



# Mechanics of Materials I: Fundamentals of Stress & Strain and Axial Loading

Dr. Wayne Whiteman Senior Academic Professional and Director of the Office of Student Services Woodruff School of Mechanical Engineering

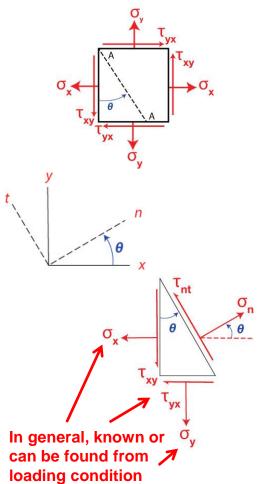




## **Module 17 Learning Outcomes**

 Derive equations for stresses on inclined planes for the case of plane stress in general





#### **Sign Convention**

#### **Normal Stress**

- (+) Tension
- -) Compression

#### **Shear Stress**

Shear stress is (+ ) if it is in the (+ ) direction of the coordinate axis

#### Angle

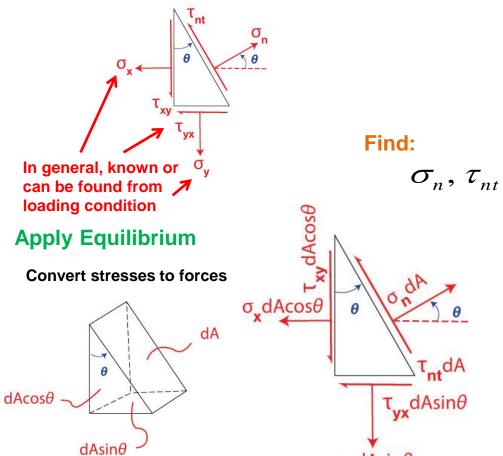
Counterclockwise angle  $\theta$  is (+ ) as measured from positive x axis as reference

#### Find:

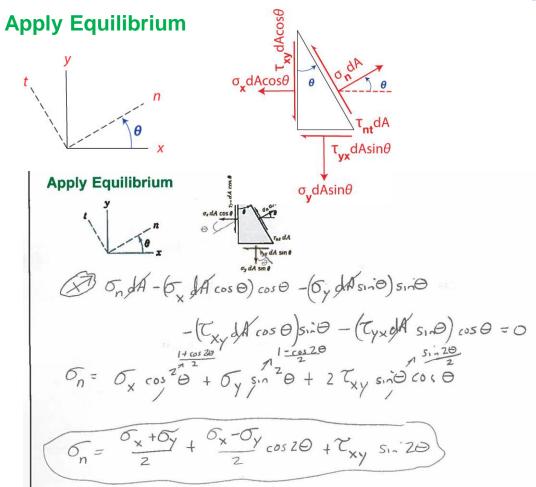
$$\sigma_n, \tau_{nt}$$

 $\sigma_{\mathbf{v}} dA sin\theta$ 



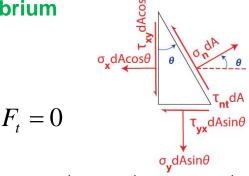








### **Apply Equilibrium**



$$+ \sum F_t = 0$$

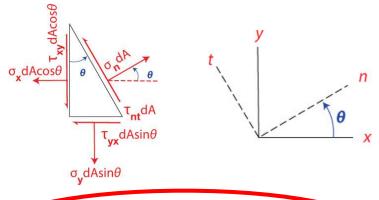
$$\tau_{nt} dA - (\tau_{xy} dA \cos \theta) \cos \theta + (\tau_{yx} dA \sin \theta) \sin \theta + (\sigma_x dA \cos \theta) \sin \theta - (\sigma_y dA \sin \theta) \cos \theta = 0$$

$$\left(\frac{\sin 2\theta}{2}\right) \quad \left(\frac{1 + \cos 2\theta}{2}\right) \left(\frac{1 - \cos 2\theta}{2}\right)$$

$$\tau_{nt} = -(\sigma_x - \sigma_y) \sin \theta \cos \theta + \tau_{xy} \left(\cos^2 \theta - \sin^2 \theta\right)$$

$$\tau_{nt} = -\left(\frac{\sigma_x - \sigma_y}{2}\right) \sin 2\theta + \tau_{xy} \cos 2\theta$$





$$\sigma_n = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta$$

$$\tau_{nt} = -\left(\frac{\sigma_x - \sigma_y}{2}\right) \sin 2\theta + \tau_{xy} \cos 2\theta$$

**Transformation equations for plane stress** 

For any plane at an angle  $\theta$ , We can find  $\sigma_n$  and  $\tau_{nt}$ 

Say transform the stress components from one set of axes to another

They were derived solely from equilibrium, therefore they are applicable to stresses for any material, whether linear or nonlinear, elastic or inelastic

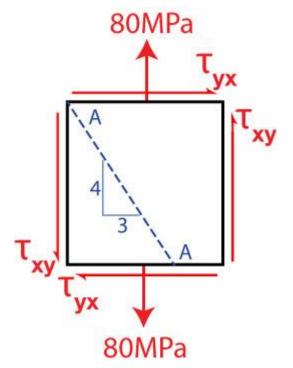
#### Worksheet:

Georgia Tech

The stresses on the horizontal and vertical planes at a point are shown in the figure below. The normal stress on a plane A-A at this point is found to be 80 MPa in tension.

#### Find:

- a) The magnitude and direction of the shear stresses,  $au_{xy} = au_{yx}$
- b) The magnitude and direction of the shear stress on the inclined plane A-A



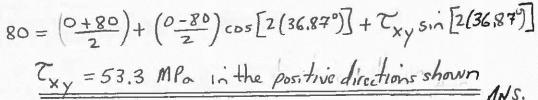
**Worksheet:** The stresses on the horizontal and vertical planes at a point are shown in the figure to the right. The normal stress on a plane A-A at this point is found to be 80 MPa in tension. Find:

a) The magnitude and direction of the shear stresses,  $\tau_{xy} = \tau_{yx}$ b) The magnitude and direction of the shear stress on the inclined plane A-A

b) The magnitude and direction of the shear stress on the inclined plane FOR THIS EXAMPLE 
$$\sigma_{x} = 0$$
  $\sigma_{y} = 80 \text{ M/p}$ 

$$\Theta = \tan^{-1}\left(\frac{3}{4}\right) = 36.87^{\circ}$$
Pa

$$\sigma_n = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \frac{\sigma_x}{2} \sin 2\theta$$



$$T_{n+} = -\left(\frac{\sigma_{x} - \sigma_{y}}{\sigma_{x}}\right) \sin 2\theta + T_{xy} \cos 2\theta$$

$$T_{nt} = -\left(\frac{0-80}{2}\right) \sin\left[2(36.87^{\circ})\right] + 53.3 \cos\left[2(36.87^{\circ})\right]$$
  
 $T_{nt} = 53.3 \, \text{MPa} \, \text{in the positive direction as shown}$ 

