

Design Calculation Sheet for IkImI

Designer:

Location:

City:

Country: Ik

Date: 2020-06-21 08:43:41

Table of Contents

1-Secondary Beams

- Design For Flexural and shear
- Design For serviceability
- Connections Design

2-Main Beams

- Design For Flexural and shear
- Design For serviceability
- Connections Design

3-Columns

- Design For Normal Stress

Secondary Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
45	(10,23.7,3)	(15,23.7,3)	5	1.64	1.31
44	(5,23.7,3)	(10,23.7,3)	5	1.64	1.31
43	(0,23.7,3)	(5,23.7,3)	5	1.64	1.31
2	(5,0,3)	(10,0,3)	5	1.87	1.5
3	(10,0,3)	(15,0,3)	5	1.87	1.5
1	(0,0,3)	(5,0,3)	5	1.87	1.5

Design Limit state:

Combo: D+L

Md: 1.87 t.m

Vd: 1.5 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -0.32 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 27.93 < 81.98 \Rightarrow$ Compact Web

$c/tf = 4.36 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 117.48 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

$f_{act} = 1.28 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.16 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$\delta_{act} = 0.95 \text{ cm} < \delta_{all} = 1.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 1.5$ ton

$R_{least} = 2.85$ ton

$N = 3$ with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.1$ t/cm² & $q = 0.07$ t/cm² $\Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.16$ t/cm² $< 1.1 * 0.2F_u = 0.79$ t/cm² \Rightarrow OK

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.07$ t/cm² & $q_{mt} = 0.1$ t/cm² $\Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.13$ t/cm² $< 0.2F_u = 0.72$ t/cm² \Rightarrow OK

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.13$ t/cm² $< 0.72 * F_y = 1.73$ t/cm² \Rightarrow OK

Plate Layout $\Rightarrow L = 189$ mm & $t_p = 10$ mm & $S_w = 6$ mm

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
39	(10,21.1,3)	(15,21.1,3)	5	3.16	2.53
37	(0,21.1,3)	(5,21.1,3)	5	3.16	2.53
36	(10,19.8,3)	(15,19.8,3)	5	3.16	2.53
35	(5,19.8,3)	(10,19.8,3)	5	3.16	2.53
34	(0,19.8,3)	(5,19.8,3)	5	3.16	2.53
38	(5,21.1,3)	(10,21.1,3)	5	3.16	2.53
40	(0,22.4,3)	(5,22.4,3)	5	3.16	2.53
42	(10,22.4,3)	(15,22.4,3)	5	3.16	2.53
41	(5,22.4,3)	(10,22.4,3)	5	3.16	2.53

6	(10,1.5,3)	(15,1.5,3)	5	3.63	2.9
9	(10,3,3)	(15,3,3)	5	3.63	2.9
8	(5,3,3)	(10,3,3)	5	3.63	2.9
7	(0,3,3)	(5,3,3)	5	3.63	2.9
5	(5,1.5,3)	(10,1.5,3)	5	3.63	2.9
4	(0,1.5,3)	(5,1.5,3)	5	3.63	2.9
31	(0,18.5,3)	(5,18.5,3)	5	3.98	3.18
32	(5,18.5,3)	(10,18.5,3)	5	3.98	3.18
33	(10,18.5,3)	(15,18.5,3)	5	3.98	3.18
10	(0,4.5,3)	(5,4.5,3)	5	4.21	3.37
11	(5,4.5,3)	(10,4.5,3)	5	4.21	3.37
12	(10,4.5,3)	(15,4.5,3)	5	4.21	3.37

Design Limit state:

Combo: D+L

Md: 4.21 t.m

Vd: 3.37 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -0.65 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 32.39 < 81.98 \Rightarrow$ Compact Web

$c/tf = 4.81 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 154.92 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

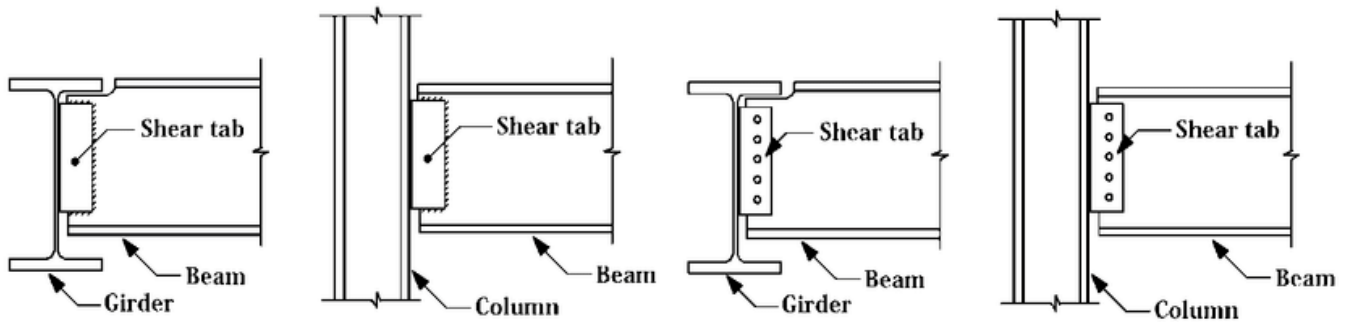
$f_{act} = 1.3 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$$q_{act} = 0.23 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$$

5-Check Deflection

$$d_{act} = 0.65 \text{ cm} < d_{all} = 1.67 \text{ cm}$$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$$V_d = 3.37 \text{ ton}$$

$$R_{least} = 2.85 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 63 \text{ mm \& Full Layout: (31;63 63 31.5)}$$

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.24 \text{ t/cm}^2 \text{ \& } q = 0.15 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.36 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.15 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.24 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.28 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.28 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 189 \text{ mm \& } t_p = 10 \text{ mm \& } S_w = 6 \text{ mm}$$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
15	(10,6.5,3)	(15,6.5,3)	5	4.8	3.84
13	(0,6.5,3)	(5,6.5,3)	5	4.8	3.84
14	(5,6.5,3)	(10,6.5,3)	5	4.8	3.84
20	(5,10.5,3)	(10,10.5,3)	5	4.8	3.84
16	(0,8.5,3)	(5,8.5,3)	5	4.8	3.84
17	(5,8.5,3)	(10,8.5,3)	5	4.8	3.84

18	(10,8.5,3)	(15,8.5,3)	5	4.8	3.84
19	(0,10.5,3)	(5,10.5,3)	5	4.8	3.84
21	(10,10.5,3)	(15,10.5,3)	5	4.8	3.84
22	(0,12.5,3)	(5,12.5,3)	5	4.8	3.84
24	(10,12.5,3)	(15,12.5,3)	5	4.8	3.84
25	(0,14.5,3)	(5,14.5,3)	5	4.8	3.84
26	(5,14.5,3)	(10,14.5,3)	5	4.8	3.84
27	(10,14.5,3)	(15,14.5,3)	5	4.8	3.84
28	(0,16.5,3)	(5,16.5,3)	5	4.8	3.84
29	(5,16.5,3)	(10,16.5,3)	5	4.8	3.84
30	(10,16.5,3)	(15,16.5,3)	5	4.8	3.84
23	(5,12.5,3)	(10,12.5,3)	5	4.8	3.84

Design Limit state:

Combo: D+L

Md: 4.8 t.m

Vd: 3.84 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -1 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 34.73 < 81.98 \Rightarrow$ Compact Web

$c/tf = 5.3 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 174.28 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

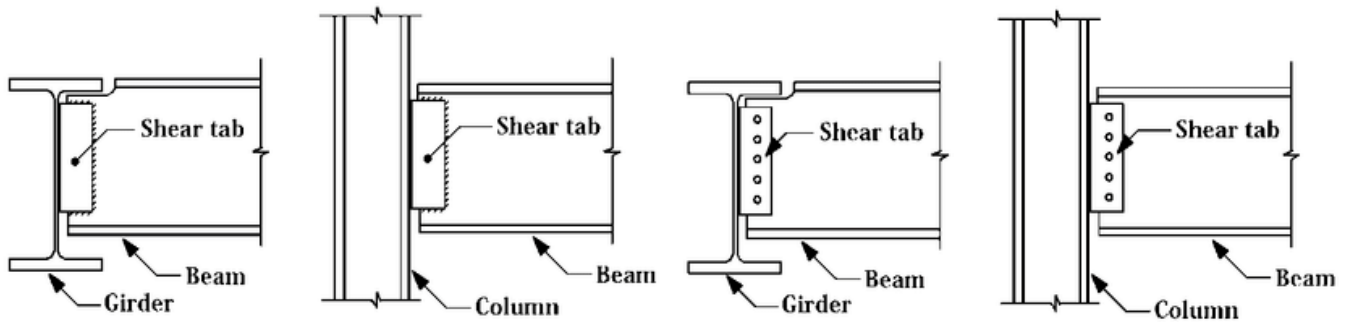
$f_{act} = 1.12 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$$q_{act} = 0.22 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$$

5-Check Deflection

$$d_{act} = 0.67 \text{ cm} < d_{all} = 1.67 \text{ cm}$$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$$V_d = 3.84 \text{ ton}$$

$$R_{least} = 2.85 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 63 \text{ mm \& Full Layout: (31;63 63 31.5)}$$

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.27 \text{ t/cm}^2 \text{ \& } q = 0.17 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.4 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.17 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.27 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.32 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.32 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 189 \text{ mm \& } t_p = 10 \text{ mm \& } S_w = 6 \text{ mm}$$

Main Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
13	(15,0,3)	(15,4.5,3)	4.5	4.45	2.98
1	(0,0,3)	(0,4.5,3)	4.5	4.45	2.98

Design Limit state:

Combo: D+L

Md: 4.45 t.m

Vd: 2.98 ton

Service Limit State

Combo: LIVE

Span: 4.5 m

Load: -0.83 t/m'

Design Checks

1-Check Local Buckling

$d_w/t_w = 32.39 < 81.98 \Rightarrow$ Compact Web

$c/t_f = 4.81 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 154.92 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

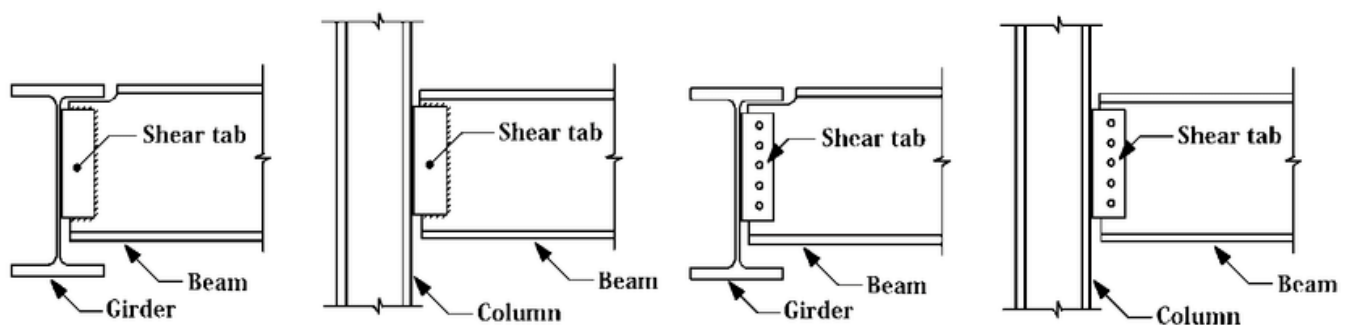
$f_{act} = 1.37 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.2 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 0.54 \text{ cm} < d_{all} = 1.5 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

Vd= 2.98 ton

Rleast= 2.85 ton

N= 3 with Pitch= 63 mm & Full Layout: (31;63 63 31.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.21 \text{ t/cm}^2$ & $q = 0.14 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.31 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.14 \text{ t/cm}^2$ & $q_{mt} = 0.21 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.25 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.25 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 189 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
16	(15,18.5,3)	(15,23.7,3)	5.2	6.69	3.89
4	(0,18.5,3)	(0,23.7,3)	5.2	6.69	3.89
14	(15,4.5,3)	(15,10.5,3)	6	7.84	3.95
2	(0,4.5,3)	(0,10.5,3)	6	7.84	3.95
9	(10,0,3)	(10,4.5,3)	4.5	8.8	5.89
5	(5,0,3)	(5,4.5,3)	4.5	8.8	5.89

Design Limit state:

Combo: D+L

Md: 8.8 t.m

Vd: 5.89 ton

Service Limit State

Combo: LIVE

Span: 6 m

Load: -0.83 t/m'

Design Checks

1-Check Local Buckling

$d_w/t_w = 37.87 < 81.98 \Rightarrow \text{Compact Web}$

$c/t_f = 5.64 < 10.91 \Rightarrow \text{Compact Flange}$

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 206.56 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE330

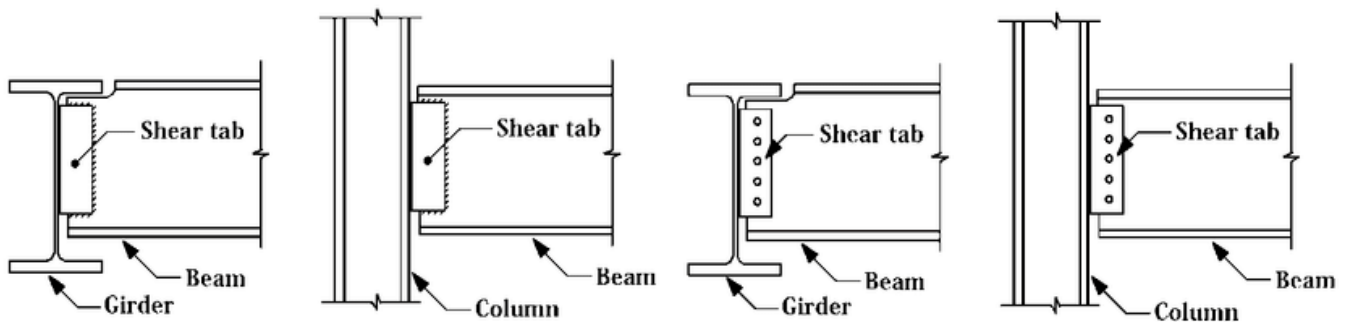
$f_{act} = 1.23 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.24 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 0.57 \text{ cm} < d_{all} = 2 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 5.89 \text{ ton}$

$R_{least} = 3.24 \text{ ton}$

$N = 3$ with Pitch = 77 mm & Full Layout: (38;77 77 38.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.28 \text{ t/cm}^2$ & $q = 0.22 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.47 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.22 \text{ t/cm}^2$ & $q_{mt} = 0.28 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.35 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.33 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 231 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
---------	-------------	-----------	----------	------------	------------

12	(10,18.5,3)	(10,23.7,3)	5.2	13.27	7.68
8	(5,18.5,3)	(5,23.7,3)	5.2	13.27	7.68
10	(10,4.5,3)	(10,10.5,3)	6	15.52	7.79
6	(5,4.5,3)	(5,10.5,3)	6	15.52	7.79
15	(15,10.5,3)	(15,18.5,3)	8	15.65	5.9
3	(0,10.5,3)	(0,18.5,3)	8	15.65	5.9

Design Limit state:

Combo: D+L

Md: 15.65 t.m

Vd: 5.9 ton

Service Limit State

Combo: LIVE

Span: 8 m

Load: -0.94 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 40.24 < 81.98 \Rightarrow$ Compact Web

$c/tf = 5.35 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 232.38 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE400

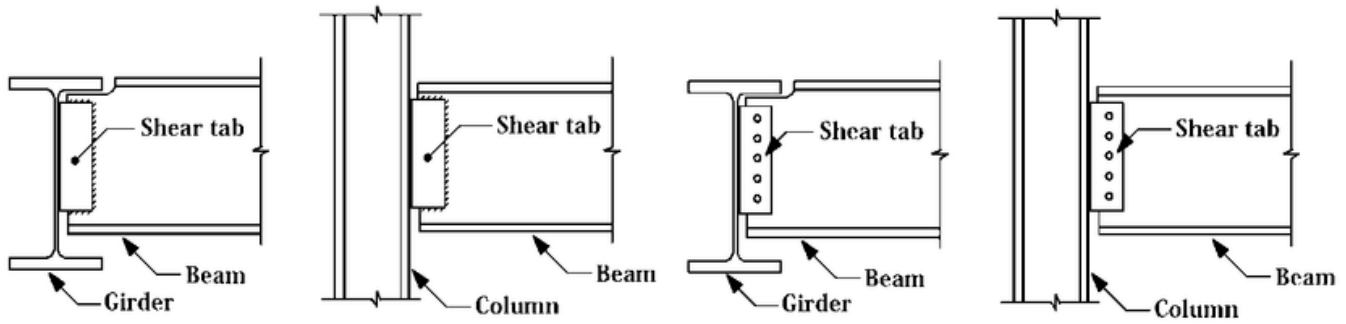
$f_{act} = 1.35 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.17 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 1.03 \text{ cm} < d_{all} = 2.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 5.9$ ton

$R_{least} = 3.72$ ton

$N = 3$ with Pitch = 93 mm & Full Layout: (46;93 93 47.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.19$ t/cm² & $q = 0.18$ t/cm² $\Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.36$ t/cm² < $1.1 * 0.2F_u = 0.79$ t/cm² \Rightarrow OK

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.18$ t/cm² & $q_{mt} = 0.19$ t/cm² $\Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.26$ t/cm² < $0.2F_u = 0.72$ t/cm² \Rightarrow OK

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.23$ t/cm² < $0.72 * F_y = 1.73$ t/cm² \Rightarrow OK

Plate Layout $\Rightarrow L = 280$ mm & $t_p = 10$ mm & $S_w = 6$ mm

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
11	(10,10.5,3)	(10,18.5,3)	8	31.01	11.67
7	(5,10.5,3)	(5,18.5,3)	8	31.01	11.67

Design Limit state:

Combo: D+L

$M_d = 31.01$ t.m

$V_d = 11.67$ ton

Service Limit State

Combo: LIVE

Span: 8 m

Load: -1.88 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 43.36 < 81.98 \Rightarrow$ Compact Web

$c/tf = 4.79 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 271.11 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE550

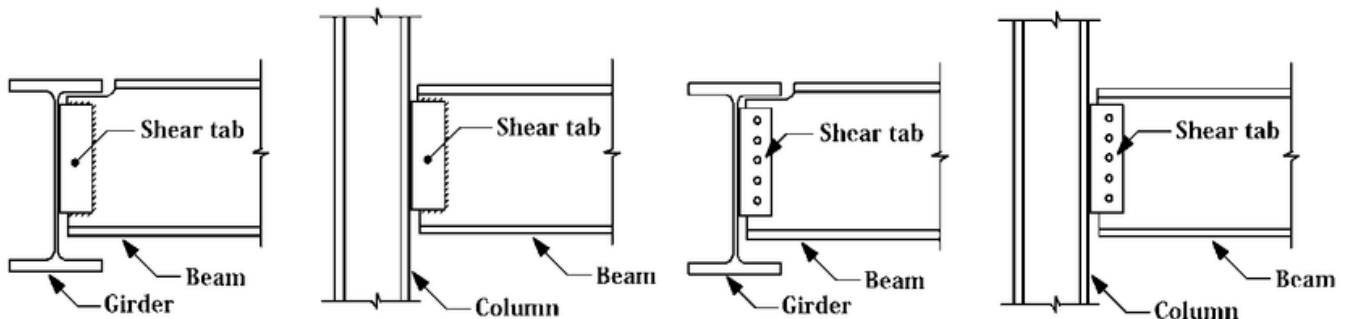
$f_{act} = 1.27 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.19 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$\delta_{act} = 0.71 \text{ cm} < \delta_{all} = 2.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 11.67 \text{ ton}$

$R_{least} = 4.8 \text{ ton}$

$N = 4$ with Pitch = 80 mm & Full Layout: (40;80 80 80 40)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.28 \text{ t/cm}^2$ & $q = 0.31 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.6 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.31 \text{ t/cm}^2$ & $q_{mt} = 0.28 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.42 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 \cdot V_d \cdot e) / (t_p \cdot L^2) = 0.34 \text{ t/cm}^2 < 0.72 \cdot F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 320 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Columns

Column ID	Start Point	End Point	Height (m)	Nmax (ton)
8	(5,10.5,0)	(5,10.5,3)	3	-27.26
13	(10,10.5,0)	(10,10.5,3)	3	-27.26
9	(5,18.5,0)	(5,18.5,3)	3	-25.84
14	(10,18.5,0)	(10,18.5,3)	3	-25.84
7	(5,4.5,0)	(5,4.5,3)	3	-20.55
12	(10,4.5,0)	(10,4.5,3)	3	-20.55
3	(0,10.5,0)	(0,10.5,3)	3	-13.82
18	(15,10.5,0)	(15,10.5,3)	3	-13.82
4	(0,18.5,0)	(0,18.5,3)	3	-13.1
19	(15,18.5,0)	(15,18.5,3)	3	-13.1
2	(0,4.5,0)	(0,4.5,3)	3	-10.43
17	(15,4.5,0)	(15,4.5,3)	3	-10.43
10	(5,23.7,0)	(5,23.7,3)	3	-10.42
15	(10,23.7,0)	(10,23.7,3)	3	-10.42
6	(5,0,0)	(5,0,3)	3	-9.01
11	(10,0,0)	(10,0,3)	3	-9.01
20	(15,23.7,0)	(15,23.7,3)	3	-5.32
5	(0,23.7,0)	(0,23.7,3)	3	-5.32
16	(15,0,0)	(15,0,3)	3	-4.61
1	(0,0,0)	(0,0,3)	3	-4.61

Design Limit state:

Combo: D+L

Nd: -27.26 ton

1-Check Local Buckling

$dw/tw = 36.23 < 37.44 \Rightarrow$ Compact Web

$c/tf = 5.68 < 10.91 \Rightarrow$ Compact Flange

2-Check Normal Stress

Section: IPE300

$\lambda = 89.55 < 100$

$f_c = 0.51 \text{ t/cm}^2 < F_c = 0.88 \text{ t/cm}^2$
