## **Automatic Control Systems**

**NYQUIST-II** 

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#### Reading:

- chapter 8 (Section 8.2)
- chapter 9 (Section 9.5)

#### Practice problems

- Study table 9.6 on pages 704-711
- Solve problems at the end of chapter 9

## The Nyquist Stability Criterion

#### Case 1

L(S) = G(s)H(s) has no open-loop poles in the RHP, (P = 0)

"A feedback system is stable <u>if and only if</u> the contour  $\Gamma_L$  in the G(s)H(s)-plane <u>DOES NOT</u> encircle the (-1,0) point when the number of poles of L(s) in the RHP is zero (P=0)"

$$Z = N + P$$

$$\therefore Z = 0$$

$$\Rightarrow N = 0$$

• For no zeros of the characteristic equation on the RHP, then there should be <u>no encirclement</u> of the point -1 in the L(s) plane

## The Nyquist Stability Criterion

# Case 2 G(s)H(s) has poles in the RHP, $(P \neq 0)$

"A feedback system is stable <u>if and only if</u>, for the contour  $\Gamma_L$ , the number of <u>counterclockwise</u> encirclements of the (-1,0) point equals to the number of poles of G(s)H(s) in the RHP of the splane"

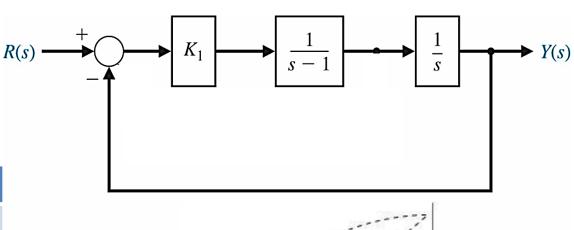
• For no zeros of the characteristic equation on the RHP, we must have P counterclockwise encirclement of the -1 point in the G(s)H(s) plane

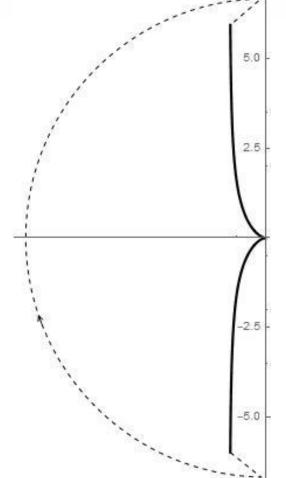
example

$$G(s) = K_1/s(s-1)$$

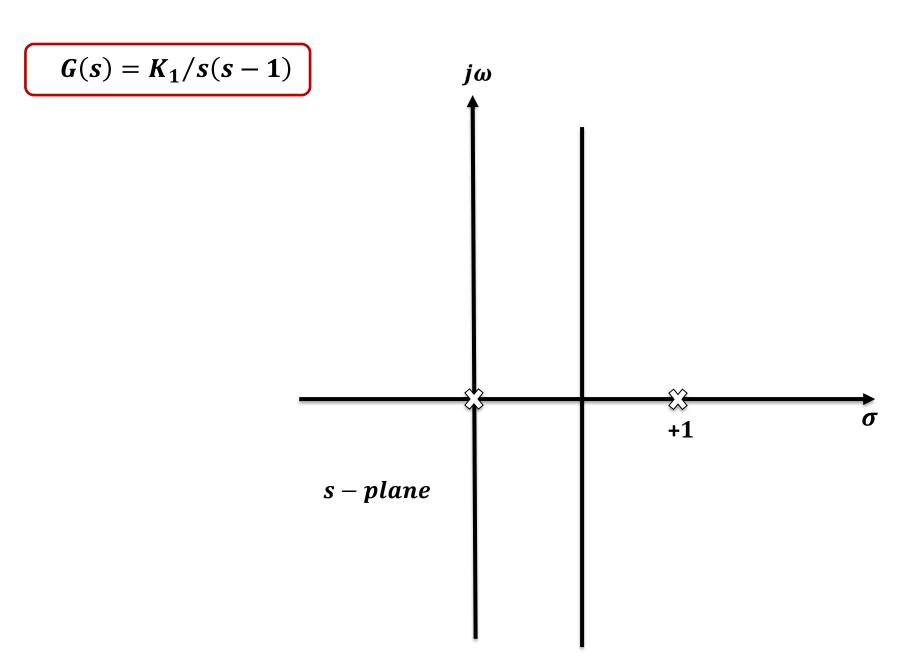
ω	$GH(j\omega)$
$j\omega \to 0_+$	$\infty$ $\angle$ $-$ 270 $^{0}$
$j\omega \to +\infty$	$0 \angle - 180^{0}$

- $\rightarrow$  This is CASE-II
- $P = 1, N = +1 \Rightarrow Z = 2$
- The <u>system is unstable</u> because there are **two roots** in the **RHS** of the s-plane regardless of the value of  $K_1$

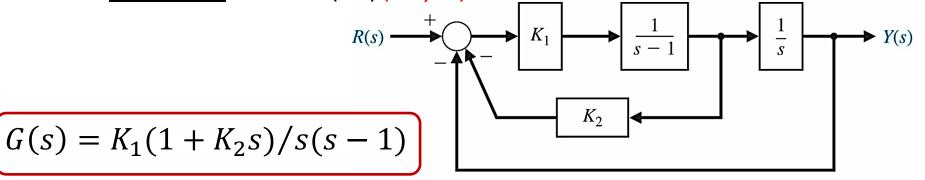




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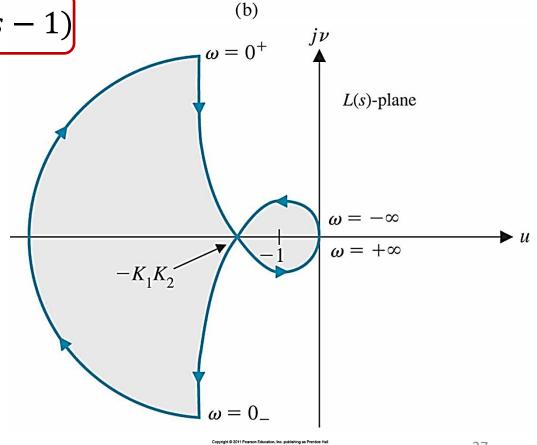
Add a <u>derivative</u> feedback (PD) (why??)



 $G(s) = K_1 K_2 (s + 1/K_2)/s(s - 1)$ 

ω	$GH(j\omega)$
$j\omega \to 0_+$	$\infty$ $\angle$ $-$ 270 $^{0}$
$j\omega \to +\infty$	$0 \angle - 90^{0}$

- P = 1 (case II)
- Condition for stability:  $N = -1 \Rightarrow K = K_1 K_2 > 1$

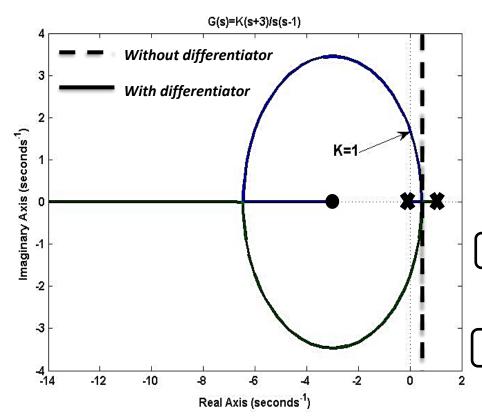


• 
$$G(s) = K_1 K_2 (s + 1/K_2)/s(s - 1)$$

• Let:  $1/K_2 = 3$ ,  $K_1K_2 = K$ 

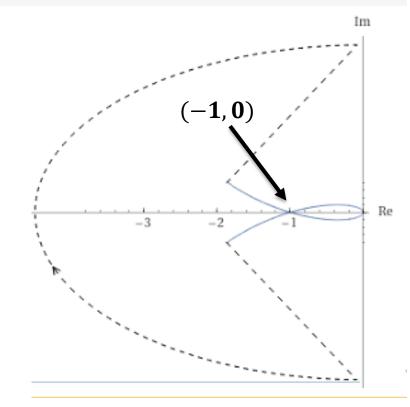
$$G(s)H(s) = \frac{K(s+3)}{s(s-1)}$$

verify using root locus



Nyquist plot transfer function  $\left(\frac{s+3}{s(s-1)}\right)$ 

Nyquist plot

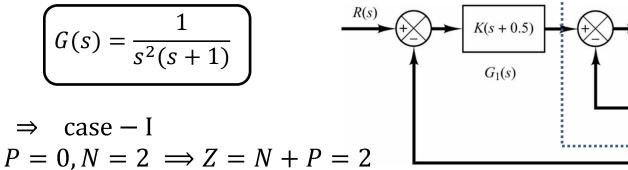


b=0.5\*[1 3]; a=[1 -1 0]; nyquist(b,a)

b=[1 3]; a=[1 -1 0]; rlocus(b,a)

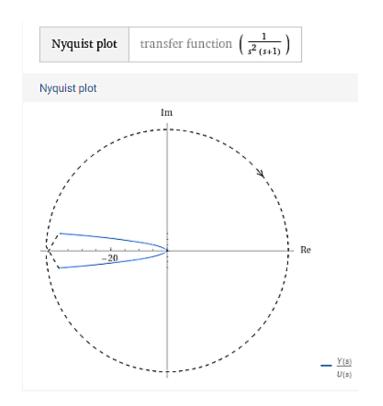
- **example**: determine the range of *K* for stability?
- *Inner-loop* stability

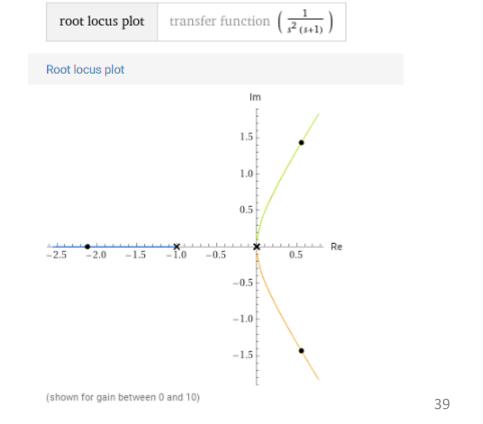
$$G(s) = \frac{1}{s^2(s+1)}$$



Inner-loop is <u>unstable</u>

case — I



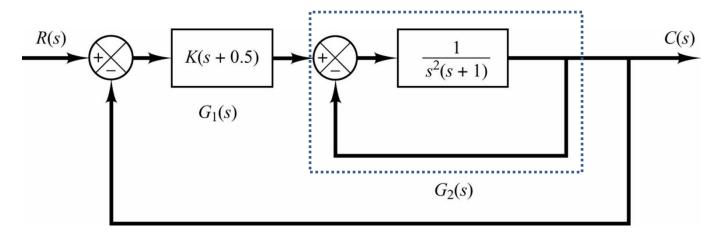


 $s^2(s+1)$ 

 $G_2(s)$ 

C(s)

After adding a differentiator



The open-loop system with PD

$$G(s) = G(s)_1 G_2(s) = \frac{K(s+0.5)}{s^3 + s^2 + 1}$$

- P=2
- For stability of the closed-loop system,

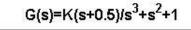
• 
$$\Rightarrow Z = 0 \Rightarrow N = Z - P = -2$$

(two **counterclockwise** encirclement of the critical point -1)

• Open-loop poles, P = 2 (case II)

C(s) - C(s) C(s) -	K(s+0.5)
$G(s) = G(s)_1 G_2(s) =$	$\frac{1}{s^3 + s^2 + 1}$

ω	$G(j\omega)$
$j\omega \rightarrow 0_+$	0.5 <i>K</i>
$j\omega \to +\infty$	$0 \angle - 180^{0}$

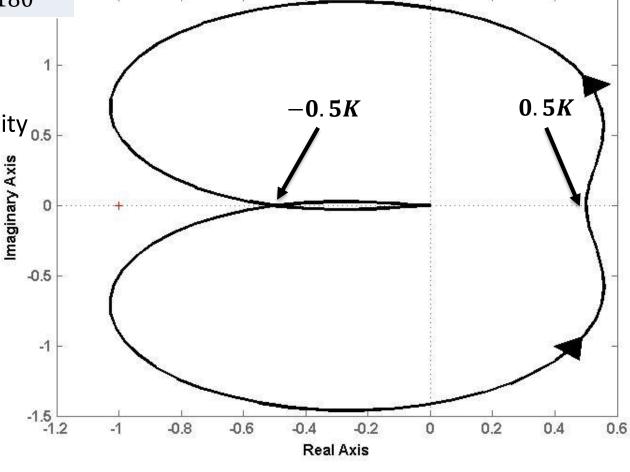


Condition for stability<sub>0.5</sub>

$$Z = 0 \Longrightarrow N = -2$$

$$\Rightarrow 0.5K > 1$$

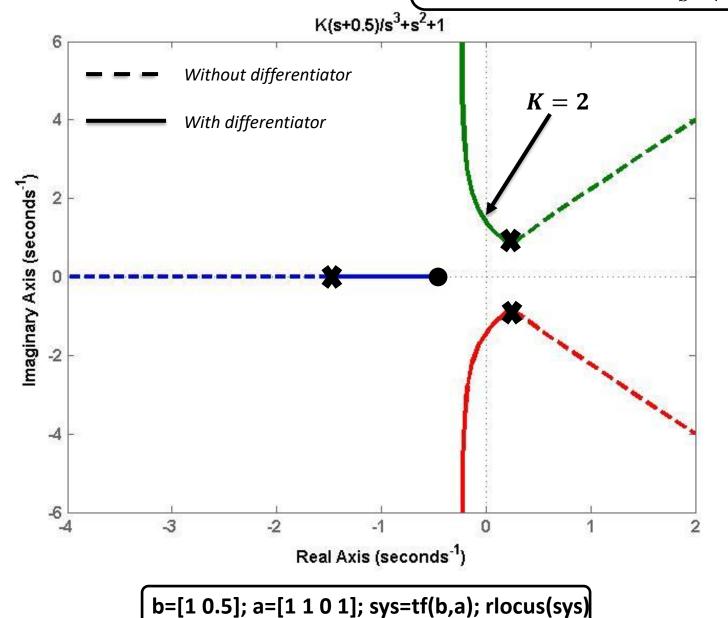
$$\Rightarrow K > 2$$



b=[1 0.5]; a=[1 1 0 1]; sys=tf(b,a); nyquist(sys)

verify using root locus

$$G(s) = G(s)_1 G_2(s) = \frac{K(s+0.5)}{s^3 + s^2 + 1}$$

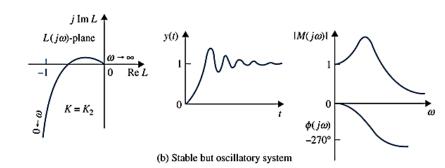


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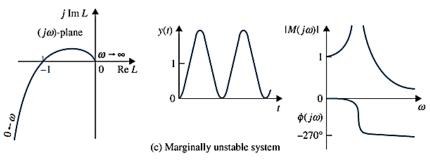
## Relative stability and the Nyquist criterion Pulled Plane

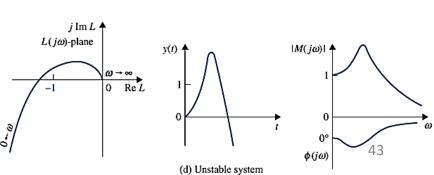
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• The proximity of the  $L(j\omega)$  - locus to the (-1,j0) critical point is a measure of the relative stability of the system.



- Case I
- Case II





## **Gain Margin (GM)**

Gain margin is the amount of gain in decibels (dB) that is allowed to be <a href="INCREASED/DECREASED">IN the loop before the closed-loop system reaches instability</a>

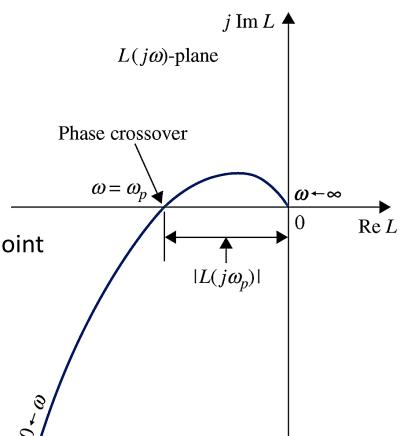
#### **PHASE-CROSSOVER POINT:**

the point at which the Nyquist plot  $L(j\omega)$  intersects the negative real axis

#### PHASE-CROSSOVER FREQUENCY

is the frequency  $\omega_p$  at the phase-crossover point  $\angle L(j\omega) = 180^o$ 

$$G.M. = 20\log_{10}\frac{1}{|L(j\omega_p)|}$$
 dB



• A rule of thumb G.M > 6dB

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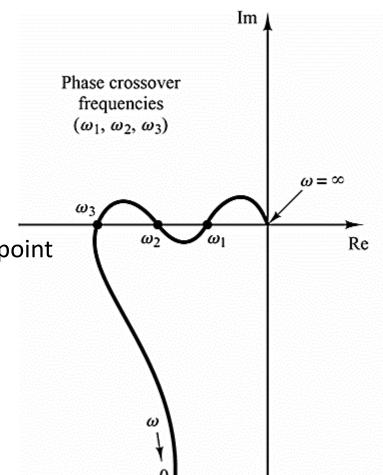
the point at which the Nyquist plot  $L(j\omega)$  intersects the negative real axis

#### PHASE-CROSSOVER FREQUENCY

is the frequency  $\omega_p$  at the phase-crossover point

$$\angle L(j\omega) = 180^{o}$$

$$G.M. = 20\log_{10} \frac{1}{|L(j\omega_p)|}$$
 dB



• A rule of thumb G.M > 6dB

• <u>example</u>

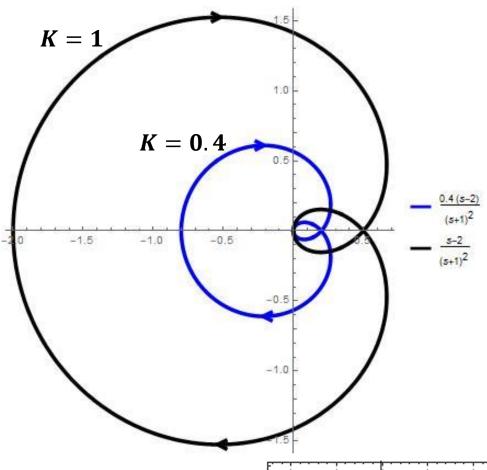
$$GH(s) = K(s-2)/(s+1)^2$$

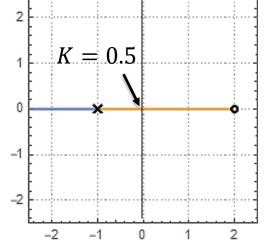
• P = 0 (case-1)

ω	$GH(j\omega)$
$j\omega \rightarrow 0_+$	-2K
$\omega = \sqrt{5}$	<i>K</i> /2
$j\omega \to +\infty$	$0 \angle - 90^{0}$

• 
$$G.M. = 20log_{10}(1/2K)$$

K	N	Z	stability
0 < K < 1/2	0	0	stable
K=1/2			Marginally stable
K > 1/2	1	1	unstable



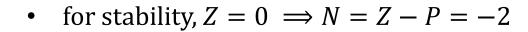


#### • Example

$$GH(s) = K(s+2)/(s-1)^2$$

• P = 2 (case-II)

ω	$GH(j\omega)$
$j\omega \rightarrow 0_+$	2 <i>K</i>
$\omega = \sqrt{5}$	-K/2
$j\omega \to +\infty$	$0 \angle - 90^{0}$

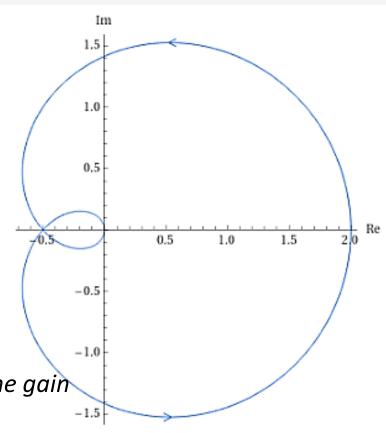


• G.M. Is defined <u>in this case</u> by how much the gain Is <u>decreased</u> to reach instability

K	N	Z	stability
0 < K < 2	0	2	UNSTABLE
K = 2			Marginally stable
K > 2	-2	0	STABLE

Nyquist plot	transfer function	$\left(\frac{s+2}{(s-1)^2}\right)$
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Nyquist plot



• Example

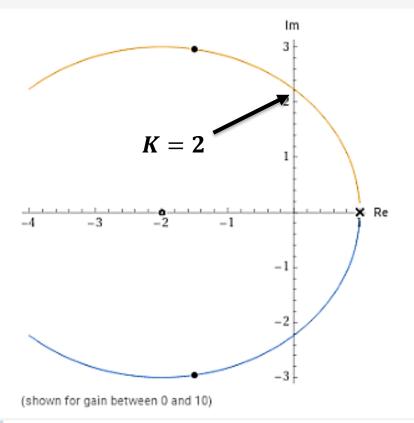
$$GH(s) = K(s+2)/(s-1)^2$$

- P = 2 (case-II)
- G.M. Is defined <u>in this case</u> by how much the gain is <u>decreased</u> to reach <u>instability</u>

K	N	Z	stability
0 < K < 2	0	2	UNSTABLE
K = 2			Marginally stable
K > 2	-2	0	STABLE

root locus plot transfer function  $\left(\frac{s+2}{(s-1)^2}\right)$ 

Root locus plot



## **Phase Margin (PM)**

Phase margin is defined as the angle in degrees in which the  $L(j\omega)$  plot must be rotated about the origin in order that the gain-crossover point on the locus passes through the (-1,j0) point

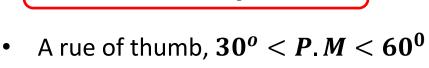
#### **GAIN-CROSSOVER POINT:**

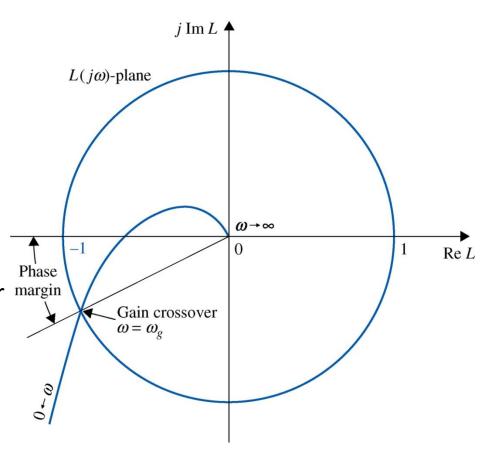
The point on the  $L(j\omega)$  plot at which  $|L(j\omega)|=1$ 

#### **GAIN-CROSSOVER FREQUENCY:**

The gain-crossover frequency  $\omega_g$  is the frequency of  $L(j\omega)$  at the gain-crossover point, i.e.,  $\left|L(j\omega_g)\right|=1$ 

$$P.M. = \angle L(j\omega_g) - 180^\circ$$





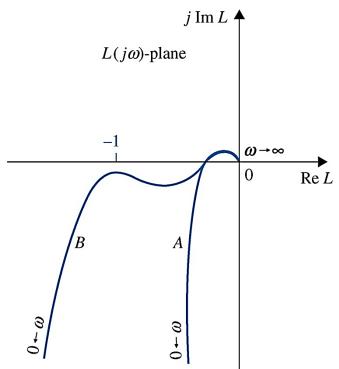
#### **Phase Margin (PM)**

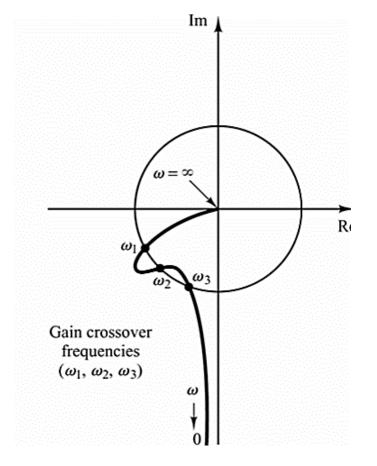
Phase margin is defined as the angle in degrees in which the  $L(j\omega)$  plot must be rotated about the origin in order that the gain-crossover point on the locus passes through the (-1,j0) point

Iocus A and Iocus B have the same

Gain margin but have different relative stability

Because they have different phase margins





<u>example</u>

$$L(s) = \frac{2500}{s(s+50)(s+5)}$$

**Phase-crossover:** 

$$\operatorname{Im}(L(j\omega)) = 0$$

$$\Rightarrow L(j\omega_p) = -0.182$$
,  $\omega_p = 15.88$  rad/s

Phase margin

 $=31.72^{\circ}$ 

**Gain margin** 

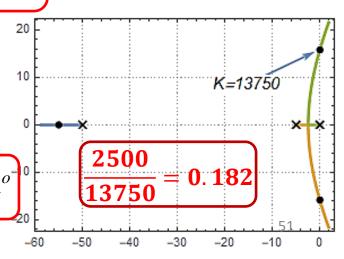
$$G.M. = 20\log_{10}\frac{1}{|L(j\omega_p)|} = 20\log_{10}\frac{1}{0.182} = 14.82 \text{ dB}$$

Gain-crossover:

$$|L(j\omega)| = 1 \implies \omega_g = 6.22 \text{ rad/s}$$

Phase margin

$$P.M. = \angle L(j\omega_g) - 180^\circ = 211.72^\circ - 180^\circ = 31.72^\circ$$



 $j \operatorname{Im} L \spadesuit$ 

 $\omega = \infty$ 

-0.182 0

 $\omega_p = 15.88 \text{ rad/sec}$ 

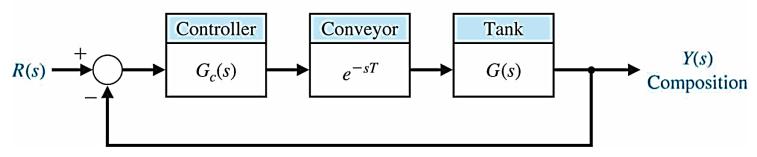
 $\omega_{o} = 6.22 \text{ rad/sec}$ 

 $L(j\omega)$ -plane

Re L

## The stability of control systems with time delays

- Usually, we cannot eliminate the effect of delay in real systems
- The Routh-Hurwitz criterion is **not applicable** for systems with time delays because **the characteristic equation is no longer algebraic**
- a pure time delay, T , is modeled as  $G_d(s) = e^{-sT}$
- the loop transfer function L(s), becomes  $L(s) = G_c(s)G(s)e^{-sT}$



## The stability of control systems with time delays

• A pure time delay, T, has no effect on the gain of  $L(j\omega)$ 

$$|L(j\omega)| = |G_c(j\omega)G(j\omega)|$$

• A pure delay, T, reduces the phase of  $L(j\omega)$ 

$$\angle L(j\omega) = \angle G_c(j\omega)G(j\omega) - \omega T$$

- Time delay <u>reduces</u> the stability of the system
- Reduces Phase Margin

