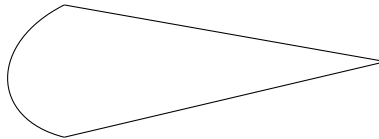
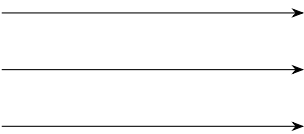


# 2013-XE-27-39

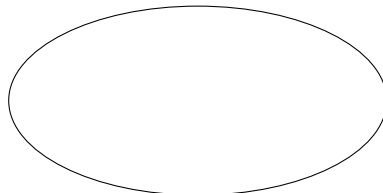
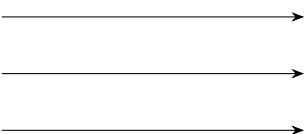
EE24BTECH11003 - Akshara Sarma Chennubhatla

- 1) Flow past a circular cylinder can be produced by superposition of the following elementary potential flows:
  - a) Uniform flow, doublet
  - b) Uniform flow, vortex
  - c) Source, vortex
  - d) Sink, vortex
- 2) Let  $\delta$ ,  $\delta_1$  and  $\delta_2$  denote respectively the boundary-layer thickness, displacement thickness and the momentum thickness for laminar boundary layer flow of an incompressible fluid over a flat plate. The correct relation among these quantities is
  - a)  $\delta < \delta_1 < \delta_2$
  - b)  $\delta > \delta_1 > \delta_2$
  - c)  $\delta > \delta_1 < \delta_2$
  - d)  $\delta < \delta_1 > \delta_2$
- 3) In the hydrodynamic entry region of a circular duct, the pressure forces balance the sum of
  - a) viscous and buoyancy forces
  - b) inertia and buoyancy forces
  - c) inertia and surface tension forces
  - d) inertia and viscous forces
- 4) Bodies with various cross-sectional shapes subjected to cross-flow of air are shown in the following figures. The characteristic dimension of all the shapes is the same. The cross-sectional shape with the largest coefficient of drag (i.e. sum of the pressure and skin-friction drags), at any moderately large Reynolds number, is

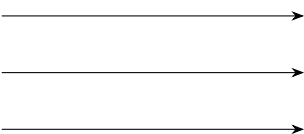
a) 



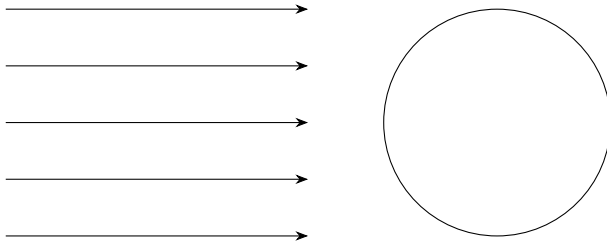
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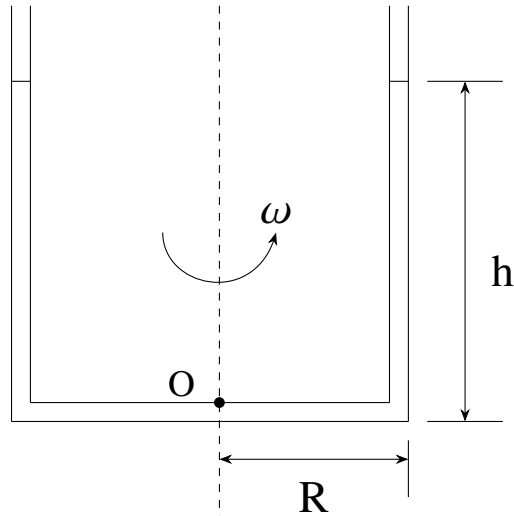
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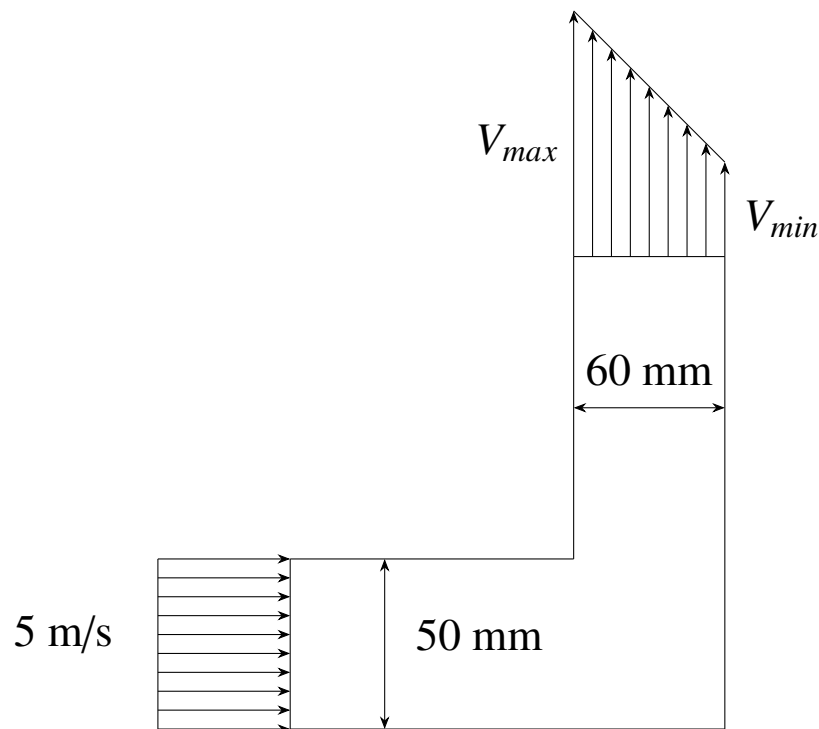
d) 



- 5) A U-tube of a very small bore, with its limbs in a vertical plane and filled with a liquid of density  $\rho$ , up to a height of  $h$ , is rotated about a vertical axis, with an angular velocity of  $\omega$ , as shown in the Figure. The radius of each limb from the axis of rotation is  $R$ . Let  $p_a$  be the atmospheric pressure and  $g$ , the gravitational acceleration. The angular velocity at which the pressure at the point  $O$  becomes half of the atmospheric pressure is given by



- a)  $\sqrt{\frac{p_a + 2\rho gh}{\rho R^2}}$   
 b)  $\sqrt{\frac{2(p_a + \rho gh)}{\rho R^2}}$   
 c)  $\sqrt{\frac{p_a + 2\rho gh}{2\rho R^2}}$   
 d)  $\sqrt{\frac{p_a + \rho gh}{2\rho R^2}}$
- 6) An incompressible fluid at a pressure of 150 kPa (absolute) flows steadily through a two-dimensional channel with a velocity of 5 m/s as shown in the Figure. The channel has a  $90^\circ$  bend. The fluid leaves the channel with a pressure of 100 kPa (absolute) and linearly-varying velocity profile.  $v_{max}$  is four times  $v_{min}$ . The density of the fluid is  $914.3 \text{ kg/m}^3$ . The velocity  $v_{min}$ , in m/s, is



- a) 25
  - b) 2.5
  - c) 2.0
  - d) 0.2
- 7) The velocity vector corresponding to a flow field is given, with usual notation, by  $\mathbf{V} = 3x\mathbf{i} + 4xy\mathbf{j}$ . The magnitude of rotation at the point (2, 2) in rad/s is
- a) 0.75
  - b) 1.33
  - c) 2
  - d) 4
- 8) The stream function for a potential flow field is given by  $\psi = x^2 - y^2$ . The corresponding potential function, assuming zero potential at the origin, is
- a)  $x^2 + y^2$
  - b)  $2xy$
  - c)  $x^2 - y^2$
  - d)  $x - y$
- 9) Fully developed flow of an oil takes place in a pipe of inner diameter 50 mm. The pressure drop per metre length of the pipe is 2 kPa. Determine the shear stress, in Pa, at the pipe wall.\_\_\_\_\_
- 10) The Darcy friction factor  $f$  for a smooth pipe is given by  $f = \frac{64}{Re}$  for laminar flow and by  $f = \frac{0.3}{Re^{0.25}}$  for turbulent flow, where  $Re$  is the Reynolds number based on the diameter. For fully developed flow of a fluid of density  $1000 \text{ kg/m}^3$  and dynamic viscosity  $0.001 \text{ Pa}\cdot\text{s}$  through a smooth pipe of diameter 10 mm with a velocity of 1 m/s, determine the Darcy friction factor.\_\_\_\_\_
- 11) Air flows steadily through a channel. The stagnation and static pressures at a point in the flow are measured by a Pitot tube and a wall pressure tap, respectively. The pressure difference is found to be 20 mm Hg. The densities of air, water and mercury, in  $\text{kg/m}^3$ , are 1.18, 1000 and 13600, respectively. The gravitational acceleration is  $9.81 \text{ m/s}^2$ . Determine the air speed in m/s.\_\_\_\_\_

**Common Data for Questions 12 and 13:**

The velocity field within a laminar boundary layer is given by the expression:

$$\mathbf{V} = \frac{Bu_{\infty}y}{x^{\frac{3}{2}}}\mathbf{i} + \frac{Bu_{\infty}y^2}{4x^{\frac{5}{2}}}\mathbf{j}$$

where  $B = 100m^{\frac{1}{2}}$  and the free stream velocity  $u_{\infty} = 0.1$  m/s.

- 12) Calculate the x-direction component of the acceleration in  $m/s^2$  at the point  $x = 0.5$  m and  $y = 50$  mm. \_\_\_\_\_
- 13) Find the slope of the streamline passing through the point  $x = 0.5$  m and  $y = 50$  mm. \_\_\_\_\_