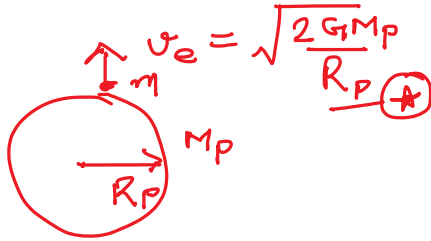


Escape Speed

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for Earth:

$$v_e = \sqrt{\frac{2GM_e}{R_e^2}} = \sqrt{\frac{2(GM_e)R_e}{(R_e^2)^2}}$$

$$(v_e)_{\text{earth}} = \sqrt{2gR_e} \approx 11.2 \text{ km/s}$$

$$\therefore v_1 = \sqrt{\frac{2GM_1}{R}} ; v_2 = \sqrt{\frac{2GM_2}{2R}}$$

$$\Rightarrow \frac{v_2}{v_1} = \sqrt{\frac{M_2}{2} \times \frac{1}{M_1}} = \sqrt{\frac{8}{2}} = 2$$

$$\therefore v_2 = 2v_0 \text{ Ans.}$$

Q) The escape velocity of an object on a planet is v_0 . What will be the escape velocity on a planet of twice radius of same density?

$$\text{Sol}^n \rightarrow R_1 = R ; R_2 = 2R$$

$$\therefore d_1 = d_2$$

$$\frac{M_1}{\frac{4}{3}\pi R_1^3} = \frac{M_2}{\frac{4}{3}\pi R_2^3}$$

$$\frac{M_1}{R^3} = \frac{M_2}{8R^3}$$

$$\therefore M_2 = 8M_1 \text{ --- (1)}$$

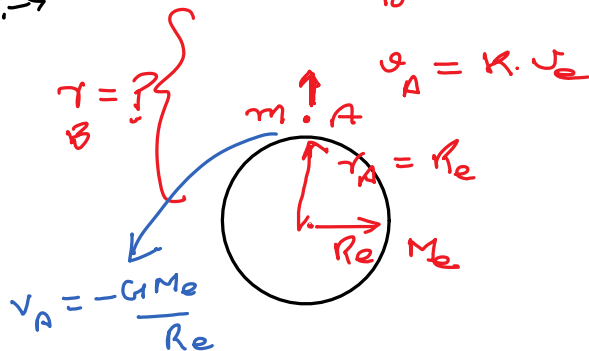
Q: A projectile is fired from the surface of earth with

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a speed $K \cdot v_e$, where K is a const $\neq 1$ & v_e is the escape speed on the Earth's surface. Neglecting the air resistance, find the max. distance of rise of projectile from earth's center.

$$V_B = -\frac{GM_e}{r_B}$$

Solⁿ \rightarrow



C.O.M.E. B/w A & B

$$K_A + v_A = K_B + v_B$$

$$\frac{1}{2} m v_A^2 + (m \cdot v_A) = 0 + (m \cdot v_B)$$

$$K^2 \frac{v_e^2}{2} = \frac{GM_e}{R_e} - \frac{GM_e}{r_B}$$

$$\therefore v_e = \sqrt{\frac{2GM_e}{R_e}}$$

$$\frac{K^2 \cdot 2 \cdot \frac{GM_e}{R_e}}{2} = \frac{GM_e}{R_e} - \frac{GM_e}{r_B}$$

$$\frac{GM_e}{r_B} = \frac{GM_e}{R_e} (1 - K^2)$$

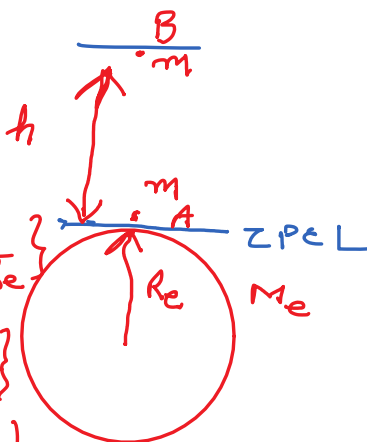
$$\frac{1}{r_B} = \left(\frac{1 - K^2}{R_e} \right)$$

$$r_B = \frac{R_e}{(1 - K^2)} m.$$

Ans.

G.P.E.(U) :

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$$\Delta U = U_B - U_A$$

$$= -G M_e m \left\{ \frac{1}{R_e + h} - \frac{1}{R_e} \right\}$$

$$= -G M_e m \left\{ \frac{R_e - R_e - h}{R_e(R_e + h)} \right\}$$

$$\Delta U = \frac{G M_e m h}{R_e(R_e + h)} = \frac{G M_e m h}{R_e^2 \left(1 + \frac{h}{R_e}\right)}$$

$$\therefore \frac{G M_e}{R_e^2} = g$$

$$\Delta U = \frac{m \cdot g \cdot h}{\left(1 + \frac{h}{R_e}\right)} \quad \text{--- (1)}$$

if $h \ll R_e$ then $1 + \frac{h}{R_e} \approx 1$

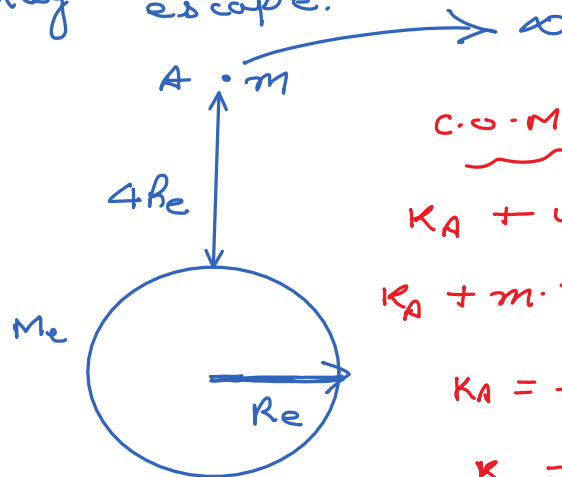
$$\Delta U = mgh \quad \text{--- (2)}$$

$$U_A = -\frac{G M_e m}{R_e} \quad \text{--- (1)}$$

$$U_B = -\frac{G M_e m}{(R_e + h)} \quad \text{--- (2)}$$

Q: A Body of mass m is situated at a Dist. $4R_e$ above the Earth's surface. How much minimum energy will be required so that it may escape?

So let \Rightarrow



C.O.M.E. B/w A & ∞

$$K_A + U_A = K_\infty + U_\infty$$

$$K_A + m \cdot V_A = 0 + 0$$

$$K_A = -m \cdot V_A = -m \cdot \left(-\frac{GM_e}{5R_e} \right)$$

$$K_A = \frac{GM_e m}{5R_e} \quad \text{Final Ans}$$

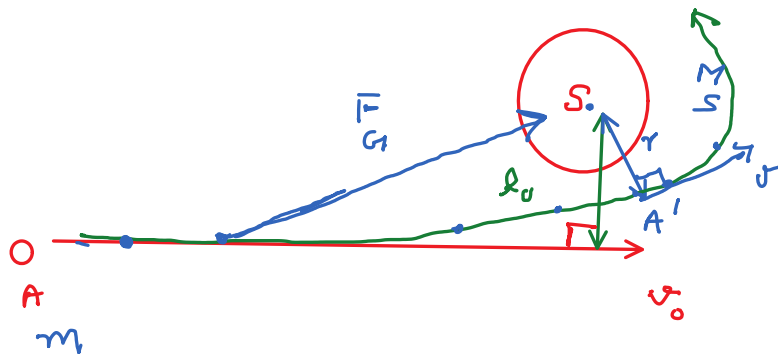
$$K_A = \frac{1}{2} m v_A^2 = \frac{GM_e m}{5R_e}$$

$$\therefore v_A = \sqrt{\frac{2}{5} \frac{GM_e}{R_e}} \text{ m/s.}$$

Escape speed at A.

Q: \rightarrow A cosmic body 'A' moves towards the sun with velocity v_0 (when it was very far from the sun) & when the impact parameter was b_0 . The direction of the initial velocity vector was as shown in the figure. Find the minimum distance of closest approach of the body from sun.

Solⁿ: \rightarrow \therefore The Gravitational force acting on the body is a central force (ie it pass from the center of sun which is $A \cdot O \cdot R$)



$$\Rightarrow \dot{r} = 0$$

Body

$$\Rightarrow L_{\text{Body}} = 0$$

$$\Rightarrow L_i = L_f$$

$$\Rightarrow m \times v_0 \times b_0 = m \cdot v \cdot r$$

$$\uparrow v = \frac{v_0 \cdot b_0}{r} \quad \text{--- (1)}$$

C.O.M.E. B/w A & A'

$$K_A + U_A = K_{A'} + U_{A'}$$

$$\Rightarrow \frac{1}{2} m v_0^2 + 0 = \frac{1}{2} m v^2 + (m \times V_{A'})$$

$$\Rightarrow \frac{v_0^2}{2} = v^2 \frac{b_0^2}{2 \cdot r^2} = \frac{G M_S}{r}$$

$$\Rightarrow r^2 \cdot v_0^2 = v_0^2 \cdot b_0^2 - 2 \cdot r G M_S$$

$$\Rightarrow v_0^2 \cdot r^2 + 2 G M_S \cdot r - v_0^2 \cdot b_0^2 = 0$$

$$\Rightarrow r = \frac{- (2 G M_S) \pm \sqrt{4 G^2 M_S^2 + 4 v_0^4 b_0^2}}{2 v_0^2}$$

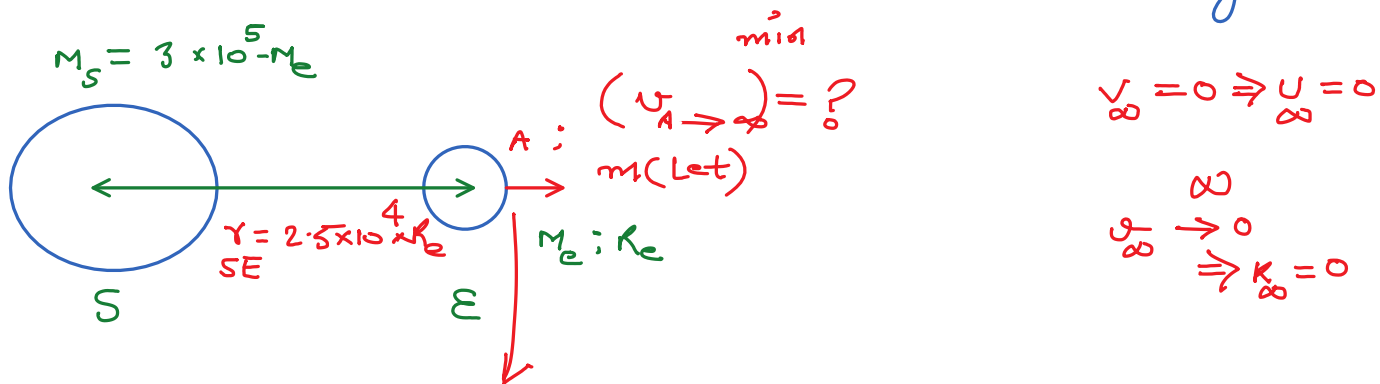
$$r = \frac{\sqrt{G^2 M_S^2 + v_0^4 b_0^2} - G M_S}{v_0^2} =$$

2017 ADV

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12:21

\rightarrow A rocket is launched normal to the Earth's surface away from the Sun, along the line joining the Sun & the Earth. The Sun is 3×10^5 times heavier than Earth & it is at a distance 2.5×10^4 times larger than the Earth's radius. Escape vel. on the Earth's surface is 11.2 km/s . find the min. launching speed for the rocket so that it escapes from Sun-Earth Gravity.



$$V_A = V_E + V_S = -\frac{GM_E}{R_E} - \frac{GM_S}{(r_{SE} + R_E)} = -\frac{GM_E}{R_E} - \frac{GM_S}{r_{SE}}$$

from C.O.M.E.

$$K_A + U_A = K_\infty + U_\infty$$

$$\frac{1}{2} m V_A^2 + m \left\{ -\frac{GM_E}{R_E} - \frac{GM_S}{r_{SE}} \right\} = 0 + 0$$

$$\frac{V_A^2}{2} = \frac{GM_E}{R_E} + \frac{G \cdot 3 \times 10^5 M_E}{2.5 \times 10^4 R_E}$$

$$\frac{V_A^2}{2} = \frac{GM_E}{R_E} \cdot \left\{ 1 + \frac{30}{2.5} \right\} = \frac{GM_E}{R_E} \times 13$$

$$\Rightarrow V_A = \sqrt{13 \times \left(\frac{2GM_E}{R_E} \right)} = \sqrt{13} \times 11.2 \text{ km/s}$$

$$= 42 \text{ km/s}$$

(approx)

$$\therefore \sqrt{\frac{2GM_E}{R_E}} = v_{\text{escape}} = 11.2 \frac{\text{km}}{\text{s}}$$

Gravitational Self Energy (U_s)

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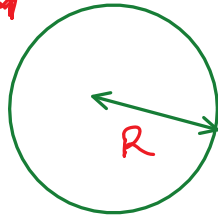
i) uniform spherical shell

Let the inst. mass deposited on the surface of the sphere is m kg

$$(dW_{\infty \rightarrow s})_{\text{ext}} = dU = dm \cdot V_s$$

$\xleftarrow{\quad dm \quad \infty \quad}$
(further incoming mass)

$0 \rightarrow m \rightarrow M$



work done by the external agents to bring dm additionally on the surface of the shell

$$dW_{\text{ext}} \text{ or } dU = dm \times V$$

$$dU = dm \times -\frac{GM}{R}$$

$$\Rightarrow \int_0^U dU = -\frac{G}{R} \cdot \int_0^M m \cdot dm$$

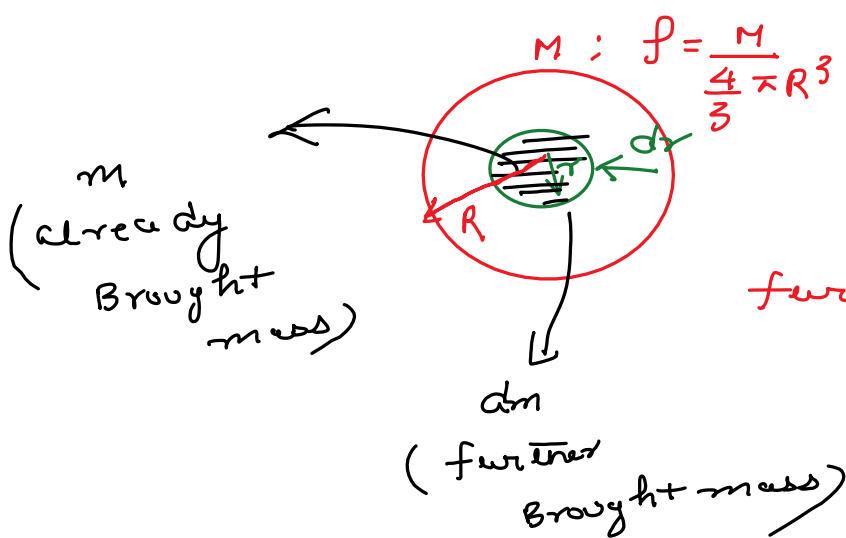
$$\boxed{U = -\frac{GM^2}{2R}} \quad \text{Joules}$$

ii) Self Energy of a uniform solid sphere:

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already Brought mass of radius

$$(m) = \rho \times \frac{4}{3} \pi r^3$$



$$= \frac{M}{\frac{4}{3} \pi R^3} \times \frac{4}{3} \pi r^3$$

$$m = \frac{M \cdot r^3}{R^3} \quad \text{--- (1)}$$

further Brought mass

$$dm = \rho \times dv$$

$$= \frac{M}{\frac{4}{3} \pi R^3} \times 4 \pi r^2 \cdot dr$$

$$dm = \frac{3M}{R^3} r^2 \cdot dr \quad \text{--- (2)}$$

additional work done by the external agents

$$dw = du = dm \times v$$

$$= dm \times \left(- \frac{Gm}{r} \right)$$

$$= - \frac{3GM^2}{R^6} \cdot \frac{r^5}{r} \cdot dr$$

$$\Rightarrow \int_0^U du = - \frac{3GM^2}{R^6} \cdot \int_0^R r^4 \cdot dr$$

$$(U)_0^U = - \frac{3GM^2}{R^6} \times \left(\frac{r^5}{5} \right)_0^R$$

$$\Rightarrow \boxed{U = - \frac{3}{5} \frac{GM^2}{R}} \quad \checkmark$$