Community Christmas Competition IV (2020) - Simulation of the backward facing step

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Background

Background

- Turbulent shear layer is important flow structure in industrial combustors
- Backward facing step combustor is a simple experimental configuration incorporating the aspects of premixed combustion stabilized in a turbulent shear layer

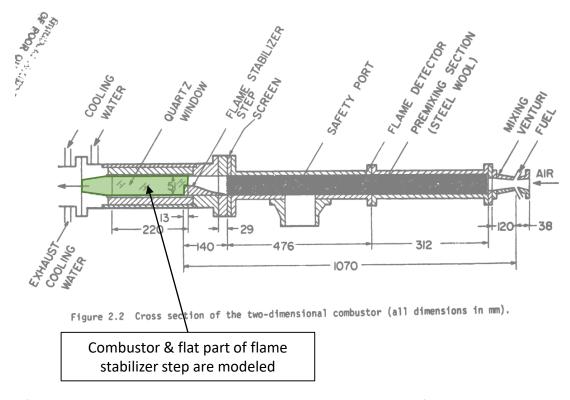
Objectives

 Compare length of re-circulation zone for non-reacting flow behind the step between RANS and LES turbulence models

For RANS, k-epsilon model is used and inlet step assuming turbulent boundary layer

For LES, k-Eqn model is used and boundary layer (y+ \sim 3) is resolved at inlet step

Model Extent



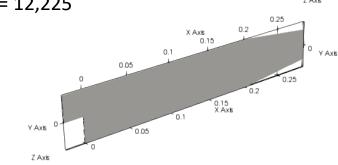
Reference: Robert W. Pitz and John W. Daily, 1981. An experimental study of combustion: The turbulent structure of a reacting shear layer formed at a rearward-facing step. NASA Contractor Report 165427.

Parameters & Mesh

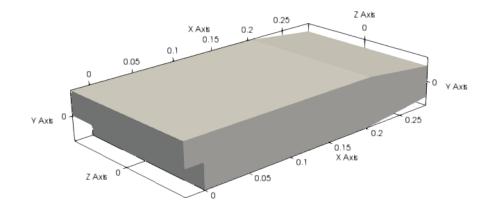
- Inlet conditions (Section 3.1 of Pitz and Daily, 1981)
- U at inlet = 9.1m/s, 13.3m/s and 22.2m/s
- Turbulence intensity at inlet = 2%
- Initial conditions are important as Reynolds number is well below 2x10⁶ (p.58 of Pitz and Daily, 1981)
- Condition of boundary layer for all three flow velocities/Reynolds numbers is best described as "transitional" (p.60 of Pitz and Daily, 1981)
- Outlet boundary conditions
- Pressure = $0m^2/s^2$
- Solver: pimpleFoam

• Mesh (2D) for RANS model





- Mesh (3D) for LES model
- No of cells = 7,335,000

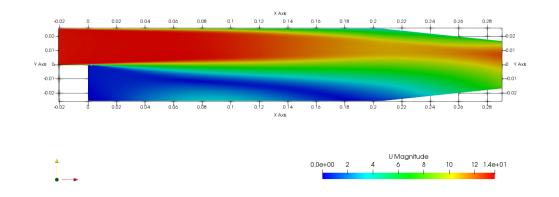


Simulation 1 – Using k-epsilon RANS 2D model

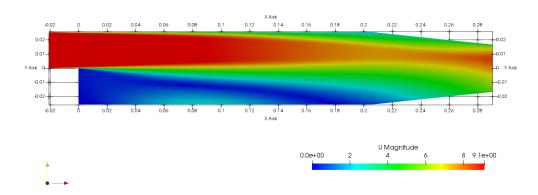
• Length of re-circulation zone

Model	U (m/s)	X _R /H (Simulated)	X _R /H (Pitz and Daily, 1981)
1a	9.1	7.0	6.5
1b	13.3	7.2	7.0
1c	22.2	7.2	6.8

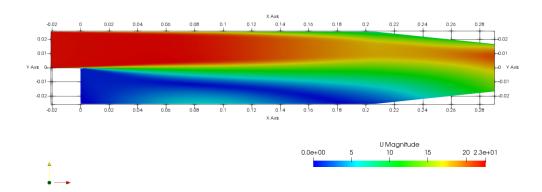
• Model 1b



• Model 1a



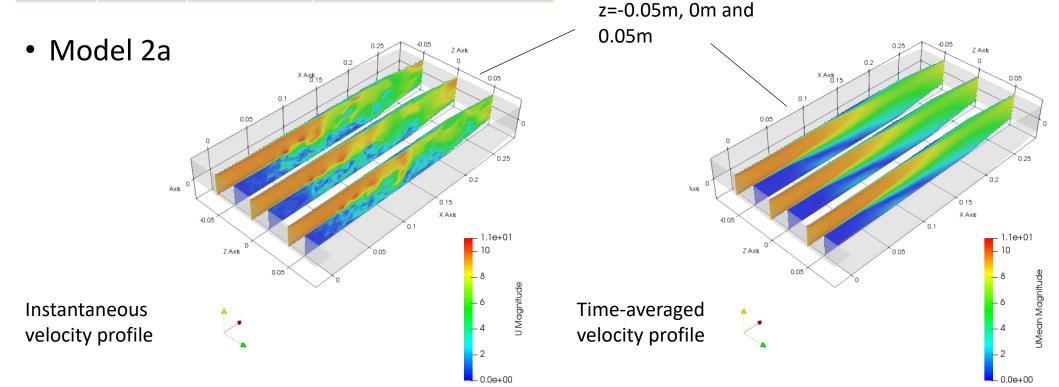
• Model 1c



Simulation 2 – Using k-Eqn LES 3D model

• Length of re-circulation zone

Model	U (m/s)	X _R /H (Simulated)	X _R /H (Pitz and Daily, 1981)
2a	9.1	6.0	6.5



3 slices are cut at

Conclusions

- As length of re-circulation zone is defined in average sense, RANS model gives reasonable estimates
- 2D RANS model is computationally very efficient
- Assuming turbulent boundary layer at inlet step by RANS model appears acceptable, may be because inlet boundary layer is actually "transitional"
- LES model, by time averaging to obtain length of re-circulation zone, gives no advantage over RANS model
- LES model gives insight into turbulent shear layer and mixing process, which is good to understand the process