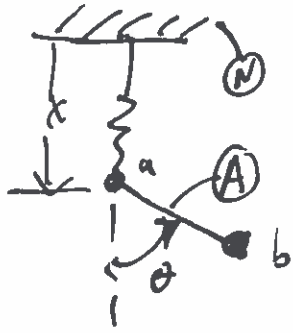
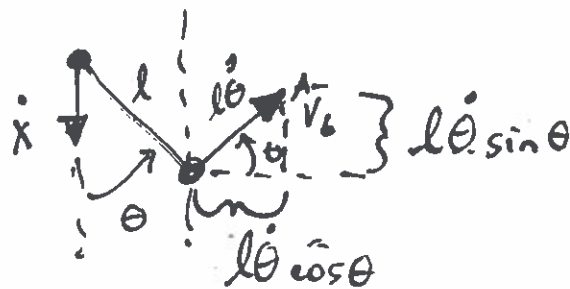


Corrected Velocity From Buggy Compound Pendulum

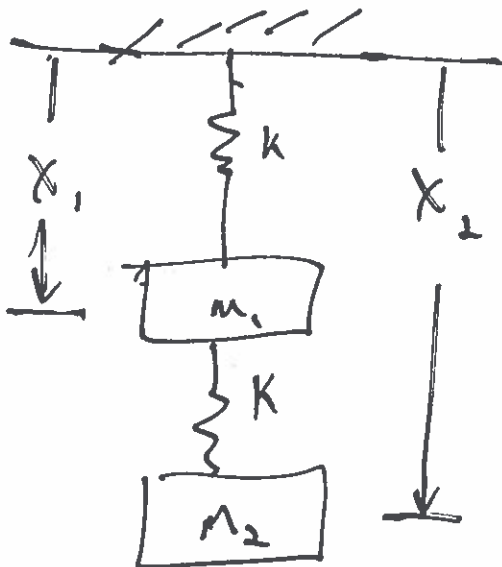
$${}^N\vec{V}_b = {}^N\vec{V}_a + \underbrace{{}^N\omega^A \times \vec{r}_{a/b}}_{{}^A\vec{V}_b}$$

$$|\vec{V}_b| = l\dot{\theta}$$



$$|\vec{V}_b| = \sqrt{(\dot{x} - l\dot{\theta}\sin\theta)^2 + (l\dot{\theta}\cos\theta)^2}$$

$$T = \frac{1}{2} m (\dot{V}_b)^2 + \dots$$

Potential Energy

$$U = \frac{1}{2} k (x_1)^2 + \frac{1}{2} K (x_2 - x_1)^2 - m_1 g x_1 - m_2 g x_2$$



$$x_1 = 0, x_2 = 0$$

## Viscous Damping

$$m\ddot{x} + c\dot{x} + kx = 0$$

or

$$\ddot{x} + 2\zeta\omega_n\dot{x} + \omega_n^2 x = 0$$

$$\zeta = \frac{c}{2\sqrt{km}}$$

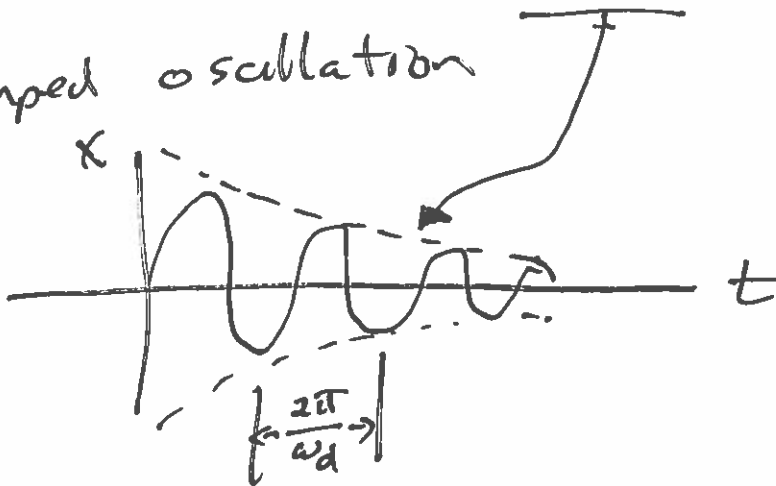
$$\omega_d = \omega_n \sqrt{1 - \zeta^2}$$

3 solutions !!

Underdamped ( $0 < \zeta < 1$ ) (complex roots)

$$x(t) = A e^{-\zeta\omega_n t} \sin(\omega_d t + \phi)$$

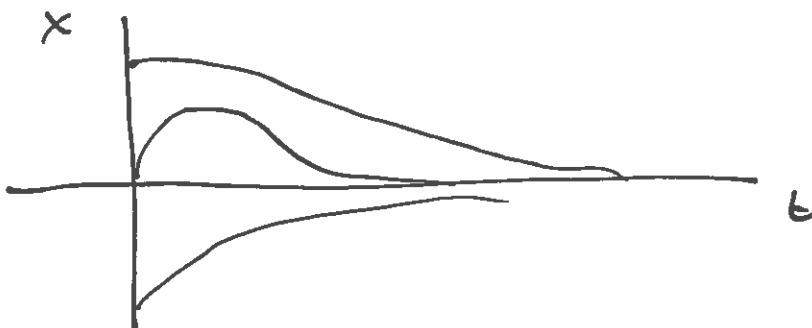
Damped oscillation



Overdamped ( $\zeta > 1$ )

$$x(t) = e^{-\zeta\omega_n t} \left( a_1 e^{-\omega_n \sqrt{\zeta^2 - 1} t} + a_2 e^{\omega_n \sqrt{\zeta^2 - 1} t} \right)$$

distinct pair of real roots



no oscillation

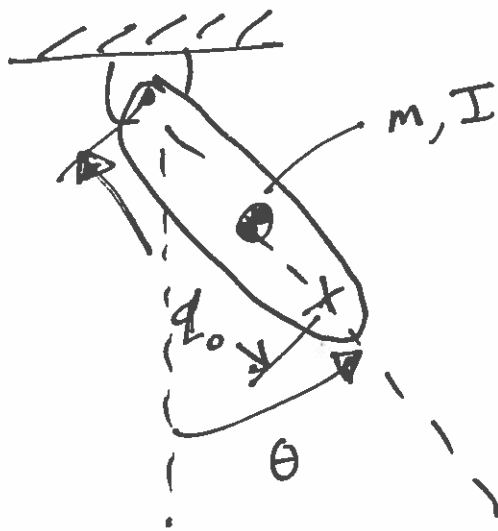
Critically damped ( $\zeta=1$ ) pair of repeated roots

$$x(t) = (a_1 + a_2 t) e^{-\omega_n t} \quad \text{no oscillation}$$



fastest return to zero without oscillation

# Compound Pendulum



There exists a simple pendulum that has the same period of oscillation as the compound pendulum. The length of this simple pendulum is referred as the center of percussion.

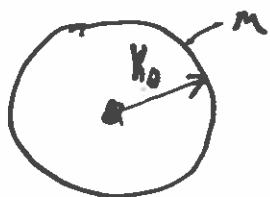
$$\left. \begin{array}{l} \text{compound } \omega_n^2 = \frac{mgl}{I} \\ \text{simple } \omega_n^2 = \frac{g}{l_0} \end{array} \right\}$$

$$l_0 = \frac{I}{ml} \quad \text{C.o.P}$$

$\hookrightarrow$  length

## Radius of gyration

radius of ring that has same moment of inertia as the object in question.



$$mk_0^2 = I \Rightarrow k_0 = \sqrt{\frac{I}{m}}$$

$$l_0 = k_0^2 \quad \uparrow \text{ length}$$

$\uparrow$  length to COM of compound pendulum

$$l_0 = \frac{mk_0^2}{ml}$$

(4)

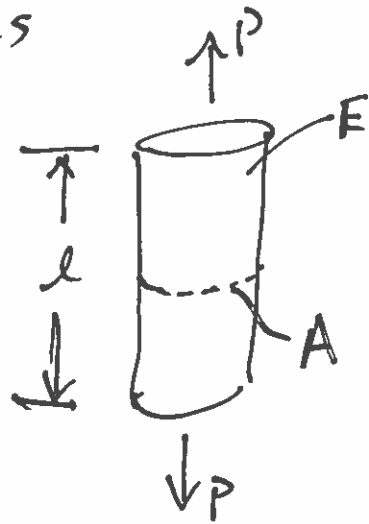
# Stiffness

Function describing how much load is required to produce a unit of deflection in a mechanical structural element.

Stiffness is a function of both the geometry and material properties.

For examples

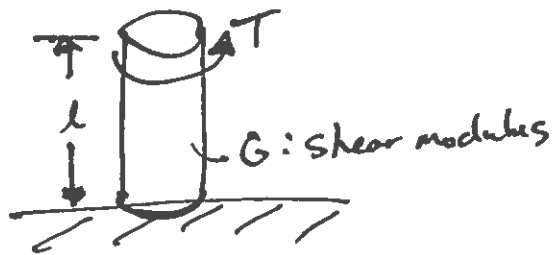
slender  
elastic  
rod



modulus of elasticity

$$K = \frac{EA}{l}$$

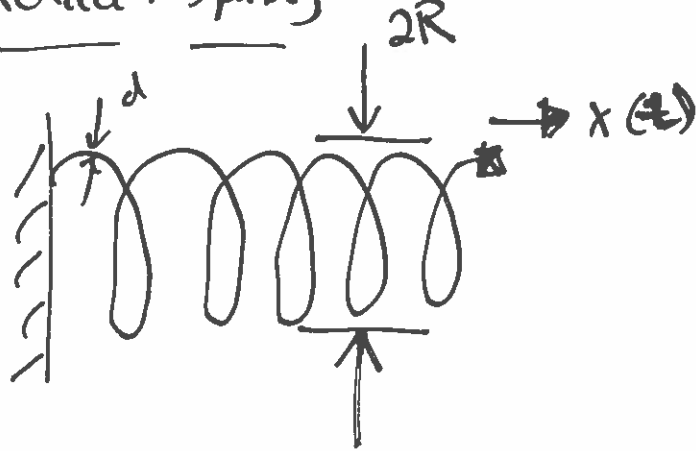
slender  
torsion  
bar



$$K = \frac{GJ}{l}$$

$J$ : second polar  
moment of  
area of  
the cross  
section

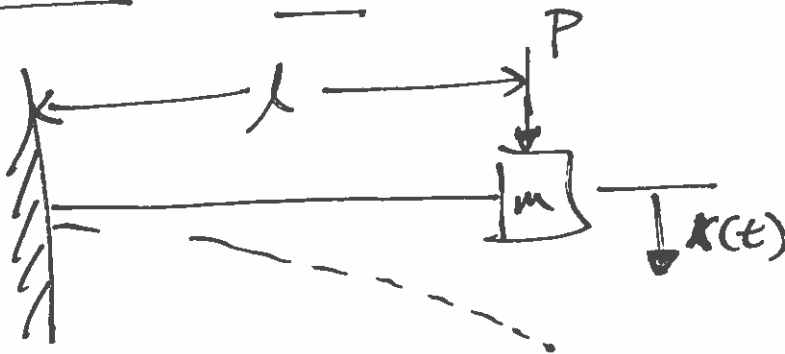
### helical spring



$$k = \frac{G d^4}{64 n R^3}$$

$n$ : number of coils

### massless beam

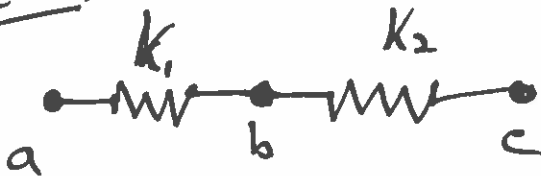


$$k = \frac{3EI}{l^3}$$

$I$ : second moment of area of the beam's cross section

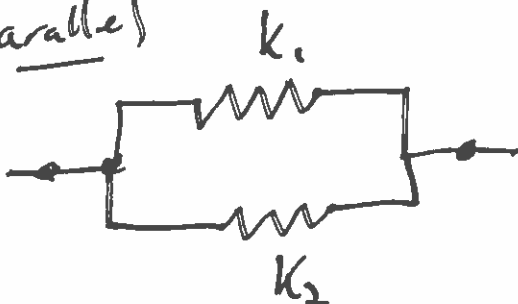
### Combinations of springs (stiffnesses)

#### series



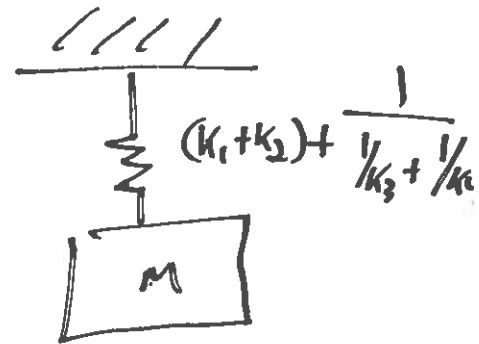
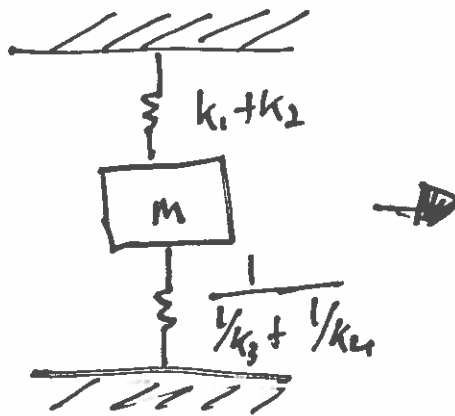
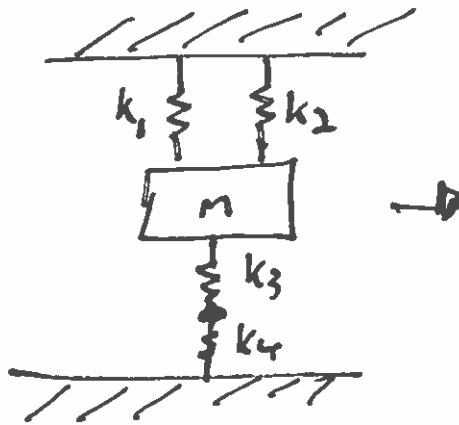
$$K_{\text{total}} = \frac{1}{\frac{1}{k_1} + \frac{1}{k_2}} = \frac{k_1 k_2}{k_1 + k_2}$$

#### parallel



$$K_{\text{total}} = k_1 + k_2$$

## Example



# Stability

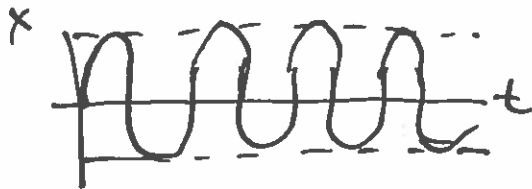
Describes behavior of a system when  $t \rightarrow \infty$ . All systems so far have been stable. A system can be classified as stable or unstable.

## Stable system

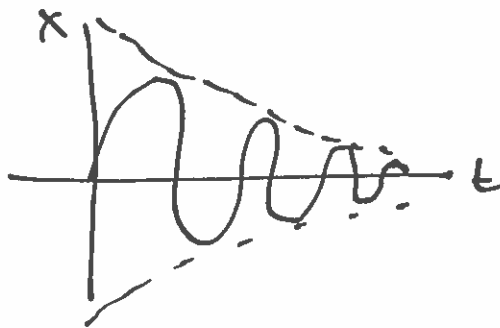
the behavior is such that as  $t \rightarrow \infty$  it is always bounded,  $\|x_{t \rightarrow \infty}\| \leq x_{\max}$ .

For unstable system the behavior is not bounded.

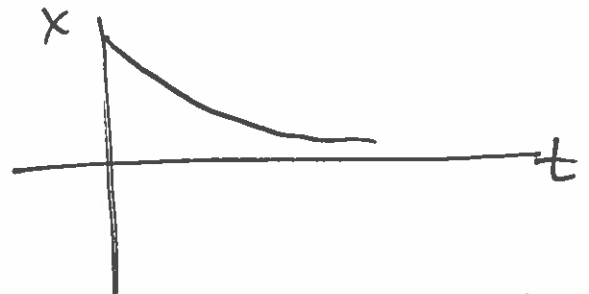
Stable



marginally stable

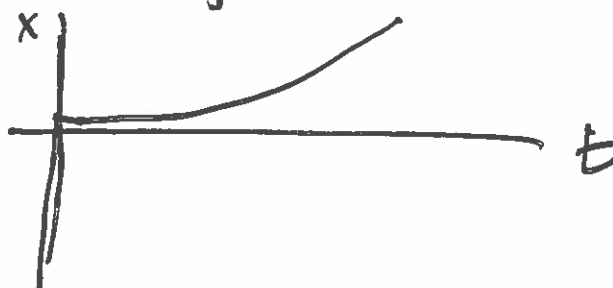


asymptotic stability

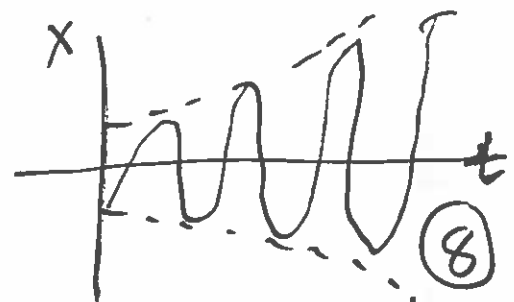


unstable

divergent instability



flutter instability





Stability is a function of the system parameters :  $m, c, k$

### Examples

$$m\ddot{x} + kx = 0 \quad (+k, m) \quad x(t) = A \sin(\omega t + \phi)$$

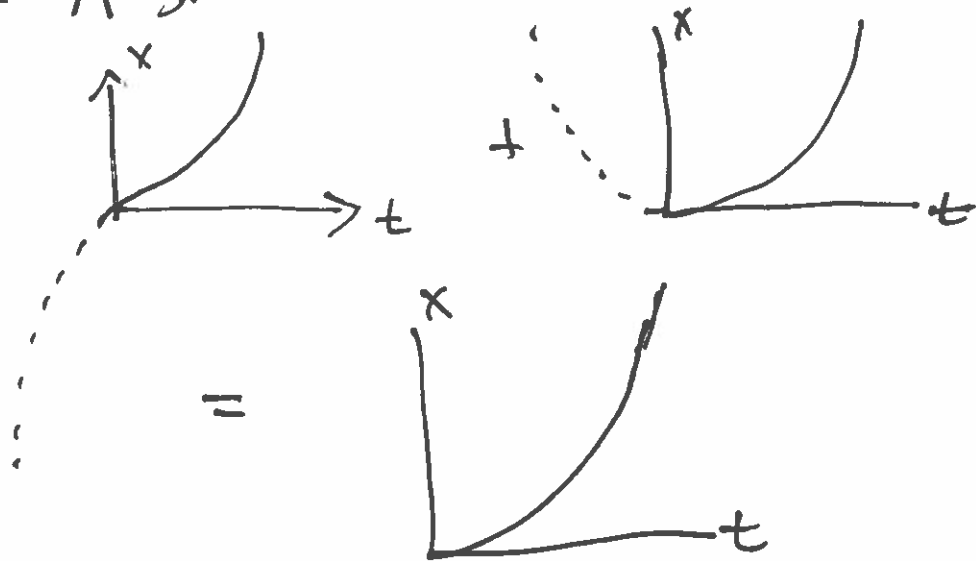
$$|x(t)| \leq |A|$$

(marginally) Stable!

$$m\ddot{x} - kx = 0 \quad (k, m +)$$

$$x(t) = a_1 e^{-\sqrt{\frac{k}{m}}t} + a_2 e^{\sqrt{\frac{k}{m}}t}$$

$$= A \sinh(\omega_n t) + B \cosh(\omega_n t)$$

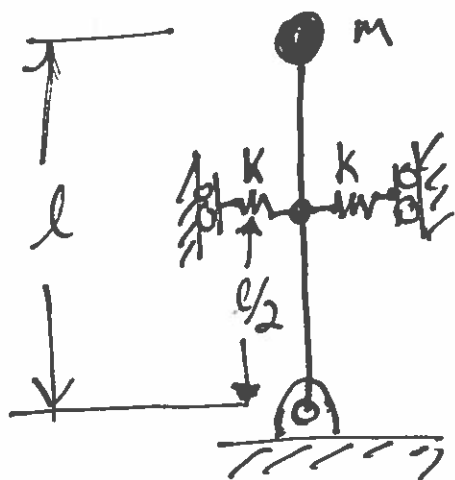


unstable!

$$m\ddot{x} + c\dot{x} + kx = 0 \quad m, c, k > 0$$

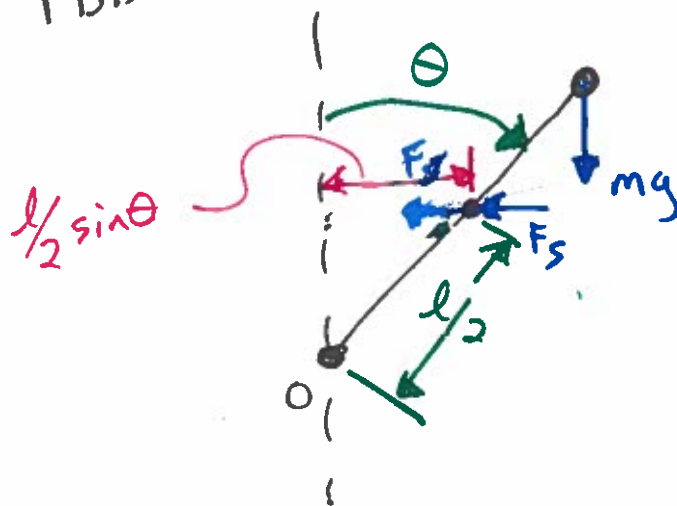
stable solution

## Example



Newton and Euler's equations

FBD



$$I_o \ddot{\theta} = \sum M_o$$

$$I_o = ml^2$$

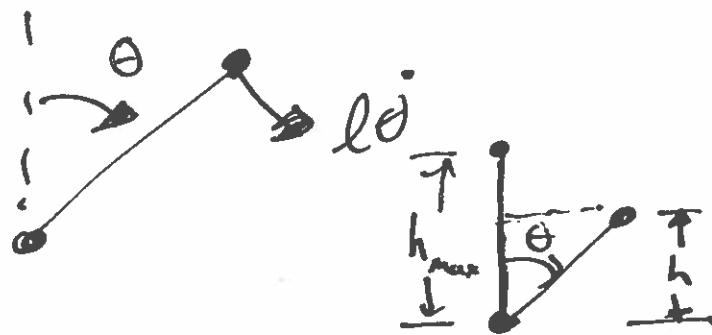
$$\begin{aligned} F_s &= k \Delta x_1 + k \Delta x_2 \\ &= 2k \left( \frac{l}{2} \sin \theta \right) \\ &= kl \sin \theta \end{aligned}$$

$$F_g = mg$$

$$\ddot{\theta} ml^2 = -kl \sin \theta \frac{l}{2} \cos \theta + mgl \sin \theta$$

Lagrange's Method

$$T = \frac{1}{2} m (\dot{\theta})^2$$



$$U = U_{s1} + U_{s2} + U_g$$

$$\begin{aligned} &= \frac{1}{2} k \left( \frac{l}{2} \sin \theta \right)^2 + \frac{1}{2} k \left( \frac{l}{2} \sin \theta \right)^2 + mgl \cos \theta \\ &= k \left( \frac{l}{2} \sin \theta \right)^2 + mgl \cos \theta \end{aligned}$$

$$L = T - U$$

$$L = \frac{1}{2} m (l \dot{\theta})^2 - K \left( \frac{l}{2} \sin \theta \right)^2 - mgl \cos \theta$$

$$\frac{d}{dt} \left( \frac{\partial L}{\partial \dot{\theta}} \right) - \frac{\partial L}{\partial \theta} = 0$$

$$\frac{d}{dt} (m l^2 \ddot{\theta}) + 2K \frac{l}{2} \sin \theta \frac{l}{2} \cos \theta - mgl \sin \theta = 0$$

$$\frac{K l^2}{2} \sin \theta \cos \theta$$

$$m l^2 \ddot{\theta} + \frac{K l^2}{2} \sin \theta \cos \theta - mgl \sin \theta = 0$$

Assume  $\sin \theta = \theta$ ,  $\cos \theta = 1$

$$m l^2 \ddot{\theta} + \frac{K l^2}{2} \theta - mgl \theta = 0$$

$$\ddot{\theta} + \underbrace{\frac{K}{2m} \theta - \frac{g}{l} \theta}_{\text{negative?}} = 0$$

$$\ddot{\theta} + \underbrace{\left( \frac{K}{2m} - \frac{g}{l} \right)}_{\text{negative?}} \theta = 0$$

$$\frac{K}{2m} - \frac{g}{l} < 0$$

$$\boxed{\frac{K}{2m} < \frac{g}{l}} \rightarrow \text{lead unstable behavior!}$$

(11)