

## Design Calculation Sheet for ITIFinal02

Designer: dfd

Location: dfdf

City: dfdf

Country: fdf

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## Secondary Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
20	(18,4,6)	(18,8,6)	4	0.07	0.07
19	(18,0,6)	(18,4,6)	4	0.07	0.07
2	(0,4,6)	(0,8,6)	4	0.07	0.07
1	(0,0,6)	(0,4,6)	4	0.07	0.07

### Design Limit state:

Combo: D+L

Md: 0.07 t.m

Vd: 0.07 ton

### Service Limit State

Combo: LIVE

Span: 4 m

Load: 0 t/m'

### Design Checks

#### 1-Check Local Buckling

$dw/tw = 15.58 < 81.98 \Rightarrow$  Compact Web

$c/tf = 3.06 < 10.91 \Rightarrow$  Compact Flange

#### 2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 59.39 \text{ m} \Rightarrow$  Supported (No LTB)

#### 3-Check Bending Stress

Section: IPE270

$f_{act} = 0.36 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### 4-Check Shear Stress

$q_{act} = 0.02 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### 5-Check Deflection

$d_{act} = 0 \text{ cm} < d_{all} = 1.33 \text{ cm}$



## Group Connection Design (Simple Shear Plate Connection)

### 1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 0.07$  ton

$R_{least} = 2.85$  ton

$N = 3$  with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.01$  t/cm<sup>2</sup> &  $q = 0$  t/cm<sup>2</sup>  $\Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.01$  t/cm<sup>2</sup>  $< 1.1 \cdot 0.2F_u = 0.79$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0$  t/cm<sup>2</sup> &  $q_{mt} = 0.01$  t/cm<sup>2</sup>  $\Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.01$  t/cm<sup>2</sup>  $< 0.2F_u = 0.72$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 4-Check Thickness of Plate

$f = (6 \cdot V_d \cdot e) / (t_p \cdot L^2) = 0.01$  t/cm<sup>2</sup>  $< 0.72 \cdot F_y = 1.73$  t/cm<sup>2</sup>  $\Rightarrow$  OK

Plate Layout  $\Rightarrow L = 189$  mm &  $t_p = 10$  mm &  $S_w = 6$  mm

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
16	(14,4,6)	(14,8,6)	4	2.07	2.07
15	(14,0,6)	(14,4,6)	4	2.07	2.07
14	(12,4,6)	(12,8,6)	4	2.07	2.07
13	(12,0,6)	(12,4,6)	4	2.07	2.07
12	(10,4,6)	(10,8,6)	4	2.07	2.07
11	(10,0,6)	(10,4,6)	4	2.07	2.07
17	(16,0,6)	(16,4,6)	4	2.07	2.07
9	(8,0,6)	(8,4,6)	4	2.07	2.07
8	(6,4,6)	(6,8,6)	4	2.07	2.07

7	(6,0,6)	(6,4,6)	4	2.07	2.07
6	(4,4,6)	(4,8,6)	4	2.07	2.07
5	(4,0,6)	(4,4,6)	4	2.07	2.07
4	(2,4,6)	(2,8,6)	4	2.07	2.07
3	(2,0,6)	(2,4,6)	4	2.07	2.07
18	(16,4,6)	(16,8,6)	4	2.07	2.07
10	(8,4,6)	(8,8,6)	4	2.07	2.07

### **Design Limit state:**

Combo: D+L

Md: 2.07 t.m

Vd: 2.07 ton

### **Service Limit State**

Combo: LIVE

Span: 4 m

Load: -1 t/m'

### **Design Checks**

#### **1-Check Local Buckling**

$dw/tw = 27.93 < 81.98 \Rightarrow$  Compact Web

$c/tf = 4.36 < 10.91 \Rightarrow$  Compact Flange

#### **2-Check Lateral Torsional Buckling**

$Lu_{act} = 0 \text{ m} < Lu_{max} = 117.48 \text{ m} \Rightarrow$  Supported (No LTB)

#### **3-Check Bending Stress**

Section: IPE270

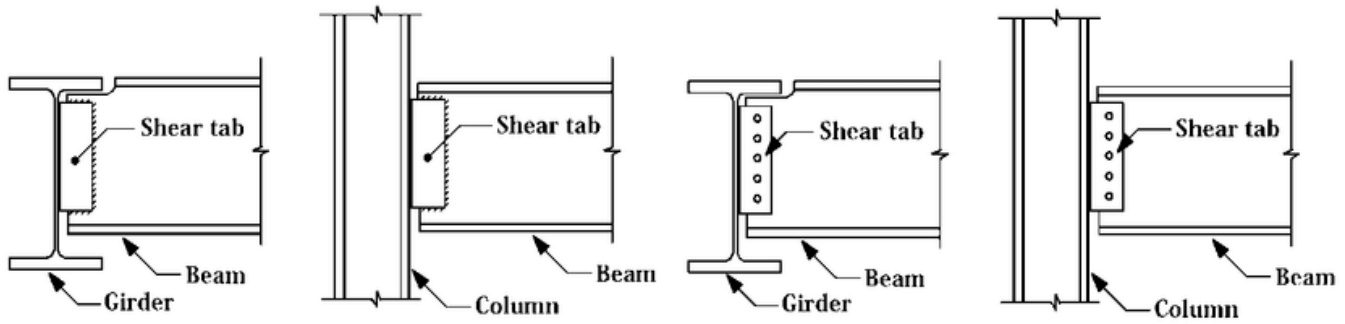
$f_{act} = 1.42 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### **4-Check Shear Stress**

$q_{act} = 0.22 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### **5-Check Deflection**

$d_{act} = 1.2 \text{ cm} < d_{all} = 1.33 \text{ cm}$



## Group Connection Design (Simple Shear Plate Connection)

### 1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 2.07$  ton

$R_{least} = 2.85$  ton

$N = 3$  with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.15$  t/cm<sup>2</sup> &  $q = 0.09$  t/cm<sup>2</sup>  $\Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.22$  t/cm<sup>2</sup> <  $1.1 * 0.2F_u = 0.79$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.09$  t/cm<sup>2</sup> &  $q_{mt} = 0.15$  t/cm<sup>2</sup>  $\Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.17$  t/cm<sup>2</sup> <  $0.2F_u = 0.72$  t/cm<sup>2</sup>  $\Rightarrow$  OK

### 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.17$  t/cm<sup>2</sup> <  $0.72 * F_y = 1.73$  t/cm<sup>2</sup>  $\Rightarrow$  OK

Plate Layout  $\Rightarrow L = 189$  mm &  $t_p = 10$  mm &  $S_w = 6$  mm

## Main Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
9	(12,8,6)	(18,8,6)	6	4.25	2.14
8	(6,8,6)	(12,8,6)	6	4.25	2.14
7	(0,8,6)	(6,8,6)	6	4.25	2.14
3	(12,0,6)	(18,0,6)	6	4.25	2.14
2	(6,0,6)	(12,0,6)	6	4.25	2.14
1	(0,0,6)	(6,0,6)	6	4.25	2.14

### Design Limit state:

Combo: D+L

Md: 4.25 t.m

Vd: 2.14 ton

### Service Limit State

Combo: LIVE

Span: 6 m

Load: -0.67 t/m'

### Design Checks

#### 1-Check Local Buckling

$d_w/t_w = 32.39 < 81.98 \Rightarrow$  Compact Web

$c/t_f = 4.81 < 10.91 \Rightarrow$  Compact Flange

#### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 154.92 \text{ m} \Rightarrow$  Supported (No LTB)

#### 3-Check Bending Stress

Section: IPE270

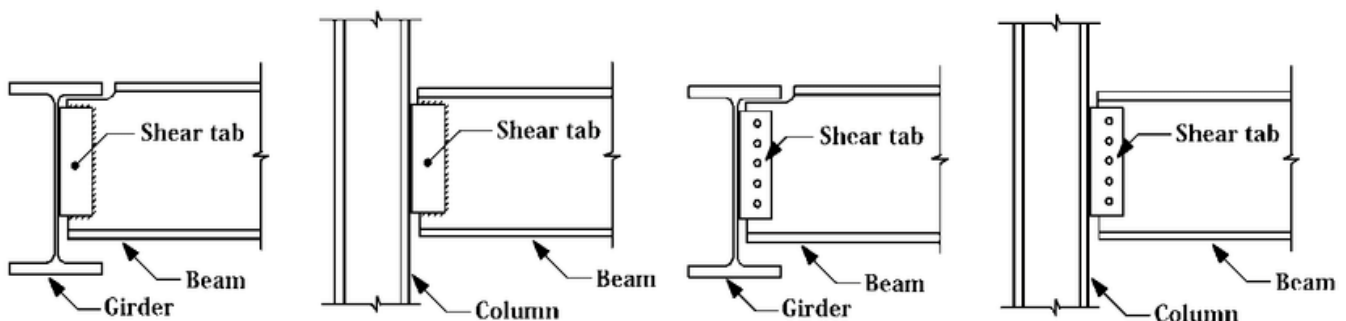
$f_{act} = 1.31 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

#### 4-Check Shear Stress

$q_{act} = 0.14 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

#### 5-Check Deflection

$d_{act} = 1.38 \text{ cm} < d_{all} = 2 \text{ cm}$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

Vd= 2.14 ton

Rleast= 2.85 ton

N= 3 with Pitch= 63 mm & Full Layout: (31;63 63 31.5)

#### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.15 \text{ t/cm}^2$  &  $q = 0.1 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.23 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.1 \text{ t/cm}^2$  &  $q_{mt} = 0.15 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.18 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

#### 4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.18 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout  $\Rightarrow L = 189 \text{ mm}$  &  $t_p = 10 \text{ mm}$  &  $S_w = 6 \text{ mm}$

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Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
6	(12,4,6)	(18,4,6)	6	8.39	4.21
5	(6,4,6)	(12,4,6)	6	8.39	4.21
4	(0,4,6)	(6,4,6)	6	8.39	4.21

#### Design Limit state:

Combo: D+L

Md: 8.39 t.m

Vd: 4.21 ton

#### Service Limit State

Combo: LIVE

Span: 6 m

Load: -1.33 t/m'

#### Design Checks

##### 1-Check Local Buckling

$d_w/t_w = 36.23 < 81.98 \Rightarrow \text{Compact Web}$

$c/t_f = 5.68 < 10.91 \Rightarrow \text{Compact Flange}$

##### 2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 193.65 \text{ m} \Rightarrow \text{Supported (No LTB)}$

##### 3-Check Bending Stress



Section: IPE300

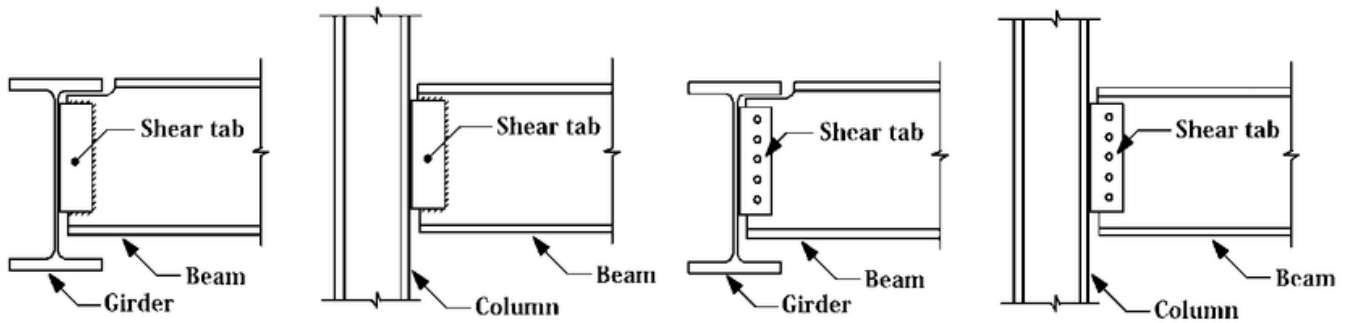
$$fact = 1.51 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$$

#### 4-Check Shear Stress

$$q_{act} = 0.2 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$$

#### 5-Check Deflection

$$d_{act} = 1.28 \text{ cm} < d_{all} = 2 \text{ cm}$$



### Group Connection Design (Simple Shear Plate Connection)

#### 1-Bolts Design

Bolts: M20 of Grade 8.8

$$V_d = 4.21 \text{ ton}$$

$$R_{least} = 3.07 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 70 \text{ mm} \text{ \& Full Layout: (35;70 70 35)}$$

#### 2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.24 \text{ t/cm}^2 \text{ \& } q = 0.17 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.38 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

#### 3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.17 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.24 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.29 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

#### 4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.29 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 210 \text{ mm} \text{ \& } t_p = 10 \text{ mm} \text{ \& } S_w = 6 \text{ mm}$$

## Columns

Column ID	Start Point	End Point	Height (m)	Nmax (ton)
6	(6,4,0)	(6,4,6)	6	-12.82

7	(12,4,0)	(12,4,6)	6	-12.82
2	(6,0,0)	(6,0,6)	6	-6.6
3	(12,0,0)	(12,0,6)	6	-6.6
10	(6,8,0)	(6,8,6)	6	-6.6
11	(12,8,0)	(12,8,6)	6	-6.6
5	(0,4,0)	(0,4,6)	6	-4.61
8	(18,4,0)	(18,4,6)	6	-4.61
1	(0,0,0)	(0,0,6)	6	-2.46
4	(18,0,0)	(18,0,6)	6	-2.46
9	(0,8,0)	(0,8,6)	6	-2.46
12	(18,8,0)	(18,8,6)	6	-2.46

### Design Limit state:

Combo: D+L

Nd: -12.82 ton

### 1-Check Local Buckling

$dw/tw = 36.23 < 37.44 \Rightarrow$  Compact Web

$c/tf = 5.68 < 10.91 \Rightarrow$  Compact Flange

### 2-Check Normal Stress

Section: IPE300

$\lambda = 179.1 > 100$

$f_c = 0.24 \text{ t/cm}^2 < F_c = 0.23 \text{ t/cm}^2$

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