

## DESIGN REPORT

### Input Parameters

- Payload Weight : 0.2 kg
- Endurance : 60.0 minutes
- Cruise Altitude : 300.0 m
- Cruise Speed : 15.0 m/s
- Wing Loading : 3 kg/m<sup>2</sup>
- Aspect Ratio : 5
- Taper Ratio : 0.5
- Fuselage Width Ratio : 0.14
- LE Sweep Angle : 35 deg
- Twist Angle : -2 deg
- Root Airfoil : hs522
- Tip Airfoil : hs522

### Calculated Parameters

- MTOW/Payload Ratio : 4.8792
- MTOW : 0.9758 kg
- Wing Area : 0.3253 m<sup>2</sup>
- Full Span : 1.2753 m
- Fuselage Width : 0.1785 m
- Wing-Only Semispan : 0.5484 m
- Root Chord : 0.3249 m
- Tip Chord : 0.1625 m
- MAC : 0.2527 m
- MAC LE X-Position : 0.1707 m

### Flight Conditions

- Air Density : 1.1911 kg/m<sup>3</sup>
- Dynamic Viscosity : 1.7797e-05 Ns/m<sup>2</sup>
- Reynolds Number : 253702
- Mach Number : 0.0442
- Lift Coefficient : 0.2196

### Q3D Analysis Results

- Angle of Attack : 2.2014 deg
- Drag Coefficient : 0.0123

### Propulsion System Specifications

- Power-to-Weight Ratio : 220 W/kg
- Minimum Motor Power : 214.6858 W