

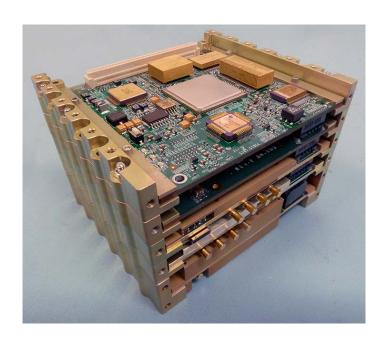
Iris V2.1 CubeSat Deep Space Transponder

X-, Ka-, S-Band, and UHF

Deep Space Telecommunications and Navigation

Features

- · Deep Space Network Compatible
- · Low Volume, Mass, and Cost
- Configurable Software Defined Coherent Transponder
- ~0.5 U Volume
- 1.2 kg Mass
- 35 W DC Power Consumption at 3.8 W Radio Frequency Output, Full Transpond
- Deep Space Network Capability at X-Band Frequencies for Command, Telemetry, and Navigation
- Ka-Band, S-Band, UHF Options with additional NRE
- Passive (Conductive) Thermal Dissipation
- Radiation Tolerant Parts for Extended Deep Space Missions
- Configurable for Earth Orbit
- Targeted for NPR 7120.8 technology demonstrations and Class-D space flight projects
- · CCSDS Compatible



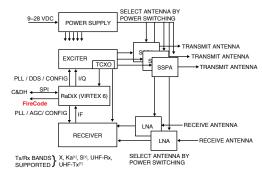
Iris Version 2.1 is a CubeSat/SmallSat compatible transponder developed by the National Aeronautics and Space Administration's (NASA's) Jet Propulsion Laboratory (JPL) as a low volume and mass, lower power and cost, software/firmware defined telecommunications subsystem for deep space. Iris is a deep-space transponder targeted for NPR 7120.8 technology demonstrations and Class-D space flight projects, utilizing COTS-grade components. Iris V2.1's features include ~0.5 U volume, 1.2 kg mass, 35 W DC power consumption when fully transponding at 3.8 W radio frequency output (12.6 W DC input for receive only), and interoperability with NASA's Deep Space Network (DSN) at X-Band frequencies (7.2 GHz uplink, 8.4 GHz downlink) for command, telemetry, and navigation.

Iris V2.1 is designed with an environmentally robust architecture including radiation tolerant parts needed for deep space missions with durations of a few years and thermal management needed for navigation tracking sessions of several hours.

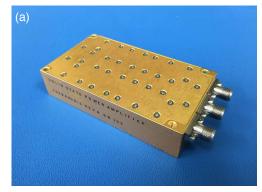
Iris uses a hardware slice architecture and reconfigurable software and firmware enabling extension and adaptation to new capabilities. Among those now planned are: Radio Science support (atmospheric and media measurements and occultations, gravity fields, radars, and radiometers); additional frequency bands (Ka-, S-, UHF); Disruption/Delay Tolerant Networking (DTN); proximity operations (at other planets such as Mars); Near Earth Network (NEN) compatibility; and Space Network (SN) compatibility.

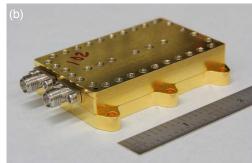
Iris V2.1

General Specifications			
Network Compatibility	DSN, NEN[1], SN[1]		
Redundancy	Single string		
Design Lifetime	3 years		
Frequency Bands	X-band, UHF receive, Ka-[1], S-[1], UHF transmit[1]		
Envelope	100.5 x 101.0 x 56.0 mm		
LNA Envelope	69.4 x 47.5 x 13.0 mm		
SSPA Envelope	86.6 x 42.7 x 17.8 mm		
Flight Operating Temperature	-20°C to +50°C		
Solid State Power Amplifier	3 RF paths, dedicated to 3 antennas, path selectable via power switching		
Low Noise Receive Amplifier	2 RF paths, dedicated to 2 antennas, path selectable via power switching		
VCO	Internal TCXO, external 10 MHz ^[1]		
TCXO Allan Deviation	10 ⁻⁹ at 1 sec (non-coherent operation)		
Ranging Delay Variation	<±30 nsec		
Telemetry Symbol Rates (downlink)	62.5 bps 8 k 1.024 M ^[1] 125 16 k 2.048 M ^[1] 250 32 k 4.046 M ^[1] 500 64 k 8.192 M ^[1] 1 k 128 k semaphores — (< 62.5 bps) ^[1] 2 k 256 k Other arbitrary rates ^[2] 4 k 512 k ^[1]		
Subcarriers, Downlink	25 kHz 281.25 kHz Arbitrary subcarriers to 10 MHz ^[2] Direct carrier modulation		
FPGA	Virtex 6 (LX130T or LX204T*)		
CPU	Gaisler LEON3-FT softcore (on Virtex 6)		
Memory	32 Mbit non-volatile NOR-Flash (radiation tolerant) 16 Mbit volatile SRAM (radiation tolerant) 4 Mbit volatile EDAC SRAM (radiation tolerant)		
Interface	Point-to-point SPI		
Launch Capability	Non-operational at launch		
Radiation, SEL Levels TID Levels	LET >37 MeV-cm²/mg (Virtex 6) 15 krad (ELDRS to 5 krad)		
Telemetry Encoder	Firmware encoder		
Command Detector	Firmware decode with FireCode (spacecraft reset direct command)		
Mounting	CubeSat stack in chassis with separate SSPA and LNA modules		
Carrier Loop BW	Configurable (100 Hz typical)		
Command uplink rates (bps)	62.5 PM/PSK/NRZ 2000 125 4000 250 8000 500 Arbitrary rates ^[2]		
Command uplink subcarriers	16 kHz Direct Carrier modulation Arbitrary subcarriers ^[2]		



Iris V2.1 Block Diagram





Iris V2.1 SSPA (a) and LNA (b) are mounted separately for thermal reasons.

 $^{{}^{[1]}\!}Capability$ under development or planned.

 $^{^{[2]}}$ Capability supportable due to software/firmware reconfigurability with additional NRE.

^{*}Select at build.

Iris V2.1

General Specifications (Contin	ued)		
Command/Telemetry Interface	Command and Telemetry Dictionary, configurable ^[2] Uplink: TC Space Data Link Protocol CCSDS 232.0-B-3 Downlink: AOS Space Data Link Protocol CCSDS 732.0-B-3		
Mass and Power			
Stack Mass	1000 g including thermal enclosure (no UHF) not including cables		
SSPA Mass	150 g		
LNA Mass	80 g		
Input Supply Voltage	9–28 VDC		
Input Supply Power	0.5–35 W (see power states)**		
	Iris Mode	DC Input (W)	
	Battery Connect	0.5	
	X-Receive Only	12.6	
	X-Transmit Only	30.8	
	X-Transmit/Receive	35.0	
Transponder Specifications			
X-Band Uplink Frequency Range	7.145 – 7.190 GHz (channel assignment programmed in firmware) 7.190 – 7.235 (near Earth supported)		
X-Band Downlink Frequency Range	8.400 – 8.450 GHz (channel assignment programmed in firmware)) 8.450 – 8.500 (near Earth supported)		
Other Bands	S-Band: Deep Space ^[1] /near Earth ^[1] Ka-Band: 32/34 GHz Deep Space ^[1] ; 26 GHz near Earth ^[1]		
Coherant Turnaround Ratio X-band	880/749 Standard S- and Ka-Band ratios ^[1] , arbitrary ratios ^[2]		
UHF Frequency Range	390-450 MHz receive, transmit	390–450 MHz receive, transmit ^[1]	
Receiver Specifications			
Noise Figure	5 dB X-Band and UHF		
Carrier Tracking Signal Range	−70 to −130 dBm		
Tracking Range	100 MHz		
Ranging Filter Type	Digital		
Ranging Filter	1500 kHz		
Exciter (X-Band)			
8.4 GHz Output Power (SSPA)	3.8 W BOL (–15 dBm drive from exciter)		
X-Band Phase Noise (1 Hz offset) (100 Hz – 100 kHz offset)	TBM (-20 dBc/Hz) TBM (-60 dBc/Hz)		
X-Band Spurious & Harmonic Outputs	<-40 dBc (-60 dBc at SSPA)	<-40 dBc (-60 dBc at SSPA)	
TLM Encoding	Convolutional 7-1/2 Manchester, Bi-Phase, and bypass (NRZ) Reed Solomon (255,223) Turbo 1/2 Turbo 1/3 Turbo 1/6, block size 8920 bits		
TLM Phase Deviation	0 to 180 degrees	0 to 180 degrees	
Diff 1-way Ranging (coh w/DL carrier)	V Rand 2E1: 10.2 MHz 17.5° typical		

For More Information Contact:

David Bell

Supervisor, Communications Systems & Operations (818) 354-8041 David.J.Bell@jpl.nasa.gov

National Aeronautics and Space Administration Jet Propulsion Laboratory

California Institute of Technology Pasadena, California

www.nasa.gov

JPL 400-1604, Initial Release Copyright 2016 California Institute of Technology. U.S. Government sponsorship acknowledged.

Diff 1-way Ranging (coh w/DL carrier)

X-Band 2F1: 19.2 MHz 17.5° typical

^[1]Capability under development or planned.

^[2]Capability supportable due to software/firmware reconfigurability with additional NRE.

^{**}Power numbers with LX130T FPGA.