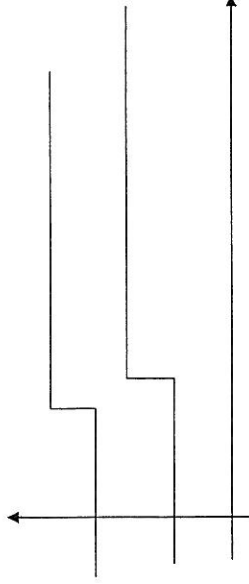


Chapter 4. Design of State-Space Servo- Controllers

4.1 Introducing the Reference Signal

It is desired that the process output vector should follow a specified vector of setpoints:



Consider the SISO process:

$$\begin{aligned}\dot{\underline{x}} &= A\underline{x} + Bu \\ y &= C\underline{x}\end{aligned}$$

For a steady state desired output y_{ss} we have:

Propose the control-law:

$$u(t) = u_{ss} - K(\underline{x}(t) - \underline{x}_{ss})$$

In the steady state:

$$\begin{aligned}\underline{0} &= A\underline{x}_{ss} + Bu_{ss} \\ y_{ss} &= C\underline{x}_{ss}\end{aligned}$$

Let us propose the simple relationships:

$$u_{ss} = N_u r_{ss} \quad \underline{x}_{ss} = N_x r_{ss}$$

This implies:

$$\begin{aligned}\underline{0} &= AN_x r_{ss} + BN_u r_{ss} \\ y_{ss} &= CN_x r_{ss}\end{aligned}$$

Of course if we want the steady state output to be r_{ss} :

Then the gains N_x and N_u can be determined as:

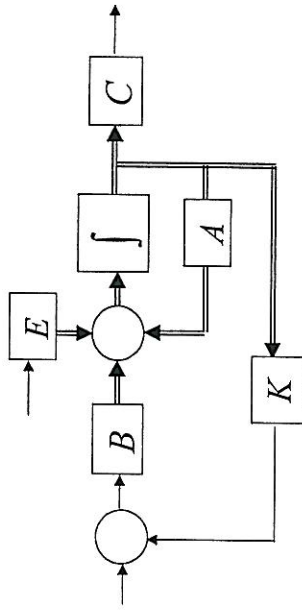
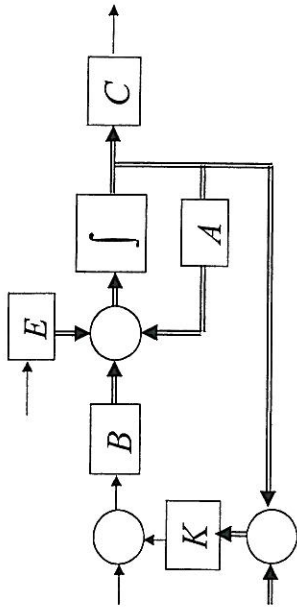
$$\begin{bmatrix} N_x \\ N_u \end{bmatrix} = \begin{bmatrix} A & B \\ C & 0 \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

The control-law is then:

$$u(t) = N_u r(t) - K(\underline{x}(t) - N_x r(t))$$

Which could of course be rewritten as:

This yields the following control structures:



Tutorial: Introduce a setpoint signal to the motor speed controller developed earlier – test in Simulink.

Some of the problems of this technique include:

- Does not increase the system type:
- Gains designed to reduce e_{ss} for setpoint changes:

4.2 State-Space Control with Integral Action

Consider again the SISO process:

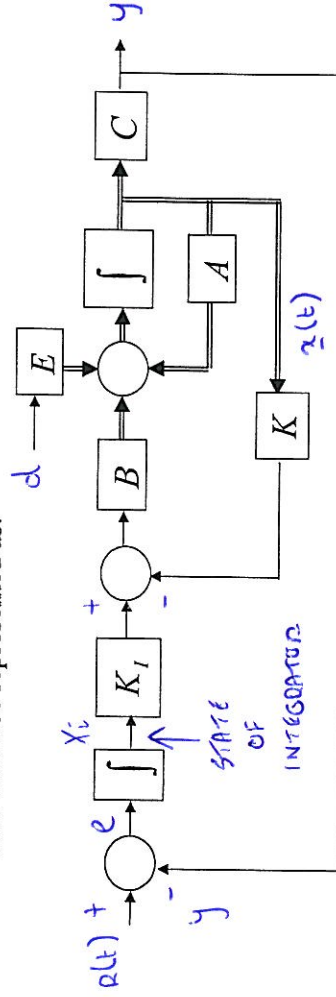
$$\dot{\underline{x}}(t) = A\underline{x}(t) + Bu(t)$$

$$y(t) = C\underline{x}(t)$$

The state control-law with integral action is:

$$u(t) = -K\underline{x}(t) + K_I \int_0^t e(\tau) d\tau$$

where: $e(t) = r(t) - y(t)$
 we want $y(t)$ to track setpoint $r(t)$
 This could be represented as:



we have introduced an outer feedback loop

Introduce another state:

$$\dot{x}_I(t) = \int_0^t e(\tau) d\tau$$

OUTPUT OF
CONTROLLER INTEGRATOR

The control-law become:

$$u(t) = -K\underline{x}(t) + K_I x_I$$

This yields the closed-loop state equation:

$$\dot{\underline{x}}(t) = A\underline{x}(t) + B(-K\underline{x}(t) + K_1 x_1(t)) + \underbrace{u(t)}_{\text{N.B.}}$$

But we know:

$$x_1(t) = \int_0^t e(\tau) d\tau$$

$$\therefore \dot{\underline{x}}_I(t) = e(t) = r(t) - y(t) = r(t) - C\underline{x}(t)$$

Hence the complete closed loop system can be represented by the coupled equations:

$$\begin{cases} \dot{\underline{x}}(t) = (A - BK)\underline{x}(t) + BK_1 x_1(t) \leftarrow \text{PROCESS} \\ \dot{x}_1(t) = -C\underline{x}(t) + r(t) \leftarrow \text{CONTROLLER} \end{cases}$$

Assign a new state vector:

$$\underline{z}(t) = \begin{bmatrix} \underline{x}(t) \\ x_1(t) \end{bmatrix} \quad \begin{matrix} \uparrow N \\ \uparrow 1 \end{matrix} \quad \begin{matrix} \varepsilon R^{N+1} \end{matrix}$$

The closed-loop equations can be written more compactly as:

$$\frac{d}{dt} \begin{bmatrix} \underline{x}(t) \\ x_1(t) \end{bmatrix} = \begin{bmatrix} A - BK & BK_1 \\ -C & 0 \end{bmatrix} \begin{bmatrix} \underline{x}(t) \\ x_1(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r(t)$$

$$\dot{\underline{z}} = A_2 \underline{z} + B_2 r(t)$$

The poles of the closed-loop system are given by the roots of:

$$\det(sI - A_2) = 0$$

N.B.: There are $(N+1)$ closed loop poles

Determine the gains to place the $N+1$ closed-loop poles to obtain the following characteristic equation:

$$C_{des}(s) = s^{N+1} + C_N s^N + \dots + C_1 s + C_0$$

Proof of Integral Action: Assume closed-loop stability

Consider an asymptotically constant setpoint signal:

$$\text{step like } \lim_{t \rightarrow \infty} r(t) = R_{ss}$$

Since the closed-loop system is stable — the states must converge to steady-state values: reach equilibrium

$$\lim_{t \rightarrow \infty} \dot{\underline{x}} = 0 \rightarrow 0 = (A - BK)\underline{x}_{ss} + BK_1 x_{1ss} \quad x(t) \rightarrow x_{ss}$$

$$\lim_{t \rightarrow \infty} \dot{x}_1 = 0 \rightarrow 0 = -C\underline{x}_{ss} + r_{ss} \quad r(t) \rightarrow r_{ss}$$

$$\text{hence: } 0 = -C\underline{x}_{ss} + R_{ss} \quad y(t) \rightarrow y_{ss}$$

$$y_{ss} = C\underline{x}_{ss}$$

$$0 = -y_{ss} + R_{ss}$$

$$y_{ss} = R_{ss} \quad e_{ss} = 0$$

EXAMPLE: The DC Motor

$$\frac{d}{dt} \begin{bmatrix} \omega(t) \\ i(t) \end{bmatrix} = \begin{bmatrix} 0 & 50 \\ -200 & -200 \end{bmatrix} \begin{bmatrix} \omega(t) \\ i(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 200 \end{bmatrix} v(t) + \begin{bmatrix} -50 \\ 0 \end{bmatrix} T_L(t)$$

$$y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} \omega(t) \\ i(t) \end{bmatrix}$$

$$\text{SPEED CONTROL } y(t) = \omega(t)$$

Use the control-law

$$v(t) = -[k_1 \quad k_2] \begin{bmatrix} \omega(t) \\ i(t) \end{bmatrix} + K_1 \int_0^t (r_o(\tau) - y(\tau)) d\tau$$

Then :

$$\dot{\underline{x}} = R_{ss}(t) - [C \quad 1 \quad 0] \begin{bmatrix} \omega \\ i \end{bmatrix}$$

SPEED ERROR

Hence the closed loop state-equation is:

$$\frac{d}{dt} \begin{bmatrix} \omega(t) \\ i(t) \\ x_1(t) \end{bmatrix} = \begin{bmatrix} 0 & 50 & 0 \\ -200 - 200k_1 & -200 - 200k_2 & 200K_I \\ -1 & 0 & 0 \end{bmatrix} \begin{bmatrix} \omega(t) \\ i(t) \\ x_1(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} r(t) + \begin{bmatrix} -50 \\ 0 \\ 0 \end{bmatrix} T_L(t)$$

The poles of the closed-loop system are given by roots of:

$$\det \begin{bmatrix} s & 0 & 0 \\ 0 & s & 0 \\ 0 & 0 & s \end{bmatrix} - \begin{bmatrix} 0 & 50 & 0 \\ -200 - 200k_1 & -200 - 200k_2 & 200K_I \\ -1 & 0 & 0 \end{bmatrix} = 0$$

which yields:

$$\det \begin{bmatrix} s & -50 & 0 \\ 200 + 200k_1 & s + 200 + 200k_2 & -200K_I \\ 1 & 0 & -s \end{bmatrix} = 0$$

the closed-loop characteristic equation is then:

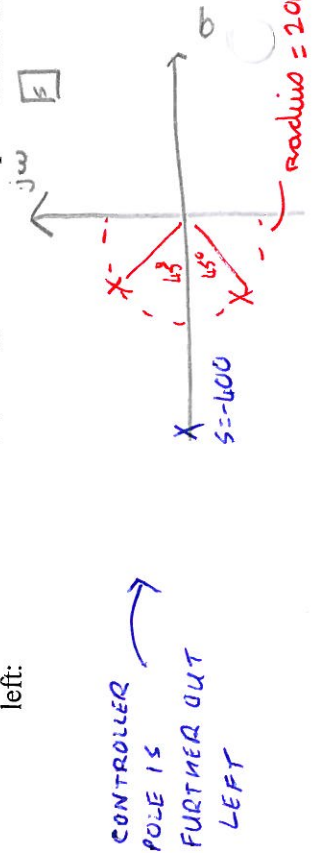
$$1 \mid \begin{matrix} -50 & 0 \\ s + 200 + 200k_2 & -200k_2 \end{matrix} + s \mid \begin{matrix} s & -50 \\ 200 + 200k_1 & s + 200 + 200k_2 \end{matrix} = 0$$

$$s^3 + 200(1+k_2)s^2 + 10,000(1+k_2)s + 10,000k_2 = 0$$

Now we still want a second order dominant response with:

$$\xi = 0.707 \quad \omega_n = 200 \text{ rad/s}$$

This can be achieved by placing the controller pole further out left:



The desired closed-loop characteristic equation is:

$$C_{des}(s) = (s + 400)(s^2 + 282.8s + 40000) = s^3 + 682.8s^2 + 153120s + 16000000$$

FAST SLOW

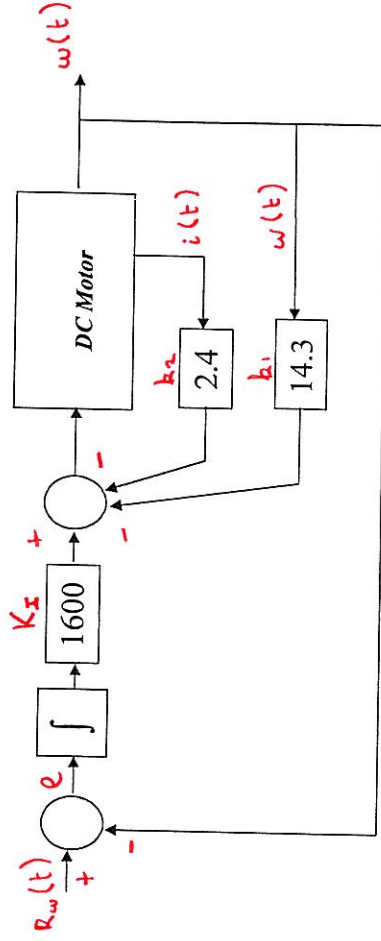
Compare with the closed-loop characteristic equation:

$$\begin{aligned} 10000K_I &= 16000000 \Rightarrow K_I = 1600 \\ 10000(1+k_1) &= 153120 \Rightarrow k_1 = 14.3 \\ 200(1+k_2) &= 682.8 \Rightarrow k_2 = 2.4 \end{aligned}$$

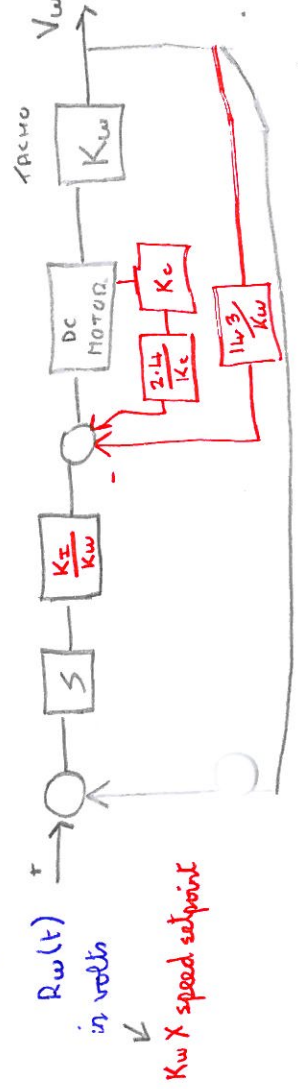
This yields the controller:

$$v(t) = -14.3\omega(t) - 2.4i(t) + 1600 \int_0^t (r_o(\tau) - \omega(\tau)) d\tau$$

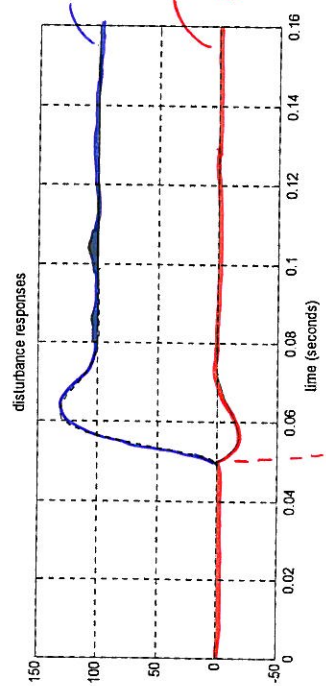
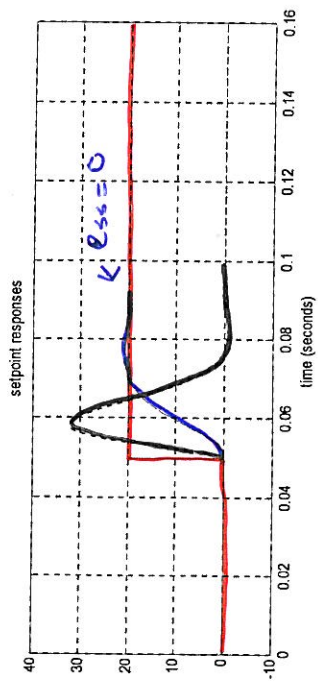
which could be built as follows:



In practice



The closed-loop responses are:



$\downarrow \uparrow 100 \text{ Nm}$ load torque applied

4.2.1 Use of Ackermann's Method to Design Controllers with Integral Action

Define the open-loop equations as:

$$\dot{\underline{x}}(t) = A\underline{x}(t) + B u(t)$$

$$\dot{x}_I(t) = r(t) - C\underline{x}(t)$$

Or in more compact form as:

$$\frac{d}{dt} \begin{bmatrix} \underline{x} \\ x_I \end{bmatrix} = \begin{bmatrix} A & 0 \\ -C & 0 \end{bmatrix} \begin{bmatrix} \underline{x} \\ x_I \end{bmatrix} + \begin{bmatrix} B \\ 0 \end{bmatrix} u(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r(t)$$

OPEN LOOP EQUATIONS

$\dot{\underline{z}} = A_2 \underline{z} + B_2 u + G_R$
Which is under the control:

$$u(t) = -K\underline{x}(t) + K_I x_I$$

or:

$$u(t) = -[K \mid -K_I] \underline{z}(t)$$

$$u = -k_2 \underline{z}(t)$$

We can then use Ackermann's formula:

$$k_2 = [K \mid -K_I] = [0 \ 0 \ \dots \ 0 \ 1] C_z^{-1} C_{des}(A_2)$$

where:

$$C_{des}(s) = s^{N+1} + C_N s^N + \dots + C_1 s + C_0 = 0$$

$$A_2 = \begin{bmatrix} A & 0 \\ -C & 0 \end{bmatrix}$$

and:

$$C_z = \begin{bmatrix} B_2 & A_2 B_2 & A_2^2 B_2 & \dots & A_2^N B_2 \end{bmatrix}$$

$$\begin{bmatrix} B \\ 0 \end{bmatrix} \nearrow \begin{bmatrix} A & 0 \\ -C & 0 \end{bmatrix}$$

The control-law becomes:

۲۵

۲۵

۲۵

۲۵



۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵



۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵

۲۵



۲۵

۲۵