Generated: 2023-05-19 16:52:00

## DESIGN REPORT

# Input Parameters

- Payload Weight : 0.5 kg
- Endurance : 60.0 minutes
- Cruise Altitude : 300.0 meters
- Cruise Speed : 15.0 m/s
- Wing Loading : 3 kg/m2
- Aspect Ratio : 7
- Taper Ratio : 0.5

- Fuselage Width Ratio : 0.14

LE Sweep AngleTwist Angle25 degrees2 degrees

## Calculated Parameters

- MTOW/Payload Ratio : 4.8792228128

- MTOW : 2.4396114064 kg - Wing Area : 0.8132038021333333 meters - Full Span : 2.385880679106424 meters - Fuselage Width : 0.3340232950748994 meters - Wing-Only Semispan : 1.0259286920157624 meters - Root Chord : 0.4341912063887943 meters - Tip Chord : 0.21709560319439716 meters - MAC : 0.3377042716357289 meters : 0.21709560319439716 meters - MAC : 0.3377042716357289 meters - MAC LE X-Position : 0.212621513692396 meters

## Freestream Conditions

- Air Density : 0.5 kg/m3 - Dynamic Viscosity : 60.0 Ns/m2 - Reynolds Number : 300.0 - Mach Number : 15.0 - AoA : 1 degrees

# Q3D Analysis Results

- Lift Coefficient : 0.39388 - Drag Coefficient : 0.017155595222749644