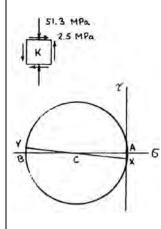
 $= 28.125 \times 10^{-6} \text{ m}^4$

PROBLEM 8.50 (Continued)

Stresses at point *K*:

$$\sigma_K = -\frac{P}{A} + \frac{M_z x}{I_z} = -\frac{120 \times 10^3}{15 \times 10^{-3}} + \frac{(-16.238 \times 10^3)(75 \times 10^{-3})}{28.125 \times 10^{-6}} = -51.3 \text{ MPa}$$

$$\tau_K = \frac{3}{2} \frac{V_z}{A} = \frac{3}{2} \frac{25 \times 10^3}{15 \times 10^{-3}} = 2.5 \text{ MPa}$$



Use Mohr's circle.

$$\sigma_{\text{ave}} = \frac{1}{2}\sigma_K = -25.65 \text{ MPa}$$

$$R = \sqrt{\left(\frac{51.3}{2}\right)^2 + (2.5)^2} = 25.77 \text{ MPa}$$

$$\sigma_a = \sigma_{\text{ave}} + R$$

$$\sigma_a = 0.12 \text{ MPa} \blacktriangleleft$$

$$\sigma_b = \sigma_{\text{ave}} - R$$

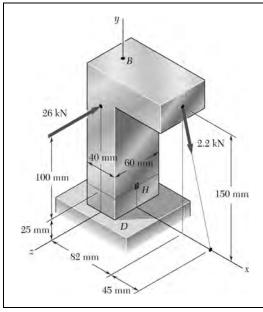
$$\sigma_b = -51.4 \text{ MPa}$$

$$\tan 2\theta_p = \frac{2\tau_K}{-\sigma_K} = 0.09747$$

$$\theta_a = 2.8^{\circ}$$
 $\theta_b = 92.8^{\circ}$

$$\tau_{\rm max} = R$$

$$\tau_{\rm max} = 25.8 \, \mathrm{MPa} \, \blacktriangleleft$$



Two forces are applied to the small post BD as shown. Knowing that the vertical portion of the post has a cross section of 40×60 mm, determine the principal stresses, principal planes, and maximum shearing stress at point H.

SOLUTION

Components of 2.2 kN force:

$$F_x = \frac{(2.2 \times 10^3)(45)}{156.6} = 632.2 \text{ N}$$

$$F_y = -\frac{(2.2 \times 10^3)(150)}{156.6} = -2107.3 \text{ N}$$

Moment arm of 2.2 kN force:

$$\vec{r} = 82\vec{i} + (150 - 25)\vec{j}$$

Moment of 2.2 kN force:

$$\overrightarrow{M} = \begin{vmatrix} \overrightarrow{i} & \overrightarrow{j} & \overrightarrow{k} \\ 82 & 125 & 0 \\ 632.2 & -21073 & 0 \end{vmatrix} = -251823.6\overrightarrow{k} \text{ N} \cdot \text{mm} = -251.8\overrightarrow{k} \text{ N} \cdot \text{m}$$

At the section containing point H,

sining point *H*,
$$P = -2107.3 \text{ N}$$
 $V_x = 632.2 \text{ N}$

$$V_z = -26 \text{ kN}, \qquad M_z = -251.8 \text{ N} \cdot \text{m}, \qquad M_x = -(0.100)(26) = -2.6 \text{ kN} \cdot \text{m}$$

$$M_x = -(0.100)(26) = -2.6 \text{ kN} \cdot \text{m}$$

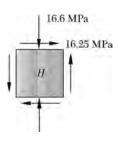
$$A = (40)(60) = 2400 \text{ mm}^2$$

$$A = (40)(60) = 2400 \text{ mm}^2$$
 $I_z = \frac{1}{12}(60)(40)^3 = 32 \times 10^4 \text{ mm}^4$

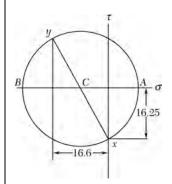
$$\sigma_H = \frac{P}{A} + \frac{M_z x}{I_z} = -\frac{2107.3}{2400 \times 10^{-6}} + \frac{(-251.8)(0.020)}{32 \times 10^{-8}} = -16.6 \text{ MPa}$$

$$\tau_H = \frac{3}{2} \frac{V_z}{A} = \frac{3}{2} \frac{-26 \times 10^3}{2400 \times 10^{-6}} = 16.25 \text{ MPa}$$

$$\sigma_c = -\frac{16.6}{2} = -8.3 \text{ MPa}$$



PROBLEM 8.51 (Continued)



$$R = \sqrt{\left(\frac{16.6}{2}\right)^2 + (16.25)^2} = 18.24 \text{ MPa}$$

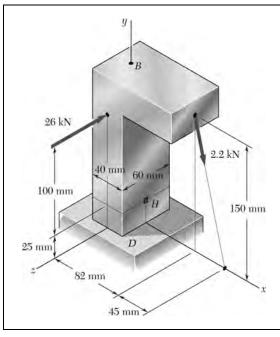
$$\sigma_a = \sigma_c + R = 9.94 \text{ MPa}$$

$$\sigma_b = \sigma_c - R = -26.54 \text{ MPa}$$

$$\tan 2\theta_p = \frac{2\tau_H}{|\sigma_H|} = \frac{(2)(16.25)}{16.6} = 1.958$$

$$\theta_a = 31.47^{\circ}, \quad \theta_b = 121.47^{\circ}$$

$$\tau_{\text{max}} = R = 18.24 \text{ MPa}$$



Solve Prob. 8.51, assuming that the magnitude of the 26 kN force is reduced to 6.5 kN.

PROBLEM 8.51 Two forces are applied to the small post BD as shown. Knowing that the vertical portion of the post has a cross section of 38×60 mm, determine the principal stresses, principal planes, and maximum shearing stress at point H.

SOLUTION

Components of 2.2 kN force:

$$F_x = \frac{(2.2 \times 10^3)(45)}{156.6} = 632.2 \text{ N}$$

 $F_y = \frac{(2.2 \times 10^3)(150)}{156.6} = -2107.3 \text{ N}$

Moment arm of 2.2 kN force:

$$\vec{r} = 82\vec{i} + (150 - 25)\vec{j}$$

$$\vec{M} = \begin{vmatrix} \vec{i} & \vec{j} & \vec{k} \\ 82 & 125 & 0 \\ 632.2 & -21073 & 0 \end{vmatrix} = -251823.6\vec{k} \text{ kN} \cdot \text{mm} = 251.8\vec{k} \text{ N} \cdot \text{m}$$

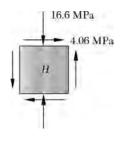
At the section containing point
$$H$$
, $P = -2107.3 \text{ N}$ $V_x = 632.2 \text{ N}$ $V_z = -6500 \text{ N}$, $M_z = -251.8 \text{ N} \cdot \text{m}$, $M_x = -(0.100)(6500) = 650 \text{ N} \cdot \text{m}$

$$A = (40)(60) = 2400 \text{ mm}^2$$
 $I_z = \frac{1}{12}(60)(40)^3 = 32 \times 10^4 \text{ mm}^4$

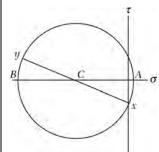
$$\sigma_H = \frac{P}{A} + \frac{M_z x}{I_z} = -\frac{2107.3}{2400 \times 10^{-6}} + \frac{(-251.8)(0.020)}{32 \times 10^{-8}} = -16.6 \text{ MPa}$$

$$\tau_H = \frac{3}{2} \frac{V_z}{A} = \frac{3}{2} \frac{6500}{2400 \times 10^{-6}} = 4.06 \text{ MPa}$$

$$\sigma_c = \frac{1}{2}\sigma_H = -8.3 \text{ MPa}$$



PROBLEM 8.52 (Continued)



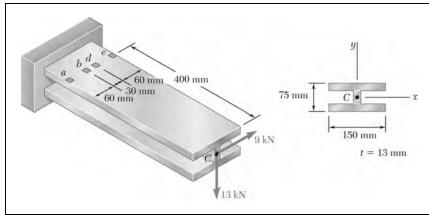
$$R = \sqrt{\left(\frac{16.6}{2}\right)^2 + (4.06)^2} = 9.24 \text{ MPa}$$

$$\sigma_a = \sigma_c + R = 0.94 \text{ MPa}$$

$$\sigma_b = \sigma_c - R = -17.54 \text{ MPa}$$

$$\tan 2\theta_p = \frac{2\tau_H}{|\sigma_H|} = \frac{(2)(4.06)}{16.6} = 0.4891$$

$$\theta_a = 13.03^{\circ}$$
 $\theta_b = 103.03^{\circ}$ $\tau_{\text{max}} = R = 9.24 \text{ MPa}$



Three steel plates, each 13 mm thick, are welded together to form a cantilever beam. For the loading shown, determine the normal and shearing stresses at points a and b.

SOLUTION

Equivalent force-couple system at section containing points a and b.

$$F_x = 9 \text{ kN}, \quad F_y = -13 \text{ kN}, \quad F_z = 0$$

$$M_x = (0.400)(13 \times 10^3) = 5200 \text{ N} \cdot \text{m}$$

$$M_y = (0.400)(9 \times 10^3) = 3600 \text{ N} \cdot \text{m}$$

$$M_z = 0$$

$$A = (2)(150)(13) + (13)(75 - 26)$$

$$= 4537 \text{ mm}^2$$

$$= 4537 \times 10^{-6} \text{m}^2$$

$$I_x = 2 \left[\frac{1}{12} (150)(13)^3 + (150)(13)(37.5 - 6.5)^2 \right] + \frac{1}{12} (13)(75 - 26)^3$$

$$= 3.9303 \times 10^6 \text{ mm}^4$$

$$= 3.9303 \times 10^{-6} \text{ m}^4$$

$$I_y = 2 \cdot \frac{1}{12} (13)(150)^3 + \frac{1}{12} (75 - 26)(13)^3$$

$$= 7.3215 \times 10^6 \text{ mm}^4 = 7.3215 \times 10^{-6} \text{ mm}^4$$

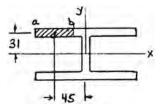
For point a,

$$Q_{\rm r} = 0, \quad Q_{\rm v} = 0$$

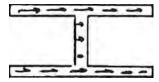
For point b,

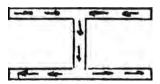
$$A^* = (60)(13) = 780 \text{ mm}^2$$

 $\overline{x} = -45 \text{ mm}$ $\overline{y} = 31 \text{ mm}$
 $Q_x = A^* \overline{y} = 24.18 \times 10^3 \text{ mm}^3 = 24.18 \times 10^{-6} \text{ m}^3$
 $Q_y = A^* \overline{x} = -35.1 \times 10^{-3} \text{ mm}^3 = -35.1 \times 10^{-6} \text{ m}^3$



PROBLEM 8.53 (Continued)





Direction of shearing stress for horizontal and for vertical components of shear.

At point *a*:

$$\sigma = \frac{M_x y}{I_x} - \frac{M_y x}{I_y}$$

$$= \frac{(5200)(37.5 \times 10^{-3})}{3.9303 \times 10^{-6}} - \frac{(3600)(-75 \times 10^{-3})}{7.3215 \times 10^{-6}}$$

$$\sigma = 86.5 \text{ MPa}$$

 $\tau = 0$

At point b:

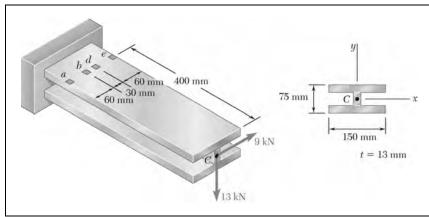
$$\sigma = \frac{M_x y}{I_x} - \frac{M_y x}{I_y}$$

$$= \frac{(5200)(37.5 \times 10^{-3})}{3.9303 \times 10^{-6}} - \frac{(3600)(-15 \times 10^{-3})}{7.3215 \times 10^{-6}} \qquad \sigma = 57.0 \text{ MPa} \blacktriangleleft$$

$$\tau = \frac{|V_x||Q_y|}{I_y t} + \frac{|V_y||Q_x|}{I_x t}$$

$$= \frac{(9 \times 10^3)(35.1 \times 10^{-6})}{(7.3215 \times 10^{-6})(13 \times 10^{-3})} + \frac{(13 \times 10^3)(24.18 \times 10^{-6})}{(3.9303 \times 10^{-6})(13 \times 10^{-3})}$$

$$= 3.32 \text{ MPa} + 6.15 \text{ MPa} \qquad \tau = 9.47 \text{ MPa} \blacktriangleleft$$



Three steel plates, each 13 mm thick, are welded together to form a cantilever beam. For the loading shown, determine the normal and shearing stresses at points d and e.

SOLUTION

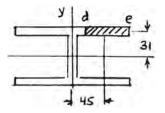
Equivalent force-couple system at section containing points d and e.

$$\begin{split} F_x &= 9 \text{ kN}, \quad F_y = -13 \text{ kN}, \quad F_z = 0 \\ M_x &= (0.400)(13 \times 10^3) = 5200 \text{ N} \cdot \text{m} \\ M_y &= (0.400)(9 \times 10^3) = 3600 \text{ N} \cdot \text{m} \\ M_z &= 0 \\ A &= (2)(150)(13) + (13)(75 - 26) \\ &= 4537 \text{ mm}^2 \\ &= 4537 \times 10^{-6} \text{ m}^2 \\ I_x &= 2 \bigg[\frac{1}{12} (150)(13)^3 + (150)(13)(37.5 - 6.5)^2 \bigg] + \frac{1}{12} (13)(75 - 26)^3 \\ &= 3.9303 \times 10^6 \text{ mm}^4 \\ &= 3.9303 \times 10^{-6} \text{ m}^4 \\ I_y &= 2 \bigg[\frac{1}{12} (13)(150)^3 \bigg] + \frac{1}{12} (75 - 26)(13)^3 \\ &= 7.3215 \times 10^6 \text{ mm}^4 = 7.3215 \times 10^{-6} \text{ m}^4 \end{split}$$

For point d,

$$A^* = (60)(13) = 780 \text{ mm}^2$$

 $\overline{x} = 45 \text{ mm}$ $\overline{y} = 31 \text{ mm}$
 $Q_x = A^* \overline{y} = 24.18 \times 10^3 \text{ mm}^3 = 24.18 \times 10^{-6} \text{ m}^3$
 $Q_y = A^* \overline{x} = 35.1 \times 10^3 \text{ mm}^3 = 35.1 \times 10^{-6} \text{ m}^3$



For point e,

$$Q_x = 0, \quad Q_y = 0$$

PROBLEM 8.54 (Continued)

At point *d*:

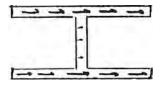
$$\sigma = \frac{M_x y}{I_x} - \frac{M_y x}{I_y}$$

$$= \frac{(5200)(37.5 \times 10^{-3})}{3.9303 \times 10^{-6}} - \frac{(3600)(15 \times 10^{-3})}{7.3215 \times 10^{-6}}$$

 σ = 42.2 MPa

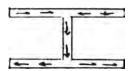
Due to V_x :

$$\tau = \frac{|V_x||Q_y|}{I_y t} = \frac{(9000)(35.1 \times 10^{-6})}{(7.3215 \times 10^{-6})(13 \times 10^{-3})}$$
$$= 3.32 \text{ MPa} \rightarrow$$



Due to V_{v} :

$$\tau = \frac{|V_y||Q_x|}{I_x t} = \frac{(13000)(24.18 \times 10^{-6})}{(3.9303 \times 10^{-6})(13 \times 10^{-3})} = 6.15 \text{ MPa} \leftarrow$$



By superposition, the net value is

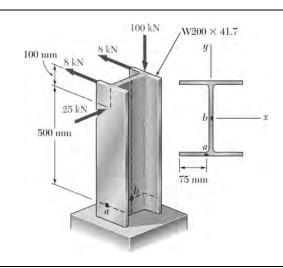
 $\tau = 2.83 \text{ MPa}$

At point *e*:

$$\sigma = \frac{M_x y}{I_x} - \frac{M_y x}{I_y} = \frac{(5200)(37.5 \times 10^{-3})}{3.9303 \times 10^{-6}} - \frac{(3600)(75 \times 10^{-3})}{7.3215 \times 10^{-6}}$$

 σ = 12.74 MPa

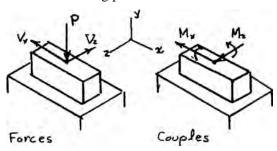
 $\tau = 0$



Four forces are applied to a W200 \times 41.7 rolled beam as shown. Determine the principal stresses and maximum shearing stress at point a.

SOLUTION

Calculate forces and couples at section containing point a.



Section properties.

$$A = 5310 \text{ mm}^2$$
 $d = 205 \text{ mm}$
 $b_f = 166 \text{ mm}$ $\tau_f = 11.8 \text{ mm}$
 $t_w = 7.2 \text{ mm}$

$$P = 100 \text{ kN}, \quad V_x = -16 \text{ kN}, \quad V_y = 25 \text{ kW}$$

$$M_x = -(500 \times 10^{-3})(25 \times 10^3)$$

$$-\left(\frac{205}{2} \times 10^{-3}\right)(100 \times 10^3)$$

$$= -122.75 \times 10^3 \text{ N} \cdot \text{m}$$

$$M_y = (500 + 100)(10^{-3})(16 \times 10^3)$$

$$= 9.6 \times 10^3 \text{ N} \cdot \text{m}$$

 $I_x = 40.9 \times 10^6 \text{ mm}^4 = 40.9 \times 10^{-6} \text{ m}^4, \quad I_y = 9.01 \times 10^6 \text{ mm}^4 = 9.01 \times 10^{-6} \text{ m}^4$

Point a:
$$x_a = -\frac{166}{2} + 75 = -8 \text{ mm}, \quad y_a = -\frac{205}{2} = -102.5 \text{ mm}$$

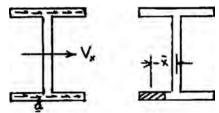
$$\sigma_a = -\frac{P}{A} + \frac{M_x y_a}{I_x} + \frac{M_y x_a}{I_y}$$

$$= -\frac{100 \times 10^3}{5310 \times 10^{-6}} + \frac{(-22.75 \times 10^3)(-102.5 \times 10^{-3})}{40.9 \times 10^{-6}} + \frac{(9.6 \times 10^3)(-8 \times 10^{-3})}{9.01 \times 10^{-6}}$$

$$= -18.83 \times 10^6 + 57.01 \times 10^6 - 8.52 \times 10^6 = 29.66 \text{ MPa}$$

PROBLEM 8.55 (Continued)

Searing stress at point a due to V_r :



$$A = (75)(11.8) = 885 \text{ mm}^2$$

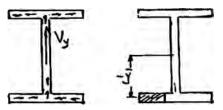
$$\overline{x} = -\frac{166}{2} + \frac{75}{2} = -45.5 \text{ mm}$$

$$Q = -A\overline{x} = 40.2675 \times 10^3 \text{ mm}^2$$

$$\tau_{xz} = \frac{V_x Q}{I_y t} = \frac{(-16 \times 10^3)(40.2675 \times 10^{-6})}{(9.01 \times 10^{-6})(11.8 \times 10^{-3})}$$

$$= -6.060 \text{ MPa}$$

Shearing stress at point a due to V_{v}



$$A = (75)(11.8) = 885 \text{ mm}^2$$

$$\overline{y} = -\frac{205}{2} + \frac{11.8}{2} = -96.6 \text{ mm}$$

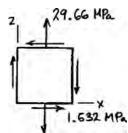
$$Q = -A\overline{y} = -85.491 \times 10^3 \text{ mm}^2$$

$$\tau_{xz} = -\frac{V_y Q}{I_x t} = -\frac{(25 \times 10^3)(-85.491 \times 10^{-6})}{(40.9 \times 10^{-6})(11.8 \times 10^{-3})}$$

$$= 4.428 \text{ MPa}$$

Combined shearing stress:

$$\tau_a = -6.060 + 4.428 = -1.632 \text{ MPa}$$



$$\sigma_{\text{ave}} = \frac{29.66 + 0}{2} = 14.83 \text{ MPa}$$

$$R = \sqrt{\left(\frac{29.66 - 0}{2}\right)^2 + (1.632)^2} = 14.92 \text{ MPa}$$

$$\sigma_{\max} = \sigma_{\text{ave}} + R$$

$$\sigma_{\min} = \sigma_{\text{ave}} - R$$

$$\sigma_{\min} = \sigma_{\text{ave}} - R$$

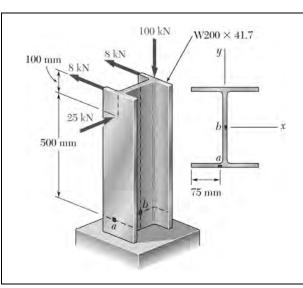
$$\tau_{\rm max} = R$$

$$\sigma_{\rm max} = 29.8 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\sigma_{\min} = -0.09 \text{ MPa} \blacktriangleleft$$

$$O_{\min} = -0.07 \text{ WH } a$$

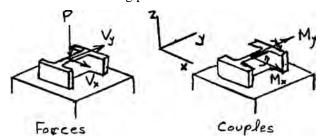
$$\sigma_{\rm max}$$
 = 14.92 MPa \blacktriangleleft



Four forces are applied to a W200 \times 41.7 rolled beam as shown. Determine the principal stresses and maximum shearing stress at point b.

SOLUTION

Calculate forces and couples at section containing point b.



Section properties.

$$A = 5310 \text{ mm}^4$$
 $d = 205 \text{ mm}$
 $b_f = 166 \text{ mm}$ $\tau_f = 11.8 \text{ mm}$
 $t_w = 7.2 \text{ mm}$

$$P = 100 \text{ kN}, \quad V_x = -16 \text{ kN}, \quad V_y = 25 \text{ kN}$$

$$M_x = -(500 \times 10^{-3})(25 \times 10^3)$$

$$-\left(\frac{205}{2} \times 10^{-3}\right)(100 \times 10^3)$$

$$= -22.75 \times 10^3 \text{ N} \cdot \text{m}$$

$$M_y = (500 + 100)(10^{-3})(16 \times 10^3)$$

$$= 9.6 \times 10^3 \text{ N} \cdot \text{m}$$

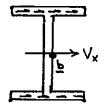
$$I_x = 40.9 \times 10^6 \text{ mm}^4 = 40.9 \times 10^{-6} \text{ m}^4, \quad I_y = 9.01 \times 10^6 \text{ mm}^4 = 9.01 \times 10^{-6} \text{ m}^4$$

Point b:
$$x_b = 0$$
, $y_b = 0$

$$\sigma_b = -\frac{P}{A} + \frac{M_x y_b}{I_x} + \frac{M_y x_b}{I_y} = -\frac{100 \times 10^3}{5310 \times 10^{-6}} = -18.83 \text{ MPa}$$

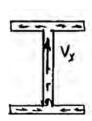
PROBLEM 8.56 (Continued)

Shearing stress at point b due to V_x .



$$\tau_{xz} = 0$$

Shearing stress at point b due to V_y .





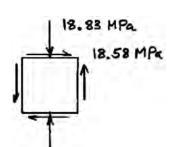
	$A (\mathrm{mm}^2)$	\overline{y} (mm)	$A\overline{y} (10^3 \text{ mm}^3)$
1	1958.8	96.6	189.220
2	653.04	45.35	29.615
Σ			218.835

$$Q = \Sigma A \overline{y} = 218.835 \times 10^{3} \text{ mm}^{3} = 218.835 \times 10^{-6} \text{ m}^{3}$$

$$\tau_{b} = \frac{V_{y}Q}{I_{x}I_{w}} = \frac{(25 \times 10^{3})(218.835 \times 10^{-6})}{(40.9 \times 10^{-6})(7.2 \times 10^{-3})} = 18.58 \text{ MPa}$$

 $\sigma_{\min} = \sigma_{\text{ave}} - R$

 $\tau_{\text{max}} = R$



18.83 MPa
$$\sigma_{\text{ave}} = \frac{-18.83 + 0}{2} = -9.415 \text{ MPa}$$

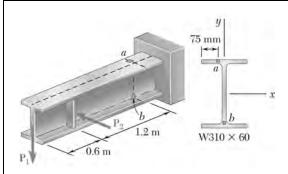
$$R = \sqrt{\left(\frac{-18.83 - 0}{2}\right)^2 + (18.58)^2} = 20.829 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R$$

$$\sigma_{\rm max}$$
 =11.41 MPa \blacktriangleleft

$$\sigma_{\min} = -30.2 \text{ MPa} \blacktriangleleft$$

$$\sigma_{\rm max} = 20.8 \ {\rm MPa} \blacktriangleleft$$



Two forces \mathbf{P}_1 and \mathbf{P}_2 are applied as shown in directions perpendicular to the longitudinal axis of a W310 × 60 beam. Knowing that $\mathbf{P}_1 = 25 \text{ kN}$ and $\mathbf{P}_2 = 24 \text{ kN}$, determine the principal stresses and the maximum shearing stress at point a.

SOLUTION

At the section containing points a and b,

$$M_x = (1.8)(25) = 45 \text{ kN} \cdot \text{m}$$

 $M_y = -(1.2)(24) = -28.8 \text{ kN} \cdot \text{m}$
 $V_x = -24 \text{ kN}$ $V_y = -25 \text{ kN}$

For W310 \times 60 rolled steel section,

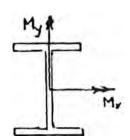
$$d = 302 \text{ mm}, \qquad b_f = 203 \text{ mm}, \qquad t_f = 13.1 \text{ mm}, \qquad t_w = 7.49 \text{ mm}$$

$$I_x = 128 \times 10^6 \text{ mm}^4 = 128 \times 10^{-6} \text{ m}^4, \qquad I_y = 18.4 \times 10^6 \text{ mm}^4 = 18.4 \times 10^{-6} \text{ m}^4$$

Normal stress at point *a*:

$$x = -\frac{b_f}{2} + 75 = -26.5 \text{ mm}$$

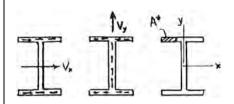
 $y = \frac{1}{2}d = 151 \text{ mm}$



$$\sigma_z = \frac{M_x y}{I_x} - \frac{M_y x}{I_y} = \frac{(45 \times 10^3)(151 \times 10^{-3})}{128 \times 10^{-6}} - \frac{(-28.8 \times 10^3)(-26.5 \times 10^{-3})}{18.4 \times 10^{-6}}$$
= 53.086 MPa - 41.478 MPa = 11.608 MPa

Shearing stress at point *a*:

$$\tau_{xz} = -\frac{V_x A^* \overline{x}}{I_y t_f} - \frac{V_y A^* \overline{y}}{I_x t_f}$$



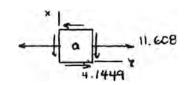
$$A^* = (75 \times 10^{-3})(13.1 \times 10^{-3}) = 982.5 \times 10^{-6} \text{ m}^2$$

$$\overline{x} = -\frac{b_f}{2} + \frac{75}{2} = -64 \text{ mm}$$

$$\overline{y} = \frac{d}{2} - \frac{t_f}{2} = 144.45 \text{ mm}$$

$$\begin{split} \tau_{_{XZ}} &= -\frac{(-24\times10^3)(982.5\times10^{-6})(-64\times10^{-3})}{(18.4\times10^{-6})(13.1\times10^{-3})} - \frac{(-25\times10^3)(982.5\times10^{-6})(144.45\times10^{-3})}{(128\times10^{-6})(13.1\times10^{-3})} \\ &= -6.2609 \text{ MPa} + 2.1160 \text{ MPa} = -4.1449 \text{ MPa} \end{split}$$

PROBLEM 8.57 (Continued)



$$\sigma_{\text{ave}} = \frac{11.608}{2} = 5.804 \text{ MPa}$$

$$R = \sqrt{\left(\frac{11.608}{2}\right)^2 + (4.1449)^2} = 7.1321 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R$$

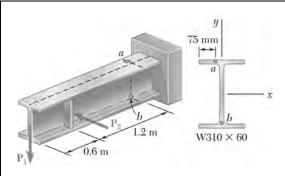
$$\sigma_{\rm max} = 12.94 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\sigma_{\min} = \sigma_{\text{ave}} - R$$

$$\sigma_{\min} = -1.33 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\tau_{\rm max} = R$$

$$\tau_{\rm max} = 7.13 \ {
m MPa}$$



Two forces \mathbf{P}_1 and \mathbf{P}_2 are applied as shown in directions perpendicular to the longitudinal axis of a W310 × 60 beam. Knowing that $\mathbf{P}_1 = 25 \text{ kN}$ and $\mathbf{P}_2 = 24 \text{ kN}$, determine the principal stresses and the maximum shearing stress at point b.

SOLUTION

At the section containing points a and b,

$$M_x = (1.8)(25) = 45 \text{ kN} \cdot \text{m}$$

 $M_y = -(1.2)(24) = -28.8 \text{ kN} \cdot \text{m}$
 $V_x = -24 \text{ kN}, \qquad V_y = -25 \text{ kN}$

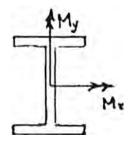
For W310 \times 60 rolled steel section,

$$d = 302 \text{ mm}, \qquad b_f = 203 \text{ mm}, \qquad t_f = 13.1 \text{ mm}, \qquad t_w = 7.49 \text{ mm}$$

$$I_x = 128 \times 10^6 \text{ mm}^4 = 128 \times 10^{-6} \text{ m}^4, \qquad I_y = 18.4 \times 10^6 \text{ mm}^4 = 18.4 \times 10^{-6} \text{ m}^4$$

Normal stress at point *b*:

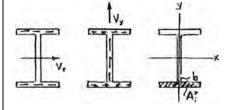
$$x \approx 0$$
, $y = -\frac{1}{2}d + t_f = -137.9 \text{ mm}$



$$\sigma_z = \frac{M_x y}{I_x} - \frac{M_y x}{I_y} = \frac{(45 \times 10^3)(-137.9 \times 10^{-3})}{128 \times 10^{-6}} - 0$$
$$= -48.480 \text{ MPa}$$

Shearing stress at point *b*:

$$\tau_{yz} = -\frac{V_y A^* \overline{y}}{I_x t_w}$$

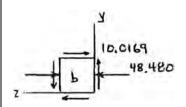


$$A^* = A_f = b_f t_f = 2659.3 \text{ mm}^2$$

= $2659.3 \times 10^{-6} \text{ m}^2$
 $\overline{x} = 0, \qquad \overline{y} = -\frac{1}{2}d + \frac{1}{2}t_f = -144.45 \text{ mm}$

$$\tau_{yz} = -\frac{(-25\times10^3)(2659.3\times10^{-6})(-144.45\times10^{-3})}{(128\times10^{-6})(7.49\times10^{-3})} = -10.0169~\mathrm{MPa}$$

PROBLEM 8.58 (Continued)



$$\sigma_{\text{ave}} = -\frac{48.480}{2} = -24.240 \text{ MPa}$$

$$R = \sqrt{\left(\frac{48.48}{2}\right)^2 + (10.0169)^2} = 26.228 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R$$

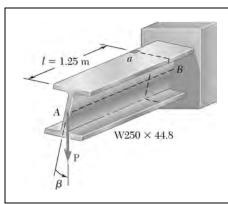
$$\sigma_{\rm max} = 1.99 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\sigma_{\min} = \sigma_{\text{ave}} - R$$

$$\sigma_{\min} = -50.5 \text{ MPa} \blacktriangleleft$$

$$\tau_{\rm max} = R$$

$$\tau_{\rm max} = 26.2 \ {
m MPa}$$



A vertical force **P** is applied at the center of the free end of cantilever beam AB. (a) If the beam is installed with the web vertical ($\beta = 0$) and with its longitudinal axis AB horizontal, determine the magnitude of the force **P** for which the normal stress at point a is +120 MPa. (b) Solve part a, assuming that the beam is installed with $\beta = 3^{\circ}$.

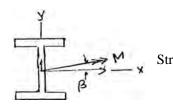
SOLUTION

For W250 \times 44.8 rolled steel section,

$$S_x = 531 \times 10^3 \text{ mm}^3 = 531 \times 10^{-6} \text{ m}^3$$

$$S_v = 94.2 \times 10^3 \text{ mm}^3 = 94.2 \times 10^{-6} \text{ m}^3$$

At the section containing point a,



$$M_x = Pl\cos\beta, \qquad M_y = Pl\sin\beta$$

$$\sigma = \frac{M_x}{S_x} + \frac{M_y}{S_y} = \frac{Pl\cos\beta}{S_x} + \frac{Pl\sin\beta}{S_y}$$

Allowable load.
$$P_{\text{all}} = \frac{\sigma_{\text{all}}}{l} \left[\frac{\cos \beta}{S_x} + \frac{\sin \beta}{S_y} \right]^{-1}$$

(a)
$$\underline{\beta = 0}$$
: $P_{\text{all}} = \frac{120 \times 10^6}{1.25} \left[\frac{1}{531 \times 10^{-6}} + 0 \right]^{-1} = 51.0 \times 10^3 \text{ N}$

 $P_{\text{all}} = 51.0 \text{ kN}$

(b)
$$\underline{\beta = 3^{\circ}}$$
: $P_{\text{all}} = \frac{120 \times 10^{6}}{1.25} \left[\frac{\cos 3^{\circ}}{531 \times 10^{-6}} + \frac{\sin 3^{\circ}}{94.2 \times 10^{-6}} \right]^{-1} = 39.4 \times 10^{3} \text{ N}$

 $P_{\rm all} = 39.4 \, \rm kN \, \blacktriangleleft$



A force **P** is applied to a cantilever beam by means of a cable attached to a bolt located at the center of the free end of the beam. Knowing that **P** acts in a direction perpendicular to the longitudinal axis of the beam, determine (a) the normal stress at point a in terms of P, b, h, l, and β , (b) the values of β for which the normal stress at a is zero.

SOLUTION

$$I_{x} = \frac{1}{12}bh^{3} \quad I_{y} = \frac{1}{12}hb^{3}$$

$$\sigma = \frac{M_{x}(h/2)}{I_{x}} - \frac{M_{y}(b/2)}{I_{y}}$$

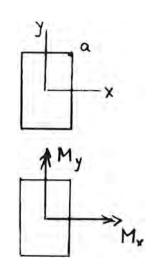
$$= \frac{6M_{x}}{bh^{2}} - \frac{6M_{y}}{hb^{2}}$$

$$P = P\sin\beta \mathbf{i} - P\cos\beta \mathbf{j} \quad \mathbf{r} = l\mathbf{k}$$

$$M = \mathbf{r} \times P = l\mathbf{k} \times (P\sin\beta \mathbf{i} - P\cos\beta \mathbf{j})$$

$$= Pl\cos\beta \mathbf{i} + Pl\sin\beta \mathbf{j}$$

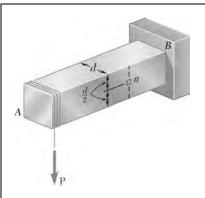
$$M_{x} = Pl\cos\beta \qquad M_{y} = Pl\sin\beta$$



(a)
$$\sigma = \frac{6 Pl \cos \beta}{bh^2} - \frac{6 Pl \sin \beta}{hb^2} = \frac{6 Pl}{bh} \left[\frac{\cos \beta}{h} - \frac{\sin \beta}{b} \right]$$

(b)
$$\sigma = 0$$
 $\frac{\cos \beta}{h} - \frac{\sin \beta}{b} = 0$ $\tan \beta = \frac{b}{h}$

$$\beta = \tan^{-1} \left(\frac{b}{h} \right)$$



PROBLEM 8.61*

A 5-kN force **P** is applied to a wire that is wrapped around bar AB as shown. Knowing that the cross section of the bar is a square of side d = 40 mm, determine the principal stresses and the maximum shearing stress at point a.

SOLUTION

Bending: Point *a* lies on the neutral axis.

$$\sigma = 0$$

Torsion:
$$\tau = \frac{T}{c_1 a b^2}$$
 where $a = b = d$

and
$$c_1 = 0.208$$
 for a square section.

Since
$$T = \frac{Pd}{2}$$
, $\tau_T = \frac{P}{0.416 d^2} = 2.404 \frac{P}{d^2}$.

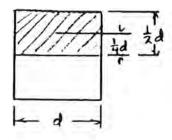
Transverse shear:

$$V = P I = \frac{1}{12}d^4$$

$$A = \frac{1}{2}d^2 \overline{y} = \frac{1}{4}d Q = A\overline{y} = \frac{1}{8}d^3$$

$$t = d$$

$$\tau_V = \frac{VQ}{It} = 1.5\frac{P}{d^2}$$

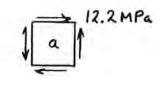


By superposition,

$$\tau = \tau_T + \tau_V = 3.904 \frac{P}{d^2}$$

$$\tau = \frac{(3.904)(5 \times 10^3)}{(40 \times 10^{-3})^2}$$

$$= 12.2 \times 10^6 \text{Pa}$$
12.2 MPa.



By Mohr's circle,

$$\sigma_{\rm max} = 12.2 \, \mathrm{MPa} \, \blacktriangleleft$$

$$\sigma_{\min} = -12.2 \text{ MPa}$$

$$\tau_{\rm max} = 12.2 \, \mathrm{MPa} \, \blacktriangleleft$$

75 mm 150 mm 250 mm 4 mm 40 kN

PROBLEM 8.62*

Knowing that the structural tube shown has a uniform wall thickness of 8 mm, determine the principal stresses, principal planes, and maximum shearing stress at (a) point H, (b) point K.

SOLUTION

At the section containing points H and K,

$$V = 40 \text{ kN}$$
 $M = (40)(0.25) = 10 \text{ kNm}$
 $T = (40)(0.075 - 0.004) = 2.84 \text{ kNm}$

Torsion:

$$a = (142)(92) = 13064 \text{ mm}^2$$

 $\tau = \frac{T}{2ta} = \frac{(2.84 \times 10^6)}{(2)(8)(13064)} = 13.59 \text{ MPa}$



$$Q_H = 0$$

$$Q_K = (75)(50)(25) - (67)(42)(21) = 34656 \text{ mm}^3$$

$$I = \frac{1}{12}(150)(100)^3 - \frac{1}{12}(134)(841)^3 = 5.88 \times 10^6 \text{ mm}^4$$

$$\tau_H = 0 \qquad \tau_K = \frac{VQ_a}{It} = \frac{(40 \times 10^3)(34656)}{(5.88 \times 10^6)(8)} = 29.47 \text{ MPa}$$

Bending:

$$\sigma_H = \frac{Mc}{I} = \frac{(10 \times 10^6)(50)}{5.88 \times 10^6} = 85 \text{ MPa}, \quad \sigma_K = 0$$

(a) Point H:
$$\sigma_c = \frac{85}{2} = 42.5 \text{ MPa}$$

$$R = \sqrt{\left(\frac{85}{2}\right)^2 + (13.59)^2} = 44.62 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_c + R = 87.1 \text{ MPa}$$

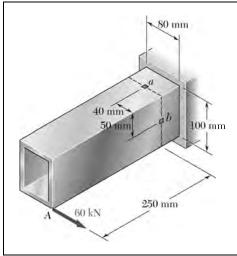
$$\sigma_{\text{min}} = \sigma_c - R = -2.1 \text{ MPa}$$

$$\tan 2\theta_p = \frac{2\tau}{\sigma} = -0.32$$

$$\tau_{\text{max}} = R = 44.6 \text{ MPa}$$

PROBLEM 8.62* (Continued)

(b) Point K: $\sigma = 0$ $\tau = 13.59 + 29.47 = 43.06 MPa$ $\sigma_{\max} = 43.1 \text{ MPa}$ $\sigma_{\min} = -43.1 \text{ MPa}$ $\theta_P = \pm 45^\circ$ $\sigma_{\max} = 43.1 \text{ MPa}$



PROBLEM 8.63*

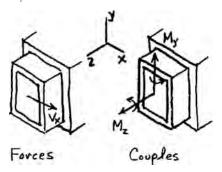
The structural tube shown has a uniform wall thickness of 8 mm. Knowing that the 60-kN load is applied 4 mm above the base of the tube, determine the shearing stress at (a) point a, (b) point b.

SOLUTION

Calculate forces and couples at section containing points a and b.

$$V_x = 60 \text{ kN}$$

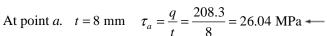
 $M_z = (50 - 4)(60) = 2760 \text{ kN} \cdot \text{mm}$
 $M_y = (60)(250) = 15000 \text{ kN} \cdot \text{mm}$



Shearing stresses due to torque $T = M_z$.

$$a = [80 - (2)(4)][100 - (2)(4)] = 6624 \text{ mm}^2$$

$$q = \frac{M_z}{2a} = \frac{2760 \times 10^3}{(2)(6624)} = 208.3 \text{ N/mm}$$



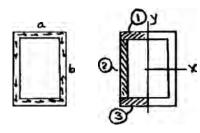
At point b.
$$t = 8 \text{ mm}$$
 $\tau_b = \frac{q}{t} = \frac{208.3}{8} = 26.04 \text{ MPa}$



PROBLEM 8.63* (Continued)

Shearing stresses due to V_x .

At point *a*:



Part	$A(\text{mm}^2)$	$\overline{x}(mm)$	$A\overline{x}(\text{mm}^3)$
1	320	-20	-6400
2	672	36	24192
3	320	-20	-6400
Σ			-36992

$$Q = |\Sigma A\overline{x}| = 36992 \text{ mm}^3$$

$$t = (2)(8) = 16 \text{ mm}$$

$$I_y = \frac{1}{12} (100)(80)^3 - \frac{1}{12} (84)(64)^3 = 2.4317 \times 10^6 \text{ mm}^4$$

$$\tau = \frac{V_x Q}{I_y t} = \frac{(60000)(36992)}{(2.4317 \times 10^6)(16)} = 57.05 \text{ MPa}$$

At point *b*:

$$\tau_b = 0$$

Combined shearing stresses.

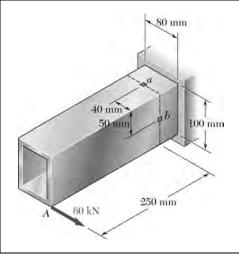
At point a.

$$\tau_a = 26.04 \longrightarrow +57.05 \longrightarrow = 31 \text{ MPa} \longrightarrow$$

$$\tau_b = 26.04 + 0 = 26 \text{ MPa}$$

At point b.

$$\tau_b = 26.04 + 0$$



PROBLEM 8.64*

For the tube and loading of Prob. 8.63, determine the principal stresses and the maximum shearing stress at point b.

PROBLEM 8.63* The structural tube shown has a uniform wall thickness of 8 mm. Knowing that the 60-kN load is applied 4 mm above the base of the tube, determine the shearing stress at (a) point a, (b) point b.

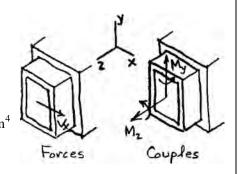
SOLUTION

Calculate forces and couples at section containing point b.

$$V_x = 60 \text{ kN}$$

 $M_z = (50 - 4)(60) = 2760 \text{ kN} \cdot \text{mm}$
 $M_y = (60)(250) = 15000 \text{ kN} \cdot \text{mm}$
 $I_y = \frac{1}{12}(100)(80)^3 - \frac{1}{12}(84)(64)^3 = 2.4317 \times 10^6 \text{ mm}^4$

$$\sigma_b = -\frac{M_y x_b}{I_y} = -\frac{(15 \times 10^6)(40)}{2.4317 \times 10^6} = -246.7 \text{ MPa}$$



Shearing stress at point b due to torque M_z .

$$a = [80 - (2)(4)][100 - (2)(4)] = 6624 \text{ mm}^2$$

$$q = \frac{M_z}{2a} = \frac{2760 \times 10^3}{(2)(6624)} = 208.3 \text{ N/mm}$$

$$\tau = \frac{q}{t} = \frac{208.3}{8} = 26.04 \text{ MPa}$$

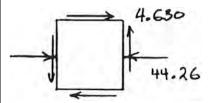
Shearing at point b due to V_r .



$$\tau = 0$$

PROBLEM 8.64* (Continued)

Calculation of principal stresses and maximum shearing stress.



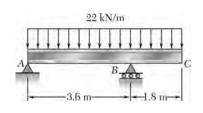
$$\sigma_{\text{ave}} = \frac{-246.7 + 0}{2} = -123.35 \text{ MPa}$$

$$R = \sqrt{\left(\frac{-246.7 - 0}{2}\right)^2 + (26.04)^2} = 126.07 \text{ MPa}$$

$$\sigma_{\text{max}} = \sigma_{\text{ave}} + R = 2.72 \text{ MPa}$$

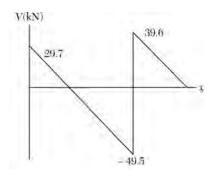
$$\sigma_{\min} = \sigma_{\text{ave}} - R = -249.4 \text{ MPa}$$

$$\tau_{\rm max} = R = 126 \text{ MPa}$$



(a) Knowing that $\sigma_{\rm all} = 165 \, {\rm MPa}$ and $\tau_{\rm all} = 100 \, {\rm MPa}$, select the most economical wide-flange shape that should be used to support the loading shown. (b) Determine the values to be expected for σ_m , τ_m , and the principal stress $\sigma_{\rm max}$ at the junction of a flange and the web of the selected beam.

SOLUTION



+)
$$\Sigma M_B = 0$$
 $-3.6 R_A + (22)(5.4)(0.9) = 0$ $R_A = 29.7 \text{ kN}$
+) $\Sigma M_A = 0$ $3.6 R_B - (22)(5.4)(2.7) = 0$ $R_B = 89.1 \text{ kN}$
| $V|_{\text{max}} = 49.5 \text{ kN}$
| $M|_{\text{max}} = 35.64 \text{ kN} \cdot \text{m}$

$$S_{\text{min}} = \frac{|M|_{\text{max}}}{\sigma_{\text{all}}} = \frac{35.64 \times 10^3}{165 \times 10^6} = 216 \times 10^{-6} \text{ m}^3$$

 Shape
 $S_x(10^3 \text{ mm}^3)$

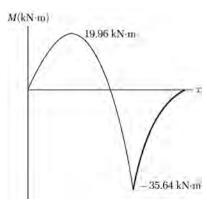
 W 310 × 23.8
 280

 W 250 × 22.3
 228

 W 200 × 26.6
 249

 W 150 × 29.8
 219

 $= 216 \times 10^3 \, \text{mm}^3$



(a) Use W 250×22.3

d = 254 mm

 $t_f = 6.9 \text{ mm}$

 $t_w = 5.8 \text{ mm}$

 $\sigma_m = \frac{|M|_{\text{max}}}{S} = \frac{35.64 \times 10^3}{228 \times 10^{-6}} = 156.31 \text{ MPa}$

 $\tau_m = \frac{|V|_{\text{max}}}{dt_w} = \frac{49.5 \times 10^3}{(0.254)(0.0058)} = 33.6 \text{ MPa}$

PROBLEM 8.65 (Continued)

$$c = \frac{1}{2}d = \frac{254}{2} = 127 \text{ mm} \quad y_b = c - t_f = 127 - 6.9 = 120.1 \text{ mm}$$

$$\sigma_b = \frac{y_b}{c}\sigma_m = \left(\frac{120.1}{127}\right)(156.3) = 147.8 \text{ MPa}$$

$$R = \sqrt{\left(\frac{\sigma_b}{2}\right)^2 + \tau_m^2} = \sqrt{(73.9)^2 + (33.6)^2} = 81.18 \text{ MPa}$$

$$\sigma_{\text{max}} = \frac{\sigma_b}{2} + R = \frac{147.8}{2} + 81.18 = 155.08 \text{ MPa}$$

P 150 mm T = 600 N · m

PROBLEM 8.66

Determine the smallest allowable diameter of the solid shaft ABCD, knowing that $\tau_{\rm all}=60$ MPa and that the radius of disk B is r=80 mm.

SOLUTION

$$\Sigma M_{\text{axis}} = 0: \quad T - Pr = 0 \quad P = \frac{T}{r} = \frac{600}{80 \times 10^{-3}} = 7.5 \times 10^{3}$$

$$R_{A} = R_{C} = \frac{1}{2}P$$

$$= 3.75 \times 10^{3} \text{ N}$$

$$M_{B} = (3.75 \times 10^{3})(150 \times 10^{-3})$$

Bending moment: (See sketch).

 $= 562.5 \text{ N} \cdot \text{m}$

Torque: (See sketch).

Critical section lies at point *B*.

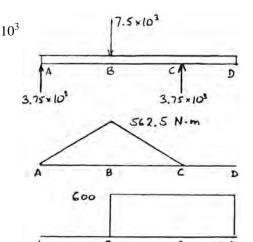
$$M = 562.5 \text{ N} \cdot \text{m}, \quad T = 600 \text{ N} \cdot \text{m}$$

$$\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\left(\sqrt{M^2 + T^2}\right)_{\text{max}}}{\tau_{\text{all}}}$$

$$c^3 = \frac{2}{\pi} \frac{\sqrt{M^2 + T^2}}{\tau_{\text{all}}} = \frac{2}{\pi} \frac{\sqrt{(562.5)^2 + (600)^2}}{60 \times 10^6}$$

$$= 8.726 \times 10^{-6} \text{m}^3$$

 $c = 20.58 \times 10^{-3} \text{m}$ $d = 2c = 41.2 \times 10^{-3} \text{m}$



d = 41.2 mm

Using the notation of Sec. 8.3 and neglecting the effect of shearing stresses caused by transverse loads, show that the maximum normal stress in a circular shaft can be expressed as follows:

$$\sigma_{\text{max}} = \frac{c}{J} \left[\left(M_y^2 + M_z^2 \right)^{1/2} + \left(M_y^2 + M_z^2 + T^2 \right)^{1/2} \right]_{\text{max}}$$

SOLUTION

Maximum bending stress:

$$\sigma_m = \frac{|M|c}{I} = \frac{\sqrt{M_y^2 + M_z^2}c}{I}$$

Maximum torsional stress:

$$\tau_m = \frac{Tc}{J}$$

$$\frac{\sigma_m}{2} = \frac{\sqrt{M_y^2 + M_z^2}c}{2J} = \frac{c}{J}\sqrt{M_y^2 + M_z^2}$$

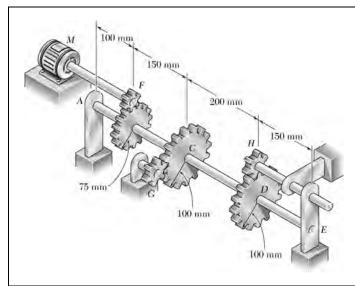
Using Mohr's circle,

$$R = \sqrt{\left(\frac{\sigma_m}{2}\right)^2 + \tau_m^2} = \sqrt{\frac{c^2}{J^2} \left(M_y^2 + M_z^2\right) + \frac{T^2 c^2}{J^2}}$$

$$= \frac{c}{J} \sqrt{M_y^2 + M_z^2 + T^2}$$

$$\sigma_{\text{max}} = \frac{\sigma_m}{2} + R = \frac{c}{J} \sqrt{M_y^2 + M_z^2} + \frac{c}{J} \sqrt{M_y^2 + M_z^2 + T^2}$$

$$= \frac{c}{J} \left[\left(M_y^2 + M_z^2\right)^{1/2} + \left(M_y^2 + M_z^2 + T^2\right)^{1/2} \right]$$



The solid shaft AE rotates at 600 rpm and transmits 45 kW from the motor M to machine tools connected to gears G and H. Knowing that $\tau_{\rm all} = 55$ MPa and that 30 kW is taken off at gear G and 15 kW is taken off at gear H, determine the smallest permissible diameter of shaft AE.

SOLUTION

$$f = \frac{600 \text{ rpm}}{60 \text{ sec/min}} = 10 \text{ Hz}$$

Torque on gear *B*:

$$T_B = \frac{P}{2\pi f} = \frac{(45000)}{2\pi (10)}$$
$$= 716.2 \text{ Nm}$$

Torques on gears *C* and *D*:

$$T_C = \frac{30}{45} T_B = 477.5 \text{ Nm}$$

 $T_D = \frac{15}{45} T_B = 238.7 \text{ Nm}$

Shaft torques:

AB:
$$T_{AB} = 0$$

BC: $T_{BC} = 716.2 \text{ Nm}$
CD: $T_{CD} = 238.7 \text{ Nm}$

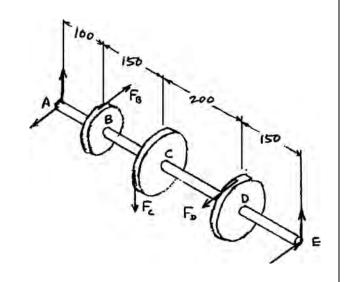
$$DE: T_{DE} = 0$$

Gear forces:

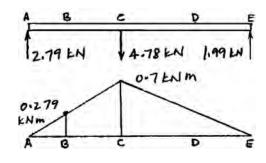
$$F_B = \frac{T_B}{r_B} = \frac{716.2}{0.075} = 9.55 \text{ kN}$$

$$F_C = \frac{T_C}{r_C} = \frac{477.5}{0.1} = 4.78 \text{ kN}$$

$$F_D = \frac{T_D}{r_D} = \frac{238.7}{0.1} = 2.39 \text{ kN}$$



Forces in vertical plane



PROBLEM 8.68 (Continued)

At
$$B^+$$
, $\sqrt{M_z^2 + M_y^2 + T^2}$
= $\sqrt{0.279^2 + 0.736^2 + 0.7162^2}$
= 1.064 kNm

At
$$C^-$$
, $\sqrt{M_z^2 + M_y^2 + T^2}$
= $\sqrt{0.7^2 + 0.4075^2 + 0.7162^2}$
= 1.081 kNm (maximum)

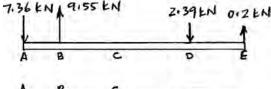
$$\tau_{\text{all}} = \frac{c}{J} \left(\sqrt{M_z^2 + M_y^2 + T^2} \right)_{\text{max}}$$

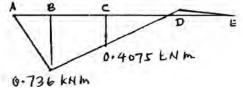
$$\left(\sqrt{M_z^2 + M_z^2 + T^2} \right)$$

$$\frac{J}{c} = \frac{\pi}{2}c^3 = \frac{\left(\sqrt{M_z^2 + M_y^2 + T^2}\right)}{\tau_{\text{all}}} = \frac{1.081}{55 \times 10^3} = 19.655 \times 10^{-6} \text{ m}^3$$

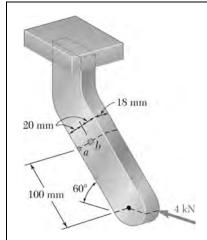
$$c = 0.0232 \text{ m}$$

Forces in horizontal plane



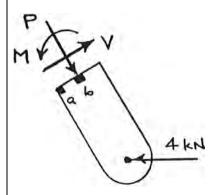


$$d = 2c = 46.4 \text{ mm}$$



For the bracket and loading shown, determine the normal and shearing stresses at (a) point a, (b) point b.

SOLUTION



Draw free body diagram of portion below section ab.

From statics:

$$P = 4000\cos 60^{\circ} = 2000 \text{ N}$$

$$V = 4000 \sin 60^{\circ} = 3464.1 \text{ N}$$

$$M = (0.1)(4000) \sin 60^{\circ} = 346.41 \,\mathrm{N} \cdot \mathrm{m}$$

Section properties:

$$A = (0.018)(0.040) = 720 \times 10^{-6} \text{ m}^2$$

$$I = \frac{1}{12}(0.018)(0.040)^3 = 96 \times 10^{-9} \text{ m}^4$$

 $\overline{y} = 0.01 \text{ m}$

$$\sigma = -\frac{P}{A} - \frac{Mc}{I} = -\frac{2000}{720 \times 10^{-6}} - \frac{(346.41)(0.02)}{96 \times 10^{-9}}$$

$$=-74.9\times10^6$$
 Pa

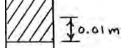
$$\sigma = -74.9 \text{ MPa} \blacktriangleleft$$

$$\tau = 0$$

0.018 m

$$\sigma = -\frac{P}{A} = -\frac{2000}{720 \times 10^{-6}}$$

$$\sigma = -2.78 \, \mathrm{MPa} \, \blacktriangleleft$$



$$A = (0.018)(0.02) = 360 \times 10^{-6} \text{ m}^3$$

$$\bar{z} = 3.6 \times 10^{-6} \text{ m}^3$$

$$A = (0.018)(0.02) = 360 \times 10^{-6} \text{ m}^3 \qquad \overline{y} = \frac{1}{100} = \frac{1}$$

 $\tau = 7.22 \text{ MPa}$

45 mm 1500 N 1200 N 75 mm

PROBLEM 8.70

Two forces are applied to the pipe AB as shown. Knowing that the pipe has inner and outer diameters equal to 35 and 42 mm, respectively, determine the normal and shearing stresses at (a) point a, (b) point b.

SOLUTION

$$c_0 = \frac{d_o}{2} = 21 \text{ mm}, \quad c_i = \frac{d_i}{2} = 17.5 \text{ mm} \quad A = \pi \left(c_o^2 - c_i^2\right) = 423.33 \text{ mm}^2$$

$$J = \frac{\pi}{2} \left(c_o^4 - c_i^4\right) = 158.166 \times 10^3 \text{mm}^4 \quad I = \frac{1}{2} J = 79.083 \times 10^3 \text{mm}^4$$

For semicircle with semicircular cutout,

$$Q = \frac{2}{3} (c_o^3 - c_i^3) = 2.6011 \times 10^3 \text{mm}^3$$



At the section containing points a and b,

$$P = -1500 \text{ N} \qquad V_z = -1200 \text{ N} \qquad V_x = 0$$

$$M_z = -(45 \times 10^{-3}) (1500) = -67.5 \text{ N} \cdot \text{m}$$

$$M_x = -(75 \times 10^{-3}) (1200) = -90 \text{ N} \cdot \text{m}$$

$$T = (90 \times 10^{-3}) (1200) = 108 \text{ N} \cdot \text{m}$$

(a)
$$\sigma = \frac{P}{A} - \frac{M_x c}{I} = \frac{-1500}{423.33 \times 10^{-6}} - \frac{(-90)(21 \times 10^{-3})}{79.083 \times 10^{-9}}$$
 $\sigma = 20.4 \text{ MPa}$

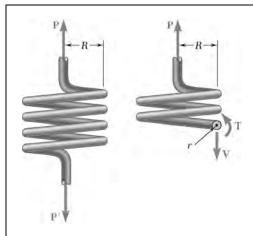
$$\tau = \frac{Tc}{J} + \frac{V_x Q}{It} = \frac{(108)(21 \times 10^{-3})}{158 \cdot 166 \times 10^{-9}} + 0$$

$$\tau = 14.34 \text{ MPa}$$

(b)
$$\sigma = \frac{P}{A} + \frac{M_z c}{I} = \frac{-1500}{423.33 \times 10^{-6}} + \frac{(-67.5)(21 \times 10^{-3})}{79.083 \times 10^{-9}}$$
 $\sigma = -21.5 \text{ MPa}$

$$\tau = \frac{Tc}{J} + \frac{|V_z|Q}{It} = \frac{(108)(21 \times 10^{-3})}{158.166 \times 10^{-9}} + \frac{(1200)(2.6011 \times 10^{-6})}{(79.083 \times 10^{-9})(7 \times 10^{-3})}$$

$$\tau = 19.98 \text{ MPa}$$



A close-coiled spring is made of a circular wire of radius r that is formed into a helix of radius R. Determine the maximum shearing stress produced by the two equal and opposite forces P and P'. (Hint: First determine the shear V and the torque T in a transverse cross section.)

SOLUTION

$$+\uparrow \Sigma F_{v} = 0$$

$$+ \uparrow \Sigma F_y = 0$$
: $P - V = 0$ $V = P$

$$V = P$$

$$+\Sigma M_C = 0:$$
 $T - PR = 0$ $T = PR$

$$T - PR = 0$$

$$T = PR$$

Shearing stress due to T.

$$\tau_T = \frac{Tc}{I} = \frac{2T}{\pi c^3} = \frac{2PR}{\pi r^3}$$

Shearing stress due to V.

For semicircle.

$$Q = \frac{2}{3}r^3, \ t = d = 2r$$

For solid circular section,

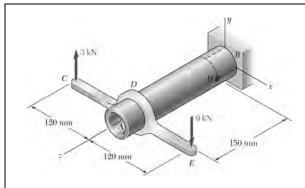
$$I = \frac{1}{2}J = \frac{\pi}{4}r^3$$

$$\tau_V = \frac{VQ}{It} = \frac{V\left(\frac{2}{3}r^3\right)}{\frac{\pi}{4}r^4(2r)} = \frac{4V}{3\pi r^2} = \frac{4P}{3\pi r^2}$$

By superposition,

$$au_{ ext{max}} = au_T + au_V$$

$$\tau_{\text{max}} = P(2R + 4r/3)/\pi r^3 \blacktriangleleft$$



The steel pipe AB has a 72-mm outer diameter and a 5-mm wall thickness. Knowing that the arm CDE is rigidly attached to the pipe, determine the principal stresses, principal planes, and the maximum shearing stress at point H.

SOLUTION

Replace the forces at C and E by an equivalent force-couple system at D.

$$F_D = 9 - 3 = 6 \text{ kN}$$

 $T_D = (9 \times 10^3)(120 \times 10^{-3}) + (3 \times 10^3)(120 \times 10^{-3})$
 $= 1440 \text{ N} \cdot \text{m}$

At the section containing point H,

$$P = 0$$
, $V = 6$ kN, $T = 1440$ N·m

$$M = (6 \times 10^3)(150 \times 10^{-3}) = 900 \text{ N} \cdot \text{m}$$

Section properties:

$$d_o = 72 \text{ mm}$$
 $c_o = \frac{1}{2}d_o = 36 \text{ mm}$ $C_i = C_o - t = 31 \text{ mm}$

$$A = \pi (C_o^2 - C_i^2) = 1.0524 \times 10^3 \text{ mm}^2 = 1.0524 \times 10^{-3} \text{ m}^2$$

$$I = \frac{\pi}{4} \left(C_o^4 - C_i^4 \right) = 593.84 \times 10^{-3} \text{ mm}^4 = 593.84 \times 10^{-9} \text{ m}^4$$

$$J = 2I = 1.877 \times 10^{-6} \,\mathrm{m}^4$$

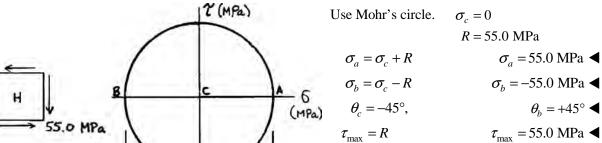
For half-pipe,

$$Q = \frac{2}{3} \left(C_o^3 - C_i^3 \right) = 11.243 \times 10^3 \text{ mm}^3 = 11.243 \times 10^{-6} \text{ m}^3$$

At point *H*.

Point H lies on the neutral axis of bending.

$$\tau_H = \frac{Tc}{J} + \frac{VQ}{It} = \frac{(1440)(36 \times 10^{-3})}{1.1877 \times 10^{-6}} + \frac{(6 \times 10^3)(11.243 \times 10^{-6})}{(593.84 \times 10^{-9})(10 \times 10^{-3})} = 55.0 \text{ MPa}$$



Use Mohr's circle. $\sigma_c = 0$

$$R = 55.0 \text{ MPa}$$

$$\sigma_a = \sigma_c + R$$

$$\sigma_a = 55.0 \text{ MPa}$$

$$\sigma_{i} = \sigma_{i} - R$$

$$\sigma_b = -55.0 \text{ MPa}$$

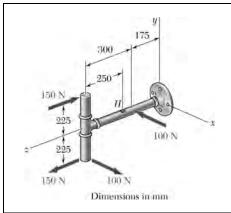
$$\theta_{\cdot} = -45^{\circ}$$
.

$$\theta_b = +45^{\circ} \blacktriangleleft$$

$$\tau = R$$

 $\tau_{\rm max} = 55.0 \text{ MPa} \blacktriangleleft$

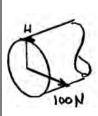
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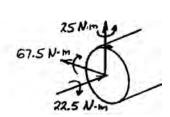


Several forces are applied to the pipe assembly shown. Knowing that each section of pipe has inner and outer diameters equal to 36 and 42 mm, respectively, determine the normal and shearing stresses at point H located at the top of the outer surface of the pipe.

SOLUTION

At the section containing points H,





$$P = 0$$
, $V_x = 100 \text{ N}$, $V_y = 0$

$$M_x = -(0.450)(150) = -67.5 \text{ N} \cdot \text{m}$$

$$M_{v} = (0.250)(100) = 25 \text{ N} \cdot \text{m}$$

$$M_z = -(0.225)(100) = -22.5 \text{ N} \cdot \text{m}$$

$$d_o = 42 \text{ mm}$$
 $d_i = 32 \text{ mm}$

$$c_o = 21 \text{ mm}$$
 $c_i = 18 \text{ mm}$

$$t = c_o - c_i = 3 \text{ mm}$$

$$A = \pi (C_o^2 - C_i^2) = 367.57 \text{ mm}^2 = 367.57 \times 10^{-6} \text{ m}^2$$

$$I = \frac{\pi}{4} \left(C_o^4 - C_i^4 \right) = 70.30 \times 10^3 \text{ mm}^4 = 70.30 \times 10^{-9} \text{ m}^4, J = 2I = 140.59 \times 10^{-9} \text{ m}^4$$

$$Q = \frac{2}{3} (C_o^3 - C_i^3) = 2.286 \times 10^3 \text{ mm}^3 = 2.286 \times 10^{-6} \text{ m}^3$$

$$\sigma_H = \frac{M_x y}{I_x} = \frac{(-67.5)(21 \times 10^{-3})}{70.30 \times 10^{-9}}$$

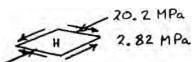
$$\sigma$$
 = -20.2 MPa

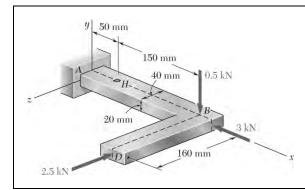
$$(\tau_H)_T = \frac{Tc}{J} = \frac{(22.5)(21 \times 10^{-3})}{140.59 \times 10^{-9}} = 3.36 \text{ MPa}$$

$$(\tau_H)_V = \frac{VQ}{It} = \frac{(100)(2.286 \times 10^{-6})}{(70.30 \times 10^{-9})(6 \times 10^{-3})} = 0.54 \text{ MPa}$$

$$\tau_H = 3.36 - 0.54$$

$$\tau = 2.82 \text{ MPa} \blacktriangleleft$$





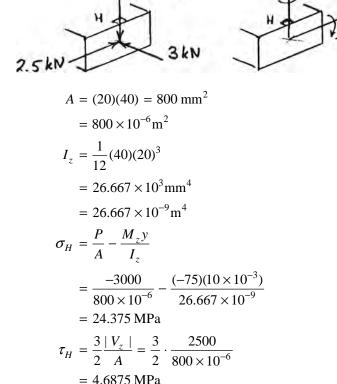
Three forces are applied to the machine component ABD as shown. Knowing that the cross section containing point H is a 20×40 -mm rectangle, determine the principal stresses and the maximum shearing stress at point H.

SOLUTION

Equivalent force-couple system at section containing point *H*:

$$F_x = -3 \text{ kN}, \quad F_y = -0.5 \text{ kN}, \quad F_z = -2.5 \text{ kN}$$

 $M_x = 0, \quad M_y = (0.150)(2500) = 375 \text{ N} \cdot \text{m}$
 $M_z = -(0.150)(500) = -75 \text{ N} \cdot \text{m}$



PROBLEM 8.74 (Continued)

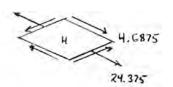
Use Mohr's circle.

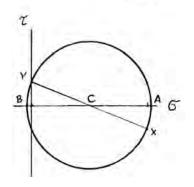
$$\sigma_{\text{ave}} = \frac{1}{2}\sigma_{H}$$

$$= 12.1875 \text{ MPa}$$

$$R = \sqrt{\left(\frac{24.375}{2}\right)^{2} + (4.6875)^{2}}$$

$$= 13.0579 \text{ MPa}$$





$$\sigma_a = \sigma_{\text{ave}} + R$$

$$\sigma_b = \sigma_{\text{ave}} - R$$

$$\tan 2\theta_p = \frac{2\tau_H}{\sigma_H} = \frac{(2)(4.6875)}{24.375} = 0.3846$$

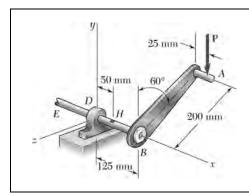
$$\theta_a = 10.5^\circ, \quad \theta_b = 100.5^\circ$$

$$\tau_{\text{max}} = R$$

$$\sigma_a = 25.2 \, \mathrm{MPa} \, \blacktriangleleft$$

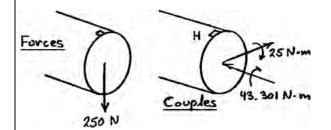
$$\sigma_b = -0.87 \, \mathrm{MPa} \, \blacktriangleleft$$

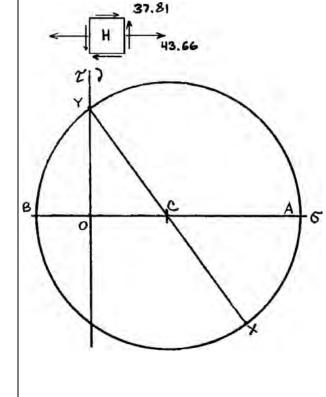
 $\tau_{\rm max} = 13.06 \, \mathrm{MPa}$



A vertical force \mathbf{P} of magnitude 250 N is applied to the crank at point A. Knowing that the shaft BDE has a diameter of 18 mm, determine the principal stresses and the maximum shearing stress at point H located at the top of the shaft, 50 mm to the right of support D.







Force-couple system at the centroid of the section containing point *H*:

$$F_x = 0, V_y = -250 \text{ N}, V_z = 0$$

$$M_z = -(125 - 50 + 25)(10^{-3})(250) = -25 \text{ N} \cdot \text{m}$$

$$M_x = -(200 \sin 60^\circ)(10^{-3})(250) = -43.301 \text{ N} \cdot \text{m}$$

$$d = 18 \text{ mm} c = \frac{1}{2}d = 9 \text{ mm}$$

$$I = \frac{\pi}{4}c^4 = 5.153 \times 10^3 \text{ mm}^4 = 5.153 \times 10^{-9} \text{ m}^4$$

$$J = 2I = 10.306 \times 10^{-9} \text{ m}^4$$

At point H,

$$\sigma_H = -\frac{M_z y}{I_x} = -\frac{(-25)(9 \times 10^{-3})}{5.153 \times 10^{-9}} = 43.66 \text{ MPa}$$

$$\tau_H = \frac{Tc}{J} = \frac{(43.301)(9 \times 10^{-3})}{10.306 \times 10^{-3}} = 37.81 \text{ MPa}$$

Use Mohr's circle,

ose Monif sericle,
$$\sigma_{C} = \frac{1}{2}\sigma_{H} = 21.83 \text{ MPa}$$

$$R = \sqrt{\left(\frac{43.66}{2}\right)^{2} + (37.81)^{2}} = 43.66 \text{ MPa}$$

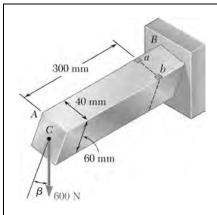
$$\sigma_{a} = \sigma_{c} + R \qquad \sigma_{\text{max}} = 65.5 \text{ MPa} \blacktriangleleft$$

$$\sigma_{b} = \sigma_{c} - R \qquad \sigma_{\text{min}} = -21.8 \text{ MPa} \blacktriangleleft$$

$$\tan 2\theta_{p} = \frac{2\tau_{H}}{\sigma_{H}} = \frac{75.62}{43.66} = 1.7320$$

$$\theta_{a} = 30.0^{\circ}, \qquad \theta_{b} = 120^{\circ}$$

$$\tau_{\text{max}} = R \qquad \tau_{\text{max}} = 43.7 \text{ MPa} \blacktriangleleft$$



The cantilever beam AB will be installed so that the 60-mm side forms an angle β between 0 and 90° with the vertical. Knowing that the 600-N vertical force is applied at the center of the free end of the beam, determine the normal stress at point a when (a) $\beta = 0$, (b) $\beta = 90°$. (c) Also, determine the value of β for which the normal stress at point a is a maximum and the corresponding value of that stress.

SOLUTION

$$S_x = \frac{1}{6} (40)(60)^2 = 24 \times 10^3 \text{ mm}^3$$

$$= 24 \times 10^{-6} \text{ m}^3$$

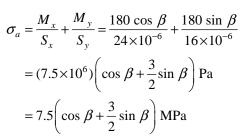
$$S_y = \frac{1}{6} (60)(40)^2 = 16 \times 10^3 \text{ mm}^3$$

$$= 16 \times 10^{-3} \text{ m}^3$$

$$M = Pl = (600)(300 \times 10^{-3}) = 180 \text{ N} \cdot \text{m}$$

$$M_x = M \cos \beta = 180 \cos \beta$$

$$M_y = M \sin \beta = 180 \sin \beta$$



(a)
$$\beta = 0$$

$$\sigma_a = 7.50 \text{ MPa}$$

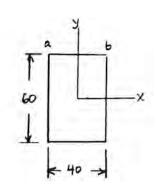
$$(b) \qquad \beta = 90^{\circ}.$$

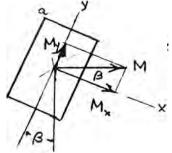
$$\sigma_a = 11.25 \text{ MPa}$$

$$\frac{d\sigma_a}{d\beta} = 7.5 \left(-\sin \beta + \frac{3}{2}\cos \beta \right) = 0$$

$$\sin \beta = \frac{3}{2}\cos \beta \qquad \tan \beta = \frac{3}{2}$$

$$\sigma_a = 7.5 \left(\cos 56.3^\circ + \frac{3}{2}\sin 56.3^\circ\right)$$





- σ = 7.50 MPa
- $\sigma = 11.25 \text{ MPa}$

$$\beta = 56.3^{\circ} \blacktriangleleft$$

$$\sigma$$
 = 13.52 MPa \triangleleft