Case name: Nessus Block 0

Description: The first ever 69lbf engine

Thu Nov 14 19:58:45 2024

Table 1. Propellant Specification

Component	Temperature, K	Pressure, MPa	Enthalpy, kJ/mol	Enthalpy, kJ/kg	Mass fraction
C2H5OH(L)	298.1	0.1013	-277.5087	-6023.834	0.4000000
O2(L)	90.2	0.1013	-12.9790	-405.608	0.6000000
Total			-96.7031	-2652.898	

Propellant exploded formula:

 $\begin{array}{cc} C_{0.633004}\,H_{1.899011}\,O_{1.683498} & \text{(based on 1 mole)} \\ C_{15.016054}\,H_{45.048163}\,O_{39.935783} & \text{(by mass \%)} \end{array}$

α: 0.7198465 (oxidizer excess coefficient)

O/F: 1.5000000

O/F₀: 2.0837775 (stoichiometric)

rho: 937.63159 kg/m³

Table 2. Combustion Properties

Parameter	Injector	Nozzle inlet	Nozzle throat	Nozzle exit Unit
Pressure	2.8958	2.8228	1.6481	0.1013 MPa
Temperature	3223.6150	3218.2576	3044.4694	2103.6377 K
Enthalpy	-59008.1683	-59362.4473	-74029.7377	-136226.8211 J/mol
	-2652.8993	-2668.4463	-3295.0344	-5919.1492 kJ/kg
Entropy	265.9365	266.0793	268.7225	275.2714 J/(mol·K)
	11.9560	11.9607	11.9607	11.9607 kJ/(kg·K)
Internal energy	-85810.8252	-86120.5597	-99342.8937	-153717.4582 J/mol
	-3857.8977	-3871.2704	-4421.7129	-6679.1294 kJ/kg
Specific heat (p=const)	5.1264	5.1243	4.5428	2.2645 kJ/(kg·K)
Specific heat (V=const)	4.4201	4.4189	3.9295	1.8945 kJ/(kg·K)
Gamma	1.1598	1.1596	1.1561	1.1953
Isentropic exponent	1.1374	1.1373	1.1399	1.1948
Gas constant	0.3738	0.3738	0.3701	0.3613 kJ/(kg·K)
Molecular weight	22.2429	22.2461	22.4671	23.0146
Density	2.4032	2.3468	1.4628	0.1333 kg/m³
Sonic velocity	1170.6983	1169.6060	1133.2560	952.8956 m/s
Mach number	0.0000	0.1508	1.0000	2.6822
Area ratio	4.0000	4.0000	1.0000	4.8648 A/At
Mass flux	413.8282	413.8282	1657.7412	$340.7642 \text{kg/(m}^2 \cdot \text{s)}$
Viscosity	1.0405	1.0393	1.0010	0.7714 x10 ⁻⁴ kg/(m·s)
Conductivity, frozen	0.3501	0.3496	0.3323	0.2359 W/(m·K)
Specific heat (p=const), frozen	2.2342	2.2338	2.2199	2.0988 kJ/(kg·K)
Prandtl number, frozen	0.6641	0.6641	0.6687	0.6864
Conductivity, effective	1.0258	1.0245	0.8692	0.2779 W/(m·K)
Specific heat (p=const), effective	5.1264	5.1243	4.5428	2.2635 kJ/(kg·K)
Prandtl number, effective	0.5200	0.5199	0.5232	0.6284

Table 3. Combustion Products

Product	Injector	Injector N	lozzle inlet N	lozzle inlet	Nozzle	Nozzle exit	Nozzle exit		
	mass	mole	mass	mole t	hroat mass t	hroat mole	mass	mole	
	fraction	fraction	fraction	fraction	fraction	fraction	fraction	fraction	
CO	0.3094818	0.2457604	0.3093108	0.2456596	0.3007642	0.2412447	0.2658618	0.2184462	
CO2	0.2779567	0.1404824	0.2782262	0.1406387	0.2916682	0.1488980	0.3465229	0.1812128	
COOH	0.0000126	0.0000062	0.0000123	0.0000061	0.0000067	0.0000033			
Н	0.0008030	0.0177214	0.0008017	0.0176947	0.0006343	0.0141391	0.0000475	0.0010857	
H2	0.0091672	0.1011497	0.0091665	0.1011564	0.0090302	0.1006416	0.0102678	0.1172241	
H2O	0.3666639	0.4527083	0.3667465	0.4528749	0.3731451	0.4653535	0.3768458	0.4814220	
H2O2	0.0000067	0.0000044	0.0000065	0.0000043	0.0000032	0.0000021			
HCHO,form	0.0000004	0.0000003	0.0000004	0.0000003	0.0000002	0.0000002			
aldehy									
HCO	0.0000101	0.0000078	0.0000099	0.0000076	0.0000053	0.0000041			
НСООН	0.0000025	0.0000012	0.0000025	0.0000012	0.0000014	0.0000007			
HO2	0.0000307	0.0000207	0.0000302	0.0000203	0.0000139	0.0000095			
0	0.0023169	0.0032210	0.0023032	0.0032025	0.0013626	0.0019134	0.0000024	0.0000035	
O2	0.0080924	0.0056252	0.0080495	0.0055961	0.0049692	0.0034890	0.0000085	0.0000061	
ОН	0.0254549	0.0332910	0.0253339	0.0331374	0.0183955	0.0243008	0.0004430	0.0005994	
Gaseous	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	1.0000000	
fraction:									
Condensed	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	0.0000000	
fraction:									

Table 4. Ideal Performance

Parameter	Sea level	Optimum expansion	Vacuum Units
Characteristic velocity		1724.9489	m/s
Effective exhaust velocity	2555.8756	2555.8756	2853.2219 m/s
Specific impulse (by mass)	2555.8756	2555.8756	2853.2219 N·s/kg
Specific impulse (by weight)	260.6268	260.6268	290.9477 s
Thrust coefficient	1.4817	1.4817	1.6541
Thrust	0.3227	0.3227	0.3603 kN
Altitude	0.0000	0.0000	km
Ambient pressure	1.0000	1.0000	0.0000 atm

Table 5. Estimated Delivered Performance

Table 3. Estillated Delivered Ferro	imanee		
Parameter	Sea level	Optimum expansion	Vacuum Units
Characteristic velocity		1683.3120	m/s
Effective exhaust velocity	2423.7054	2423.7054	2721.0517 m/s
Specific impulse (by mass)	2423.7054	2423.7054	2721.0517 N·s/kg
Specific impulse (by weight)	247.1492	247.1492	277.4701 s
Thrust coefficient	1.4398	1.4398	1.6165
Thrust	0.3060	0.3060	0.3436 kN
Altitude	0.0000	0.0000	km
Ambient pressure	1.0000	1.0000	0.0000 atm

Table 6. Altitude Performance

Table 7. Throttled Performance

Table 8. Chamber Size

	Combustion chamber s	ize		Nozzle size			
Dc	19.46	mm	Туре	TIC nozzle			
Dt	9.73	mm	Rn	1.86	mm		
Lcyl	245.07	mm	Tn	19.28	deg		
Lc	259.34	mm	Te	7.83	deg		
L*	1016.00	mm	De	21.46	mm		
R1	7.30	mm	Le	22.76	mm		
R2	14.50	mm	Le/Dt	2.34			
b	30.00	deg	Le/Lc15	104.00	%		
Ac/At	4.00		Ae/At	4.86			

Parameter		Engine	Chamber
Thrust	sea level	0.3060	0.3060 kN
	opt exp	0.3060	0.3060 kN
	vacuum	0.3436	0.3436 kN
Specific Impulse	sea level	2423.7054	2423.7054 N·s/kg
	opt exp	2423.7054	2423.7054 N·s/kg
	vacuun	2721.0517	2721.0517 N·s/kg
Mass flow rate	total	0.1263	0.1263 kg/s
	oxidizer	0.0758	0.0758 kg/s
	fuel	0.0505	0.0505 kg/s
Number of chambers		1	

Table 10. Thermal Analysis

6526 2877 910 8787 0000

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Loca	Radi	Conv (Conv	Rad.	Total	Twg,	Twi,	Twc,	Tc,	рc,	wc,	ρ, Ν	Heli	A,	de,	hc,	a,	b,	Type
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mm	mm	coeff f	flux,	flux,	flux,							m³	angl						Com
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		m²⋅K											ees						
0.00	9.73	1. 3	3204.	165.0	3369.	1400.													
		6531	1846	274	2121	0000													
3.95	9.73	1. 3	3204.	185.0	3389.	1400.													
		6530	0350	112	0462	0000													
7.90	9.73	1. 3	3203.	203.7	3407.	1400.													
		6529	8854	057	5912	0000													
11.85	9.73	1. 3	3203.	221.1	3424.	1400.													
		6528	7359	110	8469	0000													
15.81	9.73	1. 3	3203.	237.2	3440.	1400.													
		6528	5864	269	8134	0000													
19.76	9.73	1. 3	3203.	252.0	3455.	1400.													
		6527	4370	536	4906	0000													
23.71	9.73	1. 3	3203.	265.5	3468.	1400.													

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27.66 9.73 1. 3203. 277.8 3480. 1400.
          6525 1384 391 9775 0000
31.61 9.73 1. 3202. 288.7 3491. 1400.
          6524 9892 980 7872 0000
35.56 9.73 1. 3202. 298.4 3501. 1400.
          6524 8400 676 3075 0000
39.52 9.73 1. 3202. 306.8 3509. 1400.
          6523 6908 479 5387 0000
43.47 9.73 1. 3202. 313.9 3516. 1400.
          6522 5418 389 4807 0000
47.42 9.73 1. 3202. 319.7 3522. 1400.
          6521 3928 406 1334 0000
51.37 9.73 1. 3202. 324.2 3526. 1400.
          6521 2438 531 4969 0000
55.32 9.73 1. 3202. 327.4 3529. 1400.
          6520 0949 763 5712 0000
59.27 9.73 1, 3201, 329.4 3531, 1400,
          6519 9461 102 3563 0000
63.23 9.73 1. 3201. 330.0 3531. 1400.
          6518 7973 548 8521 0000
67.18 9.73 1. 3201. 330.0 3531. 1400.
          6518 6485 548 7034 0000
71.13 9.73 1. 3201. 330.0 3531. 1400.
          6517 4999 548 5547 0000
75.08 9.73 1. 3201. 330.0 3531. 1400.
          6516 3512 548 4061 0000
79.03 9.73 1. 3201. 330.0 3531. 1400.
          6515 2027 548 2575 0000
82.98 9.73 1. 3201. 330.0 3531. 1400.
          6514 0541 548 1090 0000
86.93 9.73 1. 3200. 330.0 3530. 1400.
          6514 9057 548 9605 0000
90.89 9.73 1. 3200. 330.0 3530. 1400.
          6513 7573 548 8121 0000
94.84 9.73 1. 3200. 330.0 3530. 1400.
          6512 6089 548 6638 0000
98.79 9.73 1. 3200. 330.0 3530. 1400.
          6511 4607 548 5155 0000
102.7 9.73 1. 3200. 330.0 3530. 1400.
          6511 3124 548 3673 0000
106.6 9.73 1. 3200. 330.0 3530. 1400.
         6510 1642 548 2191 0000
110.6 9.73 1. 3200. 330.0 3530. 1400.
         6509 0161 548 0710 0000
114.6 9.73 1. 3199. 330.0 3529. 1400.
         6508 8680 548 9229 0000
118.5 9.73 1. 3199. 330.0 3529. 1400.
          6508 7200 548 7749 0000
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122.5 9.73 1. 3199. 330.0 3529. 1400.

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b, Type

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Loca Radi Conv Conv Rad. Total Twg, Twi, Twc, Tc, pc, wc, ρ, N Heli A, de, hc, a,
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         kW/m^2m^2m^2
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         6507 5721 548 6269 0000
126.4 9.73 1. 3199. 330.0 3529. 1400.
        6506 4242 548 4790 0000
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130.4 9.73 1. 3199. 330.0 3529. 1400. 6505 2763 548 3312 0000 134.3 9.73 1. 3199. 330.0 3529. 1400. 6505 1285 548 1834 0000 138.3 9.73 1. 3198. 330.0 3529. 1400. 6504 9808 548 0356 0000 142.2 9.73 1. 3198. 330.0 3528. 1400. 6503 8331 548 8880 0000 146.2 9.73 1. 3198. 330.0 3528. 1400. 6502 6855 548 7403 0000 150.1 9.73 1. 3198. 330.0 3528. 1400. 6501 5379 548 5928 0000 154.1 9.73 1. 3198. 330.0 3528. 1400. 6501 3904 548 4452 0000 158.0 9.73 1. 3198. 330.0 3528. 1400. 6500 2429 548 2978 0000 162.0 9.73 1. 3198. 330.0 3528. 1400. 6499 0955 548 1504 0000 165.9 9.73 1. 3197. 330.0 3528. 1400. 6498 9482 548 0030 0000 169.9 9.73 1. 3197. 330.0 3527. 1400. 6498 8009 548 8557 0000 173.8 9.73 1. 3197. 330.0 3527. 1400. 6497 6536 548 7085 0000 177.8 9.73 1. 3197. 330.0 3527. 1400. 6496 5064 548 5613 0000 181.7 9.73 1. 3197. 330.0 3527. 1400. 6495 3593 548 4141 0000 185.7 9.73 1. 3197. 330.0 3527. 1400. 6495 2122 548 2671 0000 189.6 9.73 1. 3197. 330.0 3527. 1400. 6494 0652 548 1200 0000 193.6 9.73 1. 3196. 330.0 3526. 1400. 6493 9182 548 9731 0000 197.5 9.73 1. 3196. 330.0 3526. 1400. 6492 7713 548 8262 0000 201.5 9.73 1. 3196. 330.0 3526. 1400. 6492 6244 548 6793 0000 205.4 9.73 1. 3196. 330.0 3526. 1400. 6491 4776 548 5325 0000 209.4 9.73 1. 3196. 330.0 3526. 1400. 6490 3309 548 3857 0000 213.3 9.73 1. 3196. 330.0 3526. 1400. 6489 1842 548 2390 0000 217.3 9.73 1. 3196. 330.0 3526. 1400. 6489 0375 548 0924 0000

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tion, us, .heat .heat heat K K K MPa m/s kg/ x m<sup>2</sup> mm mm mm /
mm mm coeff flux, flux, flux,
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221.2 9.73 1. 3195. 330.0 3525. 1400.
        6488 8909 548 9458 0000
225.2 9.73 1. 3195. 330.0 3525. 1400.
         6487 7444 548 7993 0000
229.1 9.73 1. 3195. 330.0 3525. 1400.
         6486 5979 548 6528 0000
233.1 9.73 1. 3195. 330.0 3525. 1400.
        6486 4515 548 5063 0000
237.1 9.73 1. 3195. 330.0 3525. 1400.
         6485 3051 548 3600 0000
241.0 9.73 1. 3195. 330.0 3525. 1400.
         6484 1588 548 2137 0000
245.0 9.73 1. 3195. 330.0 3525. 1400.
         6483 0125 548 0674 0000
245.0 9.73 1. 3195. 330.0 3525. 1400.
        6483 0098 530 0628 0000
247.0 9.59 1. 3285. 326.2 3611. 1400.
        6949 3820 820 6641 0000
249.0 9.18 1. 3586. 328.7 3915. 1400.
         8504 6296 482 3778 0000
251.0 8.46 2. 4217. 324.2 4541. 1400.
        1757 2528 070 4598 0000
252.9 7.41 2. 5487. 320.7 5808. 1400.
        8310 4362 867 2229 0000
254.9 6.27 3. 7624. 308.3 7932. 1400.
         9333 1000 215 4215 0000
256.9 5.27 5. 1061 262.7 1087 1400.
         4758 3.847 770 6.624 0000
                3
258.9 4.88 6. 1206 195.8 1225 1400.
        2238 3.902 733 9.775 0000
                 1
259.3 4.86 6. 1202 186.5 1220 1400.
          2023 2.183 260 8.710 0000
                 9
                          0
263.2 6.08 3. 6535. 49.64 6585. 1400.
         3717 4235 22 0658 0000
267.2 7.42 2. 3877. 28.55 3906. 1400.
        0006 8186 89 3775 0000
271.1 8.57 1. 2647. 19.51 2666. 1400.
         3657 2432 24 7556 0000
275.1 9.52 1. 2015. 14.81 2030. 1400.
        0399 6022 85 4207 0000
7641 0452 65 8618 0000
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Loca Radi Conv Conv Rad. Total Twg, Twi, Twc, Tc, pc, wc, ρ, N Heli A, de, hc, a, b, Type