

# Design Calculation Sheet for WEE

Designer:

Location:

City:

Country:

Date: 2020-06-22 03:05:11

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## Secondary Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
22	(20,4,6)	(20,8,6)	4	1.12	1.12
21	(20,0,6)	(20,4,6)	4	1.12	1.12
2	(0,4,6)	(0,8,6)	4	1.12	1.12
1	(0,0,6)	(0,4,6)	4	1.12	1.12

### Design Limit state:

Combo:  $1.4 \cdot D + 1 \cdot L$

Md: 1.12 t.m

Vd: 1.12 ton

### Service Limit State

Combo: LIVE

Span: 4 m

Load:  $-0.2 \text{ t/m'}$

### Design Checks

#### 1-Check Local Buckling

$d_w/t_w = 23.92 < 81.98 \Rightarrow$  Compact Web

$c/t_f = 3.95 < 10.91 \Rightarrow$  Compact Flange

#### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 94.24 \text{ m} \Rightarrow$  Supported (No LTB)

#### 3-Check Bending Stress

Section: IPE270

$f_{act} = 1.45 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### 4-Check Shear Stress

$q_{act} = 0.17 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### 5-Check Deflection

$d_{act} = 0.59 \text{ cm} < d_{all} = 1.33 \text{ cm}$



## Group Connection Design (Simple Shear Plate Connection)

### 1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 1.12$  ton

$R_{least} = 2.85$  ton

$N = 3$  with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.08$  t/cm<sup>2</sup> &  $q = 0.05$  t/cm<sup>2</sup>  $\Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.12$  t/cm<sup>2</sup>  $< 1.1 * 0.2F_u = 0.79$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.05$  t/cm<sup>2</sup> &  $q_{mt} = 0.08$  t/cm<sup>2</sup>  $\Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.09$  t/cm<sup>2</sup>  $< 0.2F_u = 0.72$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.09$  t/cm<sup>2</sup>  $< 0.72 * F_y = 1.73$  t/cm<sup>2</sup>  $\Rightarrow$  OK

Plate Layout  $\Rightarrow L = 189$  mm &  $t_p = 10$  mm &  $S_w = 6$  mm

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
18	(16,4,6)	(16,8,6)	4	2.18	2.18
17	(16,0,6)	(16,4,6)	4	2.18	2.18
20	(18,4,6)	(18,8,6)	4	2.18	2.18
3	(2,0,6)	(2,4,6)	4	2.18	2.18
4	(2,4,6)	(2,8,6)	4	2.18	2.18
5	(4,0,6)	(4,4,6)	4	2.18	2.18
6	(4,4,6)	(4,8,6)	4	2.18	2.18
7	(6,0,6)	(6,4,6)	4	2.18	2.18
8	(6,4,6)	(6,8,6)	4	2.18	2.18

9	(8,0,6)	(8,4,6)	4	2.18	2.18
10	(8,4,6)	(8,8,6)	4	2.18	2.18
19	(18,0,6)	(18,4,6)	4	2.18	2.18
12	(10,4,6)	(10,8,6)	4	2.18	2.18
13	(12,0,6)	(12,4,6)	4	2.18	2.18
14	(12,4,6)	(12,8,6)	4	2.18	2.18
15	(14,0,6)	(14,4,6)	4	2.18	2.18
16	(14,4,6)	(14,8,6)	4	2.18	2.18
11	(10,0,6)	(10,4,6)	4	2.18	2.18

### **Design Limit state:**

Combo:  $1.4 \cdot D + 1 \cdot L$

Md: 2.18 t.m

Vd: 2.18 ton

### **Service Limit State**

Combo: LIVE

Span: 4 m

Load: -0.4 t/m'

### **Design Checks**

#### **1-Check Local Buckling**

$d_w/t_w = 29.65 < 81.98 \Rightarrow$  Compact Web

$c/t_f = 4.56 < 10.91 \Rightarrow$  Compact Flange

#### **2-Check Lateral Torsional Buckling**

$L_{uact} = 0 \text{ m} < L_{umax} = 129.1 \text{ m} \Rightarrow$  Supported (No LTB)

#### **3-Check Bending Stress**

Section: IPE270

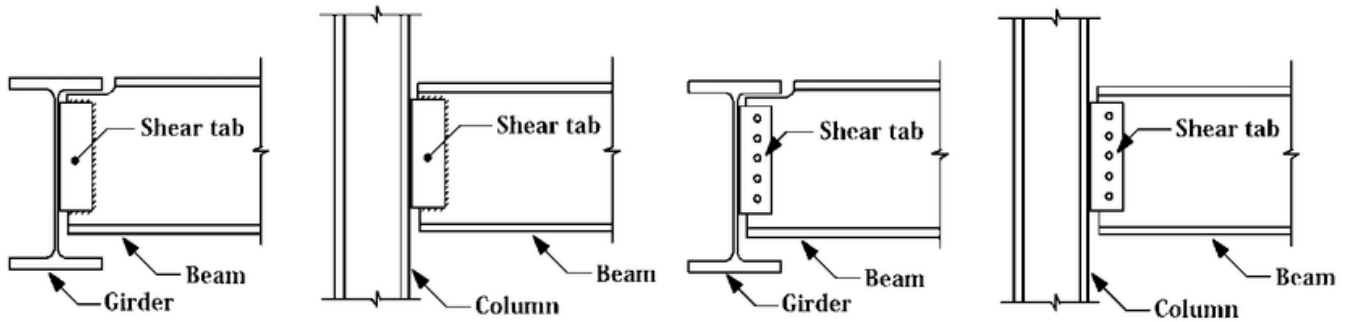
$f_{act} = 1.13 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### **4-Check Shear Stress**

$q_{act} = 0.19 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### **5-Check Deflection**

$d_{act} = 0.33 \text{ cm} < d_{all} = 1.33 \text{ cm}$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 2.18$  ton

$R_{least} = 2.85$  ton

$N = 3$  with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

#### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.15 \text{ t/cm}^2$  &  $q = 0.1 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.23 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.1 \text{ t/cm}^2$  &  $q_{mt} = 0.15 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.18 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.18 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout  $\Rightarrow L = 189 \text{ mm}$  &  $t_p = 10 \text{ mm}$  &  $S_w = 6 \text{ mm}$

## Main Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
9	(14,8,6)	(20,8,6)	6	4.59	2.33
7	(0,8,6)	(6,8,6)	6	4.59	2.33
3	(14,0,6)	(20,0,6)	6	4.59	2.33
1	(0,0,6)	(6,0,6)	6	4.59	2.33

#### Design Limit state:

Combo:  $1.4 * D + 1 * L$

$M_d = 4.59 \text{ t.m}$

Vd: 2.33 ton

### Service Limit State

Combo: LIVE

Span: 6 m

Load: -0.27 t/m'

### Design Checks

#### 1-Check Local Buckling

$d_w/t_w = 32.39 < 81.98 \Rightarrow$  Compact Web

$c/t_f = 4.81 < 10.91 \Rightarrow$  Compact Flange

#### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 154.92 \text{ m} \Rightarrow$  Supported (No LTB)

#### 3-Check Bending Stress

Section: IPE270

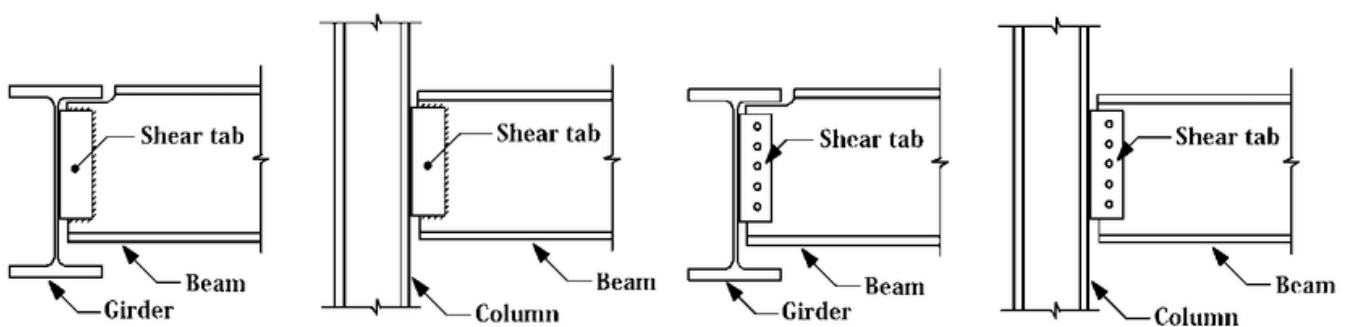
$f_{act} = 1.42 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### 4-Check Shear Stress

$q_{act} = 0.16 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### 5-Check Deflection

$\delta_{act} = 0.55 \text{ cm} < \delta_{all} = 2 \text{ cm}$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

Vd= 2.33 ton

Rleast= 2.85 ton

N= 3 with Pitch= 63 mm & Full Layout: (31;63 63 31.5)

## 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.16 \text{ t/cm}^2$  &  $q = 0.11 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.25 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

## 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.11 \text{ t/cm}^2$  &  $q_{mt} = 0.16 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.2 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

## 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.2 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout  $\Rightarrow L = 189 \text{ mm}$  &  $t_p = 10 \text{ mm}$  &  $S_w = 6 \text{ mm}$

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Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
6	(14,4,6)	(20,4,6)	6	8.96	4.52
4	(0,4,6)	(6,4,6)	6	8.96	4.52
8	(6,8,6)	(14,8,6)	8	9.14	3.48
2	(6,0,6)	(14,0,6)	8	9.14	3.48

## Design Limit state:

Combo:  $1.4 * D + 1 * L$

Md: 9.14 t.m

Vd: 3.48 ton

## Service Limit State

Combo: LIVE

Span: 8 m

Load:  $-0.3 \text{ t/m'}$

## Design Checks

### 1-Check Local Buckling

$d_w/t_w = 37.87 < 81.98 \Rightarrow \text{Compact Web}$

$c/t_f = 5.64 < 10.91 \Rightarrow \text{Compact Flange}$

### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 206.56 \text{ m} \Rightarrow \text{Supported (No LTB)}$

### 3-Check Bending Stress

Section: IPE330



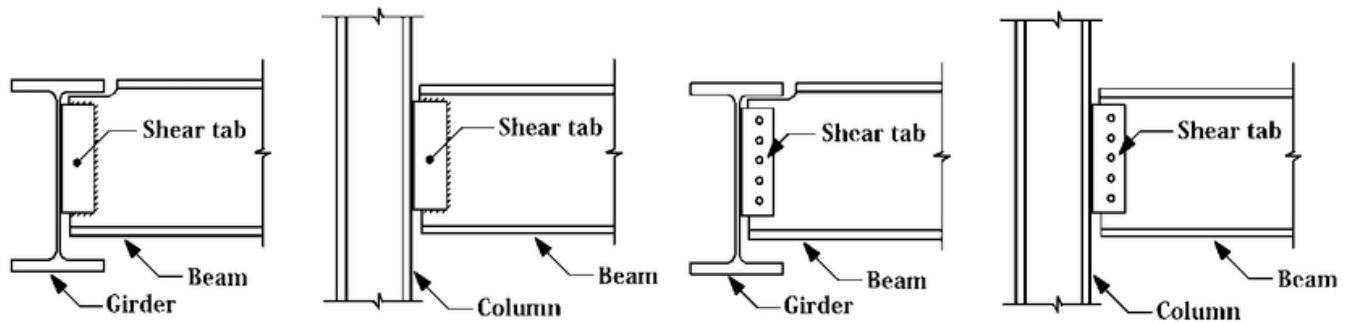
$$fact = 1.28 \text{ t/cm}^2 < Fb = 1.54 \text{ t/cm}^2$$

#### 4-Check Shear Stress

$$qact = 0.14 \text{ t/cm}^2 < qall = 0.84 \text{ t/cm}^2$$

#### 5-Check Deflection

$$dact = 0.65 \text{ cm} < dall = 2.67 \text{ cm}$$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

$$Vd = 3.48 \text{ ton}$$

$$R_{least} = 3.24 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 77 \text{ mm} \text{ \& Full Layout: (38;77 77 38.5)}$$

#### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.16 \text{ t/cm}^2 \text{ \& } q = 0.13 \text{ t/cm}^2 \Rightarrow feq = (f^2 + 3q^2)^{0.5} = 0.28 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

#### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.13 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.16 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.21 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

#### 4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.2 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 231 \text{ mm} \text{ \& } t_p = 10 \text{ mm} \text{ \& } S_w = 6 \text{ mm}$$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
5	(6,4,6)	(14,4,6)	8	17.87	6.75

#### Design Limit state:

$$\text{Combo: } 1.4 * D + 1 * L$$

Md: 17.87 t.m

Vd: 6.75 ton

### Service Limit State

Combo: LIVE

Span: 8 m

Load: -0.6 t/m'

### Design Checks

#### 1-Check Local Buckling

$d_w/t_w = 41.66 < 81.98 \Rightarrow$  Compact Web

$c/t_f = 5.19 < 10.91 \Rightarrow$  Compact Flange

#### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 245.29 \text{ m} \Rightarrow$  Supported (No LTB)

#### 3-Check Bending Stress

Section: IPE450

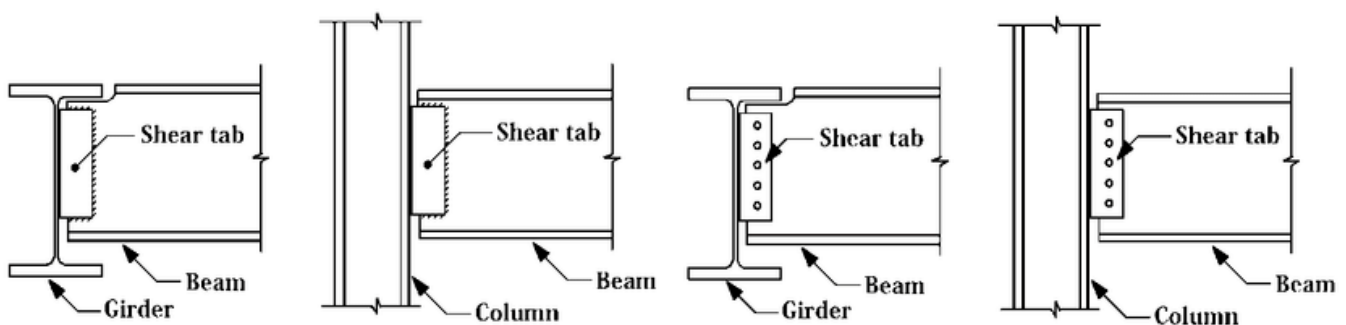
$f_{act} = 1.19 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### 4-Check Shear Stress

$q_{act} = 0.16 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### 5-Check Deflection

$\delta_{act} = 0.45 \text{ cm} < \delta_{all} = 2.67 \text{ cm}$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

Vd= 6.75 ton

Rleast= 4.06 ton

N= 3 with Pitch= 105 mm & Full Layout: (52;105 105 52.5)

#### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.17 \text{ t/cm}^2$  &  $q = 0.18 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.36 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.18 \text{ t/cm}^2$  &  $q_{mt} = 0.17 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.25 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.2 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout  $\Rightarrow L = 315 \text{ mm}$  &  $t_p = 10 \text{ mm}$  &  $S_w = 6 \text{ mm}$

## Columns

Column ID	Start Point	End Point	Height (m)	Nmax (ton)
6	(6,4,0)	(6,4,6)	6	-15.99
7	(14,4,0)	(14,4,6)	6	-15.99
2	(6,0,0)	(6,0,6)	6	-8.35
10	(6,8,0)	(6,8,6)	6	-8.35
3	(14,0,0)	(14,0,6)	6	-8.35
11	(14,8,0)	(14,8,6)	6	-8.35
8	(20,4,0)	(20,4,6)	6	-7.12
5	(0,4,0)	(0,4,6)	6	-7.12
1	(0,0,0)	(0,0,6)	6	-3.81
4	(20,0,0)	(20,0,6)	6	-3.81
9	(0,8,0)	(0,8,6)	6	-3.81
12	(20,8,0)	(20,8,6)	6	-3.81

#### Design Limit state:

Combo:  $1.4 * D + 1 * L$

Nd: -15.99 ton

#### 1-Check Local Buckling

$d_w/t_w = 36.23 < 37.44 \Rightarrow \text{Compact Web}$

$c/t_f = 5.68 < 10.91 \Rightarrow \text{Compact Flange}$

2-Check Normal Stress

Section: IPE300

$\lambda = 179.1 > 100$

$f_c = 0.3 \text{ t/cm}^2 < F_c = 0.23 \text{ t/cm}^2$

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