

ENPM 667 Controls of Robotic Systems

Info Section

```
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```

Part D

```
% Choose M = 1000Kg, m 1 = m 2 = 100Kg, l 1 = 20m and l 2 = 10m. Check that  
% the system is controllable and obtain an LQR controller. Simulate the resulting response.  
% conditions when the controller is applied to the linearized system and also to the or  
% system. Adjust the parameters of the LQR cost until you obtain a suitable response. U  
% indirect method to certify stability (locally or globally) of the closed-loop system.
```

```
% Setting Given Variables
```

```
m1 = 100;  
m2 = 100;  
M = 1000;  
l1 = 20;  
l2 = 10;  
g = 9.81;
```

```
% Check if System is controllable?
```

```
A = [0 1 0 0 0 0;  
0 0 -m1*g/M 0 -m2*g/M 0;  
0 0 0 1 0 0;  
0 0 -g*(M+m1)/(M*l1) 0 -(m2*g)/(M*l1) 0;  
0 0 0 0 0 1;  
0 0 -(m1*g)/(M*l2) 0 -(g*(M+m2))/(M*l2) 0];
```

```
% disp(A);
```

```
B = [ 0 ;  
1/M ;  
0 ;  
1/(M*l1);  
0 ;  
1/(M*l2)];
```

```
disp(rank(ctrb(A,B))); % Rank = 6(Full Rank) Hence Controllable.
```

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```
% Obtaining an LQR Controller
```

```
C=[1 0 0 0 0 0;  
0 1 0 0 0 0;
```

```

    0 0 1 0 0 0;
    0 0 0 0 1 0];

D = [0];

% Selecting appropriate Q and R Values
Q = [1 0 0 0 0 0;
     0 1 0 0 0 0;
     0 0 10 0 0 0;
     0 0 0 1000 0 0;
     0 0 0 0 200 0;
     0 0 0 0 0 2000];

R = 0.0001;

% Gain Matrix
K = lqr(A, B, Q, R);
% disp(K);

disp(eig(A - B*K)); % Lyapunov Indirect Method: All eigen Values are on the left hand s

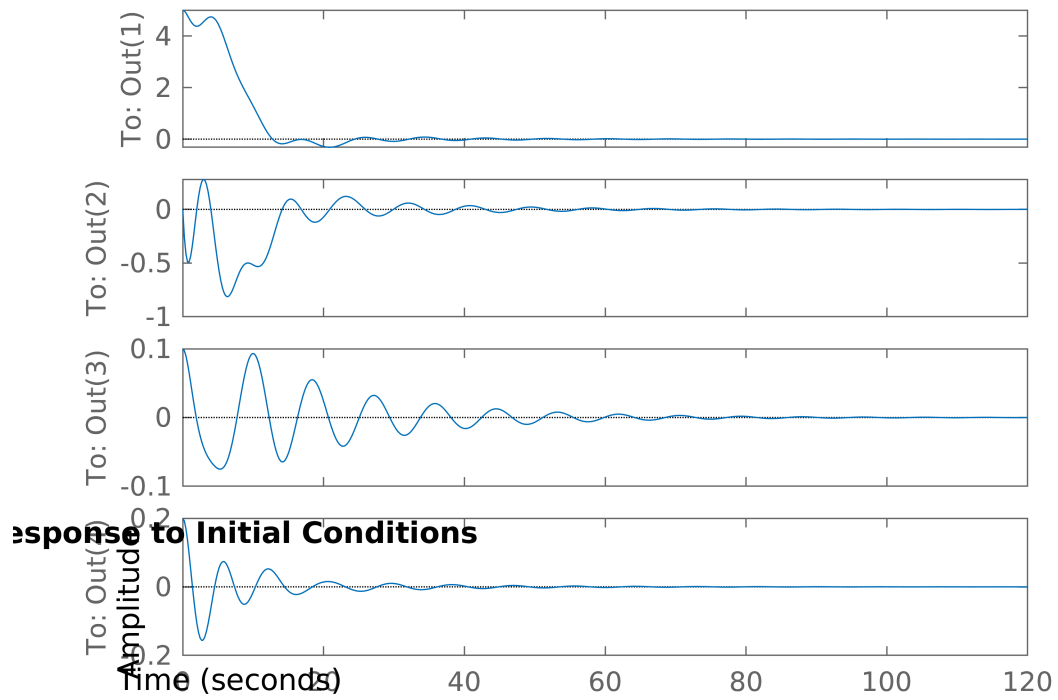
-0.2428 + 1.0205i
-0.2428 - 1.0205i
-0.2051 + 0.2028i
-0.2051 - 0.2028i
-0.0548 + 0.7229i
-0.0548 - 0.7229i

sys = ss(A-B*K, B, C, D); % Create a state-space system.

intialState = [5;
               0;
               0.1;
               0;
               0.2;
               0];

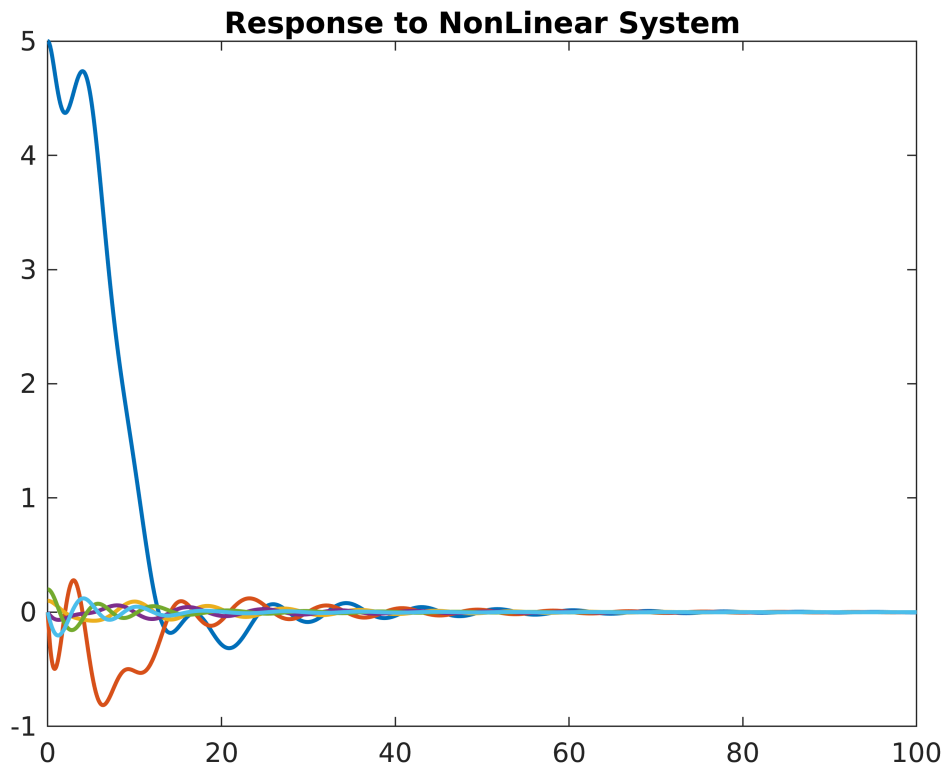
initial(sys, intialState)

```



```
% Response to Non-Linear System

t = 0:0.01:100; % TimeSteps
[t,x]=ode45(@State,t,intialState);
plot(t,x,'linewidth',1.5);
title('Response to NonLinear System');
```



Part E

```
A = [0 1 0 0 0 0;
      0 0 -m1*g/M 0 -m2*g/M 0;
      0 0 0 1 0 0;
      0 0 -g*(M+m1)/(M*l1) 0 -(m2*g)/(M*l1) 0;
      0 0 0 0 0 1;
      0 0 -(m1*g)/(M*l2) 0 -(g*(M+m2))/(M*l2) 0];

% Case I: Output Vector : x(t)
C = [1 0 0 0 0 0];
disp(rank(observ(A,C))); % Observable as Rank = 6(Full Rank)
```

6

```
% Case II: Output Vector : (θ 1 (t), θ 2 (t))
C = [0 0 0 1 0 0;
      0 0 0 0 1 0];
disp(rank(observ(A,C))); % Not Observable as Rank = 4 < 6
```

4

```
% Case III: Output Vector : (x(t), θ 2 (t))
C = [1 0 0 0 0 0;
      0 0 0 0 1 0];
```

```
disp(rank(observ(A,C)));% Observable as Rank = 6(Full Rank)
```

6

```
% Case IV: Output Vector : (x(t),  $\theta_1(t)$ ,  $\theta_2(t)$ )
C = [1 0 0 0 0 0;
     0 0 1 0 0 0;
     0 0 0 0 0 1];
disp(rank(observ(A,C)));% Observable as Rank = 6(Full Rank)
```

6

Part F

```
% Obtain your "best" Luenberger observer for each one of the output vectors for which t
% observable and simulate its response to initial conditions and unit step input. The s
% be done for the observer applied to both the linearized system and the original nonli
```

Luenberger Observer for Output $x(t)$

```
% Setting Given Variables
m1 = 100;
m2 = 100;
M = 1000;
l1 = 20;
l2 = 10;
g = 9.81;

% Check if System is controllable?

A = [0 1 0 0 0 0;
     0 0 -m1*g/M 0 -m2*g/M 0;
     0 0 0 1 0 0;
     0 0 -g*(M+m1)/(M*l1) 0 -(m2*g)/(M*l1) 0;
     0 0 0 0 0 1;
     0 0 -(m1*g)/(M*l2) 0 -(g*(M+m2))/(M*l2) 0];

% disp(A);

B= [ 0; 1/M ;0; 1/(M*l1) ;0 ;1/(M*l2)];

% disp(rank(ctrb(A,B))); % Rank = 6(Full Rank) Hence Controlable.

% Obtaining an LQR Controller

C=[1 0 0 0 0 0];

D = [0];

% Selecting appropriate Q and R Values
Q = [1 0 0 0 0 0;
     0 1 0 0 0 0;
     0 0 10 0 0 0];
```

```

0 0 0 1000 0 0;
0 0 0 0 200 0;
0 0 0 0 0 2000];

```

```
R = 0.0001;
```

```
% Gain Matrix
```

```
K = lqr(A, B, Q, R);
```

```
% disp(K);
```

```
p = [-2 -3 -4 -5 -6 -7];
```

```
L = place(A.', C.', p);
```

```
L = L.';
```

```
disp(L);
```

```
1.0e+04 *
```

```
0.0027
```

```
0.0293
```

```
-1.4999
```

```
-0.5404
```

```
1.3346
```

```
0.0686
```

```
eig(A-L*C);
```

```
Ac = [(A-B*K) (B*K); zeros(size(A)) (A-L*C)];
```

```
Bc=[ B ;zeros(size(B))];
```

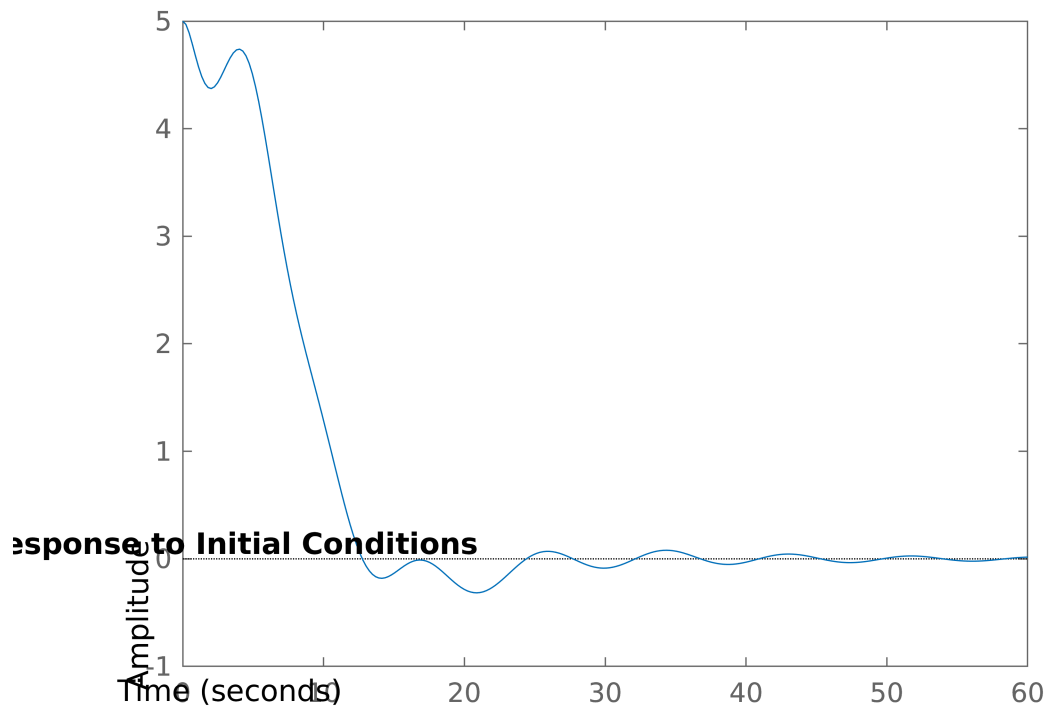
```
Cc= [C zeros(size(C))];
```

```
Dc=[0];
```

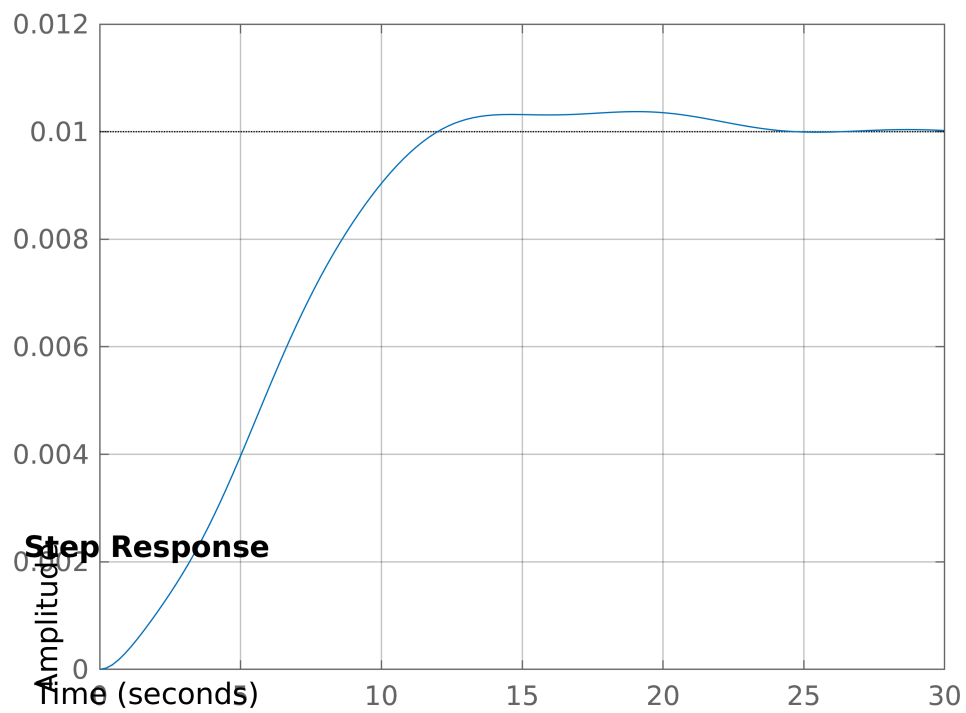
```
sys = ss(Ac,Bc,Cc,Dc);
```

```
initialState = [ 5 ; 0 ; 0.1 ; 0 ;0.2 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0];
```

```
initial(sys, initialState);
```



```
step(sys);  
grid;
```



Luenberger Observer for Output $[x(t), \theta_2(t)]$

```
C=[1 0 0 0 0 0 ; 0 0 0 0 1 0];
```

```
p = [-2 -3 -4 -5 -6 -7];
```

```
L = place(A.', C.', p);
```

```
L = L.';
```

```
disp(L);
```

Column 1

```
17.2418  
103.7782  
-256.4987  
-174.6426  
0.6195  
7.4369
```

Column 2

```
-1.9388  
-26.1469  
101.6392  
110.3637  
9.7582  
20.1529
```

```
eig(A-L*C);
```

```
Ac = [(A-B*K) (B*K); zeros(size(A)) (A-L*C)];
```

```
Bc=[ B ;zeros(size(B))];
```

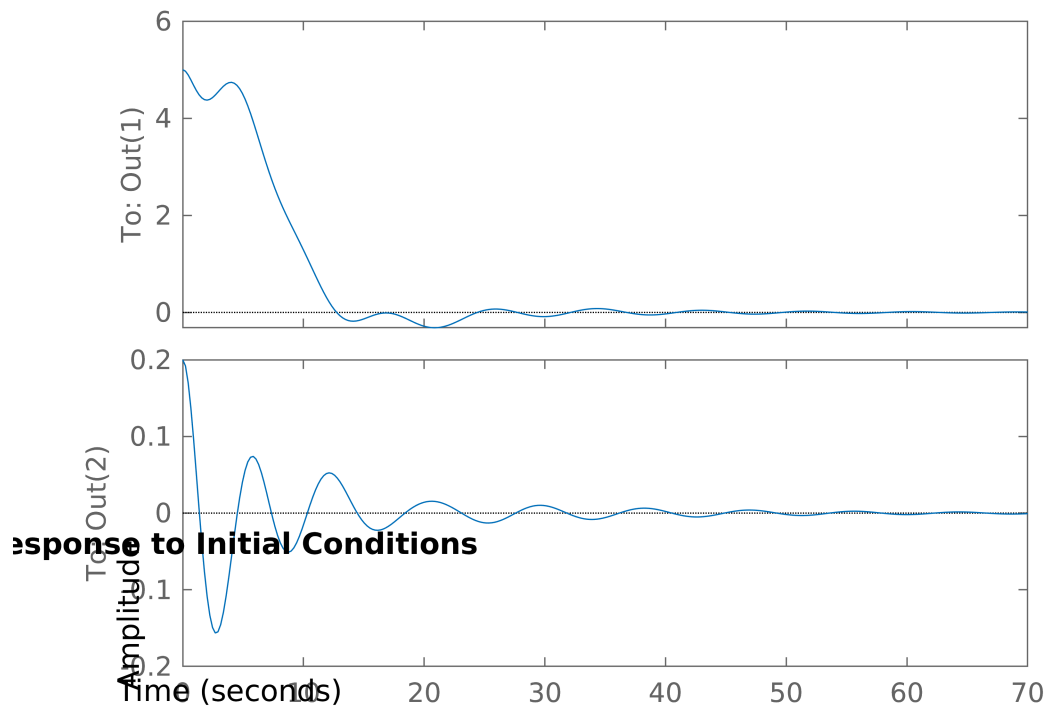
```
Cc= [C zeros(size(C))];
```

```
Dc=[0];
```

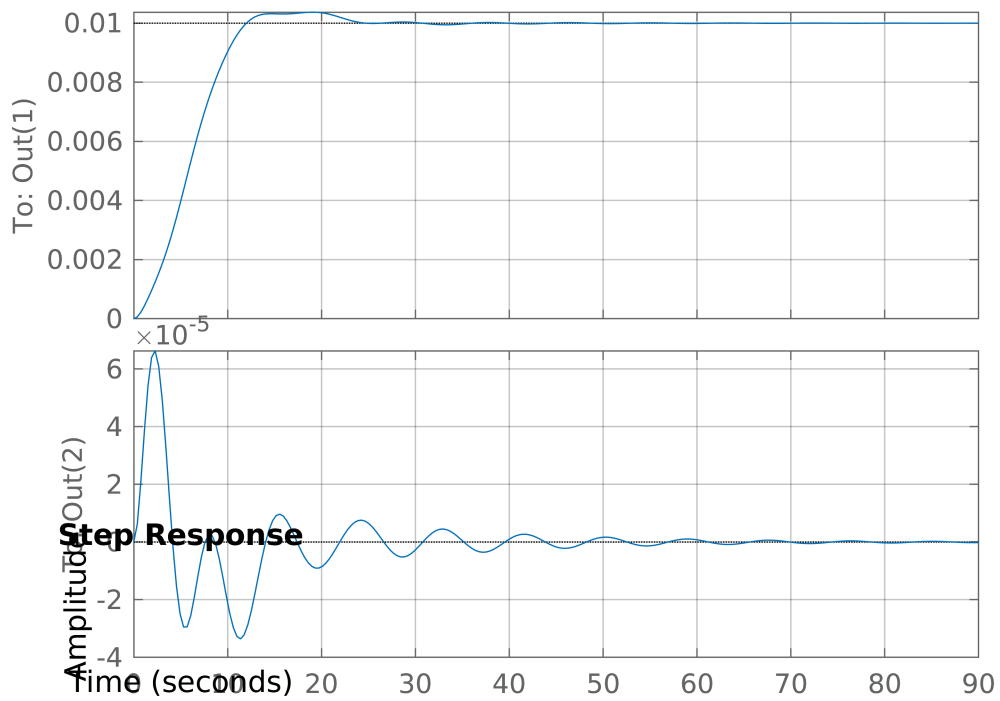
```
sys = ss(Ac,Bc,Cc,Dc);
```

```
initialState = [ 5 ; 0 ; 0.1 ; 0 ;0.2 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0];
```

```
initial(sys, initialState);
```

```
step(sys);
grid;
```



Luenberger Observer for Output $[x(t) \theta_1(t) \theta_2(t)]$

```
C=[1 0 0 0 0 0 ; 0 0 1 0 0 0; 0 0 0 0 1 0];
```

```
p = [-2 -3 -4 -5 -6 -7];
```

```
L = place(A.', C.', p);
```

```
L = L.';
```

```
disp(L);
```

Column 1

10.4331

26.3638

-0.8468

-4.6605

0.0000

0.0000

Column 2

-0.8011

-5.3594

11.5669

32.0963

-0.0000

-0.0981

Column 3

0.0000

-0.9810

-0.0000

-0.0492

5.0000

4.9209

```
eig(A-L*C);
```

```
Ac = [(A-B*K) (B*K); zeros(size(A)) (A-L*C)];
```

```
Bc=[ B ;zeros(size(B))];
```

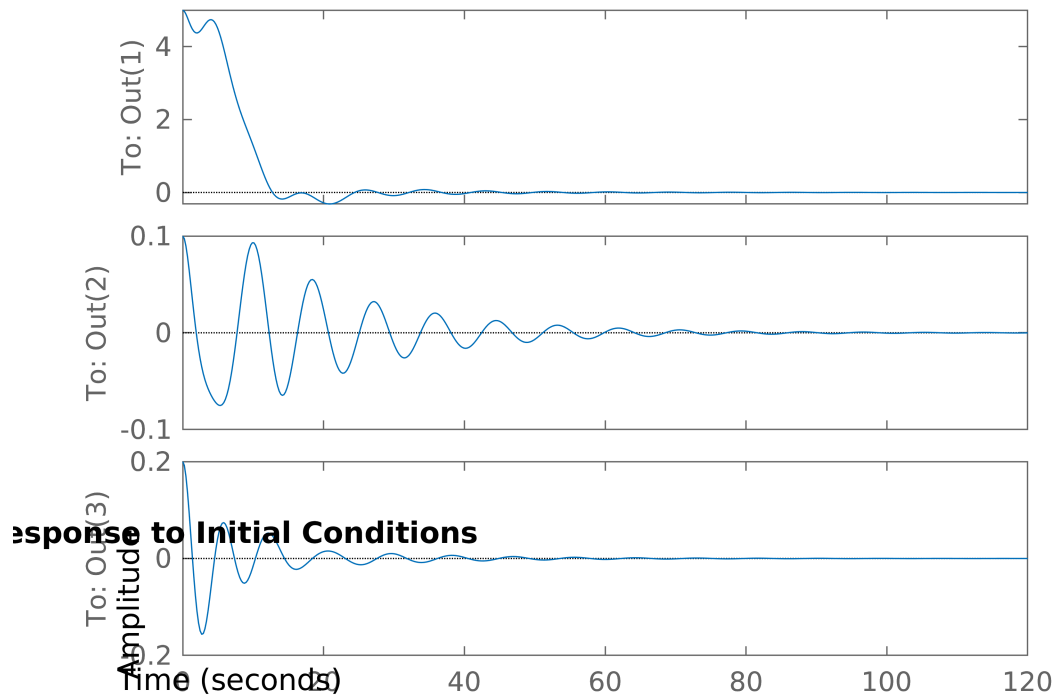
```
Cc= [C zeros(size(C))];
```

```
Dc=[0];
```

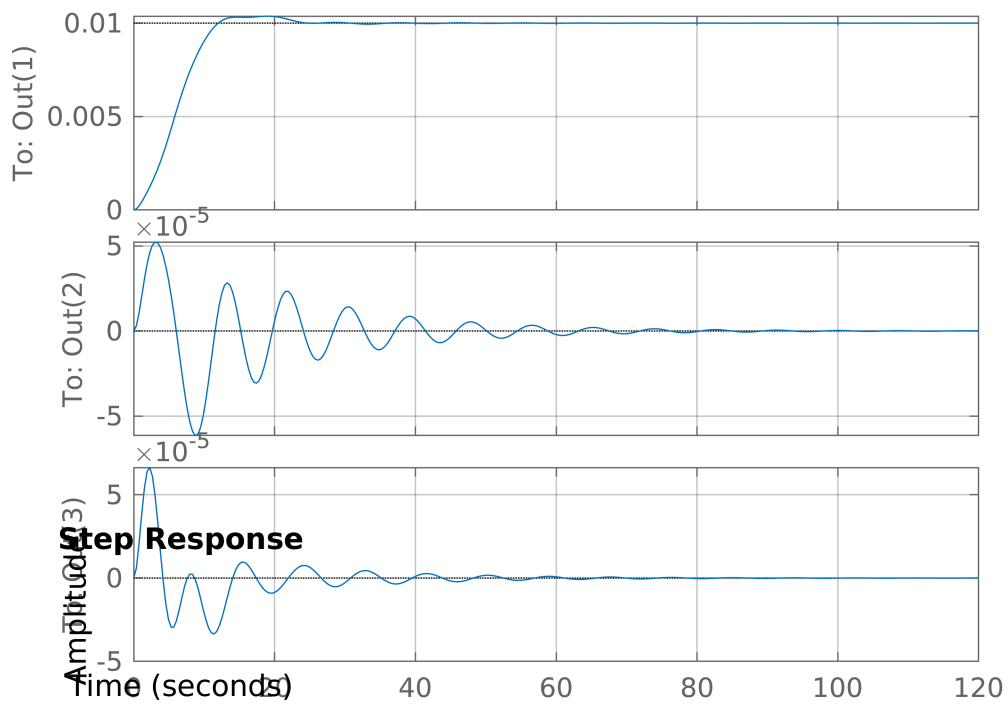
```
sys = ss(Ac,Bc,Cc,Dc);
```

```
initialState = [ 5 ; 0 ; 0.1 ; 0 ;0.2 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0];
```

```
initial(sys, initialState);
```



```
step(sys);
grid;
```



Part G

```
% Design an output feedback controller for your choice of the "smallest" output vector.  
% method and apply the resulting output feedback controller to the original nonlinear s  
% your best design and illustrate its performance in simulation. How would you reconfig  
% troller to asymptotically track a constant reference on x ? Will your design reject  
% disturbances applied on the cart ?
```

```
% Setting Given Variables
```

```
m1 = 100;  
m2 = 100;  
M = 1000;  
l1 = 20;  
l2 = 10;  
g = 9.81;
```

```
% Setting Parameters for LQE & LQR
```

```
disturbance = 0.1 * eye(1);  
noise = 1;  
sensors = [1];  
known = [1];  
states = {'x', 'x_dot', 'theta1', 'theta1_dot', 'theta2', 'theta2_dot', 'e_1', 'e_2', 'e_3', 'e_4'};  
inputs = {'F'};  
outputs = {'x'};  
initialState = [ 5 ; 0 ; 0.1 ; 0 ; 0.2 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0 ; 0];
```

```
% Check if System is controllable?
```

```
A = [0 1 0 0 0 0;  
0 0 -m1*g/M 0 -m2*g/M 0;  
0 0 0 1 0 0;  
0 0 -g*(M+m1)/(M*l1) 0 -(m2*g)/(M*l1) 0;  
0 0 0 0 0 1;  
0 0 -(m1*g)/(M*l2) 0 -(g*(M+m2))/(M*l2) 0];
```

```
% disp(A);
```

```
B = [ 0; 1/M ; 0; 1/(M*l1) ; 0 ; 1/(M*l2)];
```

```
% disp(rank(ctrb(A,B))); % Rank = 6(Full Rank) Hence Controllable.
```

```
% Obtaining an LQR Controller
```

```
C=[1 0 0 0 0 0];
```

```
D = [0];
```

```
% Selecting appropriate Q and R Values
```

```
Q = [1 0 0 0 0 0;  
0 1 0 0 0 0];
```

```

    0 0 10 0 0 0;
    0 0 0 1000 0 0;
    0 0 0 0 200 0;
    0 0 0 0 0 2000];

R = 0.0001;

% Gain Matrix
K = lqr(A, B, Q, R);
% disp(K);

% LQE Design
system = ss(A,[B B],C,[zeros(1,1) zeros(1,1)]);
[~,L,~] = kalman(system,disturbance,noise,[],sensors,known); % filter gain L
disp(L);

```

```

0.0230
0.0003
0.0000
0.0000
0.0000
0.0000
0.0000

```

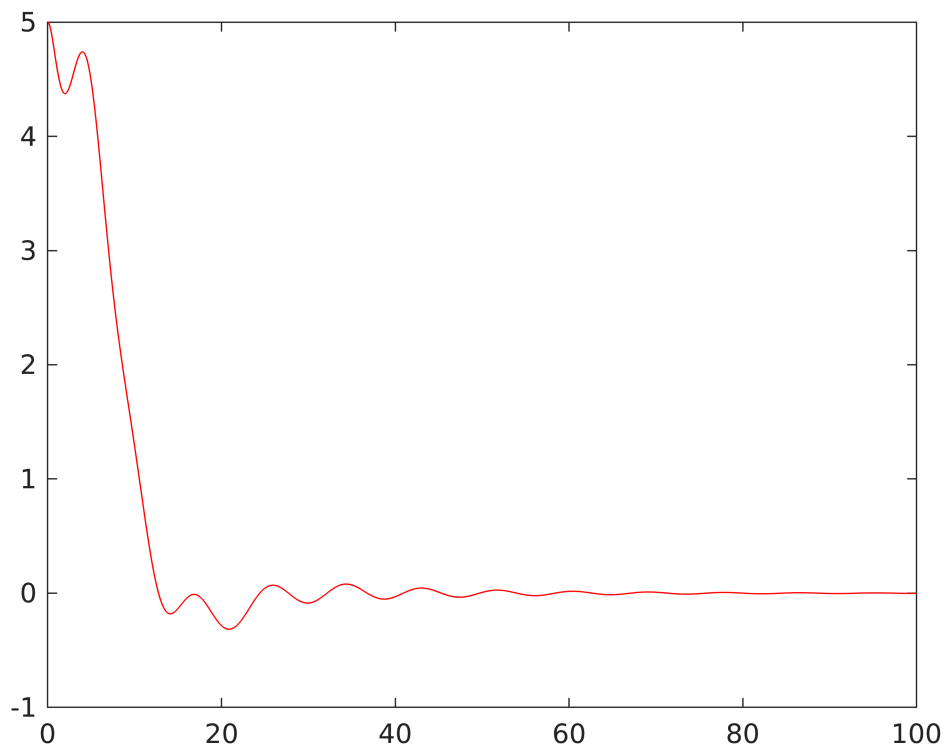
```

% LQR Design
Ac = [A-B*K B*K;zeros(size(A)) A-L*C];
Bc = zeros(12,1);
Cc = [C zeros(size(C))];
LQG = ss(Ac,Bc,Cc,D, 'statename',states,'inputname',inputs,'outputname',outputs);

time = 0:0.01:100; % Timesteps
Force = zeros(size(time)); % External Force

% Simulate Time Response
[Y,~,X] = lsim(LQG,Force,time,initialState);
figure
plot(time,Y(:,1),'r');

```



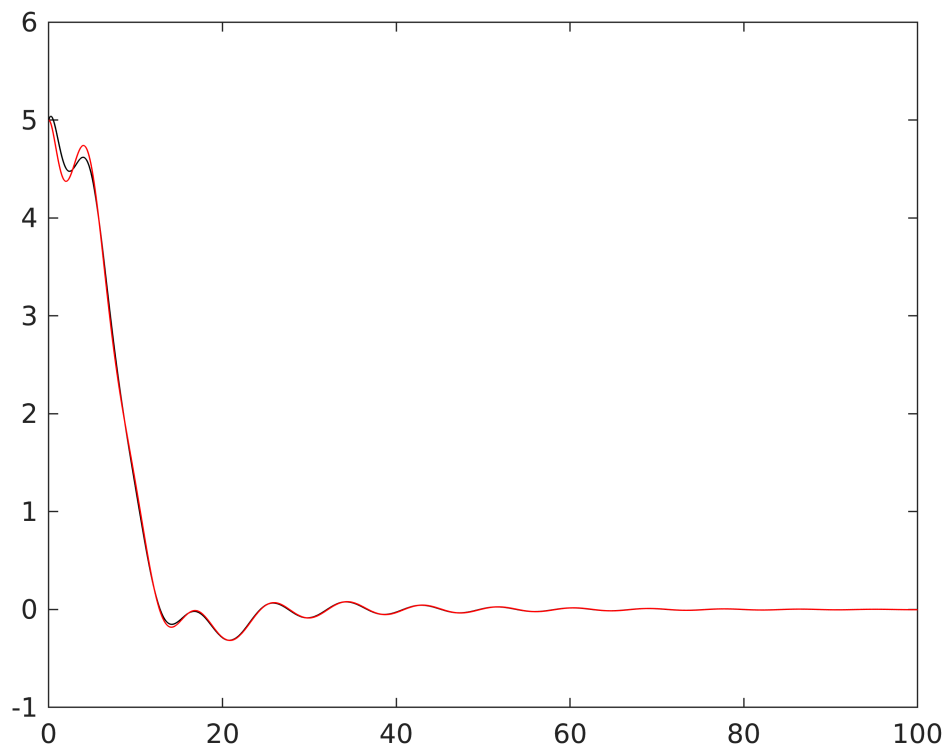
```

u = zeros(size(time)); % Input

for i = 1:size(X,1)
    u(i) = K * (X(i,1:6))';
end

Xhat = X(:,1) - X(:,6); % Error Tracking
figure;
plot(time,Xhat, 'k'), hold on, plot(time,X(:,1),'r');

```



```
function xDot = State(t,x)
```

```
% Setting Given Variables
```

```
m1 = 100;
m2 = 100;
M = 1000;
l1 = 20;
l2 = 10;
g = 9.81;
```

```
A = [0 1      0      0      0      0;
      0 0      -m1*g/M  0      -m2*g/M  0;
      0 0      0      1      0      0;
      0 0  -g*(M+m1)/(M*l1)  0  -(m2*g)/(M*l1)  0;
      0 0      0      0      0      1;
      0 0  -(m1*g)/(M*l2)  0  -(g*(M+m2))/(M*l2)  0];
```

```
% disp(A);
```

```
B= [ 0; 1/M ;0; 1/(M*l1) ;0 ;1/(M*l2)];
```

```
% Selecting appropriate Q and R Values
```

```
Q = [1 0 0 0 0 0;  
     0 1 0 0 0 0;  
     0 0 10 0 0 0;  
     0 0 0 1000 0 0;  
     0 0 0 0 200 0;  
     0 0 0 0 0 2000];
```

```
R = 0.0001;
```

```
K = lqr(A, B, Q, R);  
u = -K*x;
```

```
xDot = zeros(6,1);
```

```
xDot(1)= x(2);
```

```
xDot(2)= -((-u) + m1*l1*x(4)^2*sin(x(3)) + m1*g*sin(x(3))*cos(x(3))+ m2*l2*x(6)^2*sin(x(3))
```

```
xDot(3)= x(4);
```

```
xDot(4)= -((-u) +(M+m1)*g*sin(x(3)) + m1*l1*x(4)^2*sin(x(3))*cos(x(3)) + m2*l2*x(6)^2*sin(x(3))
```

```
xDot(5)= x(6);
```

```
xDot(6)= -((-u) + m1*l1*x(4)^2*sin(x(3))*cos(x(5)) +m1*g*sin(x(3))*cos(x(3)-x(5)) + (M+m1)*g*cos(x(3))
```

```
end
```