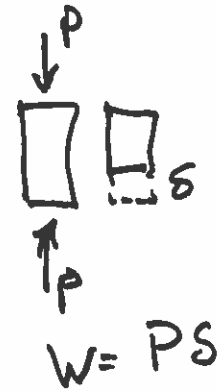


# Strain Energy

$W$ : Work done on the material by a load

$Q$  to achieve a deflection  $y$ .



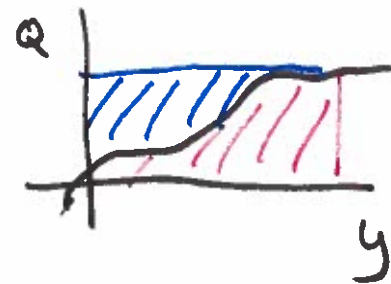
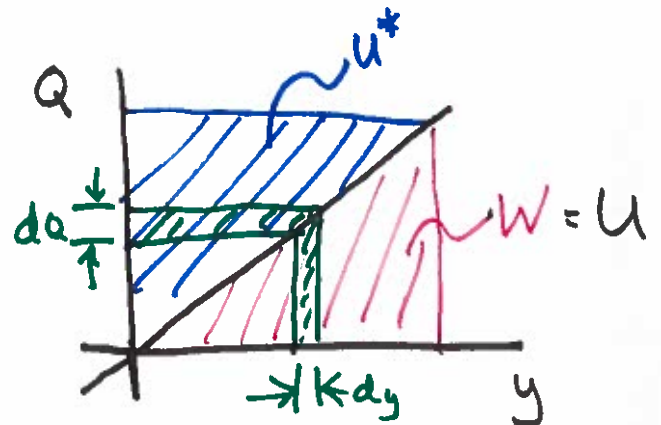
$$W = \int Q(y) dy$$

Work cause a potential energy to be stored in the material.

"Strain energy"  $\Rightarrow U$

$$W = U$$

$U^* \Rightarrow$  complementary energy



For linear only  
 $dU^* = dU$

$$U^* = \int y(Q) dQ \Rightarrow dU^* = y dQ$$

$$dU^* = dU = y dQ = Q dy \Rightarrow$$

- Valid for single applied loads on structural element

$$y = \frac{dU}{dQ}$$

①

If element is subjected to multiple loads,  $Q_i$ , all within the elastic range:

The deflection,  $y_i$ , associated with the point of application of  $Q_i$  is:

$$\boxed{y_i = \frac{\partial U}{\partial Q_i}} \Rightarrow \text{Castigliano's Theorem}$$

$U$ : total strain energy

$Q_i$ : single applied load

$y_i$ : deflection at the point of load

: When an element is elastically deflected by a combination of loads, the deflection at any point, in any direction is equal to the partial derivative of the total strain energy wrt load at that point acting in the direction.

• The applied load may or may not exist.

$L, A, E \Rightarrow$  constant wrt to  $P$

$$U^* = \int y(Q) dQ = \int \frac{PL}{AE} dP$$

$$U = U^* = \frac{P^2 L}{2AE}$$

$$\delta = \frac{PL}{AE} \Rightarrow y = \frac{QL}{AE}$$

*as func of displacement*

Load	Factors	Strain Energy Constant	Strain Energy, variable factors
Axial	$A, E, P$	$U = \frac{P^2 L}{2AE}$	$U = \int_0^L \frac{P^2}{2EA} dx$
Bending	$I, E, M$	$U = \frac{M^2 L}{2EI}$	$U = \int_0^L \frac{M^2}{2EI} dx$
Torsion	$J, G, T$	$U = \frac{T^2 L}{2GJ}$	$U = \int_0^L \frac{T^2}{2GJ} dx$
Shear	$A, G, V$	$U = \frac{CV^2 L}{2AG}$	$U = \int_0^L \frac{CV^2}{2AG} dx$

Table 4-1

Cross sections

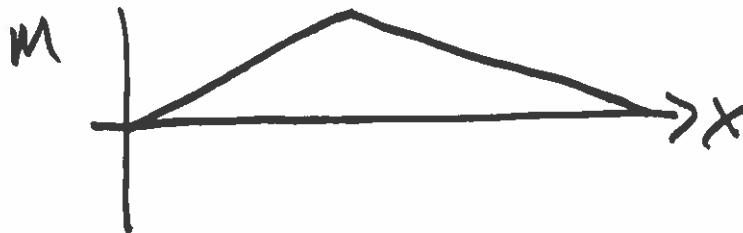
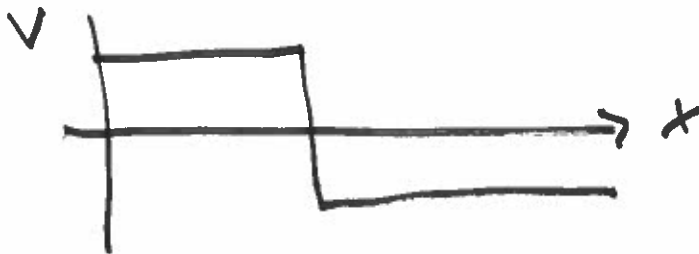
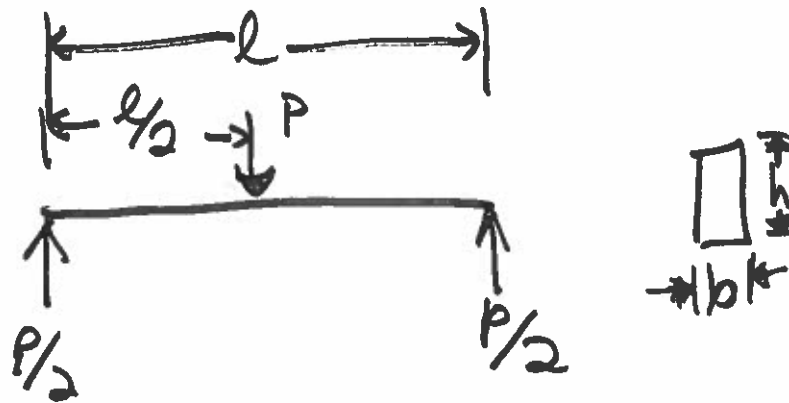
Cross sec

Rectangular

Circular

$\frac{C}{1.2}$   
 $1.11$

Example Find the deflection at the midpoint.



Solution : Beam has bending and shear.

$$V = \begin{cases} P/2 & 0 \leq x \leq \frac{l}{2} \\ -P/2 & \frac{l}{2} \leq x \leq l \end{cases}$$

$$M = \begin{cases} \frac{P}{2}x & 0 \leq x \leq \frac{l}{2} \\ \frac{Pl}{2} - \frac{Px}{2} & \frac{l}{2} \leq x \leq l \end{cases}$$

## Bending Strain Energy

$$U_{(1)} = \int_0^{L/2} \frac{M^2}{2EI} dx$$

$$U_{(1)} = U_{(2)} = \int_{L/2}^L \frac{M^2}{2EI} dx$$

↖ left                  ↗ right

$$U_{(3)} = \int_0^L \frac{CV^2}{2AG} dx$$

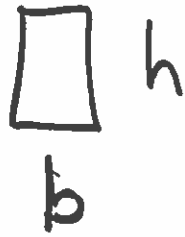
## Total

$$U = U_{(1)} + U_{(2)} + U_{(3)}$$

$$U = 2 \int_0^{L/2} \frac{M^2}{2EI} dx + \int_0^L \frac{CV^2}{2AG} dx$$

$$U = \frac{P^2 L^3}{96EI} + \frac{15P^2 L}{AG}$$

$$y\left(\frac{l}{3}\right) = \frac{\partial U}{\partial P} = \frac{PL^3}{48EI} + \frac{3PL}{AG}$$



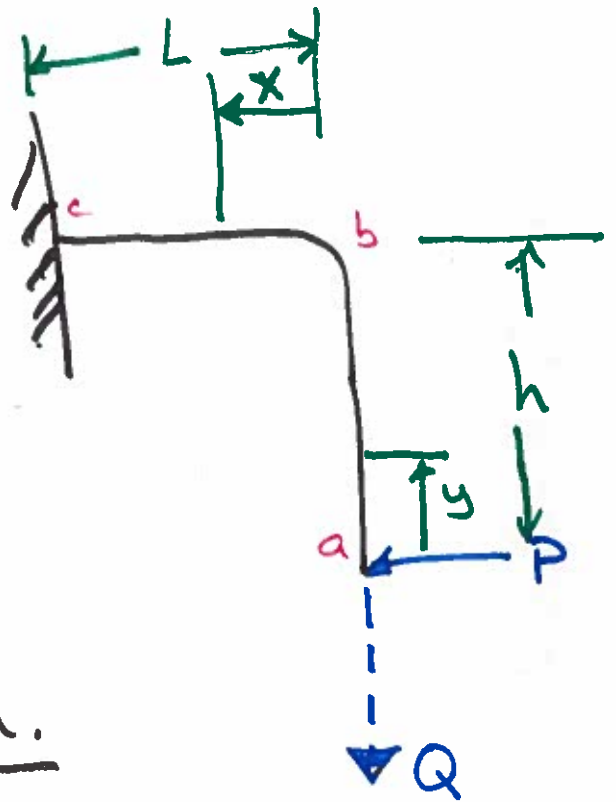
$$C = 1.2$$

Not present  
in  
table  
A-9

For  $l/h > 10 \Rightarrow$  shear typically  
negligible

## Example

Find the vertical  
deflection at the  
free end.



## Solution

Dummy load Q is needed.

- 1) Bending ab,  $M_{ab} = Py$  varying
- 2) Bending in bc,  $M_{bc} = Qx + Ph$  varying
- 3) Tension in ab,  $Q$  constant
- 4) compression cb,  $P$  constant

neglecting transverse shear

$$U = \int_0^h \frac{M_{ab}^2}{2EI} dy + \int_0^L \frac{M_{bc}^2}{2EI} dx + \frac{Q^2 h}{2EA} + \frac{P^2 L}{2EA}$$

$$U = \frac{P^2 h^3}{6EI} + \frac{Q^2 L^3}{6EI} + \frac{PQhL^2}{2EI} + \frac{P^2 h^2 L}{2EI} + \frac{Q^2 h}{2EA} + \frac{P^2 L}{2EA}$$

$$\delta_y = \frac{\partial U}{\partial Q} \Big|_{Q=0} = 0 + 0 + \frac{PhL^2}{2EI} + 0 + 0 + 0$$

$$\boxed{\delta_y = \frac{PhL^2}{2EI}}$$

Often simpler to bring derivative inside integral!

eg. Bending

See page 180 in book

$$y_i = \frac{\partial U}{\partial Q_i} = \frac{\partial}{\partial Q_i} \left( \int \frac{m^2}{2EI} dx \right)$$

$$= \int \frac{\partial}{\partial Q_i} \left( \frac{m^2}{2EI} \right) dx$$

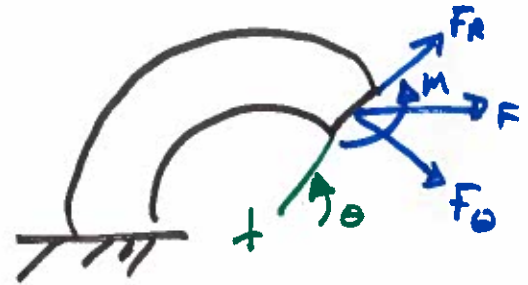
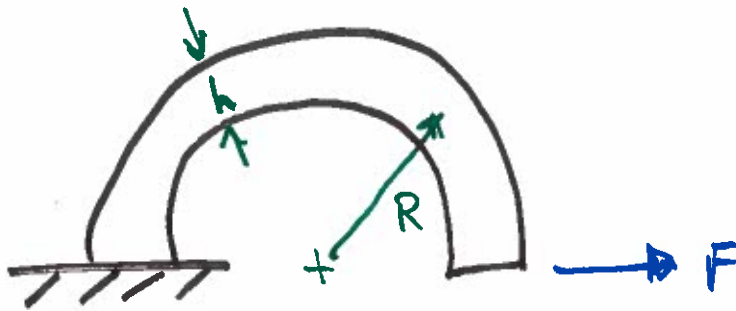
$$y_i = \int \frac{1}{EI} \left( m \frac{\partial m}{\partial Q_i} \right) dx$$

Ex 4-10 Study





## Circular



Moment alone

$$U_1 = \int \frac{m^2 d\theta}{2AeE}$$

$$e = R - r_n$$

axial

$$U_2 = \int \frac{F_\theta^2 R d\theta}{2AE}$$

Coupling

$$U_3 = - \int \frac{MF_\theta d\theta}{AE}$$

Transverse Shear

$$U_4 = \int \frac{c F_r^2 R d\theta}{2AG}$$