

Kindr Cheat Sheet v1.0

Kinematics and Dynamics for Robotics

Nomenclature

(Hyper-)complex number	Q	normal capital letter
Column vector	\mathbf{a}	bold small letter
Matrix	\mathbf{M}	bold capital letter
Identity matrix	$\mathbf{1}_{n \times m}$	$n \times m$ -matrix
Coordinate system (CS)	$\mathbf{e}_x^A, \mathbf{e}_y^A, \mathbf{e}_z^A$	Cartesian right-hand system A with basis (unit) vectors \mathbf{e}
Inertial frame	$\mathbf{e}_x^I, \mathbf{e}_y^I, \mathbf{e}_z^I$	global / inertial / world coordinate system (never moves)
Body-fixed frame	$\mathbf{e}_x^B, \mathbf{e}_y^B, \mathbf{e}_z^B$	local / body-fixed coordinate system (moves with body)
Rotation	$\mathcal{R} \in \text{SO}(3)$	generic rotation (for all parameterizations)
Machine precision	ϵ	

Operators

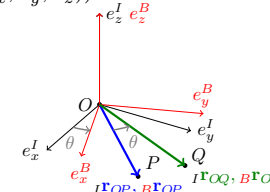
Cross product	$\mathbf{a} \times \mathbf{b} = \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} \times \begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix} \Leftrightarrow (\mathbf{a})^\wedge \mathbf{b} = \hat{\mathbf{a}} \mathbf{b} = \begin{bmatrix} 0 & -a_3 & a_2 \\ a_3 & 0 & -a_1 \\ -a_2 & a_1 & 0 \end{bmatrix} \begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix}$
Skew/uns skew	$\mathbf{a} = \hat{\mathbf{a}}^\vee$
Euclidean norm	$\ \mathbf{a}\ = \sqrt{\mathbf{a}^T \mathbf{a}} = \sqrt{a_1^2 + \dots + a_n^2}$
Exponential map for matrix	$\text{expM} : \mathbb{R}^3 \rightarrow \mathbb{R}^3, \mathbf{A} \mapsto e^{\mathbf{A}}, \quad \mathbf{A} \in \mathbb{R}^{3 \times 3}$
Logarithmic map for matrix	$\text{logM} : \mathbb{R}^3 \rightarrow \mathbb{R}^3, \mathbf{A} \mapsto \log \mathbf{A}, \quad \mathbf{A} \in \mathbb{R}^{3 \times 3}$

Position & Orientation

Position

Vector	\mathbf{r}_{OP}	from point O to point P
Position vector	${}_B \mathbf{r}_{OP} \in \mathbb{R}^3$	from point O to point P expr. in frame B
Homogeneous pos. vector	${}_B \tilde{\mathbf{r}}_{OP} = [{}_B \mathbf{r}_{OP} \quad 1]^T$	from point O to point P expr. in frame B

Orientation/Rotation

- Active Rotation: $\mathcal{R}^A : {}_A \mathbf{r}_{OP} \mapsto {}_A \mathbf{r}_{OQ}$ (rotates the vector \mathbf{r}_{OP})
- Passive Rotation: $\mathcal{R}^P : {}_A \mathbf{r}_{OP} \mapsto {}_B \mathbf{r}_{OP}$ (rotates the frame $(\mathbf{e}_x^I, \mathbf{e}_y^I, \mathbf{e}_z^I)$)
- Elementary Rotations
 ${}_I \mathbf{r}_{OP} = \mathbf{C}_{IB} {}_B \mathbf{r}_{OP}$
 around z-axis: $\mathbf{C}_{IB} = \begin{bmatrix} \cos \theta & -\sin \theta & 0 \\ \sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{bmatrix}$
 around y-axis: $\mathbf{C}_{IB} = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix}$
 around x-axis: $\mathbf{C}_{IB} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & -\sin \theta \\ 0 & \sin \theta & \cos \theta \end{bmatrix}$

- Inversion: $\mathcal{R}^{A^{-1}}(\mathbf{r}) = \mathcal{R}^P(\mathbf{r})$
 $\mathcal{R}_2^A (\mathcal{R}_1^A(\mathbf{r})) = (\mathcal{R}_2^A \otimes \mathcal{R}_1^A)(\mathbf{r}) = (\mathcal{R}_1^{A^{-1}} \otimes \mathcal{R}_2^{A^{-1}})^{-1}(\mathbf{r})$
- Concatenation:
 $\mathcal{R}_2^P (\mathcal{R}_1^P(\mathbf{r})) = (\mathcal{R}_2^P \otimes \mathcal{R}_1^P)(\mathbf{r}) = (\mathcal{R}_1^{P^{-1}} \otimes \mathcal{R}_2^{P^{-1}})^{-1}(\mathbf{r})$
- Exponential map: $\text{exp} : \mathbb{R}^3 \rightarrow \text{SO}(3), \mathbf{v} \mapsto \text{expM}(\hat{\mathbf{v}}), \quad \mathbf{v} \in \mathbb{R}^3$
- Logarithmic map: $\text{log} : \text{SO}(3) \rightarrow \mathbb{R}^3, \mathcal{R} \mapsto \text{logM}(\mathcal{R})^\vee, \quad \mathcal{R} \in \text{SO}(3)$
- Box plus: $\mathcal{R}_2 = \mathcal{R}_1 \boxplus \mathbf{v} = \text{exp}(\mathbf{v}) \otimes \mathcal{R}_1, \quad \mathcal{R}_1, \mathcal{R}_2 \in \text{SO}(3), \mathbf{v} \in \mathbb{R}^3$
- Box minus: $\mathbf{v} = \mathcal{R}_1 \boxminus \mathcal{R}_2 = \text{log}(\mathcal{R}_1 \otimes \mathcal{R}_2^{-1}), \quad \mathcal{R}_1, \mathcal{R}_2 \in \text{SO}(3), \mathbf{v} \in \mathbb{R}^3$
- Discrete integration: $\mathcal{R}_{IB}^{k+1} = \mathcal{R}_{IB}^k \boxplus ({}_I \boldsymbol{\omega}_{IB}^k \Delta t), \quad \mathcal{R}_{BI}^{k+1} = \mathcal{R}_{BI}^k \boxminus (-{}_B \boldsymbol{\omega}_{IB}^k \Delta t)$
- Discrete differential: ${}_I \boldsymbol{\omega}_{IB}^k = (\mathcal{R}_{IB}^{k+1} \boxminus \mathcal{R}_{IB}^k) / \Delta t, \quad {}_B \boldsymbol{\omega}_{IB}^k = -(\mathcal{R}_{BI}^{k+1} \boxminus \mathcal{R}_{BI}^k) / \Delta t$
 $\mathcal{R}_t = \mathcal{R}_0 \boxplus ((\mathcal{R}_1 \boxminus \mathcal{R}_0)t), \quad \mathcal{R}_t = \mathcal{R}(t), \mathcal{R}_0 = \mathcal{R}(0), \mathcal{R}_1 = \mathcal{R}(1)$
 $= (\mathcal{R}_1 \otimes \mathcal{R}_0^{-1})^t \otimes \mathcal{R}_0$
- (Spherical) linear interpolation $t \in [0, 1]$:

Rotation Parameterizations

Rotation Matrix	$\mathbf{C}_{BA} \in \text{SO}(3)$ ${}_B \mathbf{r}_{OP} = \mathbf{C}_{BA} {}_A \mathbf{r}_{OP}$ $\mathbf{C}_{BA} = \mathbf{C}_{AB}^T$	The rotation matrix (Direction Cosine Matrix) is a coordinate transformation matrix, which transforms vectors from frame A to frame B .
Rotation Quaternion	\mathbf{q}_{BA} $\mathbf{q} = [q_0 \ q_1 \ q_2 \ q_3]^T$	Hamiltonian unit quaternion (hypercomplex number) $Q = q_0 + q_1 i + q_2 j + q_3 k \in \mathbb{H}, \quad q_i \in \mathbb{R}, \quad \ Q\ = 1$
Angle-axis	$(\theta, \mathbf{n})_{BA}$	Rotation with unit rotation axis \mathbf{n} and angle $\theta \in [0, \pi]$.
Rotation Vector	$\boldsymbol{\phi}_{BA}$	Rotation with rotation axis $\mathbf{n} = \frac{\boldsymbol{\phi}}{\ \boldsymbol{\phi}\ }$ and angle $\theta = \ \boldsymbol{\phi}\ $.
Euler Angles ZYX Euler Angles YPR	$(\psi, \theta, \phi)_{BA}$	Tait-Bryan angles (Flight conv.): $z - y' - x''$, i.e. yaw-pitch-roll. Singularities are at $\theta = \pm \frac{\pi}{2}$. $\psi \in [-\pi, \pi], \theta \in [-\frac{\pi}{2}, \frac{\pi}{2}], \phi \in [-\pi, \pi]$
Euler Angles XYZ Euler Angles RPY	$(\alpha, \beta, \gamma)_{BA}$	Cardan angles (Glocker conv.): $x - y' - z''$, i.e. roll-pitch-yaw. Singularities are at $\beta = \pm \frac{\pi}{2}$. $\alpha \in [-\pi, \pi], \beta \in [-\frac{\pi}{2}, \frac{\pi}{2}], \gamma \in [-\pi, \pi]$

Rotation Quaternion

A rotation quaternion is a Hamiltonian unit quaternion:

$$Q = q_0 + q_1 i + q_2 j + q_3 k \in \mathbb{H}, \quad q_i \in \mathbb{R}, i^2 = j^2 = k^2 = ijk = -1, \quad \|Q\| = \sqrt{q_0^2 + q_1^2 + q_2^2 + q_3^2} = 1$$

Note that Q_{AB} and $-Q_{AB}$ represent the same rotation, but not the same unit quaternion.

Rot. quaternion as tuple: $Q = (q_0, q_1, q_2, q_3) = (q_0, \tilde{\mathbf{q}})$ with $\tilde{\mathbf{q}} := (q_1, q_2, q_3)^T$

Rot. quaternion as vector: $\mathbf{q} = [q_0 \ q_1 \ q_2 \ q_3]^T$

Conjugate: $Q^* = (q_0, -\tilde{\mathbf{q}})$

Inverse: $Q^{-1} = Q^* = (q_0, -\tilde{\mathbf{q}})$

Quaternion multiplication:

$$Q \cdot P = (q_0, \tilde{\mathbf{q}}) \cdot (p_0, \tilde{\mathbf{p}}) = (q_0 p_0 - \tilde{\mathbf{q}}^T \tilde{\mathbf{p}}, q_0 \tilde{\mathbf{p}} + p_0 \tilde{\mathbf{q}} + \tilde{\mathbf{q}} \times \tilde{\mathbf{p}}) \Leftrightarrow$$

$$\underbrace{\mathbf{q} \otimes \mathbf{p}}_{\text{quaternion matrix}} = \begin{pmatrix} q_0 & -\tilde{\mathbf{q}}^T \\ \tilde{\mathbf{q}} & q_0 \mathbf{1}_{3 \times 3} + \hat{\tilde{\mathbf{q}}} \end{pmatrix} \begin{pmatrix} p_0 \\ p_1 \\ p_2 \\ p_3 \end{pmatrix} = \begin{pmatrix} q_0 & -q_1 & -q_2 & -q_3 \\ q_1 & q_0 & -q_3 & q_2 \\ q_2 & q_3 & q_0 & -q_1 \\ q_3 & -q_2 & q_1 & q_0 \end{pmatrix} \begin{pmatrix} p_0 \\ p_1 \\ p_2 \\ p_3 \end{pmatrix}$$

$$= \underbrace{\tilde{\mathbf{Q}}(\mathbf{p})}_{\text{conjugate quat. matrix}} \mathbf{q} = \begin{pmatrix} p_0 & -\tilde{\mathbf{p}}^T \\ \tilde{\mathbf{p}} & p_0 \mathbf{1}_{3 \times 3} - \hat{\tilde{\mathbf{p}}} \end{pmatrix} \begin{pmatrix} q_0 \\ q_1 \\ q_2 \\ q_3 \end{pmatrix} = \begin{pmatrix} p_0 & -p_1 & -p_2 & -p_3 \\ p_1 & p_0 & p_3 & -p_2 \\ p_2 & -p_3 & p_0 & p_1 \\ p_3 & p_2 & -p_1 & p_0 \end{pmatrix} \begin{pmatrix} q_0 \\ q_1 \\ q_2 \\ q_3 \end{pmatrix}$$

Rotation Quaternion \Leftrightarrow Rotation Angle-Axis

$$\mathbf{q}_{IB} = \begin{bmatrix} \cos \frac{\theta}{2} \\ \mathbf{n} \sin \frac{\theta}{2} \end{bmatrix} \Leftrightarrow (\theta, \mathbf{n})_{IB} = \begin{cases} (2 \arccos(p_0), \frac{\tilde{\mathbf{q}}}{\|\tilde{\mathbf{q}}\|}) & \text{if } \|\tilde{\mathbf{q}}\|^2 \geq \epsilon^2 \\ (0, [1 \ 0 \ 0]^T) & \text{otherwise} \end{cases}$$

Rotation Quaternion \Leftrightarrow Direction Cosine Matrix

$$\mathbf{C}_{IB} = \mathbf{1}_{3 \times 3} + 2q_0 \hat{\tilde{\mathbf{q}}} + 2\hat{\tilde{\mathbf{q}}}^2 = (2q_0^2 - 1)\mathbf{1}_{3 \times 3} + 2q_0 \hat{\tilde{\mathbf{q}}} + 2\hat{\tilde{\mathbf{q}}} \hat{\tilde{\mathbf{q}}}^T$$

$$= \begin{bmatrix} p_0^2 + p_1^2 - p_2^2 - p_3^2 & 2p_1 p_2 - 2p_0 p_3 & 2p_0 p_2 + 2p_1 p_3 \\ 2p_0 p_3 + 2p_1 p_2 & p_0^2 - p_1^2 + p_2^2 - p_3^2 & 2p_2 p_3 - 2p_0 p_1 \\ 2p_1 p_3 - 2p_0 p_2 & 2p_0 p_1 + 2p_2 p_3 & p_0^2 - p_1^2 - p_2^2 + p_3^2 \end{bmatrix}$$

$$\mathbf{C}_{IB}^{-1} = \mathbf{C}_{BI} = \mathbf{1}_{3 \times 3} - 2p_0 \hat{\tilde{\mathbf{p}}} + 2\hat{\tilde{\mathbf{p}}}^2$$

$$= \begin{bmatrix} p_0^2 + p_1^2 - p_2^2 - p_3^2 & 2p_0 p_3 + 2p_1 p_2 & 2p_1 p_3 - 2p_0 p_2 \\ 2p_1 p_2 - 2p_0 p_3 & p_0^2 - p_1^2 + p_2^2 - p_3^2 & 2p_0 p_1 + 2p_2 p_3 \\ 2p_0 p_2 + 2p_1 p_3 & 2p_2 p_3 - 2p_0 p_1 & p_0^2 - p_1^2 - p_2^2 + p_3^2 \end{bmatrix}$$

Euler Angles ZYX \Leftrightarrow Direction Cosine Matrix

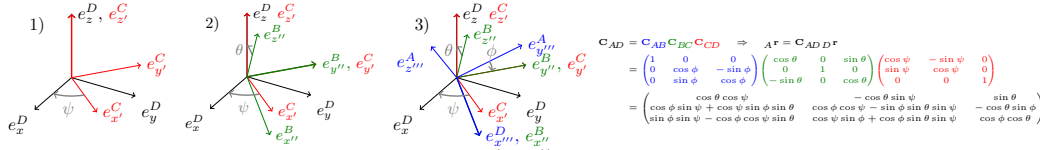


Figure 1: Rotation from A-frame to D-frame: $(z - y' - x'')$ - (yaw-pitch-roll) - $(\psi - \theta - \phi)$ - $(50^\circ - 25^\circ - 30^\circ)$

Euler Angles XYZ \Leftrightarrow Direction Cosine Matrix

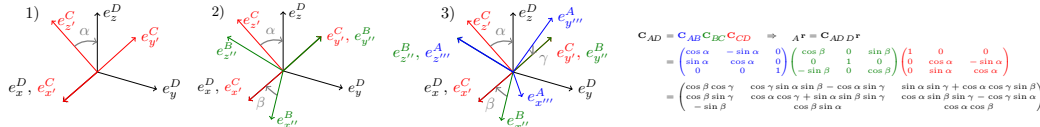


Figure 2: Rotation from A-frame to D-frame: $(x - y' - z'')$ - (roll-pitch-yaw) - $(\alpha - \beta - \gamma)$ - $(50^\circ - 25^\circ - 30^\circ)$

Pose

Homogeneous Transformation Matrix

$$\begin{bmatrix} I \mathbf{r}_{IP} \\ 1 \end{bmatrix} = \mathbf{T}_{IB} \begin{bmatrix} B \mathbf{r}_{BP} \\ 1 \end{bmatrix}, \quad \mathbf{T}_{IB} = \begin{bmatrix} \mathbf{C}_{IB} & I \mathbf{r}_{IB} \\ \mathbf{0}^T & 1 \end{bmatrix} \quad \mathbf{T}_{IB}^{-1} = \mathbf{T}_{BI} = \begin{bmatrix} \mathbf{C}_{IB}^T & -\mathbf{C}_{IB}^T I \mathbf{r}_{IB} \\ \mathbf{0}^T & 1 \end{bmatrix}$$

Time Derivatives of Position & Orientation

Linear Velocity

Velocity of point P expressed in frame B w.r.t. to the inertial frame I :

$$B \mathbf{v}_P = B \mathbf{v}_A + B \dot{\mathbf{r}}_{AP} + B \boldsymbol{\omega}_{IB} \times B \mathbf{r}_{AP}$$

Velocity of point Q on rigid body B that rotates with $B \boldsymbol{\Omega}$, where point P is on the same rigid body B :

$$B \mathbf{v}_Q = B \mathbf{v}_P + B \boldsymbol{\Omega} \times B \mathbf{r}_{PQ}, \quad B \boldsymbol{\Omega} = B \boldsymbol{\omega}_{IB}$$

Angular Velocity

$B \boldsymbol{\omega}_{IB} =: B \boldsymbol{\Omega}$	(local) absolute angular velocity of rigid body B expr. in frame B
$B \boldsymbol{\omega}_{IB} = -B \boldsymbol{\omega}_{BI}$	inverse of angular velocity
$I \boldsymbol{\omega}_{IB} = \mathbf{C}_{IB} B \boldsymbol{\omega}_{IB}$	(global) angular velocity from frame B to frame I
$I \dot{\boldsymbol{\omega}}_{IB} = \mathbf{C}_{IB} B \dot{\boldsymbol{\omega}}_{IB} \mathbf{C}_{IB}^T$	coord. transformation of angular velocity from frame B to frame I
$D \boldsymbol{\omega}_{AD} = D \boldsymbol{\omega}_{AB} + D \boldsymbol{\omega}_{BC} + D \boldsymbol{\omega}_{CD}$	composition of (relative) angular velocity

Time Derivative of Direction Cosine Matrix \Leftrightarrow Angular Velocity

$$I \dot{\boldsymbol{\omega}}_{IB} = \dot{\mathbf{C}}_{IB} \mathbf{C}_{IB}^T = \dot{\mathbf{C}}_{BI}^T \mathbf{C}_{BI} \Leftrightarrow \dot{\mathbf{C}}_{IB} = I \dot{\boldsymbol{\omega}}_{IB} \mathbf{C}_{IB}$$

$$B \dot{\boldsymbol{\omega}}_{IB} = \mathbf{C}_{IB}^T \dot{\mathbf{C}}_{IB} = \mathbf{C}_{BI} \dot{\mathbf{C}}_{BI}^T \Leftrightarrow \dot{\mathbf{C}}_{IB} = \mathbf{C}_{IB} B \dot{\boldsymbol{\omega}}_{IB}$$

Time Derivative of Rotation Quaternion \Leftrightarrow Angular Velocity

$$\begin{aligned} I \boldsymbol{\omega}_{IB} &= 2\mathbf{H}(\mathbf{q}_{IB}) \dot{\mathbf{q}}_{IB} & \Leftrightarrow & \dot{\mathbf{q}}_{IB} = \frac{1}{2} \mathbf{H}(\mathbf{q}_{IB})^T I \boldsymbol{\omega}_{IB} \\ B \boldsymbol{\omega}_{IB} &= 2\tilde{\mathbf{H}}(\mathbf{q}_{IB}) \dot{\mathbf{q}}_{IB} & \Leftrightarrow & \dot{\mathbf{q}}_{IB} = \frac{1}{2} \tilde{\mathbf{H}}(\mathbf{q}_{IB})^T B \boldsymbol{\omega}_{IB} \\ \mathbf{H}(\mathbf{q}) &= [-\hat{\mathbf{q}} \quad \hat{\mathbf{q}} + q_0 \mathbf{1}_{3 \times 3}] \in \mathbb{R}^{3 \times 4} & \mathbf{H}(\mathbf{q}) &= [-\hat{\mathbf{q}} \quad -\hat{\mathbf{q}} + q_0 \mathbf{1}_{3 \times 3}] \in \mathbb{R}^{3 \times 4} \\ &= \begin{bmatrix} -q_1 & q_0 & -q_3 & q_2 \\ -q_2 & q_3 & q_0 & -q_1 \\ -q_3 & -q_2 & q_1 & q_0 \end{bmatrix} & & = \begin{bmatrix} -q_1 & q_0 & q_3 & -q_2 \\ -q_2 & -q_3 & q_0 & q_1 \\ -q_3 & q_2 & -q_1 & q_0 \end{bmatrix} \end{aligned}$$

Time Derivative of Angle-Axis \Leftrightarrow Angular Velocity

$$\begin{aligned} I \boldsymbol{\omega}_{IB} &= \mathbf{n} \dot{\theta} + \dot{\mathbf{n}} \sin \theta + \mathbf{n} \dot{\mathbf{n}} (1 - \cos \theta) \\ B \boldsymbol{\omega}_{IB} &= \mathbf{n} \dot{\theta} + \dot{\mathbf{n}} \sin \theta - \mathbf{n} \dot{\mathbf{n}} (1 - \cos \theta) \\ \dot{\theta} &= \mathbf{n}^T I \boldsymbol{\omega}_{IB}, \quad \dot{\mathbf{n}} = \left(-\frac{1}{2} \frac{\sin \theta}{1 - \cos \theta} \dot{\mathbf{n}}^2 - \frac{1}{2} \dot{\mathbf{n}} \right) I \boldsymbol{\omega}_{IB} \quad \forall \theta \in \mathbb{R} \setminus \{0\} \\ \dot{\theta} &= \mathbf{n}^T B \boldsymbol{\omega}_{IB}, \quad \dot{\mathbf{n}} = \left(-\frac{1}{2} \frac{\sin \theta}{1 - \cos \theta} \dot{\mathbf{n}}^2 + \frac{1}{2} \dot{\mathbf{n}} \right) B \boldsymbol{\omega}_{IB} \quad \forall \theta \in \mathbb{R} \setminus \{0\} \end{aligned}$$

Time Derivative of Rotation Vector \Leftrightarrow Angular Velocity

$$I \boldsymbol{\omega}_{IB} = \left(\mathbf{1}_{3 \times 3} + \hat{\phi} \left(\frac{1 - \cos \|\phi\|}{\|\phi\|^2} \right) + \hat{\phi}^2 \left(\frac{\|\phi\| - \sin \|\phi\|}{\|\phi\|^3} \right) \right) \dot{\phi} \quad \forall \|\phi\| \in \mathbb{R} \setminus \{0\}$$

$$B \boldsymbol{\omega}_{IB} = \left(\mathbf{1}_{3 \times 3} - \hat{\phi} \left(\frac{1 - \cos \|\phi\|}{\|\phi\|^2} \right) + \hat{\phi}^2 \left(\frac{\|\phi\| - \sin \|\phi\|}{\|\phi\|^3} \right) \right) \dot{\phi} \quad \forall \|\phi\| \in \mathbb{R} \setminus \{0\}$$

$$\dot{\phi} = \left(\mathbf{1}_{3 \times 3} - \frac{1}{2} \hat{\phi} + \hat{\phi}^2 \frac{1}{\|\phi\|^2} \left(1 - \frac{\|\phi\|}{2} \frac{\sin \|\phi\|}{1 - \cos \|\phi\|} \right) \right) I \boldsymbol{\omega}_{IB} \quad \forall \|\phi\| \in \mathbb{R} \setminus \{0\}$$

$$\dot{\phi} = \left(\mathbf{1}_{3 \times 3} + \frac{1}{2} \hat{\phi} + \hat{\phi}^2 \frac{1}{\|\phi\|^2} \left(1 - \frac{\|\phi\|}{2} \frac{\sin \|\phi\|}{1 - \cos \|\phi\|} \right) \right) B \boldsymbol{\omega}_{IB} \quad \forall \|\phi\| \in \mathbb{R} \setminus \{0\}$$

Time Derivative of Euler Angles ZYX \Leftrightarrow Angular Velocity

$$\begin{bmatrix} \dot{\psi} \\ \dot{\theta} \\ \dot{\phi} \end{bmatrix} = \begin{bmatrix} -\cos \psi \sin \theta & (\sin \theta \sin \psi) & 1 \\ \cos \theta & \cos \psi & 0 \\ \sin \psi & -\sin \psi & 0 \end{bmatrix} B \boldsymbol{\omega}_{IB} \quad \forall \theta \in \mathbb{R} \setminus \left\{ \frac{\pi}{2} + k\pi \right\}, k \in \mathbb{Z}$$

$$B \boldsymbol{\omega}_{IB} = \begin{bmatrix} 0 & \sin \psi & \cos \theta \cos \psi \\ 0 & \cos \psi & -\cos \theta \sin \psi \\ 1 & 0 & \sin \theta \end{bmatrix} \begin{bmatrix} \dot{\psi} \\ \dot{\theta} \\ \dot{\phi} \end{bmatrix}$$

Dynamics of a Multi-Rigid-Body System

n	Number of bodies in system
n_j	Number of DoFs of the joints
n_q	Number of generalized coordinates
n_u	Number of generalized velocities
\mathbf{M}	Mass matrix
\mathbf{g}	Gyroscopic and Coriolis forces
\mathbf{f}	Generalized external forces and torques
\mathbf{h}	Combined force vector
\mathbf{J}_P	Jacobi matrix for translation of point P
\mathbf{J}_R	Jacobi matrix for rotation
\mathbf{F}_Q^A	External forces on point Q
\mathbf{M}^A	External torques
m	Mass
Θ	Inertia tensor
$(\dots)^-$	Variable before impact
$(\dots)^+$	Variable after impact
$(\dots)^\pm$	Variable before/after impact
Δt	Time step duration
$\Delta \mathbf{u}$	Velocity change over one time step
\mathbf{W}	Generalized force directions for contact forces
λ	Lebesgue-measurable contact forces
Λ	Purely atomic impact impulses
\mathbf{P}	Contact percussions

Generalized Coordinates of a Floating-Base System with Rotational Joints

Recommended set of generalized coordinates \mathbf{q} with quaternion \mathbf{q}_{IB} and generalized velocities \mathbf{u} :

$$\mathbf{q} = \begin{pmatrix} I \mathbf{r}_{IB} \\ \mathbf{q}_{IB} \\ \varphi_1 \\ \vdots \\ \varphi_{n_j} \end{pmatrix} \in \mathbb{R}^{7+n_j} = \mathbb{R}^{n_q} \quad \mathbf{u} = \begin{pmatrix} I \mathbf{v}_B \\ B \boldsymbol{\omega}_{IB} \\ \dot{\varphi}_1 \\ \vdots \\ \dot{\varphi}_{n_j} \end{pmatrix} \in \mathbb{R}^{6+n_j} = \mathbb{R}^{n_u} \quad \dot{\mathbf{u}} = \begin{pmatrix} I \mathbf{a}_B \\ B \boldsymbol{\psi}_{IB} \\ \ddot{\varphi}_1 \\ \vdots \\ \ddot{\varphi}_{n_j} \end{pmatrix} \in \mathbb{R}^{6+n_j}$$

$$\dot{\mathbf{q}} = \mathbf{F}\mathbf{u}, \quad \mathbf{F} = \begin{pmatrix} \mathbb{1}_{3 \times 3} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \frac{1}{2} \bar{\mathbf{H}}^\top & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathbb{1}_{n_j \times n_j} \end{pmatrix} \Leftrightarrow \mathbf{u} = \bar{\mathbf{F}} \dot{\mathbf{q}}, \quad \bar{\mathbf{F}} = \begin{pmatrix} \mathbb{1}_{3 \times 3} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & 2\bar{\mathbf{H}} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathbb{1}_{n_j \times n_j} \end{pmatrix}$$

Equations of Motion with Contacts and no Impulses

Projected Newton-Euler Equations

$$\mathbf{M} = \sum_{i=1}^n \left[(\mathbf{J}_S^\top m \mathbf{J}_S + \mathbf{J}_R^\top \Theta_S \mathbf{J}_R) \right]_i$$

$$\boxed{\mathbf{M} \dot{\mathbf{u}} - \mathbf{h} = \mathbf{W} \lambda} \text{ with } \mathbf{h} := \mathbf{f} - \mathbf{g}, \text{ and } \mathbf{g} = \sum_{i=1}^n \left[(\mathbf{J}_S^\top m \mathbf{J}_S \mathbf{u} + \mathbf{J}_R^\top (\Theta_S \dot{\mathbf{J}}_R \mathbf{u} + \Omega \times \Theta_S \Omega)) \right]_i$$

$$\mathbf{f} = \sum_{i=1}^n \left[(\mathbf{J}_Q^\top \mathbf{F}_Q^A + \mathbf{J}_R^\top \mathbf{M}^A) \right]_i$$

Equations of Motion with Contacts and Impulses

$$\boxed{\mathbf{M} \Delta \mathbf{u} - \mathbf{h} \Delta t = \mathbf{W} \mathbf{P}} \quad \left\{ \begin{array}{l} \mathbf{M}(\mathbf{u}^+ - \mathbf{u}^-) = \mathbf{W} \Lambda \\ \mathbf{M}(\underbrace{\dot{\mathbf{u}} dt + (\mathbf{u}^+ - \mathbf{u}^-) d\eta}_{d\mathbf{u}}) - \mathbf{h} dt = \mathbf{W}(\underbrace{\lambda dt + \Lambda d\eta}_{d\mathbf{P}}) \end{array} \right.$$

Transformation of Equations of Motion

Transformation from $\bar{\mathbf{M}}(\bar{\mathbf{q}}), \bar{\mathbf{h}}(\bar{\mathbf{q}}, \bar{\mathbf{u}})$ to $\mathbf{M}(\mathbf{q}), \mathbf{h}(\mathbf{q}, \mathbf{u})$, where $\bar{\mathbf{u}} = \mathbf{B}\mathbf{u}$:
 $\mathbf{M} = \mathbf{B}^\top \bar{\mathbf{M}} \mathbf{B}$
 $\mathbf{h} = \mathbf{B}^\top \bar{\mathbf{h}} - \mathbf{B}^\top \bar{\mathbf{M}} \dot{\mathbf{B}} \mathbf{u}$

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