


# **PHY 101: Mechanics and Properties of Matter**


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**(Lecture : C19-1)**



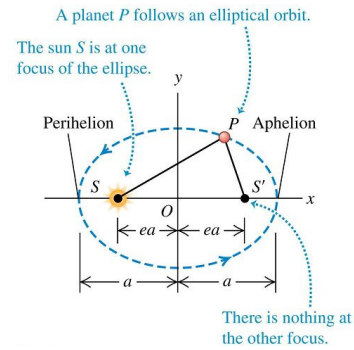
# GRAVITATION

## Learning Goals:

- Understanding the laws that governs the relationship between heavenly bodies
  - How to calculate the gravitational forces that any two bodies exert on each other.
  - How to relate the weight of an object to the general expression for gravitational force (Calculating acceleration due to gravity on, above and inside the earth crust).
  - How to calculate the speed, orbital period, and mechanical energy of a satellite in a circular orbit.
  - How to apply and interpret Kepler's three laws that describe the motion of planets.
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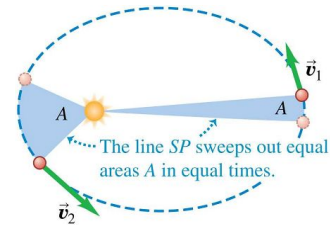
# Keplers's Law of Planetary Motion

1. The planets describes motion in elliptical orbit about the sun as a focus



1<sup>st</sup> Law

2. The line joining the sun and each planet sweeps out equal areas in equal times (i.e constant period) due to conservation of angular momentum



2<sup>nd</sup> Law

3. The square of the period  $T$  is proportional to the cube of the radius of the ellipse of each planet i.e  $T^2 \propto r^3$ .

$$T = \frac{2\pi r^{3/2}}{\sqrt{GM_e}}$$

3<sup>rd</sup> Law

# Newton's Law of Universal Gravitation


States that:

Every particle of matter attracts every other particle with a force ( $F_g$ ) that is directly proportional to the product of their masses ( $m_1$  &  $m_2$ ) and inversely proportional to the square of the distance ( $r$ ) between them. That is

$$F_g = \frac{Gm_1 m_2}{r^2}$$

Where  $G$  is the gravitational constant.

The gravitational constant  $G = 6.67 \times 10^{-11} \text{ N}\cdot\text{m}^2/\text{kg}^2$  is a fundamental physical constant that has the same value for any two particles

Consider the sun of mass,  $M$  and a planet of mass,  $m$ , the gravitational force which provide the centripetal force is given by: 

$$F = \frac{GMm}{r^2} = m\omega^2 r$$

$$\omega^2 r = \frac{GM}{r^2} \quad \text{or} \quad \frac{4\pi^2}{T^2} = \frac{GM}{r^3}$$

$$\therefore T^2 = \left(\frac{4\pi^2}{GM}\right) r^3 \quad \text{i.e. } T^2 \propto r^3.$$

This is Kepler's 3<sup>rd</sup> law and it implies 2<sup>nd</sup> law since for a particular planet moving in an orbit, the radius  $r$  is constant

# Gravitational Attraction

- Our solar system is part of a spiral galaxy like this one, which contains roughly  $10^{11}$  stars as well as gas, dust, and other matter (mostly dark matter).
- The entire assemblage is held together by the mutual gravitational attraction of all the matter in the galaxy



# Weight and Acceleration Due to Gravity

- The **weight** of an object on the earth is the gravitational force that the earth exerts on the object. This force gives the object an acceleration of  $g$  towards the centre of the earth.

Hence  $F = \frac{GM_E m}{R_E^2} = mg$

Therefore, the acceleration due to gravity at the earth's surface is:

$$g = \frac{Gm_E}{R_E^2}$$

$$\therefore g \propto \frac{1}{r^2}$$

This explains why  $g$  is not constant all over the world (It is more at the pole than at the equator). **The average value of  $g$  is  $9.8 \text{ ms}^{-2}$**

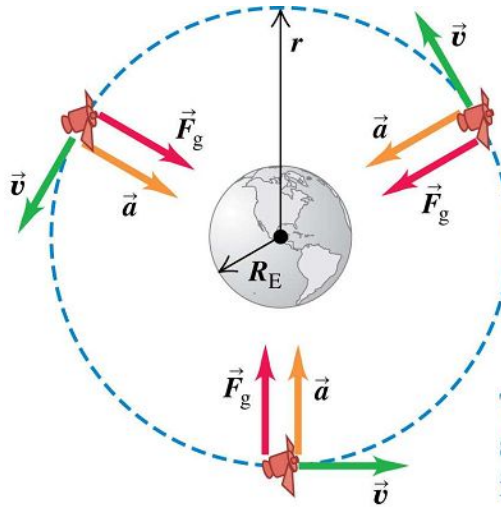
- The acceleration due to gravity,  $g'$  of an object at an height  $h$  **above the earth** is **given by**
- The acceleration due to gravity,  $g''$  of an object at an height  $h$  **beneath the earth surface** is **given by**

$$g' = \frac{R_E^2}{(R_E + h)^2} g$$

$$g'' = \left( \frac{R_E - h}{R_E} \right) g$$

# Orbits Round the Earth

- Satellites are objects that revolve other objects. For example, the moon around the earth is a natural satellite.
- The force due to the earth's gravitational attraction provides the centripetal acceleration/ force that keeps a satellite in orbit.



The satellite is in a circular orbit: Its acceleration  $\vec{a}$  is always perpendicular to its velocity  $\vec{v}$ , so its speed  $v$  is constant.

- Using the Newton's law of gravitation, we have:

$$F = \frac{GM_E m}{r^2} = m \frac{v^2}{r}$$

$$\therefore V = \sqrt{GM_E/r}$$



- For a circular orbit of radius  $r$ , the Speed,  $V$  of a satellite is just right to keep its distance from the center of the earth constant.



For one revolution of a satellite, the period is given by:  $T = 2\pi r / v$

Substituting  $V = \sqrt{GM_E/r}$  we will have

$$T = \frac{2\pi r^{3/2}}{\sqrt{GM_e}}$$

This confirms **Kepler's 3<sup>rd</sup> Law**



- A satellite is said to be parked if its period is 24 Hrs ( the time taken by earth to complete one rotation



# Earth's Gravitational Potential Energy

The earth's gravitational potential energy at any point in the earth gravitational field can be defined as the total work done per unit mass on a body to move it from the point which is at a distance  $r$  from the center of the earth to infinity (where  $F_g = 0$ )

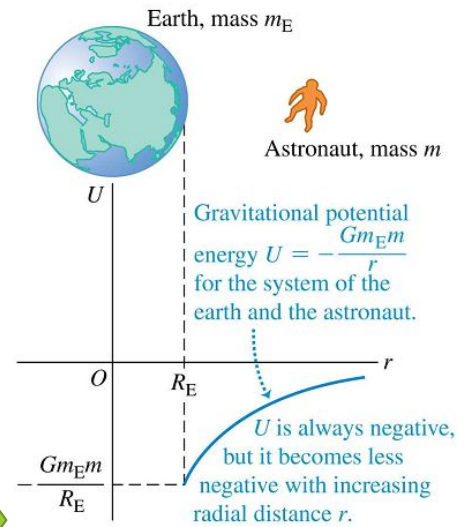
$$U = -\frac{Gm_E m}{r}$$

Gravitational constant  $G$   
Mass of the earth  $m_E$   
Mass of body  $m$   
Distance of body from the earth's center  $r$   
Gravitational potential energy (general expression)  $U$

Hence, the Gravitational potential,  $\phi$  is thus given by

$$\phi = \frac{\text{work}}{\text{mass}} = \frac{-GM_E}{r}$$

U depend on distance,  $r$  ➡



# Escape Velocity

If a body is given an energy which is equal to the gravitational potential energy on the earth surface, it will go to infinity. In other words, if a body leaves the earth surface with a **velocity,  $v$**  such that its kinetic is equal to the gravitational potential energy, it will never return. Thus, It is said to be escaped. That is :

$$\frac{1}{2}mv^2 = \frac{GM_E m}{r} \quad \text{hence,}$$

$$v = \sqrt{2GM_E/r} = \sqrt{2gr} \quad \text{is called the escape velocity}$$

**Example:**

The mass of the moon is about one eighty-first, and its radius one-fourth, that of the earth. Calculate the acceleration due to gravity on the surface of the moon.

**Solution**

$$mg = \frac{GmM}{R^2} \rightarrow g = \frac{GM}{R^2}$$

$$\text{for earth: } 9.8 = \frac{GM_E}{R_E^2}$$

$$\text{for moon: } g = \frac{G(M_E/81)}{(R_E/4)^2}$$

$$\therefore g = 9.8 \times \frac{16}{81} = \underline{1.94 \text{ ms}^{-2}}$$