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MAGISTERSKA PRACA DYPLOMOWA

# PROGRAM STAR-TRACKER DLA SATELITÓW TYPU CUBE-SAT

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# 1 Introduction

## 1.1 Motivation

The goal of this work is to make fully operational star-tracker program, that could be used on Cubesat satellites. Such program could be used on space missions and could start Polish state-of-the-art technology in growing space technology sector.

## 1.2 Outline of thesis

This thesis consists of several chapters. Here they are shortly summarized:

**Chapter 1** serves as introduction to this thesis and describes the motivation and goal of this work. It also describes the background of the topic.

**Chapter 2** describes all the important foundations for the fully understanding given work.

**Chapter 3** is the main part of this thesis. It describes how the star-tracker program works and goes through detailed comparison of different approaches.

**Chapter 4** describes the created prototype of star-tracker in Python language.

**Chapter 5** talks about the implementation of star-tracker on the existing prototype of on-board computer.

**Chapter 6** describes how the finished program is performing.

**Chapter 7** contains conclusions about this work and created star-tracker program.

### 1.3 Cubesat

Cubesat was designed on Caltech in 1999. Dimensions of satellite are measured in units. Each unit (often described simply as u) can be 10x10x10cm and can weight up to 1.33 kg. Satellites can be 1u, 2u, 3u, 6u or even 12u.

Such small satellites are susceptible to noise from densly packed electronics.

### 1.4 Means of attitude estimation

There exist many different types of attitude estimation: sun sensors, star-trackers, magnetometers, etc. However star-tracker gives the best possible accuracy for nowadays and is not susceptible to electrical nor magnetic noise.

Table 1

### 1.5 On-board computer

This section will describe the on-board computer which was done as part of other thesis.

## 2 Preliminaries

### 2.1 Coordinate frames

#### 2.1.1 ECI frame

#### 2.1.2 ECEF frame

#### 2.1.3 NED frame

#### 2.1.4 BODY frame

### 2.2 Attitude representations

#### 2.2.1 Euler angles

#### 2.2.2 Quaternions

### 2.3 Quaternion properties

#### 2.3.1 Advantages of quaternions

#### 2.3.2 Multiplication of quaternions

#### 2.3.3 Quaternions and rotations

### 2.4 Wahba's problem

Hello [1]

$$\sum_j^n ||r_j - Mb_j|| \tag{1}$$

## 2.5 Cholesky factorization

## 2.6 Lyapunov analysis

### **3 Star-tracker program**

#### **3.1 Centroid - start recognition**

$$f(x) = x^2 \tag{2}$$



## **3.2 Star identification**

### **3.2.1 Angle Matching**

### **3.2.2 Spherical Triangle Matching**

### **3.2.3 Planar Triangle**

### **3.2.4 Rate Matching**

### **3.2.5 Pyramid**

### **3.2.6 Voting**

### **3.2.7 Grid**

### **3.2.8 Different elements of algorithm**

## **3.3 Star Catalogue Generation**

## **3.4 Candidate Matching**

## **3.5 Result Verification**

## **3.6 k-vector**

## **3.7 Attitude Determination**

### **3.7.1 QUEST**

### **3.7.2 TRIAD**

### **3.7.3 The Fast Optimal Attitude Matrix**

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### **3.7.4 q-method**

### **3.7.5 DCM (direction cosine matrix) - (Singular Value Decomposition?)**

## **4    Prototype**

## 5 Complete program

## **6    Testing of star-tracker**

## References

- [1] Cheng Yang and Shuster Malcolm D., “Improvement to the Implementation of the QUEST Algorithm,” *Journal of Guidance, Control, and Dynamics*, vol. 37, no. 1, pp. 301–305, 2013. doi: 10.2514/1.62549.

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Many angles result in the same rotation Hard to quantify differences between  
two Euler Angles Unit-Quaternion Solves the problems of singularities with  
the Euler Angles Easier to compute differences of orientations Important if  
we want to control the orientation of the end-effector See Siciliano or Spong  
Textbook!

Polar moment