



# **8x6 Supersonic Wind Tunnel**

## **A PROJECT REPORT ON INDUSTRIAL VISIT**

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# CERTIFICATE

This is to certify that the project report titled "**8x6 Supersonic Wind Tunnel**" is a bonafide work carried out by **Abhishek Kumar Vishwakarma** in partial fulfillment of the requirements for his academic curriculum. The content and results presented in this report are authentic and based on studies conducted through trusted resources, particularly NASA's official information. This project represents the sincere effort, research, and dedication put forth by the student. The findings and discussions presented in this report reflect the student's understanding of the subject and commitment towards academic excellence.

**Signature:** \_\_\_\_\_

**Name of Instructor:** \_\_\_\_\_

**Date:** \_\_\_\_\_

# Acknowledgement

I would like to express my sincere gratitude to my project guide and faculty members for their constant support and encouragement. I am also thankful to NASA for providing publicly accessible virtual tours and resources that significantly helped in understanding the working of the **8x6 Supersonic Wind Tunnel**. Finally, I extend heartfelt thanks to my peers and family members who supported me throughout this project.

# Abstract

The **8x6 Supersonic Wind Tunnel** at NASA Glenn Research Center is one of the most sophisticated wind tunnels in the world, designed to simulate airspeeds ranging from subsonic to supersonic levels. This project report highlights the key features, structural components, working principles, and significance of this wind tunnel in modern aerospace research. The study explores how the tunnel contributes to the development of aerospace vehicles by providing accurate aerodynamic testing environments.

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# **1. Introduction**

Supersonic wind tunnels are vital in the field of aerospace engineering for testing the aerodynamic properties of objects moving at speeds greater than the speed of sound. The 8x6 Supersonic Wind Tunnel, located at NASA Glenn Research Center, plays a key role in experimental research and testing of aerospace components under various flow conditions.

## **2. Overview of Supersonic Wind Tunnels**

Supersonic wind tunnels differ from subsonic ones in terms of airflow speed and structural complexity. These tunnels are designed to achieve Mach numbers greater than 1 and are used to study the behavior of objects subjected to high-speed airflow.

## **3. About the 8x6 Supersonic Wind Tunnel**

The tunnel gets its name from the dimensions of its test section—8 feet high by 6 feet wide. It is a closed-circuit, variable-pressure wind tunnel capable of operating at both subsonic and supersonic speeds. It is primarily used for propulsion system testing, aerodynamic flow measurements, and structural evaluations.

## **4. Design Specifications**

- **Test Section Dimensions:** 8 ft x 6 ft
- **Mach Number Range:** 0.36 to 2.0
- **Air Supply:** Compressed air system
- **Nozzle Types:** Fixed-geometry and flexible nozzles

- **Pressure Range:** Adjustable based on test requirements

## 5. Working Principle

The tunnel operates by accelerating air through a convergent-divergent nozzle to reach supersonic speeds. Test models are placed in the test section, where instrumentation collects data on pressure, temperature, lift, drag, and other aerodynamic parameters. High-speed sensors and data acquisition systems ensure precise measurements.

## 6. Key Components

- **Settling Chamber:** Ensures steady airflow
- **Convergent-Divergent Nozzle:** Accelerates air to supersonic speeds
- **Test Section:** Houses the test model
- **Diffuser:** Slows down the air post-test section
- **Drive System:** Provides energy to move air through the tunnel

## 7. Testing Capabilities and Applications

This wind tunnel supports a variety of tests including:

- Engine inlet testing
- Supersonic propulsion studies
- Shockwave and boundary layer analysis
- Flow visualization

Applications range from aircraft design and missile testing to space vehicle research and component validation.

## **8. Safety and Control Systems**

Safety is ensured via automated pressure relief valves, emergency shutdown mechanisms, and real-time monitoring systems. The control system allows precise adjustments in pressure, temperature, and Mach number.

## **9. Importance in Aerospace Engineering**

The tunnel is indispensable in validating CFD models and enhancing our understanding of high-speed aerodynamics. Its use in pre-flight validation minimizes real-world testing costs and enhances safety.

## **10. Recent Advancements and Upgrades**

NASA has recently integrated more advanced flow diagnostics, such as laser-based measurement systems and high-speed schlieren imaging, into the 8x6 tunnel to increase test accuracy and visual analysis.

## **11. Conclusion**

The 8x6 Supersonic Wind Tunnel continues to serve as a cornerstone in aerodynamic testing. Its ability to simulate realistic flight conditions makes it a crucial asset in aerospace research and development.

## 12. References

- NASA Glenn Research Center Virtual Tour: <https://www.nasa.gov/nasa-glenn-virtual-tours/>
- NASA Official Publications and Technical Papers
- Aerodynamics textbooks and scientific articles on wind tunnel testing

*End of Report*

