Reference

Jochen Guenzel 2020

Xfoil_worker is a small command line tool to do various tasks around airfoil optimization. It is especially useful for automization of the optimization process.

Usage:

```
Xfoil_worker -w worker_action [Options]
```

Following actions are supported

-w polars Generate polars for an airfoil

-w norm Repanel and normalize an airfoil

-w smooth Repanel, normalize and smooth an airfoil

The options for the various actions are described in the following sections.

Generating polars (-w polars)

A defined polar for an airfoil will be generated in Xfoils polar format. The polar file is ready to be imported into xflr5 or flow5 via the menu function "Polars / Import Xfoil Polar(s). The definition for the polar and Xfoils initial parameters are defined in an Xoptfoil input file. Please find the description of the parameters in the Xoptfoil(-JX) documentation.

Command line options used for polar generation:

-i input_file	Name of an Xoptfoil input file to read the settings for the polar generation. Only the following namelists are needed for polar generation: &polar_generation Settings for polar: polar type, based on cl oder alpha, value range &xfoil_run_options Options for Xfoils aerodynamic calculations. Normally the default values are fine, so this namelist is optional Options for Xfoils repaneling which is done prior to the polar generation. Normally the default values are fine, so this namelist is optional	
[-r xxxxxx]	The desired Reynolds number of the polar. If the Reynolds number is not defined in the input file, the value is taken from this command line option. This is useful to have a single input file for different Reynolds numbers	
-a airfoil_file	The name of the airfoil file. If option -o is omitted the generated polar will be written to the folder <airfoil_name>_polars</airfoil_name>	
[-o output_prefix]	The generated polar will be written to the folder <pre>coutput_prefix>_polars</pre>	

Example:

```
Xfoil_worker -i iPolars.txt -r 600000 -a RG15.dat -o RG15
Xfoil_worker -i iPolars.txt -a Clarky.dat
```

Repanel and normalize an airfoil (-w norm)

An airfoil will be repaneled by Xfoils PANGEN routine and normalized to have the leading edge at 0,0 and the trailing edge at 1,0. The new airfoil will have 7 decimals in the .dat file.

The default number of data points of the new airfoil will be 200. The value can be changed if a Xoptfoil inputs file is with namelist <code>&xfoil_paneling_options</code> is defined. Depending on the "micro situation" at the leading edge an additional data point at 0,0 will be inserted. So the final number of data points can be <code>npan + 1</code>.

If option '/v' is set an additional file is generated to visualize the results of repanel and normalize in Xoptfoil_visualizer-JX. In addition to the new airfoil .dat file, a file <code>output_prefix_design_coordinates.dat</code> will be generated which allows to inspect the result of repanel and normalize with Xoptfoil_visualizer-JX.

Command line options used for polar generation:

[-i input_file]	- optional - Name of a Xoptfoil input file to read the settings for repaneling. Only the following namelists are used for repaneling:	
	<pre>&xfoil_paneling_options - optional -</pre>	
	Options for Xfoils repaneling. Normally the default values are fine.	
	If the input file or a namelist is missing default values will be taken.	
-a airfoil_file	The name of the airfoil file	
[-o output_prefix]	Name of the smoothed airfoil. Following file will be generated	
	<pre><output_prefix>.dat</output_prefix></pre>	
	If option -o is omitted, the name of the output file will be <airfoil_name>_norm.dat</airfoil_name>	
[/v]	- optional - In addition to the new airfoil .dat file, a file <pre><output_prefix_design>_coordinates.dat will be generated allowing to inspect the result of repanel and normalize with Xoptfoil_visualizer-JX.</output_prefix_design></pre>	

Example:

```
Xfoil_worker -w norm -a RG15.dat -o RG15_normalized
```

Smooth an airfoil (-w smooth)

An airfoil will be repaneled by Xfoils PANGEN routine, normalized and smoothed by Xoptfoil-JX smoothing algorithm ('Chaikin Corner Cut'). This action is useful either in preparing an airfoil for optimization or to improve the geometric quality of your airfoil collection.

In addition to the new airfoil .dat file, a file <code>output_prefix_design_coordinates.dat will</code> be generated which allows to inspect the result of repaneling, normalizing and smoothing with Xoptfoil_visualizer-JX.

Command line options used for polar generation:

[-i input_file]	- optional - Name of an Xoptfoil input file to read the settings for repaneling and smoothing. Only the following namelists are used for smoothingn:		
	&smoothing_options	- optional	
	Define spike_thresh	old.	
	&constraints	- optional	
	Define curv_threshold and highlow_threshold &xfoil_paneling_options - optional Options for Xfoils repaneling which is done prior to smoothing. Normally the default values are fine, so this namelist is optional		
-a airfoil_file	The name of the airfoil file		
[-o output_prefix]	Name of the smoothed airfoil	. Following file will be generated	
	output_prefix.dat	=	
	If option -o is omitted, the na <airfoil_name>_smoothed.</airfoil_name>	•	
[/v]		new airfoil .dat file, a file coordinates .dat will be generated allowing to and normalize with Xoptfoil_visualizer-JX.	

Example:

```
Xfoil_worker -w smooth -i iSmooth.txt -a RG15.dat -o RG15_smoothed
Xfoil_worker -w smooth -a AG40.dat
```