Chapters to review
1.1-1.6, 1.8-1.10
2.1-2.5, 2.7-2.9
3.1

## Main Topics

- deriving equations of motion for I DoF Systems using Lagrange's method. We've linearized about equilibrium points.
- Stability of linear form
- numerically simulated non-linear & linear Folks
- unforced (free) response with and without damping
- harmonically forced systems! With turthout damping
- non-linear: coulomb, aero, etc
- equivalent mass, stiffness, and damping
- Specific models: base excitation and unbalanced rotating masses
- -impulse response

## Votes

deriving EOM

unbalanced mass

base excitation

what egs?

impulse

compound pendulum

Equations

$$mx + cx + kx = F$$
 $x + 23 cmx + cmx = F/m = f$ 
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 $x + 23 cmx + cmx = G/m$ 
 $x + 23 cmx = G/m$ 
 $x + 23$ 

Solutions to ODES

Equirclent damping: table of Ceq S(t-T) H(t-T)Airac Delta Heavside unit impulse unit step

Unbalanced Mass

offset distance

offset distance

offset distance

offset distance

offset distance

offset distance

of the last driving

frequency

of the

system

 $X_{\rho}(t) = X \sin(\omega_{r}t - \theta)$   $X_{\rho}(t) = \frac{X}{m} \sin(\omega_{r}t - \theta)$   $X_{\rho}(t) = \frac{W}{m} \cos(\omega_{r}t - \theta)$ 

Mae => 10 non-dimensional

$$\frac{d}{dt}\left(\frac{\partial L}{\partial \dot{\theta}}\right) - \frac{\partial L}{\partial \dot{\theta}} = Q_{i}$$

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$$Q_{i} = \frac{\partial R}{\partial \dot{\theta}}$$

$$T = \frac{1}{2}m(l_3\theta)^2$$

$$U = \frac{1}{2}k(l_1 \sin \theta)^2$$

$$L = \frac{1}{2}m(l_0\theta)^2$$

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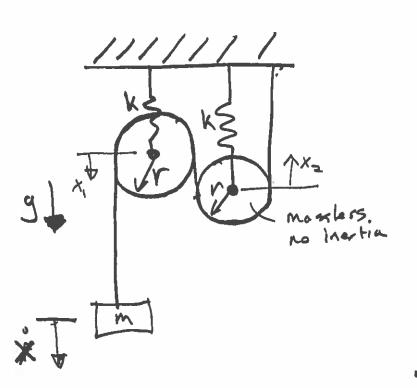
$$U = \frac{1}{3} \times (l, \theta)^{2} + mg l_{3} \sin \theta \qquad -\frac{1}{3} \times (l, \theta)^{2}$$

$$U = \frac{1}{3} \times (l, \theta)^{2} + mg l_{3} \theta \qquad \frac{1}{2} = 1$$

$$U = \frac{1}{3} \times (l, \theta)^{2} + mg l_{3} \theta \qquad \frac{1}{2} = 1$$

$$R = \frac{1}{3}c(l_0\theta)^2$$

$$\frac{d}{dt}\left(ml_3^2\dot{\theta}\right) - \left(-Kl_1^2\theta - mgl\right) = -Cl_2^2\dot{\theta}$$



$$T = \frac{1}{2} m \dot{x}^{2}$$

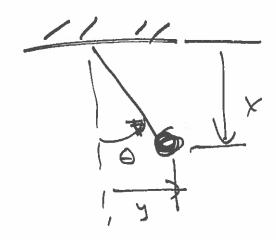
$$U = \frac{1}{2} k (\frac{1}{4} x)^{2} + \frac{1}{2} k (\frac{1}{4} x)^{2}$$

$$- Mg x$$

$$X = 3x' + 3x' = 4x'$$

$$X' = x^{2}$$

$$X = 5x^{2} + 5x'$$



Buse Excitation

mx + cx + Kx = c Twb coswb t + K Ism wb t magnitude dequency of input of excitation K& JC YSIN(wst) m=100 kg C = 50 kg/s  $X(t) = X sin(\omega t - \Theta)$  K = 1000 N/m $X = \frac{1 + (25r)^2}{(1-r^3)^2 + (34r)^3}$   $W_b = \frac{3 \text{ rad/5}}{(1-r^3)^2 + (34r)^3}$  What is the displacent of the mass?Wn= \[ \frac{h}{m} = \frac{1000}{10} = 3.16 rays underdamped S= C = 50 kg/s = 0.079 2/1008.(3,16 m/s) 01521  $\Gamma = \frac{\omega_b}{\omega_n} = 0.949$  $X = \frac{1 + (2(0.079)(0.949))^{2}}{(1-0.949^{2})^{2} + [2(0.079)(0.949)]^{2}} = 5.62$  Extra Notes

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#### Main Topics

- deriving the equations of motion for I DoF systems using Layrange's method.
- un forced (free) response with and without damping
- Stability of linear systems
- numerical simulation of non-lin systems
- Non-linear dumping: coulomb, aero, etc
- harmonically forced systems: dumped + undamped
- specific models: base excitation, unbalanced mass in notating machines
- equivalent mass, stiffness, damping
- impulse response

### Equations

$$\omega_n = \sqrt{\frac{1}{m}} \quad \mathcal{J} = \frac{c}{2\omega_n m} \quad \omega_d = \omega_n \sqrt{1 - \mathcal{J}^2}$$

$$T = \frac{a\pi}{\omega_n} \qquad f_n = \frac{\omega_n}{a\pi}$$

$$\frac{d}{dt}\left(\frac{\partial L}{\partial q_i}\right) - \frac{\partial L}{\partial q_i} = Q_i \qquad L = T - U$$

Torsional Helical Tenson/Comp
$$K = \frac{GJP}{I}$$

$$K = \frac{Gd^4}{64nR^3}$$

Cantilever

$$K = \frac{k_1 k_2}{k_1 + k_2}$$

ODE solution forms

Damping models tuble

Undamped Free Harmonic Motion

$$MX' + KX = O$$
 $X(t) = a, e^{j\omega_n t} + a_2 e^{j\omega_n t}$ 
 $\omega_n = \sqrt{\frac{1}{m}}, j = \sqrt{1}$ 

or

 $X(t) = A \sin(\omega_n t + \alpha)$ 
 $\Delta = \frac{(\omega_n^2 X^2 + v_0^2)}{\omega_n}$ 
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Vo+Ban Xo

Overdamped \$7|

X(t) = e^{-Junt}(a\_1e^{-U\_1V\_2^2-1}t\_2) + a\_2e^{+U\_1V\_2^2-1}t\_2

$$a_1 = \frac{-V_0 + (-J + V_2^2)}{2U_1V_2^2-1}U_1X_0$$

$$a_2 = \frac{V_0 + (J + [J^2-1])U_1X_0}{2U_1V_2^2-1}$$

Critically damped  $J = 1$ 

X(t) =  $(a_1 + a_2t_1)e^{-U_1t_2}$ 

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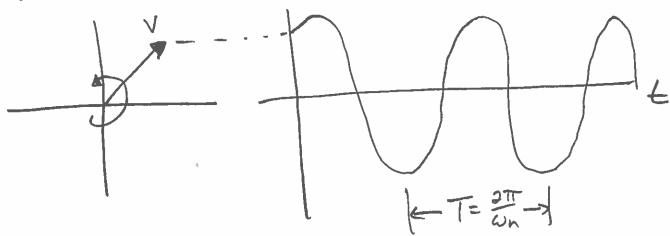
Stability

If effective stiffness is negative for undamped system =) in stability.

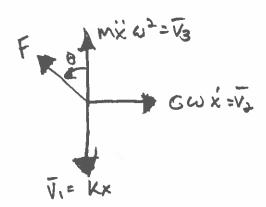
effective damping to could lead to instability.

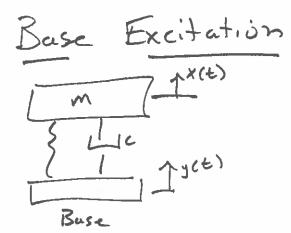
mx + umg sgn(x) + kx = 0





Sum of phasors is just vector addition





 $M\ddot{x} + c\dot{x} + kx = cY\omega_b \cos\omega_b + kY \sin\omega_b + c(\dot{x}-\dot{y}) + k(x-y) = 0$   $y(t) = Y \sin\omega_b t$ 

Un bulanced mass

mx+cx+Kx=m,ewn'sin wrt

Xo(t) = X sin(wrt-6)

 $X = \frac{M_0e}{M_0} \frac{N_2}{\sqrt{(1-N_2)^2 + (32N_2)^2}}$ 

Q = arctan 2Jn

Equivalent mass, stiffness

Ten = Tact Ven = Wact Stiffress

AF & Facx = 1 cxxdt

DE=Trank Viscous

DE = DEact

aen: Cey = 8 XWX

Coulombi Ceq = 4 mg

hystertic

DE: over in loop

Ceq = KB = h

Impulse Response

Kinderdamped:

mx + cx + Kx = F(+)

F(t)=FAE

X(t)= Fe Sin adt

X(t)= Fh(t)

h(t) = 1 e sin out

X sind=X 0 is smill |J|=lô Vx= locos G \Vx1= |V| cos6 = 1√1 U= = = Kx2+=Kx2-mg(1/2-=cos0) d-light = (d-lesso)2+ (d-lsino)2 (d-1)2+(1+2=d212-2d1.

$$\frac{2.58}{Y} < 0.55 \quad \frac{X}{Y} = \frac{1 + (2 \times n)^2}{(1 - r^2)^2 + (2 \times r)^2}$$

$$= \frac{1 + (2 \times r)^2}{(1 - r^2)^2 + (2 \times r)^2}$$

$$= \frac{1 + (2 \times r)^2}{(1 - r^2)^2 + (2 \times r)^2}$$

$$= \frac{1 + (2 \times r)^2}{(2 \times r)^2} < 0.55^2 (1 - r^2)^2 + 0.55^2 (2 \times r)^2$$

$$= \frac{1 + (2 \times r)^2}{(2 \times r)^2} < 0.55^2 (1 - r^2)^2 - 1$$

$$= \frac{1 + (2 \times r)^2}{(2 \times r)^2} < 0.55^2 (1 - r^2)^2 - 1$$

$$= \frac{1 + (2 \times r)^2}{(2 \times r)^2} = \frac{1 + ($$