State space representation of dynamical systems

State and output equation

$$\left\{ \begin{array}{ll} \dot{x}(t) = f(x(t), u(t)) & \quad x \in \mathbb{R}^n, u \in \mathbb{R}^p, \\ y(t) = g(x(t), u(t)) & \quad y \in \mathbb{R}^q \end{array} \right.$$

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) & A \in \mathbb{R}^{n,n}, B \in \mathbb{R}^{n,p}, \\ y(t) = Cx(t) + Du(t) & C \in \mathbb{R}^{q,n}, D \in \mathbb{R}^{q,p} \end{cases}$$

Solution of LTI systems

$$\begin{split} x(t) &= e^{At}x(0) + \int_0^t e^{A(t-\tau)}Bu(\tau)d\tau \\ y(t) &= C\left[e^{At}x(0) + \int_0^t e^{A(t-\tau)}Bu(\tau)d\tau\right] + Du(t) \end{split}$$

$$\begin{split} X(s) &= (sI - A)^{-1}x(0) + (sI - A)^{-1}BU(s) \\ Y(s) &= C(sI - A)^{-1}x(0) + \left[C(sI - A)^{-1}B + D\right]U(s) \end{split}$$

$$H(s) = [C(sI - A)^{-1}B + D]$$
transfer matrix

Steady state analysis

Step input $u(t) = A_u \varepsilon(t)$

$$y_{ss}(t) = A_y \varepsilon(t)$$

$$A_y = \lim_{s \to 0} sY(s) = \lim_{s \to 0} sH(s)U(s)$$

Sinusoidal input $u(t) = A_u \sin \omega t$

$$y_{ss}(t) = A_u \sin(\omega t + \phi)$$

$$A_y = A_y(j\omega) = A_u|H(j\omega)|$$
 $\phi = \phi(j\omega) = \angle H(j\omega)$

Linearization of nonlinear systems in a neighborhood of an equilibrium point (\bar{x}, \bar{u})

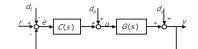
$$\delta x(t) = x(t) - \bar{x}, \delta u(t) = u(t) - \bar{u}, \delta y(t) = y(t) - \bar{y}$$

$$\begin{cases} & \dot{\delta x}(t) = A\delta x(t) + B\delta u(t) \\ & \delta y(t) = C\delta x(t) + D\delta u(t) \end{cases} \quad \delta x(0) = x(0) - \bar{x}$$

$$A = \frac{\partial f(x, u)}{\partial x} \begin{vmatrix} x = \bar{x} \\ u = \bar{u} \end{vmatrix} = \frac{\partial f(x, u)}{\partial u} \begin{vmatrix} x = \bar{x} \\ u = \bar{u} \end{vmatrix}$$

$$C = \frac{\partial g(x, u)}{\partial x} \begin{vmatrix} x = \bar{x} \\ u = \bar{u} \end{vmatrix} = \frac{\partial g(x, u)}{\partial u} \begin{vmatrix} x = \bar{x} \\ u = \bar{u} \end{vmatrix}$$

$$u = \bar{u}$$



Feedback control system

L(s) Loop function

 $S(s) = \frac{1}{1 + L(s)}$ Sensitivity function

 $T(s) = \frac{L(s)}{1+L(s)}$ Complementary sensitivity fn

 $R(s) = \frac{C(s)}{1 + L(s)}$ Control sensitivity fn

 $Q(s) = \frac{G(s)}{1+L(s)}$ Actuator disturbance

sensitivity fn

 S_p S(s) resonant peak T_n T(s) resonant peak

Stability analysis

Nyquist stability criterion: Z = P + N where

Z = # poles of T(s)

P = # poles of L(s) inside the Nyquist contour

N = number of encirclement of L(s) around (-1,0)

N > 0 if clockwise

N < 0 if counterclockwise

Steady state analysis (polynomial input)

	$\begin{array}{c} \mathrm{h} \rightarrow \\ g \downarrow \end{array}$	$r(t) = \rho \epsilon(t)$	$1 \\ r(t) = \rho t \epsilon(t)$	$r(t) = \rho \frac{t^2}{2} \epsilon(t)$
$ e_r^{\infty} $	0	$\left \frac{\rho}{1+K_0}\right $	∞	∞
	1	0	$\left rac{ ho}{K_1} ight $	∞
	2	0	0	$\left rac{ ho}{K_2} ight $

$$K_g = \lim_{s \to 0} s^g L(s)$$

$ y_{dy}^{\infty} $	$h ightarrow g \downarrow$	$\begin{array}{c} 0 \\ dy(t) = \delta_y \epsilon(t) \end{array}$	$d_{y}(t) = \delta_{y} t \epsilon(t)$	$d_{y}(t) = \delta_{y} \frac{t^{2}}{2} \epsilon(t)$
	0	$\left \frac{\delta_y}{1+K_0}\right $	∞	∞
	1	0	$\left rac{\delta y}{K_1} ight $	8
	2	0	0	$\left rac{\delta_y}{K_2} ight $

$$K_g = \lim_{s \to 0} s^g L(s)$$

$ y_{d_a}^{\infty} $	$h \rightarrow g_c \downarrow$	$d_a(t) = \delta_a \epsilon(t)$	$\begin{array}{c} 1 \\ d_a(t) = \delta_a t \epsilon(t) \end{array}$	$d_{a}(t) = \frac{2}{\delta_{a} \frac{t^{2}}{2} \epsilon(t)}$
	0	$\left \frac{\delta_a}{K_0} \right $	∞	∞
	1	0	$\left rac{\delta_a}{K_1} ight $	∞
	2	0	0	$\left rac{\delta_a}{K_2} ight $

$$K_0 = \begin{cases} K_C & \text{if } G(s) \text{ has poles in } 0\\ \frac{1+K_CK_G}{K_C} & \text{if } G(s) \text{ has not poles in } 0 \end{cases}$$

$$K_{g_c} = \lim_{s \to 0} s_c^g C(s), \quad g_c \ge 1$$

Feedback systems design

Prototype 2nd order model

$$T(s) = K \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

Step output response $(u(t) = \bar{u}\varepsilon(t))$

$$y_{\infty} = K\bar{u}$$

$$\hat{S} = \exp^{-\frac{\pi\zeta}{\sqrt{1-\zeta^2}}} \qquad \zeta = \frac{|\ln(\hat{S})|}{\sqrt{\pi^2 + \ln^2(\hat{S})}}$$

$$t_r = \frac{1}{\omega_n \sqrt{1-\zeta^2}} (\pi - \arccos\zeta)$$

$$t_{s,\alpha\%} = \frac{1}{\omega_n \zeta} \ln\left(\frac{\alpha}{100}\right)^{-1}$$

$$\hat{t} = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}}$$

Frequency response

$$T_{p} = \frac{1}{2\zeta\sqrt{1-\zeta^{2}}}$$

$$\omega_{B} = \omega_{n}\sqrt{1-2\zeta^{2}+\sqrt{2-4\zeta^{2}+4\zeta^{4}}}$$

$$S_{p} = \frac{2\zeta\sqrt{2+4\zeta^{2}+2\sqrt{1+8\zeta^{2}}}}{\sqrt{1+8\zeta^{2}}+4\zeta^{2}-1}$$

$$\omega_{c} = \omega_{n}\sqrt{\sqrt{1+4\zeta^{4}}-2\zeta^{2}}$$

Lead network / Lag network

$$C_D(s) = \frac{1 + \frac{s}{\omega_D}}{1 + \frac{s}{m_D \omega_D}} \qquad m_D > 1$$

$$C_I(s) = \frac{1 + \frac{s}{m_I \omega_I}}{1 + \frac{s}{\omega_I}} \qquad m_I > 1$$

Negative real zero

$$C_z(s) = 1 + \frac{s}{\omega_z}$$

PID standard controllers

PI

$$C(s) = \frac{K_c(1 + \frac{s}{\omega_z})}{s} \qquad C(s) = K_p \left(1 + \frac{1}{T_i s}\right)$$

PID

$$C(s) = \frac{K_c(1 + \frac{s}{\omega_{z1}})(1 + \frac{s}{\omega_{z2}})}{s} \qquad C(s) = K_p \left(1 + \frac{1}{T_i s} + T_d s\right)$$

PD

$$C(s) = K_c \left(1 + \frac{s}{\omega_z}\right)$$
 $C(s) = K_p(1 + T_d s)$

Sinusoidal disturbances

$$d(t) = \delta \sin \omega t \qquad \omega \in \left[\omega^L, \omega^H\right]$$
$$y_{ss}(t) = \delta |W_{dy}(j\omega)| \sin \left(\omega t + \angle W_{dy}(j\omega)\right)$$
$$|y_d^{\infty}| = \max |y_{perm}(t)| = \delta \cdot \max |W_{dy}(j\omega)|$$
$$\omega \in \left[\omega^L, \omega^H\right]$$

Digital Control Design

$$G_{\rm ZOH}(s) = \frac{1 - e^{Ts}}{s} \simeq \frac{1}{1 + sT/2}$$

Butterworth filter design

$$\omega_f = \left(\omega_h \frac{\gamma^2}{1 - \gamma^2}\right)^{\frac{1}{2n}} \simeq \omega_h \gamma^{\frac{1}{n}}$$

