

Design Calculation Sheet for sdss

Designer:

Location:

City:

Country:

Date: 2020-06-21 08:41:00

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- Design For Flexural and shear
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Secondary Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
45	(23.7,10,3)	(23.7,15,3)	5	0.87	0.7
44	(23.7,5,3)	(23.7,10,3)	5	0.87	0.7
43	(23.7,0,3)	(23.7,5,3)	5	0.87	0.7
2	(0,5,3)	(0,10,3)	5	0.99	0.79
3	(0,10,3)	(0,15,3)	5	0.99	0.79
1	(0,0,3)	(0,5,3)	5	0.99	0.79

Design Limit state:

Combo: D+L

Md: 0.99 t.m

Vd: 0.79 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -0.16 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 23.92 < 81.98 \Rightarrow$ Compact Web

$c/tf = 3.95 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 94.24 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

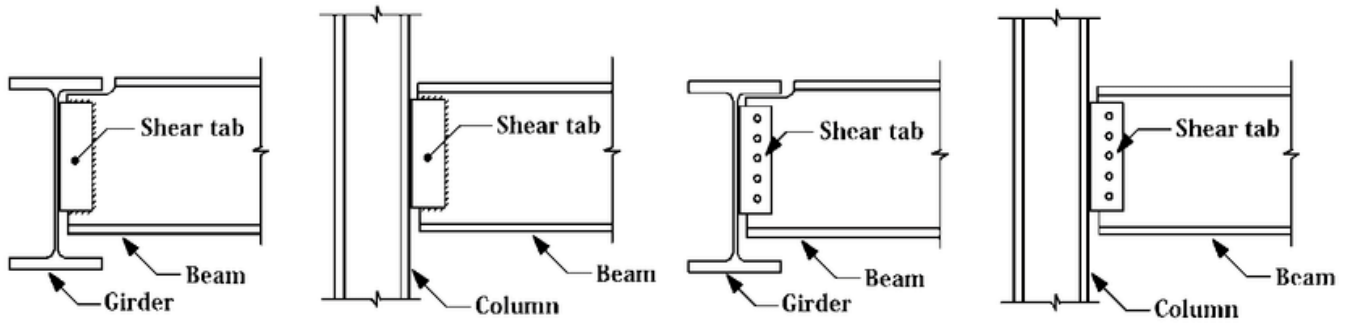
$f_{act} = 1.28 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.12 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 1.16 \text{ cm} < d_{all} = 1.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 0.79$ ton

$R_{least} = 2.85$ ton

$N = 3$ with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.06 \text{ t/cm}^2$ & $q = 0.04 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.08 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.04 \text{ t/cm}^2$ & $q_{mt} = 0.06 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.07 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.07 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 189 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
39	(21.1,10,3)	(21.1,15,3)	5	1.64	1.31
37	(21.1,0,3)	(21.1,5,3)	5	1.64	1.31
36	(19.8,10,3)	(19.8,15,3)	5	1.64	1.31
35	(19.8,5,3)	(19.8,10,3)	5	1.64	1.31
34	(19.8,0,3)	(19.8,5,3)	5	1.64	1.31
38	(21.1,5,3)	(21.1,10,3)	5	1.64	1.31
40	(22.4,0,3)	(22.4,5,3)	5	1.64	1.31
42	(22.4,10,3)	(22.4,15,3)	5	1.64	1.31
41	(22.4,5,3)	(22.4,10,3)	5	1.64	1.31

6	(1.5,10,3)	(1.5,15,3)	5	1.87	1.5
9	(3,10,3)	(3,15,3)	5	1.87	1.5
8	(3,5,3)	(3,10,3)	5	1.87	1.5
7	(3,0,3)	(3,5,3)	5	1.87	1.5
5	(1.5,5,3)	(1.5,10,3)	5	1.87	1.5
4	(1.5,0,3)	(1.5,5,3)	5	1.87	1.5
31	(18.5,0,3)	(18.5,5,3)	5	2.05	1.64
32	(18.5,5,3)	(18.5,10,3)	5	2.05	1.64
33	(18.5,10,3)	(18.5,15,3)	5	2.05	1.64
10	(4.5,0,3)	(4.5,5,3)	5	2.16	1.73
11	(4.5,5,3)	(4.5,10,3)	5	2.16	1.73
12	(4.5,10,3)	(4.5,15,3)	5	2.16	1.73

Design Limit state:

Combo: D+L

Md: 2.16 t.m

Vd: 1.73 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -0.32 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 29.65 < 81.98 \Rightarrow$ Compact Web

$c/tf = 4.56 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 129.1 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

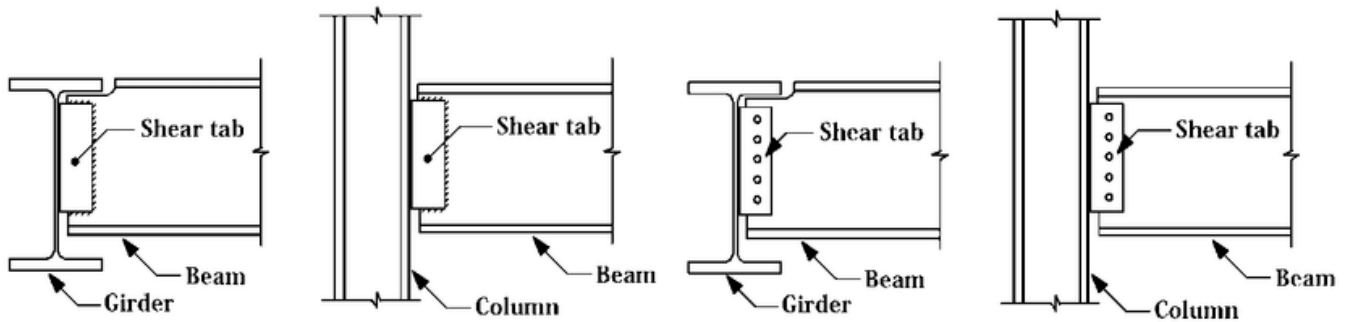
$f_{act} = 1.12 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$$q_{act} = 0.15 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$$

5-Check Deflection

$$\delta_{act} = 0.65 \text{ cm} < \delta_{all} = 1.67 \text{ cm}$$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$$V_d = 1.73 \text{ ton}$$

$$R_{least} = 2.85 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 63 \text{ mm \& Full Layout: (31;63 63 31.5)}$$

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.12 \text{ t/cm}^2 \text{ \& } q = 0.08 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.18 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.08 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.12 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.14 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.15 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 189 \text{ mm \& } t_p = 10 \text{ mm \& } S_w = 6 \text{ mm}$$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
15	(6.5,10,3)	(6.5,15,3)	5	2.46	1.97
13	(6.5,0,3)	(6.5,5,3)	5	2.46	1.97
14	(6.5,5,3)	(6.5,10,3)	5	2.46	1.97
20	(10.5,5,3)	(10.5,10,3)	5	2.46	1.97
16	(8.5,0,3)	(8.5,5,3)	5	2.46	1.97
17	(8.5,5,3)	(8.5,10,3)	5	2.46	1.97

18	(8.5,10,3)	(8.5,15,3)	5	2.46	1.97
19	(10.5,0,3)	(10.5,5,3)	5	2.46	1.97
21	(10.5,10,3)	(10.5,15,3)	5	2.46	1.97
22	(12.5,0,3)	(12.5,5,3)	5	2.46	1.97
24	(12.5,10,3)	(12.5,15,3)	5	2.46	1.97
25	(14.5,0,3)	(14.5,5,3)	5	2.46	1.97
26	(14.5,5,3)	(14.5,10,3)	5	2.46	1.97
27	(14.5,10,3)	(14.5,15,3)	5	2.46	1.97
28	(16.5,0,3)	(16.5,5,3)	5	2.46	1.97
29	(16.5,5,3)	(16.5,10,3)	5	2.46	1.97
30	(16.5,10,3)	(16.5,15,3)	5	2.46	1.97
23	(12.5,5,3)	(12.5,10,3)	5	2.46	1.97

Design Limit state:

Combo: D+L

Md: 2.46 t.m

Vd: 1.97 ton

Service Limit State

Combo: LIVE

Span: 5 m

Load: -0.5 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 29.65 < 81.98 \Rightarrow$ Compact Web

$c/tf = 4.56 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 129.1 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

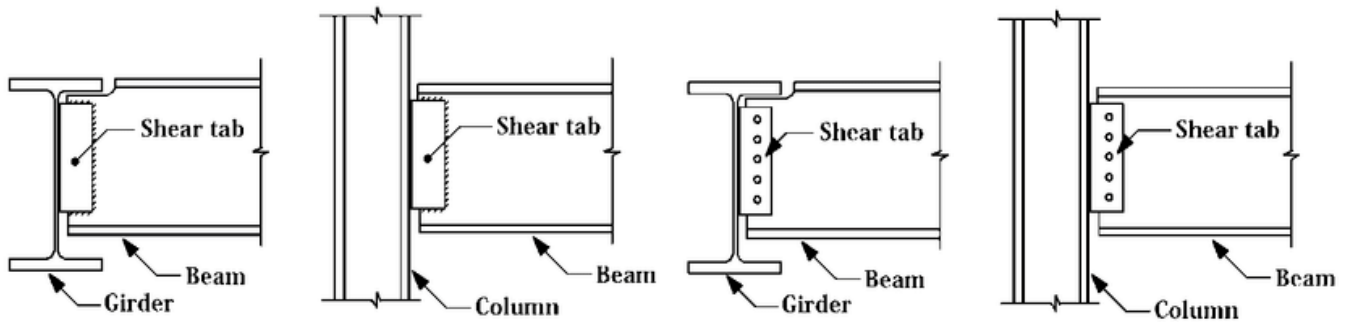
$f_{act} = 1.27 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$$q_{act} = 0.18 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$$

5-Check Deflection

$$\delta_{act} = 1 \text{ cm} < \delta_{all} = 1.67 \text{ cm}$$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$$V_d = 1.97 \text{ ton}$$

$$R_{least} = 2.85 \text{ ton}$$

$$N = 3 \text{ with Pitch} = 63 \text{ mm \& Full Layout: (31;63 63 31.5)}$$

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$$f = 0.14 \text{ t/cm}^2 \text{ \& } q = 0.09 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.21 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$$q = 0.09 \text{ t/cm}^2 \text{ \& } q_{mt} = 0.14 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.16 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$$

4-Check Thickness of Plate

$$f = (6 * V_d * e) / (t_p * L^2) = 0.17 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$$

$$\text{Plate Layout} \Rightarrow L = 189 \text{ mm \& } t_p = 10 \text{ mm \& } S_w = 6 \text{ mm}$$

Main Beams

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
13	(0,15,3)	(4.5,15,3)	4.5	2.34	1.58
1	(0,0,3)	(4.5,0,3)	4.5	2.34	1.58

Design Limit state:

Combo: D+L

Md: 2.34 t.m

Vd: 1.58 ton

Service Limit State

Combo: LIVE

Span: 4.5 m

Load: -0.42 t/m'

Design Checks

1-Check Local Buckling

$d_w/t_w = 29.65 < 81.98 \Rightarrow$ Compact Web

$c/t_f = 4.56 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 129.1 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

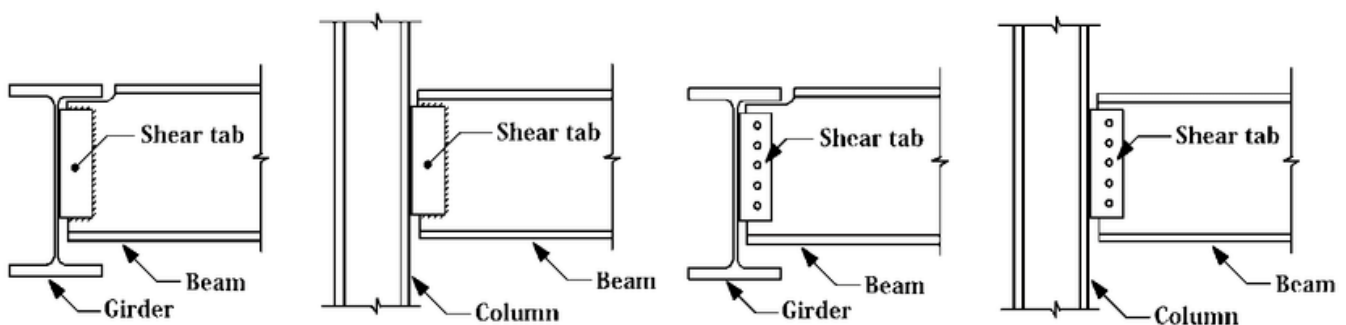
$f_{act} = 1.2 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.14 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 0.55 \text{ cm} < d_{all} = 1.5 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

Vd= 1.58 ton

Rleast= 2.85 ton

N= 3 with Pitch= 63 mm & Full Layout: (31;63 63 31.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.11 \text{ t/cm}^2$ & $q = 0.07 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.17 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.07 \text{ t/cm}^2$ & $q_{mt} = 0.11 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.13 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.13 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 189 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
16	(18.5,15,3)	(23.7,15,3)	5.2	3.53	2.06
4	(18.5,0,3)	(23.7,0,3)	5.2	3.53	2.06
14	(4.5,15,3)	(10.5,15,3)	6	4.09	2.07
2	(4.5,0,3)	(10.5,0,3)	6	4.09	2.07
9	(0,10,3)	(4.5,10,3)	4.5	4.58	3.07
5	(0,5,3)	(4.5,5,3)	4.5	4.58	3.07

Design Limit state:

Combo: D+L

Md: 4.58 t.m

Vd: 3.07 ton

Service Limit State

Combo: LIVE

Span: 6 m

Load: -0.42 t/m'

Design Checks

1-Check Local Buckling

$d_w/t_w = 32.39 < 81.98 \Rightarrow \text{Compact Web}$

$c/t_f = 4.81 < 10.91 \Rightarrow \text{Compact Flange}$

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 154.92 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE270

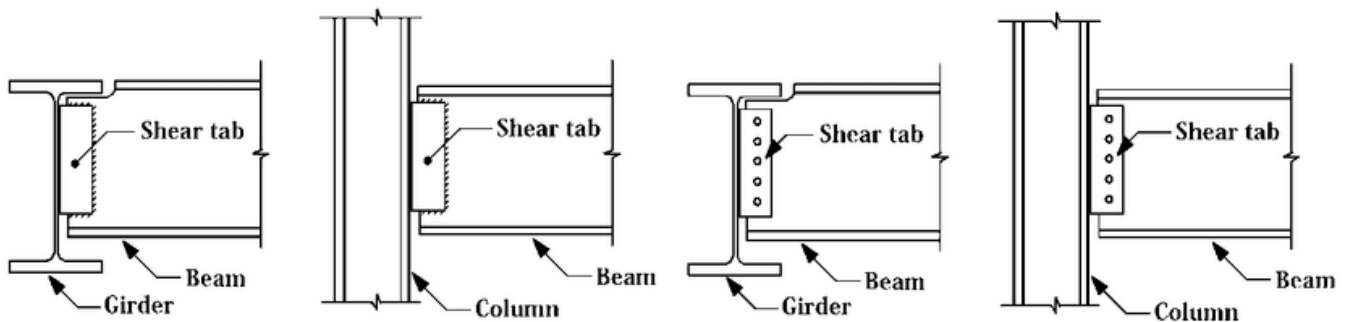
$f_{act} = 1.41 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.21 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 0.86 \text{ cm} < d_{all} = 2 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 3.07 \text{ ton}$

$R_{least} = 2.85 \text{ ton}$

$N = 3$ with Pitch = 63 mm & Full Layout: (31;63 63 31.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.22 \text{ t/cm}^2$ & $q = 0.14 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.32 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.14 \text{ t/cm}^2$ & $q_{mt} = 0.22 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.26 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.26 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 189 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
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12	(18.5,10,3)	(23.7,10,3)	5.2	6.93	4.02
8	(18.5,5,3)	(23.7,5,3)	5.2	6.93	4.02
10	(4.5,10,3)	(10.5,10,3)	6	8.02	4.04
6	(4.5,5,3)	(10.5,5,3)	6	8.02	4.04
15	(10.5,15,3)	(18.5,15,3)	8	8.15	3.09
3	(10.5,0,3)	(18.5,0,3)	8	8.15	3.09

Design Limit state:

Combo: D+L

Md: 8.15 t.m

Vd: 3.09 ton

Service Limit State

Combo: LIVE

Span: 8 m

Load: -0.47 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 36.23 < 81.98 \Rightarrow$ Compact Web

$c/tf = 5.68 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$Lu_{act} = 0 \text{ m} < Lu_{max} = 193.65 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE300

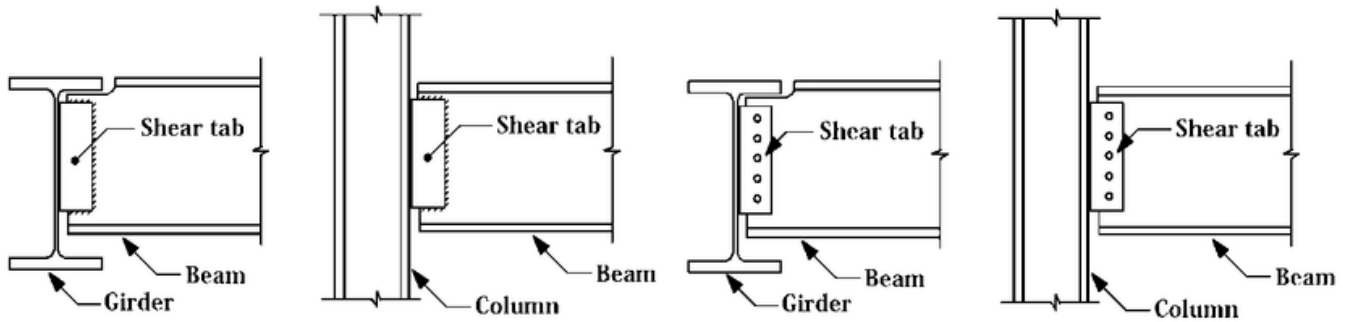
$f_{act} = 1.46 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.15 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$d_{act} = 1.42 \text{ cm} < d_{all} = 2.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 3.09$ ton

$R_{least} = 3.07$ ton

$N = 3$ with Pitch = 70 mm & Full Layout: (35;70 70 35)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.18 \text{ t/cm}^2$ & $q = 0.12 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.28 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.12 \text{ t/cm}^2$ & $q_{mt} = 0.18 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.21 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 * V_d * e) / (t_p * L^2) = 0.21 \text{ t/cm}^2 < 0.72 * F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 210 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Beam ID	Start Point	End Point	Span (m)	Mmax (t.m)	Vmax (ton)
11	(10.5,10,3)	(18.5,10,3)	8	16.01	6.04
7	(10.5,5,3)	(18.5,5,3)	8	16.01	6.04

Design Limit state:

Combo: D+L

$M_d = 16.01 \text{ t.m}$

$V_d = 6.04$ ton

Service Limit State

Combo: LIVE

Span: 8 m

Load: -0.94 t/m'

Design Checks

1-Check Local Buckling

$dw/tw = 40.24 < 81.98 \Rightarrow$ Compact Web

$c/tf = 5.35 < 10.91 \Rightarrow$ Compact Flange

2-Check Lateral Torsional Buckling

$L_{uact} = 0 \text{ m} < L_{umax} = 232.38 \text{ m} \Rightarrow$ Supported (No LTB)

3-Check Bending Stress

Section: IPE400

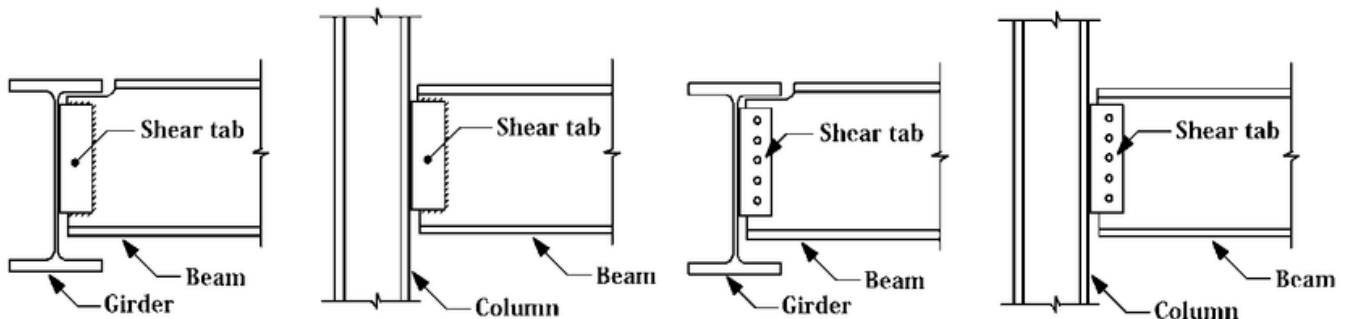
$f_{act} = 1.38 \text{ t/cm}^2 < F_b = 1.54 \text{ t/cm}^2$

4-Check Shear Stress

$q_{act} = 0.18 \text{ t/cm}^2 < q_{all} = 0.84 \text{ t/cm}^2$

5-Check Deflection

$\delta_{act} = 1.03 \text{ cm} < \delta_{all} = 2.67 \text{ cm}$



Group Connection Design (Simple Shear Plate Connection)

1-Bolts Design

Bolts: M20 of Grade 8.8

$V_d = 6.04 \text{ ton}$

$R_{least} = 3.72 \text{ ton}$

$N = 3$ with Pitch = 93 mm & Full Layout: (46;93 93 47.5)

2-Stresses Induced in Fillet Weld Lines at Plane(1-1)

$f = 0.19 \text{ t/cm}^2$ & $q = 0.18 \text{ t/cm}^2 \Rightarrow f_{eq} = (f^2 + 3q^2)^{0.5} = 0.37 \text{ t/cm}^2 < 1.1 * 0.2F_u = 0.79 \text{ t/cm}^2 \Rightarrow \text{OK}$

3-Stresses Induced in Fillet Weld Lines at Plane(2-2)

$q = 0.18 \text{ t/cm}^2$ & $q_{mt} = 0.19 \text{ t/cm}^2 \Rightarrow q_{res} = (q^2 + q_{mt}^2)^{0.5} = 0.27 \text{ t/cm}^2 < 0.2F_u = 0.72 \text{ t/cm}^2 \Rightarrow \text{OK}$

4-Check Thickness of Plate

$f = (6 \cdot V_d \cdot e) / (t_p \cdot L^2) = 0.23 \text{ t/cm}^2 < 0.72 \cdot F_y = 1.73 \text{ t/cm}^2 \Rightarrow \text{OK}$

Plate Layout $\Rightarrow L = 280 \text{ mm}$ & $t_p = 10 \text{ mm}$ & $S_w = 6 \text{ mm}$

Columns

Column ID	Start Point	End Point	Height (m)	Nmax (ton)
8	(10.5,5,0)	(10.5,5,3)	3	-14.14
13	(10.5,10,0)	(10.5,10,3)	3	-14.14
9	(18.5,5,0)	(18.5,5,3)	3	-13.46
14	(18.5,10,0)	(18.5,10,3)	3	-13.46
7	(4.5,5,0)	(4.5,5,3)	3	-10.7
12	(4.5,10,0)	(4.5,10,3)	3	-10.7
3	(10.5,0,0)	(10.5,0,3)	3	-7.26
18	(10.5,15,0)	(10.5,15,3)	3	-7.26
4	(18.5,0,0)	(18.5,0,3)	3	-6.91
19	(18.5,15,0)	(18.5,15,3)	3	-6.91
10	(23.7,5,0)	(23.7,5,3)	3	-5.55
15	(23.7,10,0)	(23.7,10,3)	3	-5.55
2	(4.5,0,0)	(4.5,0,3)	3	-5.51
17	(4.5,15,0)	(4.5,15,3)	3	-5.51
6	(0,5,0)	(0,5,3)	3	-4.79
11	(0,10,0)	(0,10,3)	3	-4.79
20	(23.7,15,0)	(23.7,15,3)	3	-2.88
5	(23.7,0,0)	(23.7,0,3)	3	-2.88
16	(0,15,0)	(0,15,3)	3	-2.5
1	(0,0,0)	(0,0,3)	3	-2.5

Design Limit state:

Combo: D+L

Nd: -14.14 ton

1-Check Local Buckling

$dw/tw = 36.23 < 37.44 \Rightarrow$ Compact Web

$c/tf = 5.68 < 10.91 \Rightarrow$ Compact Flange

2-Check Normal Stress

Section: IPE300

$\lambda = 89.55 < 100$

$f_c = 0.26 \text{ t/cm}^2 < F_c = 0.88 \text{ t/cm}^2$
