

Homework 1

ME5659 Spring 2024

Due: See Canvas, turn in on Gradescope

Problem 1 (6 points)

Describe the dynamical systems in state-space representations.

(a) **3 points.** For the following system described by the given transfer function, derive valid state-space realization (define the state variables and derive the state-space representation):

$$G(s) = \frac{Y(s)}{U(s)} = \frac{4s^4 - s^2 + 5s - 1}{2s^4 + 2s^2 - 4s + 6}$$

(b) **3 points.** Given the following differential equations, derive valid state-space realization with $u = [u_1 \ u_2]^T$ and $y = [y_1 \ y_2]^T$ (define the state variables and derive the state-space representation):

$$\begin{aligned}\ddot{y}_1(t) + 2\dot{y}_1(t) - 5(y_2(t) - y_1(t)) &= u_1(t) \\ \ddot{y}_2(t) + \dot{y}_1(t) - 4\dot{y}_2(t) - 3(y_2(t) - y_1(t)) &= u_2(t)\end{aligned}$$

SOLUTIONS

a) METHOD 1:

Note that the transfer function $G(s)$ is a 4th-order proper transfer function, meaning that the highest orders of the polynomials in both the numerator and the denominator are the same. Due to this property, we cannot directly convert $G(s)$ back into the time domain to find $y(t)$ in terms of $u(t)$, as this would introduce derivatives of $u(t)$ in the terms. One viable approach to deriving the state-space representation is to introduce an intermediate variable, such as $X(s)$. By multiplying $X(s)$ into both the numerator and the denominator of $G(s)$, the equation can be reformulated as:

$$G(s) = \frac{Y(s)}{U(s)} = \frac{Y(s)X(s)}{U(s)X(s)} = \frac{Y(s)}{X(s)} \cdot \frac{X(s)}{U(s)} = \frac{4s^4 - s^2 + 5s - 1}{2s^4 + 2s^2 - 4s + 6}$$

This allows us to rewrite the relationship as follows:

$$Y(s) = (4s^4 - s^2 + 5s - 1)X(s)$$

$$Y(s) = 4s^4X(s) - s^2X(s) + 5sX(s) - X(s)$$

$$U(s) = (2s^4 + 2s^2 - 4s + 6)X(s)$$

$$U(s) = 2s^4X(s) + 2s^2X(s) - 4sX(s) + 6X(s)$$

into the time domain yields:

$$y = 4x^4 - \ddot{x} + 5\dot{x} - x$$

$$u = 2x^4 + 2\ddot{x} - 4\dot{x} + 6x$$

Now, let's define the state variables in vector form as $x = [x_1, x_2, x_3, x_4]^T$, where $x_1 = x$, $x_2 = \dot{x}$, $x_3 = \ddot{x}$, $x_4 = \dddot{x}$. With the defined state variables, the previous equations can be written as follows:

$$\dot{y} = 4\dot{x}_4 - x_3 + 5x_2 - x_1$$

$$\dot{u} = 2\dot{x}_4 + 2x_3 - 4x_2 + 6x_1$$

$$\Rightarrow \dot{x}_4 = \frac{u - 2x_3 + 4x_2 - 6x_1}{2} = -3x_1 + 2x_2 - x_3 + \frac{u}{2}$$

Substituting the latter in the previous one, we obtain

$$y = 2(u - 2x_3 + 4x_2 - 6x_1) - x_3 + 5x_2 - x_1$$

$$y = 2u - 4x_3 + 8x_2 - 12x_1 - x_3 + 5x_2 - x_1$$

$$y = 2u - 5x_3 + 13x_2 - 13x_1$$

So, we can build the state-space representation as:

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -3 & 2 & -1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1/2 \end{bmatrix} u \quad \text{STATE EQUATION}$$

$$y = \begin{bmatrix} -13 & 13 & -5 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + 2u \quad \text{OUTPUT EQUATION}$$

METHOD 2)

Since $G(s)$ is a proper transfer function, one can initially use long division to reduce the order of the numerator. As a result, the transfer function can be reexpressed in the following form:

$$G(s) = \frac{Y(s)}{U(s)} = 2 + \frac{-5s^2 + 13s - 13}{2s^4 + 2s^2 - 4s + 6}$$

↓
D mat 21x

Now, the second transfer function on the right-hand side is a strictly proper transfer function. Therefore, we can directly derive the state-space model by utilizing the controller canonical form. This leads us to the following representation:

$$\dot{x} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -3 & 2 & -1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1/2 \end{bmatrix} u$$

Given the extra scalar term, the output equation in state-space representation will take on the form $y = Cx + 2u$. This can be further expressed as follows:

$$y = \begin{bmatrix} -13 & 13 & -5 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + 2u$$

This yields results consistent with those obtained using the first method

b) $\ddot{y}_1(+) + 2\dot{y}_1(+) - 5(y_2(+) - y_1(+)) = u_1(+)$

$$\ddot{y}_2(+) + \dot{y}_1(+) - 4\dot{y}_2(+) - 3(y_2(+) - y_1(+)) = u_2(+)$$

Given that the system incorporates two inputs, u_1 and u_2 , and the highest-order derivative terms in each equation are \ddot{y}_1 and \ddot{y}_2 respectively. We can then define the state variable as:

$$x = [y_1, \dot{y}_1, y_2, \dot{y}_2]^T$$

$$u = [u_1, u_2]^T \text{ INPUT VECTOR}$$

Therefore, the given equations can be reformulated as:

$$\ddot{y}_1 = u_1 - 2\dot{y}_1 + 5(y_2 - y_1)$$

$$\ddot{y}_2 = u_2 - \dot{y}_1 + 4\dot{y}_2 + 3(y_2 - y_1)$$

STATE EQUATION IN ITS STATE - SPACE REPRESENTATION

$$\dot{x} = \begin{bmatrix} \dot{y}_1 \\ \dot{y}_2 \\ \ddot{y}_1 \\ \ddot{y}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -5 & -2 & 5 & 0 \\ 0 & 0 & 0 & 1 \\ -3 & -1 & 3 & 4 \end{bmatrix} \begin{bmatrix} y_1 \\ \dot{y}_1 \\ y_2 \\ \dot{y}_2 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

OUTPUT EQUATION IN ITS STATE - SPACE REPRESENTATION

$$y = [y_1 \ y_2]^T$$

$$y = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} y_1 \\ \dot{y}_1 \\ y_2 \\ \dot{y}_2 \end{bmatrix}$$

Problem 2 (6 points)

Consider a pendulum as shown in Fig. 1. We assume that the mass m is concentrated at pendulum end, with length as l . Gravity should be considered. The pendulum is driven by a torque input T at the base, and the base rotation joint is subject to rotational damping b . The equation of motion of this pendulum is

$$ml^2\ddot{\theta} + b\dot{\theta} + mgl \sin \theta = T,$$

where T is input and pendulum angle θ is output.

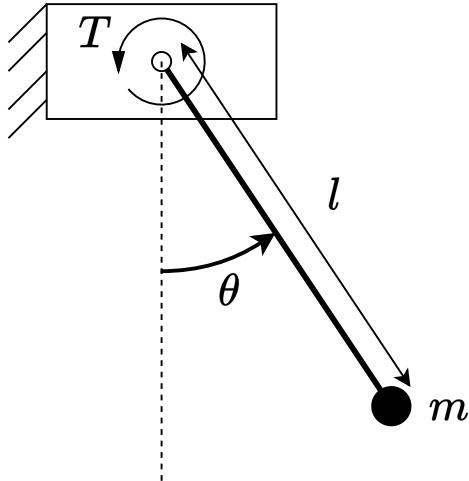


Figure 1: Simple pendulum

(a) **2 points.** Define the 2 state variables of the system. The input is $u = T$. Put the equations of motion in nonlinear state-space form, $\dot{\mathbf{x}} = f(\mathbf{x}, \mathbf{u})$.

(b) **4 points.** Consider the initial angle of pendulum is θ_0 and initial torque $T_0 = 0$ for a passive system. A small torque input δT is at the base joint, leading to perturbation angle of $\delta\theta$. Linearize the nonlinear state-space model about θ_0 and T_0 to obtain the linear state-space models for $\theta_0 = 0$ and $\theta_0 = \pi$.

SOLUTIONS

a) $ml^2\ddot{\theta} + b\dot{\theta} + mgl \sin \theta = T$ θ output, T input $T = u$

The state variable are θ and $\dot{\theta}$. In fact, the mass m storages kinetic energy.

$$\mathbf{x} = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix}; \quad \dot{\mathbf{x}} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} \dot{\theta} \\ \ddot{\theta} \end{bmatrix}$$

Rewriting the equation using the state variables defined above, we obtain:

$$\ddot{\theta} = \frac{T - b\dot{\theta} - mgl \sin \theta}{ml^2}$$

$$\dot{x}_2 = \frac{1}{ml^2} u - \frac{b}{ml^2} x_2 - \frac{mgl}{ml^2} \sin x_1$$

$$\begin{cases} \dot{x}_1 = x_2 \\ \dot{x}_2 = \frac{1}{ml^2} u - \frac{b}{ml^2} x_2 - \frac{g}{l} \sin x_1 \end{cases}$$

$$\dot{\tilde{x}} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} x_2 \\ \frac{1}{me^2} u - \frac{b}{me^2} x_2 - \frac{g}{e} \sin x_1 \end{bmatrix}; \quad \dot{x} = f(x, u)$$

where $\sin x_1$ is the term that makes our system non linear

b) θ_0 and $T_0 = 0$

EQUILIBRIUM POINTS $\theta_0 = 0$ and $\theta_0 = \pi$

$$\tilde{x} = \begin{bmatrix} \tilde{x}_1 \\ \tilde{x}_2 \end{bmatrix}; \quad \dot{\tilde{x}} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \text{ noting changes anymore}$$

$$\dot{\tilde{x}} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} = \begin{bmatrix} \tilde{x}_2 \\ \frac{1}{me^2} \tilde{u} - \frac{b}{me^2} \tilde{x}_2 - \frac{g}{e} \sin \tilde{x}_1 \end{bmatrix}$$

$\tilde{x}_{1,1} = 0$ pendulum hanging directly down

$\tilde{x}_{1,2} = \pi$ pendulum pointing directly up.

LET'S LINEARIZE about $\theta_0 = 0$

$$\tilde{x} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \quad \tilde{u} = 0$$

A matrix is given by the JACOBIAN and so:

$$A = \left. \frac{\partial f}{\partial x} \right|_{\tilde{x}, \tilde{u}} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} \end{bmatrix}_{\tilde{x}, \tilde{u}}$$

$$A = \begin{bmatrix} 0 & 1 \\ -\frac{g}{e} \cos x_1 & -\frac{b}{me^2} \end{bmatrix}_{\tilde{x}, \tilde{u}} \quad \text{where } \tilde{x} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \text{ and } \tilde{u} = 0$$

$$A = \begin{bmatrix} 0 & 1 \\ -\frac{g}{e} & -\frac{b}{me^2} \end{bmatrix}$$

$$B = \left. \frac{\partial f}{\partial u} \right|_{\tilde{x}, \tilde{u}} = \begin{bmatrix} \frac{\partial f_1}{\partial u} \\ \frac{\partial f_2}{\partial u} \end{bmatrix} = \begin{bmatrix} 0 \\ \frac{1}{me^2} \end{bmatrix}$$

$$\delta \dot{x} = \begin{bmatrix} 0 & \frac{1}{m} \\ -\frac{g}{e} & -\frac{b}{me^2} \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ \frac{1}{me^2} \end{bmatrix} \delta u$$

LET'S LINEARIZE about $\theta_0 = \pi$

$$\tilde{x} = \begin{bmatrix} \pi \\ 0 \end{bmatrix} \quad \tilde{u} = 0$$

A matrix is given by the JACOBIAN and so:

$$A = \frac{\partial f}{\partial x} \Big|_{\tilde{x}, \tilde{u}} = \begin{bmatrix} \frac{\partial p_1}{\partial x_1} & \frac{\partial p_1}{\partial x_2} \\ \frac{\partial p_2}{\partial x_1} & \frac{\partial p_2}{\partial x_2} \end{bmatrix} \Big|_{\tilde{x}, \tilde{u}}$$

$$A = \begin{bmatrix} 0 & \frac{1}{m} \\ -\frac{g \cos x_1}{e} & -\frac{b}{me^2} \end{bmatrix} \Big|_{\tilde{x}, \tilde{u}} \quad \text{where } \tilde{x} = \begin{bmatrix} \pi \\ 0 \end{bmatrix} \text{ and } \tilde{u} = 0$$

$$A = \begin{bmatrix} 0 & \frac{1}{m} \\ \frac{g}{e} & -\frac{b}{me^2} \end{bmatrix}$$

$$B = \frac{\partial f}{\partial u} \Big|_{\tilde{x}, \tilde{u}} = \begin{bmatrix} \frac{\partial p_1}{\partial u} \\ \frac{\partial p_2}{\partial u} \end{bmatrix} \Big|_{\tilde{x}, \tilde{u}} = \begin{bmatrix} 0 \\ \frac{1}{me^2} \end{bmatrix}$$

$$\delta \dot{x} = \begin{bmatrix} 0 & \frac{1}{m} \\ \frac{g}{e} & -\frac{b}{me^2} \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ \frac{1}{me^2} \end{bmatrix} \delta u$$

Problem 3 (6 points)

A single-wheel chair cart (unicycle) moving on the plane with linear velocity v and angular velocity ω can be modeled by the nonlinear system

$$\dot{p}_x = v \cos \theta, \quad \dot{p}_y = v \sin \theta, \quad \dot{\theta} = \omega,$$

where (p_x, p_y) denote the Cartesian coordinates of the wheel and θ its orientation. Regard this as a system with input $u = [v \ \omega]^T$

(a) (3 points) Construct a state-space model for this system with state

$$x = \begin{bmatrix} x_1(t) \\ x_2(t) \\ x_3(t) \end{bmatrix} = \begin{bmatrix} p_x \cos \theta + (p_y - 1) \sin \theta \\ -p_x \sin \theta + (p_y - 1) \cos \theta \\ \theta \end{bmatrix}$$

and output $y = [x_1 \ x_2]^T$.

(b) (3 points) Compute a linearization for this system around the equilibrium point $x_{eq} = 0, u_{eq} = 0$.

a)
$$(*) \begin{cases} \dot{x}_1 = \dot{p}_x \cos \theta - p_x \sin \theta \dot{\theta} + \dot{p}_y \sin \theta + p_y \cos \theta \dot{\theta} - \cos \theta \dot{\theta} \\ \dot{x}_2 = -\dot{p}_x \sin \theta - p_x \cos \theta \dot{\theta} + \dot{p}_y \cos \theta - p_y \sin \theta \dot{\theta} + \sin \theta \dot{\theta} \\ \dot{x}_3 = \dot{\theta} \end{cases}$$

Substituting $(**)$ into $(*)$

$$\begin{aligned} \dot{x}_1 &= v \cos^2 \theta - p_x \sin \theta \dot{\theta} + v \sin^2 \theta + p_y \cos \theta \dot{\theta} - \cos \theta \dot{\theta} \\ \dot{x}_2 &= v - p_x \sin \theta \dot{\theta} + p_y \cos \theta \dot{\theta} - \cos \theta \dot{\theta} \\ \dot{x}_3 &= v + (-p_x \sin \theta + p_y \cos \theta - \cos \theta) \dot{\theta} = v + (-p_x \sin \theta + (p_y - 1) \cos \theta) \dot{\theta} \end{aligned}$$

$$(**) \begin{cases} \dot{p}_x = v \cos \theta \\ \dot{p}_y = v \sin \theta \\ \dot{\theta} = \omega \end{cases}$$

$$\begin{aligned} \dot{x}_1 &= v + x_2 \dot{\theta} \\ \dot{x}_2 &= -v \cos \theta \sin \theta + v \sin \theta \cos \theta + (-p_x \cos \theta - p_y \sin \theta + \sin \theta) \dot{\theta} \\ \dot{x}_2 &= (-p_x \cos \theta - (p_y - 1) \sin \theta) \dot{\theta} \\ \dot{x}_2 &= -x_1 \dot{\theta} \quad \dot{x}_3 = \omega \\ \dot{x}_3 &= \omega \end{aligned}$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} v + \omega x_2 \\ -\omega x_1 \\ \omega \end{bmatrix} \quad \text{NON LINEAR}$$

$$y = [x_1 \ x_2]^T$$

b)

$$A = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \frac{\partial f_1}{\partial x_3} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} & \frac{\partial f_2}{\partial x_3} \\ \frac{\partial f_3}{\partial x_1} & \frac{\partial f_3}{\partial x_2} & \frac{\partial f_3}{\partial x_3} \end{bmatrix}$$

$$A = \begin{bmatrix} 0 & w & 0 \\ -w & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad \left| \begin{array}{l} v_{ep}=0 \\ x_{ep}=0 \end{array} \right. = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$$B = \begin{bmatrix} \frac{\partial f_1}{\partial v} & \frac{\partial f_1}{\partial w} \\ \frac{\partial f_2}{\partial v} & \frac{\partial f_2}{\partial w} \\ \frac{\partial f_3}{\partial v} & \frac{\partial f_3}{\partial w} \end{bmatrix} = \begin{bmatrix} 1 & x_2 \\ 0 & -x_1 \\ 0 & 1 \end{bmatrix} \quad \left| \begin{array}{l} x_{ep}=0 \\ v_{ep}=0 \end{array} \right. = \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix}$$

$$C = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \quad D = 0$$

Problem 4 (7 points)

Consider the following model for a DC motor:

$$J \frac{d^2\theta}{dt^2} + b \frac{d\theta}{dt} = K_t i \quad (1)$$

$$L \frac{di}{dt} + Ri + K_b \frac{d\theta}{dt} = V_s, \quad (2)$$

where J is the mass-moment of inertia of the load on the motor, which is damped by rotary damper with linear damping constant b . The torque delivered by the motor is $K_t i$, where K_t is the motor torque constant and i is the motor current. The motor has internal series resistance R and inductance L , and a motor speed constant K_b . The voltage supplied to the motor is V_s . In all parts, consider the input $u = V_s$ and the output $y = \theta$, motor shaft angle.

(a) (2 points) In a coupled system it may not be clear at first what the order of the system is. In this problem, we have, effectively, a first-order system in i and a second-order system in θ , giving us three states. Making the choice for states $x_1 = \theta$, $x_2 = \dot{\theta}$, and $x_3 = i$, calculate the \mathbf{A} and \mathbf{B} matrices for a state-space representation.

(b) (2 point) To better illustrate that the system is third-order, find a single third-order differential equation in terms of θ and its derivatives (the current will not appear in the equation). Laplace transforming the ODEs for manipulation, or using a differential operator will make this easier.

(c) (2 points) State-space representations of dynamical systems are *not unique*. Making the choice for states $x_1 = \theta$, $x_2 = \dot{\theta}$, and $x_3 = \ddot{\theta}$, calculate the \mathbf{A} and \mathbf{B} matrices for a state-space representation.

(d) (1 point) While state-space representations of systems are not unique, *they represent the same systems* if originating from the same set of differential equations. Assume that all constant parameters are equal to 1, and use MATLAB to calculate the eigenvalues of both \mathbf{A} -matrices (from parts (a) and (c)), and show that they are the same. List your MATLAB code, and the program/command outputs.

a) $x = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = \begin{bmatrix} \theta \\ \dot{\theta} \\ \ddot{\theta} \end{bmatrix} \quad u = V_s$

FIRST OF ALL, LET'S REWRITE THE MATRICES IN THE FOLLOWING WAY:

$$J \ddot{\theta} + b \dot{\theta} = K_t i$$

$$L \dot{i} + R i + K_b \dot{\theta} = u$$

$$\Rightarrow \begin{cases} \ddot{\theta} = \frac{K_t}{J} i - \frac{b}{J} \dot{\theta} \\ \dot{i} = \frac{1}{L} u - \frac{R}{L} i - \frac{K_b}{L} \dot{\theta} \end{cases}$$

$$\dot{x} = \begin{bmatrix} \dot{\theta} \\ \ddot{\theta} \\ \ddot{i} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & -\frac{b}{J} & \frac{kt}{J} \\ 0 & -\frac{kb}{L} & -\frac{R}{L} \end{bmatrix} \begin{bmatrix} \theta \\ \dot{\theta} \\ i \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{L} \end{bmatrix} u$$

b) LAPLACE TRANSFORMING THE ODES

1ST EQUATION BECOMES:

$$(Js^2 + bs)\Theta(s) = kt I(s)$$

2ND EQUATION BECOMES:

$$(ls + R)I(s) + kb s \Theta(s) = \frac{Vs}{s}$$

FROM THE FIRST WE GET:

$$I(s) = \frac{(Js^2 + bs)\Theta(s)}{kt}$$

SUBSTITUTING THE LATTER IN THE 1ST, WE OBTAIN:

$$(ls + R) \left[\frac{(Js^2 + bs)\Theta(s)}{kt} \right] + kb s \Theta(s) = \frac{Vs}{s}$$

$$\left[\frac{Jls^3 + Lbs^2 + Rjs^2 + Rbs}{kt} + kb s \right] \Theta(s) = \frac{Vs}{s}$$

$$\left(Jls^3 + Lbs^2 + Rjs^2 + Rbs + kbkt s \right) \Theta(s) = kt \frac{Vs}{s}$$

BACK IN TIME:

$$Jl\ddot{\theta} + Lb\ddot{\theta} + Rj\ddot{\theta} + Rb\dot{\theta} + kbkt\dot{\theta} = kt \frac{Vs}{s} \quad u = Vs$$

SINGLE THIRD-ORDER DIFFERENTIAL EQUATION IN TERMS OF THETA AND ITS DERIVATIVES

c)

$$\begin{aligned} x_1 &= \theta \\ x_2 &= \dot{\theta} \\ x_3 &= \ddot{\theta} \end{aligned}$$

FROM b we can rewrite the last equation as follows:

$$\ddot{\Theta} = \frac{Kt}{JL} v_s - \frac{Lb}{JL} \ddot{\Theta} - \frac{RJ}{JL} \dot{\Theta} - \frac{Rb}{JL} \dot{\Theta} - \frac{kbkt}{JL} \dot{\Theta}$$

$$v_s = u \quad x_1 = \theta \quad x_2 = \dot{\theta} \quad x_3 = \ddot{\theta} \quad \dot{x}_3 = \dddot{\theta}$$

$$\Rightarrow \dot{x}_3 = \frac{Kt}{JL} u - \frac{b}{J} x_3 - \frac{R}{L} x_3 - \frac{Rb}{JL} x_2 - \frac{kbkt}{JL} x_2$$

$$\dot{x}_3 = \frac{Kt}{JL} u + \left(-\frac{b}{J} - \frac{R}{L} \right) x_3 + \left(-\frac{Rb}{JL} - \frac{kbkt}{JL} \right) x_2$$

$$\ddot{X} = Ax + Bu$$

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & -\frac{Rb}{JL} - \frac{kbkt}{JL} & -\frac{b}{J} - \frac{R}{L} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{Kt}{JL} \end{bmatrix} u$$

d)

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untitled *

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1 close all; clc; clear all;
2 % System 1
3 A1=[0 1 0; 0 -1 1; 0 -1 -1];
4 eig(A1)
5 % System 2
6 A2=[0 1 0; 0 0 1; 0 -2 -2];
7 eig(A2)

```

Command Window

```

ans =
0.0000 + 0.0000i
-1.0000 + 1.0000i
-1.0000 - 1.0000i

ans =
0.0000 + 0.0000i
-1.0000 + 1.0000i
-1.0000 - 1.0000i

```

SOME EIGENVALUES