

Homework 6

ME5659 Spring 2024

Due: See Canvas, turn in on Gradescope

Problem 1 (9 points)

Consider the following linear systems, where

$$(i) A = \begin{bmatrix} -4 & 0 \\ 0 & -5 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 0 \end{bmatrix}, \quad C = [1 \ 0].$$

$$(ii) A = \begin{bmatrix} 2 & 0 \\ 0 & 1 \end{bmatrix}, \quad B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}, \quad C = [1 \ 1].$$

$$(iii) A = \begin{bmatrix} 3 & 6 & 4 \\ 9 & 6 & 10 \\ -7 & -7 & -9 \end{bmatrix}, \quad B = \begin{bmatrix} 1/3 & 4/3 \\ 4/3 & 1/3 \\ -2/3 & 1/3 \end{bmatrix}, \quad C = \begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \end{bmatrix}.$$

(a) **3 points.** Use the PBH test for observability assessment.

(b) **6 points.** If possible, compute the observer canonical form of these systems. If you cannot, put the systems into Kalman observable canonical form and evaluate their detectability.

SOLUTIONS

A) PBH TEST

$$\text{rank} \begin{bmatrix} C \\ AI - A \end{bmatrix} = n$$

i) $A = \begin{bmatrix} -4 & 0 \\ 0 & -5 \end{bmatrix} \quad B = \begin{bmatrix} 1 \\ 0 \end{bmatrix} \quad C = [1 \ 0] \quad \lambda_1 = -4 \quad \lambda_2 = -5 \Rightarrow \begin{array}{l} \text{THE SYSTEM} \\ \text{IS NOT} \\ \text{OBSERVABLE} \end{array}$

$$AI - A = \begin{bmatrix} \lambda + 4 & 0 \\ 0 & \lambda + 5 \end{bmatrix}$$

$$\text{rank} \begin{bmatrix} 1 & 0 \\ \lambda + 4 & 0 \\ 0 & \lambda + 5 \end{bmatrix} \quad \lambda_1 = -4 \quad \text{rank} \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} = 2 = n \quad \text{observable}$$

$$\lambda_2 = -5 \quad \text{rank} \begin{bmatrix} 1 & 0 \\ -1 & 0 \\ 0 & 0 \end{bmatrix} = 1 < n \quad \text{not observable}$$

ii) $A = \begin{bmatrix} 2 & 0 \\ 0 & 1 \end{bmatrix} \quad C = [1 \ 1] \quad \lambda_1 = 2 \quad \lambda_2 = 1 \quad \text{eigenvalues}$

$$\text{rank} \begin{bmatrix} 1 & 1 \\ \lambda - 2 & 1 \\ 0 & \lambda - 1 \end{bmatrix} \quad \lambda_1 = 2 \quad \text{rank} \begin{bmatrix} 1 & 1 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} = 2 = n \quad \text{observable}$$

$$\lambda_2 = 1$$

rank $\begin{bmatrix} 1 & 1 \\ -1 & 0 \\ 0 & 0 \end{bmatrix} = 2 = n$ observable THE SYSTEM IS OBSERVABLE

iii) $A = \begin{bmatrix} 3 & 6 & 4 \\ 9 & 6 & 10 \\ -7 & -7 & -9 \end{bmatrix}$ $B = \begin{bmatrix} 1/3 & 4/3 \\ 4/3 & 1/3 \\ -2/3 & 1/3 \end{bmatrix}$ $C = \begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \end{bmatrix}$ $\lambda_1 = 5$
 $\lambda_2 = -2$
 $\lambda_3 = -3$

$$\lambda_1 = 5$$

rank $\begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \\ 2 & -6 & -4 \\ -9 & -1 & -10 \\ 7 & 7 & 14 \end{bmatrix} = 2 < 3$ undetectable

$$\begin{bmatrix} \lambda - 3 & -6 & -4 \\ -9 & \lambda - 6 & -10 \\ 7 & 7 & \lambda + 9 \end{bmatrix}$$

$$\lambda_2 = -2$$

rank $\begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \\ -5 & -6 & -4 \\ -9 & -8 & -10 \\ 7 & 7 & 7 \end{bmatrix} = 3 = n$ observable

THE SYSTEM
IS NOT
OBSERVABLE

$$\lambda_3 = -3$$

rank $\begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \\ -6 & -6 & -4 \\ -9 & -9 & -10 \\ 7 & 7 & 6 \end{bmatrix} = 3 = n$ observable

b)

i) NOT OBSERVABLE

KALMAN CANONICAL FORM

$$Q = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ -4 & 0 \end{bmatrix} \quad T^{-1} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \quad C = CT$$

$$C = \begin{bmatrix} 1 & 0 \end{bmatrix}$$

$$A_K = T^{-1} A T$$

$$A_K = \begin{bmatrix} -4 & 0 \\ 0 & -5 \end{bmatrix} = \begin{bmatrix} A_{11} & 0 \\ A_{21} & A_{22} \end{bmatrix}$$
 but since $\operatorname{Re}(\lambda_1(A_{22})) < 0$
it is detectable

(i) OBSERVABILITY \Rightarrow CCF

$$(SI - A) = \left(\begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{bmatrix} 2 & 0 \\ 0 & 1 \end{bmatrix} \right) = \begin{bmatrix} s-2 & 0 \\ 0 & s-1 \end{bmatrix} = (s-2)(s-1) = s^2 - 2s - s + 2 = s^2 - 3s + 2$$

$\frac{\text{II}}{a_1}$ $\frac{\text{II}}{a_0}$

$$Q_{CCF}^{-1} = \begin{bmatrix} -3 & 1 \\ 1 & 0 \end{bmatrix} \quad Q = \begin{bmatrix} 1 & 1 \\ 2 & 1 \end{bmatrix}$$

$$T_{OCF} = (Q_{OCF}^{-1} Q)^{-1} = \begin{bmatrix} 1 & 2 \\ -1 & -1 \end{bmatrix} \quad A_{OCF} = T_{OCF}^{-1} A T_{OCF} = \begin{bmatrix} 0 & -2 \\ 1 & 3 \end{bmatrix}$$

$$C_{OCF} = C T_{OCF} = \begin{bmatrix} 0 & 1 \end{bmatrix}$$

(ii)

$$T^{-1} = \begin{bmatrix} 1 & 2 & 3 \\ 3 & 3 & 6 \\ 0 & 0 & 1 \end{bmatrix} \quad A_{kalman} = \begin{bmatrix} -3 & 1 & 0 \\ 0 & -2 & 0 \\ 0 & -2.3 & s \end{bmatrix}$$

Not Detectable
since $\operatorname{Re}(\lambda_i(Au)) > 0$

$$C_{kalman} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}$$

A_{22}

Problem 2 (4 points)

An approximate linear model of the lateral dynamics of an aircraft, for a particular set of flight conditions, has the state and control vectors in the perturbation quantities

$$x = [p \quad r \quad \beta \quad \phi]^T, \quad u = [\delta_a \quad \delta_r]^T$$

where p and r are incremental roll and yaw rates, β is an incremental sideslip change, and ϕ is an incremental roll angle. The control inputs are the incremental changes in the aileron angle δ_a and in the rudder angle δ_r , respectively. This linearized model has

$$A = \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix}, \quad B = \begin{bmatrix} 20 & 2.8 \\ 0 & -3.13 \\ 0 & 0 \\ 0 & 0 \end{bmatrix}$$

If you could have either a rate gyro that measures the incremental roll rate or a bank indicator that measures incremental roll angle, which would you choose and why?

LET'S assume we have a rate gyro that allows us to measure the incremental roll rate $\dot{\phi}$

$y = [1 \ 0 \ 0 \ 0]x$ Because we are able to measure β by hypothesis.

LET'S SEE OUR \mathbf{Q} (observability matrix).

$$Q = \begin{bmatrix} 1 & 0 & 0 & 0 \\ -10 & 0 & -10 & 0 \\ 100 & 10 & 107 & 0 \\ -1000 & -114 & -984.9 & 0 \end{bmatrix}$$

$$CA = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix}$$

$$CA \cdot A = \begin{bmatrix} -10 & 0 & -10 & 0 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 100 & 10 & 107 & 0 \end{bmatrix}$$

$$CAAA = \begin{bmatrix} 100 & 10 & 107 & 0 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} -1000 & -114 & -984.9 & 0 \end{bmatrix}$$

clearly $\det(Q) = 0$ for the presence of the east column full of zeros.
 $\text{rk}(Q) = 3$ So the system is not fully observable.

LET'S SEE IN TERMS OF ROLL ANGLE ϕ

$$\dot{x} = \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} x + \begin{bmatrix} 20 & 2.8 \\ 0 & -3.13 \\ 0 & 0 \\ 0 & 0 \end{bmatrix} u$$

where

$$x = \begin{bmatrix} p \\ r \\ \beta \\ \phi \end{bmatrix} \quad u = \begin{bmatrix} \delta a \\ \delta r \end{bmatrix} \quad \text{so} \quad \dot{x} = \begin{bmatrix} \dot{p} \\ \dot{r} \\ \dot{\beta} \\ \dot{\phi} \end{bmatrix} \quad \text{OUTPUT EQUATION} \Rightarrow y = Cx + Du$$

$$y = \begin{bmatrix} 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} p \\ r \\ \beta \\ \phi \end{bmatrix}$$

WE WANT TO MEASURE THE INCREMENTAL ROLL RATE

$$Q = \begin{bmatrix} 0 & 0 & 0 & 1 \\ 1 & 0 & 0 & 0 \\ -10 & 0 & -10 & 0 \\ 100 & 10 & 107 & 0 \end{bmatrix} \quad CA = \begin{bmatrix} 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \end{bmatrix}$$

$$CAA = \begin{bmatrix} 1 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} -10 & 0 & -10 & 0 \end{bmatrix}$$

$$CAA A = \begin{bmatrix} -10 & 0 & -10 & 0 \end{bmatrix} \begin{bmatrix} -10 & 0 & -10 & 0 \\ 0 & -0.7 & 9 & 0 \\ 0 & -1 & -0.7 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 100 & 10 & 107 & 0 \end{bmatrix}$$

$\text{rk}(Q) = ?$ $\det(Q) = -100$ and so the $\text{rk}(Q) = 4$ the system is fully observable so we can reconstruct every angle state from the output.

\therefore PICK THE BANK INDICATOR THAT MEASURES INCREMENTAL ROLL ANGLE, SINCE WITH THAT OUTPUT ALL THE STATES ARE OBSERVABLE

Problem 3 (12 points)

Given

$$G(s) = \frac{s-1}{s^3 + 2s^2 - s - 2}$$

(a) **3 points.** Find a three-dimensional controllable realization. Is it observable? Is it detectable?

(b) **3 points.** Find a three-dimensional observable realization. Is it controllable? Is it stabilizable?

(c) **2 points.** What causes the apparent difference in controllability and observability of the same system? Derive an irreducible transfer function for $G(s)$.

(d) **4 points.** Please derive a realization that is both controllable and observable, represent it in both controllable canonical form and observable canonical form

Q) $G(s) = \frac{s-1}{s^3 + 2s^2 - s - 2}$

$$A_{CCF} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -Q_0 & -Q_1 & -Q_2 \end{bmatrix} \quad C_{CCF} = \begin{bmatrix} b_0 - a_0 b_3 & b_1 - a_1 b_3 & b_2 - a_2 b_3 \end{bmatrix}$$

$$A_{OCF} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 2 & 1 & -2 \end{bmatrix} \quad B_{CCF} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \quad C_{CCF} = \begin{bmatrix} -1 & 1 & 0 \end{bmatrix}$$

$$Q = \begin{bmatrix} C \\ CA \\ CA^2 \end{bmatrix} = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 1 \\ 2 & 1 & -3 \end{bmatrix} \quad \text{rank } Q = 2 \quad \text{Not observable}$$

IN Kalman canonical form,

$$\tau^{-1} = \begin{bmatrix} -1 & 1 & 0 \\ 0 & -1 & 1 \\ 0 & 0 & 1 \end{bmatrix} \quad A = \tau^{-1} A \tau = \begin{bmatrix} 0 & 1 & 0 \\ -2 & -3 & 0 \\ -2 & -3 & 1 \end{bmatrix} \quad \text{Not detectable since } \text{Re}(\lambda(A_{22})) = 1 > 0$$

$$B = \tau^{-1} B = \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix} \quad C = C \tau = \begin{bmatrix} 1 & 0 | 0 \end{bmatrix}$$

b)

$$G(s) = \frac{s-1}{s^3 + 2s^2 - s - 2}$$

$$A_{OCF} = \begin{bmatrix} 0 & 0 & -a_0 \\ 1 & 0 & -a_1 \\ 0 & 1 & -a_2 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 2 \\ 1 & 0 & 1 \\ 0 & 1 & -2 \end{bmatrix} \quad C_{OCF} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix}$$

$$B_{OCF} = \begin{bmatrix} b_0 - Q_0 b_3 \\ b_1 - Q_1 b_3 \\ b_2 - Q_2 b_3 \end{bmatrix} = \begin{bmatrix} -1 \\ 1 \\ 0 \end{bmatrix}$$

$$P = \begin{bmatrix} B & AB & AB^2 \end{bmatrix} = \begin{bmatrix} -1 & 0 & 2 \\ 1 & -1 & 1 \\ 0 & 1 & -3 \end{bmatrix} \quad \text{rank } P = 2 \quad \text{Not controllable}$$

In Kalman Canonical form:

$$T = \begin{bmatrix} -1 & 0 & 0 \\ 1 & -1 & 0 \\ 0 & 1 & 1 \end{bmatrix} \quad A = T^{-1}AT = \left[\begin{array}{c|cc} 0 & -2 & -2 \\ \hline 1 & -3 & -3 \\ 0 & 0 & 1 \end{array} \right] \quad \begin{array}{l} \text{Not} \\ \text{Stabilizable} \\ \text{since } \operatorname{Re}(\lambda_1(A_{22})) > 0 \end{array}$$

$$B = T^{-1}B = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} \quad C = CT = \begin{bmatrix} 0 & 1 & 1 \end{bmatrix}$$

C) DUALITY

(A, B) is controllable iff (A^T, B^T) is observable

(A, C) is observable iff (A^T, C^T) is controllable

$$G(s) = \frac{s-1}{s^3+2s^2-s-2} = \frac{s-1}{(s+1)(s+1)(s+2)} = \frac{1}{(s+1)(s+2)} \quad \begin{array}{l} \text{NOT} \\ \text{MINIMAL} \\ \text{representation} \end{array}$$

MINIMAL \Rightarrow irreducible $G(s)$ with n degree of $a(s)$ is MINIMAL iff is both controllable and observable.

d) $G(s) = \frac{1}{s^2 + 3s + 2} \quad \begin{array}{l} Q_0 = 2 \\ Q_1 = 3 \\ Q_2 = 1 \end{array}$

$$A_{CCF} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \quad B_{CCF} = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \quad C_{CCF} = \begin{bmatrix} 1 & 0 \end{bmatrix} \quad \text{controllable}$$

$$A_{OCF} = \begin{bmatrix} 0 & -2 \\ 1 & -3 \end{bmatrix} \quad C_{OCF} = \begin{bmatrix} 0 & 1 \end{bmatrix} \quad B_{OCF} = \begin{bmatrix} 1 \\ 0 \end{bmatrix} \quad \text{observable}$$