3rd Design Iteration

Team Name: الثُّرَيًا

□Note: lengths should be in (cm), angles in (deg).

Wing Information				
Airfoil	NACA 6415			
	152 (cm)			
Span Root Chard	30 (cm)			
Root Chord				
Tip Chord	30 (cm)			
Offset	40 (cm)			
AR	5.07			
S _w	4600 (cm ²)			
Incidence Angle	0			
Twist	0			
Dihedral	2 (deg)			
Vertical Tail Information				
Airfoil	NACA 0009			
Semi Span (Total Length)	23 (cm)			
Root Chord	20 (cm)			
Tip Chord	18 (cm)			
Offset	2 (cm)			
S_v/S_w	0.087			
AR_{ν}	4.84			
V _v	0.04288			
Incidence Angle	0			
Tail Arm	75 (cm)			
Shifted Length in Z-Direction	-4 (cm)			
Horizontal Ta	il Information			
Airfoil	NACA 0009			
Span	64 (cm)			
Root Chord	20 (cm)			
Tip Chord	20 (cm)			
Offset	0			
SH/Sw	0.283			
AR_{H}	3.2			
V _H	0.139			
Incidence Angle	-1.5 (deg)			
Tail Arm	75 (cm)			
Shifted Length in Z-Direction	-4 (cm)			

Propulsion System Information

Input					
Model Weight (Drive included or without)	2100g include drive				
Desired Flight speed	44.3 km/h				
Brushless Motor	Manufac turer	Model	Voltage Constan t [KV]	No Load Current [A]	Resistanc e [ohm]
	ElectriFl y	RimFir e 0.15	1200	1.7	0.0415
Battery	Manufac turer	No. Of Cells	Voltage	Capacity [mAh]	C-Rating
Dattel y	HRB	3	11.1	5200	50
Propeller Size (Diameter x Pitch)	10 x 4.7 in.				
Speed Controller (Current Rating Value)	60A				
Output					
Load			6.7	С	
Mixed flight Time			10.5 n	nin	
Max. Current			33.43	ВА	
Max. Power			383.4	W	
Static Thrust		1618 g			
Available Thrust [at the desired flight speed]		1151 g			
Drive Weight		636 g			
All Up Weight		2100 g			
(Power/Weight) Ratio [Watt/lb]		85 W/lb - 187 W/kg			
(Thrust/Weight) Ratio	0.77				

Flight Phases

Phase 1 (payload isn't deployed)			
MTOW	2.1 (Kg)		
X_{CG}	10.9 (cm)		
Static margin (%)	0.2		
CL_{Cruise}	0.5932		
V _{Stall}	690 (cm/s)		
V Cruise	1230 (cm/s)		
$lpha_{\mathit{Trim}}$	1.5 (deg)		
Required Static Thrust	788.997 (g)		
Required Dynamic Thrust (at V Cruise)	170.81 (g)		
Phase 2 (payloa	d is deployed)		
Phase 2 (payloa Mass	d is deployed)		
Mass	1.6 (Kg)		
Mass X _{CG}	1.6 (Kg) 8.9 (cm)		
Mass χ_{cg} Static margin (%)	1.6 (Kg) 8.9 (cm) 0.2		
Mass X _{CG} Static margin (%) CL _{Cruise}	1.6 (Kg) 8.9 (cm) 0.2 0.4900		
Mass X_{CG} Static margin (%) CL_{Cruise} V_{Stall}	1.6 (Kg) 8.9 (cm) 0.2 0.4900 690 (cm/s)		
Mass X _{CG} Static margin (%) CL _{Cruise} V _{Stall} V cruise	1.6 (Kg) 8.9 (cm) 0.2 0.4900 690 (cm/s) 1220 (cm/s)		