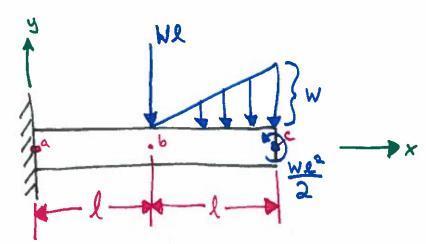
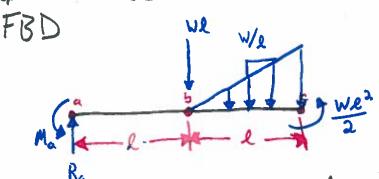
EMEISUA FALL2016 LECTURE 16 Wednesday, Oct 26, 2016
Chapters in book to review
1-1 to 1-16
3-1 to 3-14, 3-18, 3-19
4-1 to 4-6
Main Topics
- Uncertainity, reliability, tolerances
- Design Factor and Factor of Safety
- State of stress/strain (prinicipal, max stear,
- Deflection of busic elements (stiffness)
- Stress and Deflection of Beams (method of sections, [SMpe-possition, Lsing, functions)
Voted Topics
5 multi-axial stress (stress tensor)
3 Sing, functions
3 transverse shear
a strain
2 Mohr's circle
2 equations



Find the slope and deflection at the end (x=2l) using Singularity functions.

Step 1: Draw FBD



Step 2: Write singularity function for loads, include all loads!

$$-\frac{Wl^{2}}{2} < x-2l > 2$$

Step 3: Integrate twice to find V(x) and M(x). Ignove constants of integration, because we will use the lt trick to find reactions.

$$V(x) = R_{a}(x-0)^{\circ} - M_{a}(x-0)^{-1} - Wl(x-1)^{\circ} - \frac{W}{2l}(x-1)^{2}$$

$$- \frac{Wl^{2}}{2}(x-2l)^{-1}$$

M(x)= Ra <x-0>'- Ma <x-0>'- Wl <x-l>'- 6l <x-l>'

Step 4: Solve for reactions by substituting x=21t, i.e. location just outside the beam. V(21+) = Ra <21+-07 - Ma <21+-0> - We <21+-12> - W < 22+-e>2 - We2 < 2e+-2e>1 $V(2e^{t})=0=R_{a}(1)-M_{a}(0)-WL(1)-\frac{W}{5e}(L)^{2}$ - We (0) 0= Ra-Wl- Wl => [Ra= 3wl] M(2l+)=0= Ra <2l+-0>'- Ma <2l+-0>"- wl <2l+-0>" - W < 22+-27 - W22 < 21+-22) 0 = Ra (21) - Ma (1) - Wl(21) - W(1) - Wl2 (1) Ma = 3we2 - 2we2 - 8e2 - we2 Ma= We2 Step 5: Substitute in reactions and drop singularities at X=21.

V(x) = 3 we <x-0> - We <x-e> - We <x-e> = M(x) = 3we(x-0)'- 42 <x-0> - we(x-e)- 2 <x-e>3 Step 6: Integrate twice more to get OCX), V(X).
Make sure to include EI and constants of
integration.

EIO(x) =
$$\frac{3}{4}$$
 wl < x-0> - $\frac{We^2}{3}$ < x-0> - $\frac{W}{2}$ < x-2> = $\frac{W}{242}$ < x-2> + Co = $\frac{3}{12}$ wl < x-0> - $\frac{W}{6}$ < x-0> - $\frac{W}{6}$ < x-2> = \frac{W}

Step 7: Solve for constants of integration using the boundary conditions.

$$\Theta(0) = 0 = \frac{3}{4}wl(0-0)^{2} - \frac{wl^{2}}{3}(0-0)^{1} - \frac{wl}{3}(0) - \frac{wl}{24l}(0) + C_{0}$$

$$C_{0} = 0$$

$$Y(0) = 0 = \frac{3}{12}wl(0-0)^{3} - \frac{wl^{2}}{6}(0-0)^{2} - \frac{wl}{6}(0) - \frac{wl}{120l}(0) + C_{0}(0)$$

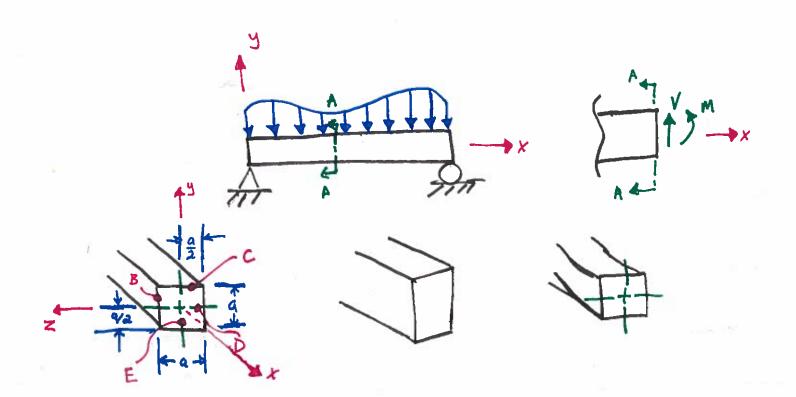
$$+ C_{y}$$

Step 8: Write expressions for O(21) and y (21)

$$\Theta(2l) = \frac{1}{EI} \left(\frac{3}{4} w l (2l)^{2} - \frac{w l^{2}}{3} (2l)^{2} - \frac{W l}{3} (2l)^{2} - \frac{W}{3} (2l-l)^{2} - \frac{W}{34l} (2l-l)^{4} \right)
\Theta(2l) = \frac{1}{EI} \left(\frac{3}{4} w l^{3} - \frac{2}{3} w l^{3} - \frac{1}{2} w l^{3} - \frac{1}{24} w l^{3} \right) = \frac{1}{EI} \frac{43}{24} w l^{3}$$

$$y(2l) = \frac{1}{EI} \left(\frac{1}{4} w l (2l)^{3} - \frac{w l^{2}}{6} (2l)^{2} - \frac{W l}{6} (2l-l)^{3} - \frac{W}{120l} (2l-l)^{5} \right)$$

$$y(2l) = \frac{1}{EI} \left(\frac{2}{4} w l^{4} - \frac{2}{3} w l^{4} - \frac{w l^{3}}{6} - \frac{w l^{4}}{120} \right) = \frac{1}{EI} \frac{139}{120} w l^{4}$$

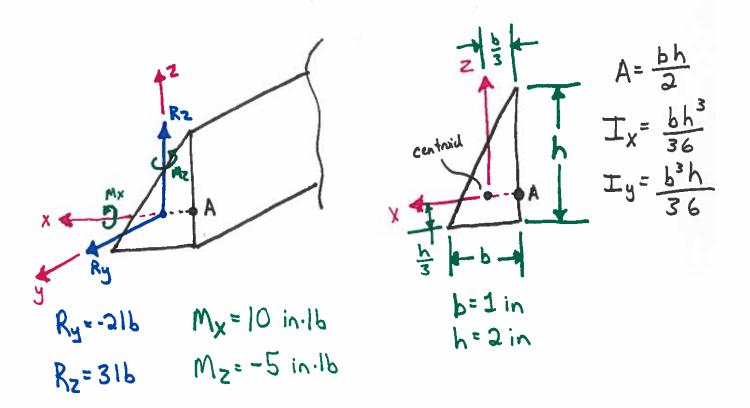


B: Compression and transverse shear

C: compression

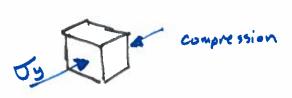
D: transverse shear

E: tension and transverse shear



Write the Cauchy Stress Tensor for point A and draw the correspond stress cudoe.

Axial



$$|\nabla_y| = \frac{|R_y|}{A} = \frac{|-216|}{1 \text{ in}^2} = 2 \text{ psi}$$

Normal due to bending about x A is on the neutral axis 50 Ty=0

Normal due to bending about Z at A there is tension | 541 = (5 in/b) (13") = 3.75 psi (2')3(2in)

Sheer

We have shear in the 2 direction on the y face and the is a moment about X, so we have to compute the transverse shear stress.

$$|T| = \frac{|R|Q}{|T|} = \frac{1}{3} \cdot \frac{2}{3} \cdot \frac{2}{3} \cdot \frac{2}{9} \cdot \frac{2$$

$$A' = \frac{2}{3}b \cdot \frac{2}{3}h = \frac{2}{9}bh$$

$$|\mathcal{T}_{zy}| = |316|\overline{9}'A'| = |316|\overline{9}(2'')\overline{9}(1'')(2'')| = |31\frac{16}{9}|$$

$$\overline{(1'')(2'')^3} \cdot \frac{3}{3}(1'') = \frac{8}{36} \cdot \frac{3}{3}$$