

Introduction

* Electromagnetic spectrum for remote sensing:

↳ Microwave: P, L, S, C, X, K

↳ Infrared (IR):

◦ Visible light: 380 - 750 nm

◦ Near Infrared: 0,75 - 8 μm or (0,75 - 3) μm

◦ Thermal Infrared: 8 - 15 μm

◦ Far Infrared: 15 μm - 1 mm

* Transparency of Earth's atmosphere as a function of λ

↳ Three transparent windows, the rest opaque

◦ Visible and near infrared (VNIR): $\lambda = 0,38 - 3 \mu\text{m}$

◦ Thermal infrared radiation (TIR): $\lambda = 8 - 15 \mu\text{m}$

◦ Microwave region (μW): $\lambda = 1 \text{ mm} - 1 \text{ m}$

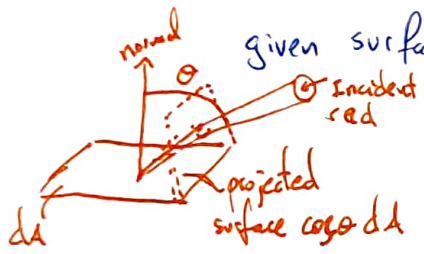
Natural EM Radiation

radiant energy: Energy of electromagnetic or gravitational radiation [J]

radiant flux: radiant energy emitted, reflected, transmitted or received per unit time [J/s] = [W]
(or power)

* Radiation Quantities

↳ Radiance: Is the radiant ^{power} ^{incident} flux emitted, reflected, transmitted or received by a given surface per unit solid angle per unit projected area

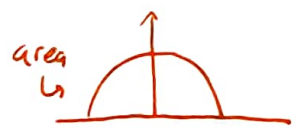


→ Power incident: $dP = L \cos\theta dA d\Omega$

→ Solid angle: $d\Omega = \sin\theta d\theta d\phi$

→ Radiance:
$$L = \frac{dP}{\cos\theta dA d\Omega} \quad [W \cdot m^{-2} \cdot sr^{-1}] \quad \sim dP(\theta, \phi) \sim L(\theta, \phi)$$

↳ Irradiance: Is the total incident power per unit area



$$E = \int_{\theta=0}^{\pi/2} \int_{\phi=0}^{2\pi} L_{in} \cos\theta d\Omega \quad [W \cdot m^{-2}]$$

↳ Radiant Exitance: Is the total emitted power per unit area

$$M = \int_{\theta=0}^{\pi/2} \int_{\phi=0}^{2\pi} L_{out} \cos\theta d\Omega \quad [W \cdot m^{-2}]$$

•) Isotropic radiation: Radiance (L) is independent of the direction, i.e. it's constant $L = L_0$

Therefore, radiant exitance:

$$M = L_0 \int_0^{\pi/2} \int_0^{2\pi} \cos\theta d\Omega = L_0 \int_0^{\pi/2} \int_0^{2\pi} \cos\theta \sin\theta d\theta d\phi = \pi L_0 \Rightarrow M = \pi L_0$$

* Thermal Radiation

- Thermal radiation is emitted by all the objects above 0K
- Is the radiation detected by majority of passive remote sensing systems

↳ Spectral radiance: L_λ or L_f

Is the radiance contained in a small range of wavelengths (freqs) $\Delta\lambda$ (Δf)

$$\Delta L = L_\lambda \Delta\lambda$$

$$\Delta L = L_f \Delta f$$

$$L_\lambda = \text{W m}^{-2} \text{sr}^{-1}$$

↓ interval λ_1, λ_2

$$L = \int_{\lambda_1}^{\lambda_2} L_\lambda d\lambda$$

$$L_f = \text{W m}^{-2} \text{Hz}^{-1} \text{sr}^{-1}$$

↓

$$L = \int_{f_2}^{f_1} L_f df$$

$$\lambda = \frac{c}{f} \rightarrow L_\lambda d\lambda = L_\lambda \left(-\frac{c}{f^2} \right) df = L_f df$$

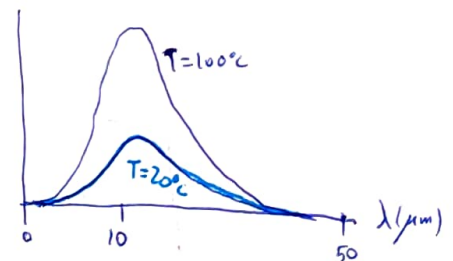
$$\boxed{L \frac{d\lambda}{L_f} = \frac{f^2}{c} = \frac{c}{\lambda^2}}$$

* Black Body

- Object that absorb all the radiation incident on it.
- Spectral radiance does not depend on θ, ϕ but it depends on wavelength λ
- Planck formula

$$L_\lambda = \frac{2hc^2}{\lambda^5} \frac{1}{\exp\left[\frac{hc}{\lambda k_B T}\right] - 1} \Rightarrow L_f = L_\lambda (c/f)$$

$$\rightarrow \text{At long } \lambda \rightarrow \infty \Rightarrow \lim_{\lambda \rightarrow \infty} L_\lambda \approx \frac{2k_B T c}{\lambda^4} \quad \text{Rayleigh-Jeans approx.}$$



⊗ Increasing T shifts to a shorter wavelengths

$$\rightarrow \text{Integrating total radiance of Black-body} \rightarrow L = \int_0^\infty L_\lambda d\lambda = \frac{2\pi^4 k_B^4}{15c^2 h^3} T^4$$

$$\Rightarrow \text{Total radiant exitance B-B} \rightarrow M = \pi L \Rightarrow \boxed{M = \sigma T^4} \quad \text{Stefan's Law}$$

radiation from B-B is isotropic

→ Wavelength at which spectral radiance is maximum.

$$\max(L_\lambda) \Rightarrow \frac{dL_\lambda}{d\lambda} = 0 \Rightarrow \boxed{\lambda_{\max} = \frac{A}{T}} \quad \text{Wien's displacement Law}$$

→ Black Body model is useful bc any radiance from a body can be modeled as B-B with correction factor: emissivity ϵ .

$$\epsilon(\lambda) \rightarrow \boxed{L_{\lambda, \epsilon} = \epsilon(\lambda) L_\lambda}$$

→ Brightness temperature: T_b .

Is the temperature of the equivalent B-B would give the same radiance @ a λ

$$\epsilon L_\lambda(\lambda, T) = L_\lambda(\lambda, T_b) \xrightarrow{\lambda \rightarrow \infty} \boxed{T_b = \epsilon T}$$

⊗ Solar Radiation

$$M = \sigma T^4$$

Wm^{-2}

$$P = 4\pi r^2 M$$

power radiated

$$E = \frac{P}{4\pi D^2}$$

Wm^{-2}

sun's irradiance
D → Earth-Sun dist

$$\Delta \Omega = \frac{\pi r^2}{D^2}$$

sr

$$L = \frac{E}{\Delta \Omega} = \frac{\sigma T^4}{\pi}$$

$\text{Wm}^{-2}\text{sr}^{-1}$

⊗ Absorption in the atmosphere

$$\rightarrow \epsilon_r = \epsilon' - j\epsilon'' = \epsilon'(1 - j\tan\delta)$$

$$\rightarrow \text{Intensity decays exponentially: } I(z) = I_0 e^{-z/L_a}$$

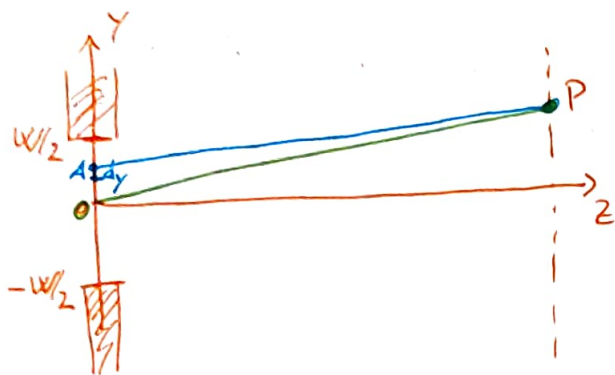
$$\rightarrow \text{Absorption length } L_a = \frac{1}{4\pi\kappa}$$

$$\left. \begin{array}{l} \text{If} \\ \kappa_a = \frac{1}{L_a} \end{array} \right\} \Rightarrow I(z) = I_0 e^{-\tau}$$

$$\tau = \int_0^z \kappa_a(z') dz'$$

optical thickness

* Diffraction



↳ Contribution on P of an element dy is proportional to $\propto \exp(jk y \sin \theta) dy$

↳ All the contributions:

$$a(\theta) = \int_{-w/2}^{w/2} \exp(jk y \sin \theta) dy$$

↳ In general there is a transmittance function $f(y)$ and we get a Fourier transf.

$$a(\theta) = \int_{-\infty}^{\infty} f(y) e^{jk y \sin \theta} dy$$

conjugated magnitudes y , $\frac{k \sin \theta}{2\pi}$

* Slit case:

$$f(y) = \text{rect}\left(\frac{y}{w}\right) \Rightarrow a(\theta) = w \text{sinc}\left(\frac{wk \sin \theta}{2\pi}\right)$$

↳ Direction of two zeros of main peak

$$\sin \theta = \pm \frac{\lambda}{w} \quad \xRightarrow[\text{if } w \gg \lambda]{} \theta \approx \pm \frac{\lambda}{w}$$

* Diffraction for circular aperture

↳ amplitude is proportional to first-order Bessel function

↳ The first zero: $\theta_r = 1.22 \frac{\lambda}{D}$ D \rightarrow diameter of aperture

↳ Approximation valid if: $\frac{w^2}{8z} < \frac{\lambda}{4}$

↳ Accurate if: $z > \frac{w^2}{2\lambda} = z_F \rightarrow$ Fresnel distance

\Rightarrow Diffraction puts a limit in the resolution and it depends on the ratio between λ and D

Aerial Photography

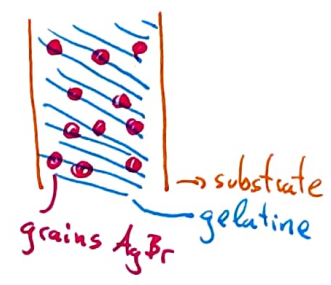
* Context.

- Photographic films sensitive to visible and near infrared.
- Aerial photography is a passive technique \Rightarrow It detects existing radiation and give 2D representation

Film-based preferred over digital due to better resolution

→ Traditional process: chemical reaction

- ① Film made of many crystals of a salt embedded in gelatin with plastic base
- ② Absorption of sufficient energetic photon converts the grain into metallic silver
- ③ Unexposed grains are removed by chemical process
- ④ The result: Negative \Rightarrow Bc areas that received light during exposure stage will appear dark
 - ↳ Visible photography: $0,3 - 0,7 \mu m$
 - ↳ True infrared photography: $0,7 - 0,9 \mu m$



→ Two Limits

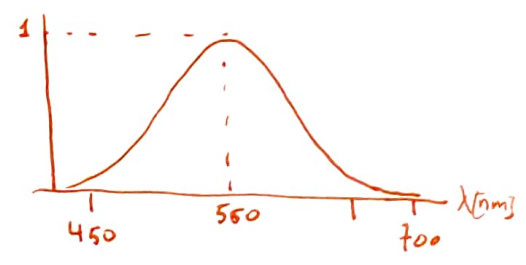
- ↳ Diffraction: I can resolve two points if the angular separation is greater than $1,22 \frac{\lambda}{D}$
- ↳ Quality of material: Size of grain, distribution of the grains in the gelatine.

→ Response of photographic film.

- ↳ In terms of photometric units: weighted with nominal spectral sensitivity of human eye
- ↳ Photometric unit corresponding to irradiance is \rightarrow illuminance

$$E_v = K \int E_\lambda V(\lambda) d\lambda$$

E_v \swarrow illuminance
 E_λ \swarrow spectral irradiance
 $V(\lambda)$ \swarrow nominal sensitivity of the light adapted to human eye
Max at green



* Speed of a film

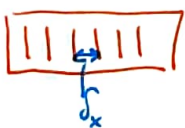
- It is the time duration a film has to be exposed to light of a given illuminance to get a significant change of opacity after processing
- Aerial Film Speed (AFS) index is used
 - larger numbers \Rightarrow Faster films \Rightarrow Shorter exposure times
- The grain size controls the speed:
 - High speed films \Rightarrow large grains

* Contrast of a film

- It is the effect of changing the exposure time (or illuminance)
- if small change of illuminance \Rightarrow large change of opacity \Rightarrow film with high contrast
- The grain size controls the contrast:
 - High contrast \Rightarrow specific range of grain size

* Spatial resolution

- The ability ^{of a system} to distinguish two points: line-pairs (lp)
- lp per unit length \Rightarrow greatest number of line-pairs per unit length that can be resolved

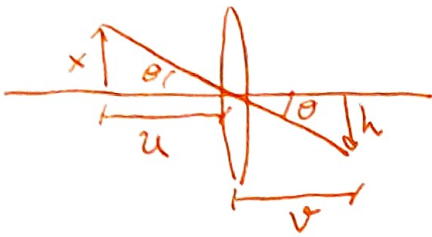


$$\delta_x = \frac{1}{2f}$$

$\delta_x \rightarrow$ smallest distance between 2 points that can be resolved

$r \rightarrow$ resolution [1/length] $10 \text{ lp/mm} = 10^4 \text{ m}^{-1}$

* Optics of photography systems



$$\frac{1}{u} + \frac{1}{v} = \frac{1}{f}$$

$$h = v \tan \theta$$

$u \rightarrow \infty$
remote sensing

$$\left. \begin{array}{l} v = f \\ \tan \theta = \frac{x}{u} \end{array} \right\} h = f \frac{x}{u} \approx f \theta$$

↳ What happens with radiance:

- Object with uniform exitance so radiance (luminance) incident at lens $\rightarrow L$

> Irradiance at lens: $L \Omega$

- Object subtends small solid angle Ω

⇒ Total power by lens diameter D : $P_{\text{Tot}} = L \Omega \cdot \pi \frac{D^2}{4}$

⇒ Solid angle for object and image is the same: $\Omega = \frac{A_{\text{image}}}{f^2}$

⇒ Irradiance on the film: $E_{\text{film}} = \frac{P_{\text{Tot}}}{A_{\text{image}}} \Rightarrow E_{\text{film}} = \frac{\pi}{4} L \left(\frac{D}{f} \right)^2$

⇒ Ratio irradiance (illuminance) at film to radiance (luminance) at lens

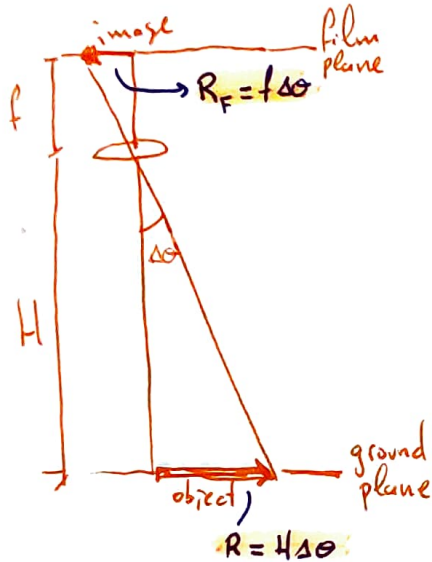
$$\frac{E_{\text{film}}}{L} = \frac{\pi}{4} \left(\frac{D}{f} \right)^2$$

↳ $f/\text{number of lens } \left(\frac{f}{D} \right)$

- the smaller f/number the larger the lens the brighter the image

⊛ Scale of vertical aerial photography

↳ Scale of the image: The ratio of the size of the representation of object on the map to the size of real object



$$S = \frac{f}{H}$$

$$S < 1$$

$$S = \frac{R_F}{R}$$

↳ Negative with width w

$$S = \frac{w}{w_g} \Rightarrow$$

↳ region of the ground

$$w_g = w \frac{H}{f}$$

↳ Spatial resolution

a) Distance on film: $\delta_x = \frac{1}{2r}$

a) Distance on the ground:

$$\delta x_g = \frac{1}{2r} \cdot \frac{H}{f}$$

$$s = \frac{\delta_x}{\delta x_g}$$

Electro-Optical Systems

* Detectors for visible and near infrared

→ Minimum photon energy \Rightarrow max photon λ depend on semiconductor

↳ Photodiode

Silicon (Si) \rightarrow up to $1.1\mu\text{m}$

Germanium (Ge) \rightarrow up to $1.7\mu\text{m}$

Lead sulfide (Pbs) \rightarrow up to $3\mu\text{m}$

Indium Antimonide (Insb) \rightarrow up to $5\mu\text{m}$

Mercury Cadmium Telluride (MCT) \rightarrow up to $15\mu\text{m}$

Mercury doped Germanium (Ge:Hg) \rightarrow up to $15\mu\text{m}$

↳ Thermal detectors

- Wider spectral range \Rightarrow when $\lambda > 15\mu\text{m}$
- Lower sensitivity

↳ Charge Coupled Devices (CCDs)

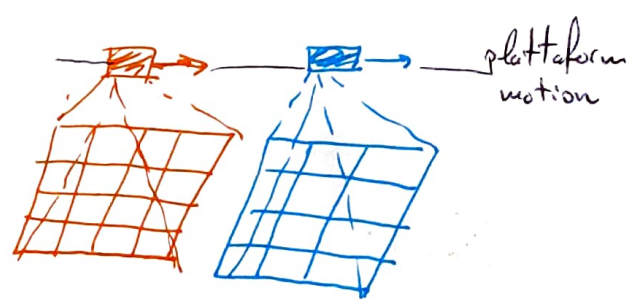
Si $\rightarrow 1.1\mu\text{m}$

* The electro-optical system can be in motion respect to the target so it may be necessary to compensate for the movements to avoid blurring the image.

* Step-Stare Imaging

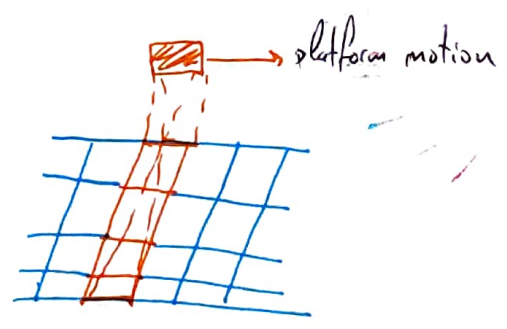
The detector array stares at a scene and then moves on to stare at next scene

We get a strip picture (frame)



* Push-Broom Imaging

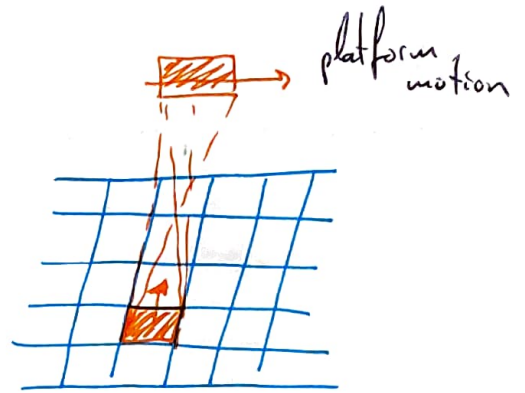
If the detector is a linear array and we synchronise the speed of the plane and the detector we get the full picture



* Whisk-Broom Imaging

There is a single detector scanning in the direction perpendicular to the motion of platform.

We have to synchronize two speeds, the one of the flight and the scanner.

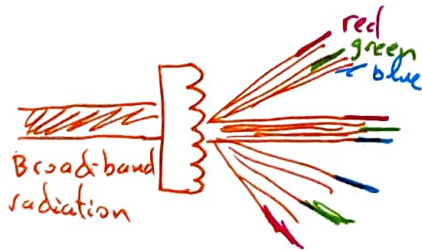


* Spin-Scan Imaging

* Diffraction Grating

It is difficult to scan a wide spectrum so the detectors are often optimized for a specific bandwidth. Using a diffraction grating we can focus on a

BW



$$\sin \theta_n = n \frac{\lambda}{d}$$

Satellites

* Atmosphere Layers

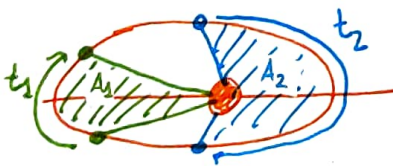
- Troposphere (0-12 km) \longrightarrow commercial aircrafts (~10 km)
(more gas density \Rightarrow we can fly)
- Stratosphere (12-50 km)
- Mesosphere (50-80 km)
- Thermosphere (80-700 km) \longrightarrow satellites (~400 km)
(low gas density \Rightarrow low friction)
(too much friction \Rightarrow satellite fall)
- Exosphere (700-10000 km)

* Kepler's Laws

- ① Planets move in a plane. Elliptical orbits being the Sun at one focus



- ② Law of areas



$$A_1 t_1 = A_2 t_2$$

A line segment joining a planet and the Sun sweeps out equal areas during equal intervals of time.

- ③

$$\frac{T^2}{a^3} = \text{Constant}$$

$T \rightarrow$ period of revolution of a planet around the Sun

$a \rightarrow$ semi-major axis

* Law of Gravitation

$$\vec{F} = -G \frac{Mm}{r^2} \hat{r}$$

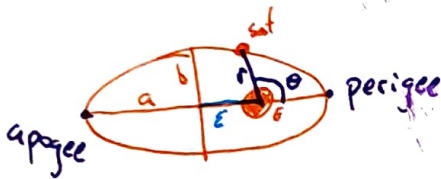
• Earth radius: $R = 6371 \text{ km}$

• Earth mass: $M = 5974 \cdot 10^{24} \text{ kg}$

• Gravitational constant: $G = 6,672 \cdot 10^{-11} \text{ m}^3 \text{ kg}^{-1} \text{ s}^{-2}$

• $\mu = GM$

* Elliptical orbit



• Distance of the satellite
w respect major axis
↳ from perigee

$$r = \frac{a(1-e^2)}{1+e \cos \theta}$$

• Satellite velocity

$$v = \sqrt{\mu \left(\frac{2}{r} - \frac{1}{a} \right)}$$

• Period of satellite motion

$$T = 2\pi \sqrt{\frac{a^3}{\mu}}$$

$$e = \sqrt{1 - \frac{b^2}{a^2}}$$

→ $v_{\text{perigee}} > v_{\text{apogee}} \Rightarrow 2^{\text{nd}} \text{ Kepler Law}$

* Circular orbit



$$r = R + h$$

$$g = \frac{GM}{R^2}$$

$$F_c = \frac{mv^2}{r}$$

$$F_g = G \frac{\mu m}{r^2}$$

• Velocity: Balance of forces

$$v = \sqrt{\frac{gR^2}{R+h}}$$

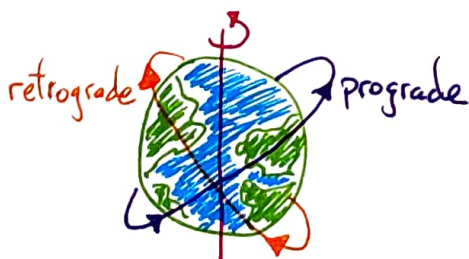
• Period of satellite

- Time of one round: T

$$v = \frac{\text{distance}}{\text{time}} \Rightarrow T = \frac{2\pi r}{v} \rightarrow$$

$$T = 2\pi \sqrt{\frac{(R+h)^3}{gR^2}}$$

* Prograde and Retrograde orbits

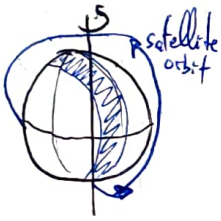


* Useful Orbits

a) Sub-satellite position: The point on the Earth's surface which is directly below the satellite.

* Polar orbit

The satellite orbits the Earth in such a way to cover the north and the south polar regions.



Uses: Earth-mapping
Weather satellites

* Ground-track

It is the path on the surface of a planet directly below a satellite trajectory

* Geostationary satellite

It is a satellite in a circular orbit above the equator. The period is equal to a Earth's rotational period, one sidereal day: $86164 \text{ seconds} = T$

It is also referred as geosynchronous equatorial orbit (GEO)

It has a fixed position in the sky so its ground-track is just a fixed point on the Earth surface

* Geosynchronous satellite

It is a satellite with a period of a sidereal day but it rotates in any axis, not necessarily the same rotation axis of the Earth

sub-satellite path traces a lemniscate ∞ ...

⊛ Additional things

-) Velocity of the lowest altitude $h=0$
-) Escape velocity: It is the minimum speed needed to stop escape from the gravitational influence of a planet.

↳ Energy balance

$$E_c = \frac{1}{2} m v^2$$

$$E_p = -\frac{GMm}{r}$$

$$U_{in} = U_{fin}$$

$$E_c + E_p = 0 \Rightarrow \frac{1}{2} m v^2 = \frac{GMm}{r}$$

$$v = \sqrt{\frac{2GM}{r}}$$