

**The University of New Mexico**  
**School of Engineering**  
**Electrical and Computer Engineering Department**  
  
**ECE 535 Satellite Communications**

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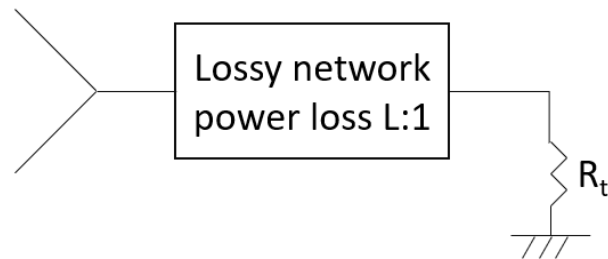
Student SN: 201

Module # 12-2: 12.15, 12.19, 12.21, 12.23, 12.25

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**12.15** An amplifier having a noise temperature of 200 K has a 4-dB attenuator connected at its input. Calculate the effective noise temperature referred to the attenuator input.



$$T_{NW,i} = T_x(L - 1)$$

$$T = 200K; \quad L = 4dB$$

$$T_S = T_O(L - 1) + \frac{T_e}{\frac{1}{L}} = T_O(L - 1) + LT_e$$

$$L = 10^{4/10} = 2.511$$

$$T_S = 290 * (2.511 - 1) + (2.511 * 200) = 438.19 + 502.2 = 940.39K$$

**12.19 Explain what is meant by carrier-to-noise ratio. At the input to a receiver the received carrier power is 400 pW and the system noise temperature is 450 K. Calculate the carrier-to-noise density ratio in dBHz. Given that the bandwidth is 36 MHz, calculate the carrier-to-noise ratio in decibels.**

Definition: Carrier-to-noise ratio is a measurement of the performance of a satellite link to the received input. Link budgets are typically concerned with solving this ratio (C/N).

$$CW_{Rx} = 400 \times 10^{-12} \text{ W}; T_1 = 450 \text{ K}; BW = 36 \text{ MHz}; k = 1.38 \times 10^{-23} \text{ J/K}$$

**(a) Carrier-to-Noise Density Ratio in dBHz**

$$P_{Rx \text{ dB}} = 10 \log(CW_{Rx}) = -93.98 \text{ dBW}$$

$$\frac{C}{N_0} = \frac{P_{Rx \text{ dB}}}{k * T_1} = \frac{-93.98 \text{ dBW}}{(1.38 \times 10^{-23} \text{ J/K}) * 450 \text{ K}} = 6.44 \times 10^{10} = 10 \log(6.44 \times 10^{10}) = 108.1 \text{ dB}$$

**(b) Calculate Carrier-to-Noise Ratio in Decibels at 36MHz Bandwidth**

$$CN_0 \text{ dBHz} = \frac{C}{N_0} \text{ dBHz} - \text{dBHz} = 108.1 \text{ dB} - 10 \log(36 \text{ MHz}) = 108.1 \text{ dB} - 75.56 \text{ dBHz} = 32.5 \text{ dBHz}$$

**12.21 In a satellite link the propagation loss is 200 dB. Margins and other losses account for another 3 dB. The receiver [G/T] is 11 dB, and the [EIRP] is 45 dBW. Calculate the received [C/N] for a system bandwidth of 36 MHz.**

$$L_{Prop} = 200dB; L_a = 2 = 3dB; G_r = 11dB; EIRP = 45dBW$$

(a) Calculate the received [C/N] for a system BW of 36MHz

Boltzmann's constant (J/K dB) for temp = 228.6 decilogs

$$C/N_0 = EIRP + k + G_r - L_{Prop} - L_a = 81.6dB$$

$$[C/N_0] = C/N_0 - BW = 81.6dB - 10\log(36MHz) = 6 \text{ dB}$$

**12.23 Explain what is meant by saturation flux density. The power received by a 1.8-m parabolic antenna at 14 GHz is 250 pW. Calculate the power flux density (a) in W/m<sup>2</sup> and (b) in dBW/m<sup>2</sup> at the antenna.**

Definition: Saturation flux density ( $\Psi_s$ ) is required at the receiving antenna to produce saturation of the TWTA (traveling-wave tube amplifier). It is the maximum flux that a material can support before it becomes saturated. Antennas can hold a calculated amount of magnetic field before saturating.

$$D = 1.8m, f = 14GHz, P_R = 250pW$$

(a) Power flux density in W/m<sup>2</sup>

$$A_{ett} = \eta_I A_0$$

$$A_0 = \pi \frac{d^2}{4}$$

$$\eta_I = 0.55$$

$$P_R = \Psi_M A_{ett} \xrightarrow{\text{yields}} \Psi_M = \frac{P_R}{A_{ett}}$$

$$A_{ett} = \eta_I A_0 = 0.55 * \left( \frac{\pi * (1.8)^2}{4} \right) = 1.399$$

$$\Psi_M = \frac{250}{1.399} = 178.6 \text{ pw/m}^2$$

(b) Power flux in dBW/m<sup>2</sup>

$$[\Psi_M] = 10\log(\Psi_M) = 10\log(178.6 \times 10^{-12}) = -97.5 \text{ dBW/m}^2$$

**12.25 A satellite transponder requires a saturation flux density of -110 dBW/m<sup>2</sup>, operating at a frequency of 14 GHz. Calculate the earth station [EIRP] required if total losses amount to 200 dB.**

$$[EIRP] = [\Psi_M] + [A_0] + [FSL]$$

$$A_0 = 10 \log \left( \frac{\lambda^2}{4\pi} \right)$$

$$A_0 = 10 \log \left( \frac{(c/f)^2}{4\pi} \right) = -44.38$$

$$EIRP = (-110 \text{ dBW/m}^2) + (-44.38 \text{ dB}) + 200 \text{ dB} = 45.6 \text{ dBW}$$