

Processed Text

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c arnaldo leal ufes br | j abstract
smart sensing for aeronautical applications is a multidisciplinary process that involves the
development of various sensor elements and advancements in the nanomaterials field the expansion
research fueled development commercial military aircraft aeronautical field
optical technology is one of the supporting pillars for this as well as the fact that the unique high
tech qualities of aircrafts align with sustainability criteria in this study a multidisciplinary
investigation of airplane monitoring systems employing optical technologies based on optical fiber
and nanomaterials that are incorporated into essential systems is presented this manuscript reports
the multifunctional integration of optical fibers and nanomaterials for aircraft sector discussing topics
such as airframe monitoring flight environments sensing from temperature and humidity to pressure sensing
sensors for navigation such as gyroscopes and displacement or position sensors pilot vital health monitoring
and novel nanomaterials for aerospace applications the primary objective of this
review is to provide researchers with direction and motivation to design and fabricate the future of
the aeronautical industry based on the actual state of the art of such vital technology thereby aiding
their future research keywords structural health monitoring aviation smart materials optical fiber sensors
aircraft security flight light citation marque c leal júnior kumar multifunctional integration
of optical fibers and nanomaterials for aircraft systems materials 2023 1 introduction 16 1433 <http://doi.org/10.3390/ma16041433> more than even sustainability is a strong word used in different areas of research in
order to orientate in the right way the progress of digitalization for obtaining unique and academic editors
theodore smar high tech qualities in different industrial sectors one of the most critical sectors is e
matikas and patricia krawczak aerospace where the aeronautical industry needs to align the progress
sustainability received 11 december 2022 photonics in this way
advances in smart sensors and nanomaterials are crucial in the revised 6 february 2023 last year
photonics has brought a growing role in aeronautical and aviation fields accepted 7 february 2023 head
up display to onboard optical fiber networks improving aircraft monitoring and published 8 february 2023
maintenance in a mean that is not possible with copper based electrical systems although
numerous developments in commercial aircraft take place in zones of a plane that travelers do not see
still some enhancements that occur inside the passenger cabin are required as led illumination
communication between crew and passengers or electrochromatic copyright 2023 author
dimming of the windows licensee mdpi basel switzerland the implementation of the fly wire
perception has significantly reduced article open access article
the weight and complexity of traditional electro mechanical systems also in increasing the distributed term
stability safety airplane 1 typical commercial military aircraft involve conditions of the creative commons
attribution cc by license <http://huge.numbercontrol.monitoring.system> demanding big quantity sensor
creativecommons.org license distributed along the aircraft from a copper cable network
the output of electrical signals 4 0 is routed for the commanding computers which activate the actions
over the last few materials 2023 16 1433 <http://doi.org/10.3390/ma16041433> <http://www.mdpi.com/journal/materials> materials 2023 16 1433 2 of 29 year the aeronautical industry has changed a lot
due to the climate concerns growing order decrease emission progress fuel efficiency hard job
been achieved by the progress of composite fuselage to replace the traditional metallic material
thus building lightweight aircraft models while this is a serious change also takes
reduced manufacturing and operating costs composite structures subject to avionic systems
to a bit of a severe level of lightning induced voltage and current actually one of the main factor leading airplane
failure subsequent electrical wire damage 2 context

optical technologies have endorsed the migration of copper harness to optical fiber based systems following the fly light perception due to the high electromagnetic immunity [3, 4] whereas such a concept has been brought under huge affirmation in terms of reliability and security [4]. The modern aircrafts have higher operational requests driven by the increase not only in terms of safety on critical systems for flight and engine control but also in non safety critical systems such as cabin environmental control systems, structural engine health monitoring (EHM), system structural health monitoring (SHM) [5] etc. which are answered more than just in the optical domain [4, 8]. Presently, the optical fiber sensors (OFS) are seen in general by the scientific community and industrial and end user communities to be the technology with the maximum potential for the continuous real time monitoring of aircraft structures in different topologies [4, 5, 9, 11] from flight environments sensing or sensors for navigation to pilot vital health monitoring. The additional potential for integrating OFS into composite materials during the layup process would also enable the monitoring of composite structures during their entire lifecycle improving their safety, cost efficiency, reliability and also spreading the operational life. Addition of OFS largely known particular characteristics such as electromagnetic fields immunity, compactness, multiplexing capabilities, passive operation, biocompatibility and chemical stability [11]. Particular profit measure report data using sensor include reducing aircraft weight, replacing bulky system, decreasing cost cheaper than other types due to particular characteristics such as multiplexing or multi point features [11] and providing predictive data that can help the flights become further fuel efficient. Such sensors and systems could offer capacities and measurements that are not conceivable with other components [5, 9, 10]. In addition, OFS can be installed in a smaller portion of time compared with their alternatives from the nature of how optical fiber works. Numerous sensors can be combined using a unique optical fiber which means non complexity in installation involving just one fiber to be attached to the aircraft structure. Consequently, linked to the data acquisition equipment in this way, this paper reviews some critical aspects in order to see that the future is expected to see an enlarged effort on the use of OFS and nanophotonic materials potential in the aircraft to expand on the industry's safety and also in the human staff that are linked with such industry such as flight attendants, pilot, the airport maintenance crew, etc. well as to extend the aircraft's life, decrease the requirement for time consumption and expensive maintenance actions and increase the flights efficiency in order to pave the way for sustainable structures and transportation. See figure 1. Materials 2023, 16, 1433. 3 of 29. Figure 1. Illustration overall content considering problem advance. Photonics and nanomaterials for aeronautical sector. 2. Airframe monitoring. As is well known, the strong use of composite structures in the aerospace industry over the last few decades with newer aircrafts such as Boeing 787 and Airbus A350 made largely respectively 50% and 50% [12, 13] of composite materials brought considerable number of new successful contributions to the progress of the SHM field for composite structure. This development can be attributed to several research performed until now. Many researchers achieved material particular mechanical property instance high strength, weight ratio and corrosion resistance. Therefore, such use of composite structures in aircraft permit to condense the operational costs due to less fuel consumption and less maintenance protocols. Although several advantages can be reached when is used composite structures for aircrafts, their performance or comportment pose challenge namely in the inspection and monitoring of damage in the aircraft structure where the damage may be present inside but not discernable upon visual checkup of the outside layers from aircraft structure through maintenance protocols. In aeronautical and aerospace industries, the application of SHM has still immature [14] and specifically its application to composite structures is very challenging for a real aircraft maintenance situation. A full damaged diagnostic on all four SHM levels rather than a single level. See figure 2. Is required. It means damaged detection, b) damage localization, c) damage type identification, damage severity. The SHM with successful application to aircraft needs two situations: a) profitable process, airline operator reducing economic loss caused unproductive downtimes and provide accurate and reliable data about the condition of the structure growing the security of critical components to consequently avoid disasters. Therefore

from industry and academia, a nextensive range of potential SHM technologies is being developed to satisfy these conditions [15]. Where the most capable options are acoustic emission method, micro electro mechanical system (MEMS), electrical strain gauge, crack wires, optical based technologies and comparative vacuum monitoring [16]. Ever from the extensive works performed until now, the optical technologies have been the future for the aircrafts' airframe in terms of SHM materials [2023, 16, 1433, 4 of 29]. Figure 2 diagram with four SHM levels for a full damage diagnostic to obtain the prognostic of such as the ones based on fiber Bragg grating (FBG) and optical measurement systems have previously found a range of exciting practical applications in the damage and load monitoring of aircraft composite structures, namely in ground tests and design well known from the literature [9]. The FBG sensor's operation principle is based on the Bragg wavelength shift related to variations on grating period and refractive index (n_i). Generally, these variations occur due to strain (ϵ) and temperature effects on the FBG, as shown in equation (1). The temperature variations lead to thermal expansion, leading to a grating period variation proportional to the temperature and thermal expansion coefficient (α) and the thermo-optic effect (variation proportional to the temperature and thermo-optic coefficient ξ). The strain leads to the variation on the grating period from the applied strain and the photoelastic effect (variation proportional to the strain and photoelastic constant $p_{e\lambda}$). $p_{e\lambda} = \epsilon \alpha \xi \lambda$ [1, b, e, b]. Some aircraft are in action with integrated fiber networks achieving measurements during flights. Airbus reported in 2013 that the long term idea is that all new aircraft will fly with distributed FBG optical sensors [17]. Indeed, some of the developed solutions are considered relatively mature at a technology readiness level (TRL) of 5-6. For instance, it has proved the viability and effectiveness using FBG based local damage detection when applied to composite parts of aircraft as bonded repairs [1]. Nevertheless, the broader acceptance of the use of such sensing systems is still hindered by the following issues: sensor performance when embedded, capability of detection, maintainability, available interrogation equipment, a slow cost solution, lack of standardization and certification framework. One main challenge: multipoint FBG based distributed detection technique. Point view ability to detect damage continues development of controlled methods to monitor the main parameters related to the onset and growth of damage over time of large structures with adequate physical and spatial resolution even position damage known precisely enough.

Aeronautic companies and entities address the issues of reducing operating costs, achieving greater aerodynamic efficiency and improving the safety and the reliability of future aircraft. Aircraft health monitoring involves using sensors to monitor the integrity of advanced structural materials expected to become backbone next generation airframes: graphite reinforced composites give the potential for larger strength and stiffness, weight ratio than aluminum alloys. It offers the potential for high density sensor coverage, slight weight penalty. Numerous sensors used: load, strain, shape, monitoring [9, 11]. Wing shape measurement, landing gear reported optical devices for applications relating to the health monitoring of composite materials [2023, 16, 1433, 5 of 29]. Materials have also been reported [18]. Graphite epoxy panels were manufactured with integrated optical fibers of different types. The panels were thermally and mechanically tested as composite strength sensor durability. Experimental results evaluate the ability of surface mounted and embedded optical fibers to measure strain and temperature were reported. The sensor performance was compared directly by the results from traditional instrumentation. Such experimental results indicate that of integrated composite potential application: monitoring structural integrity, critical parts in aerospace and aeronautical vehicles. In this way, fibers were successfully embedded in high performance graphite, polyimide and graphite epoxy parts, achieving the potential for aircraft structural applications [18] and demonstrating the use of single mode optical fibers to measure strain and temperature in composites. Figure 3 shows a representative overview about sensor elements along the optical fiber that can be useful to measure the elongation of an aircraft flap, wing shape, strain distribution and structural damage condition as well as all data collected by a data logger. It is possible that the understanding of composite aging uses the capability to measure and correlate chemical and physical property changes from this information. Predictive decisions can be made taking into account the state of health of a composite part and understanding what procedures to do as a requirement.

thus avoiding catastrophic failure figure 3

schematic representation of sensor elements along the optical fiber that can be useful to measure the elongation of an aircraft flap strain distribution and structural damage condition and all data collected by data logger 3 flight environments sensing 3 1 critical environmental sensing different engineering structures are exposed to environmental conditions related to the operation or even the natural aspects of the environment the environmental sensing plays an important role not only on the structural analysis but also in the navigation and control systems 19 it is also important to mention that the dynamics of the environmental condition such as temperature and relative humidity directly affect the operational and structural analysis in aircrafts in both corrective and predictive maintenance conditions 20 furthermore the environmental monitoring is critical for the stress strains sensors since many sensors present temperature cross sensitivity where the structural failures can be also related to the temperature distribution dynamics 21 materials 2023 16 1433 6 of 29 in aircraft operation the environmental monitoring is crucial for the navigation and instrumentation equipment where the air data system is responsible to provide the flight data to the crew 22 considering an important technology for the system navigation especially external measurement speed pitot tube technology widely employed 23 such as sensor uses the air pressure in dynamic and static intake conditions in which the operation in the harsh environments of the flight conditions can lead to freezing of the device so risk that can ultimately lead to critical issues in the sensor reading and even accident 22 to that extent the federal aviation administration provides instructions for the atmospheric conditions that can lead to such critical issues in the instrumentation which also indicate the necessity of measuring the atmospheric environmental conditions flight 22 it is also important to notice that the environmental conditions including temperature moisture concentration and pH can lead to increases in maintenance costs and downtime due to corrosion in different parts of the aircraft 24 for this reason environmental analysis also plays an important role in the structural defects due to environmental effects mainly the corrosion of the aircraft structure and components 25 in summary the corrosion is an electrochemical deterioration of metallic structures which is caused by the chemical reaction of the material with the environmental conditions 26 the extreme environmental conditions in aircraft operation such as the freezing conditions in conjunction with extreme heating of some parts exposes the aircraft structure and components to a variety of corrosion mechanisms 25 such as scenario includes additional criteria on the aircraft materials selections where the design can be performed while also considering the corrosion resistance 24 however achieving such corrosion resistance in conjunction with the critical performance aspects such as stiffness weight and strength lead to the multi objective problem of optimization in the material features that may not be completely fulfilled with a single material to that extent the corrosion detection on the structures and components have been investigated 24 in general the detection is based on vision systems which have drawbacks on the analysis of inaccessible areas for this reason the use of environmental sensors for indirect corrosion detection where the environmental or local conditions can be used to correlate the corrosion parameters such as location time and rates 27 in this case environmental parameters monitoring namely temperature humidity and even chemical compounds concentration can be associated with corrosion as well as its early detection or estimation 28 despite the influence of the atmospheric pollutants and compounds such as acids sulfates and acid chlorides as well as sea salts diffused into the moisture moisture detection and temperature play a critical role in the corrosion analysis 24 in the dynamic measurement of moisture the time at which there are atmospheric conditions for the surface layer of moisture is known as the time of wetness and it is used as a factor for corrosion growth and initiation 28 in addition the temperature is an important parameter in the corrosion analysis since it not only relates to the corrosion initiation but also on the type of the corrosion 24 different temperature thresholds were analyzed in the literature for their correlation with the type of corrosion which indicate the necessity of continuous temperature monitoring in aircraft structures and components especially the ones with direct contact with moisture or atmospheric pollutants 24

it is also important to mention that the climate conditions are important in the aircraft application not only on the external environmental conditions assessment but also the internal environmental condition since thermal comfort cockpit crew members and passengers in the case of commercial flights are important parameters 29 many case exposure long period seated position crew member passenger lead skin maceration general injury due to the microclimate conditions 30 in the literature three regions of thermal comfort in microclimate conditions e interface between the limb and the seat were defined such regions include the comfort temperatures from 29 °C to 34 °C and relative humidity 70% neutral comfort temperature 27 °C 36 °C relative humidity materials 2023 16 1433 7 of 29 below 80% and discomfort temperatures lower than 27 °C or higher than 36 °C with humidity higher than 80% 31 therefore the application of temperature and humidity sensor used important indicator thermal comfort cabin interface limb seat considering necessity temperature and humidity sensors in different positions and scenarios in an aircraft instrumentation figure 4a present schematic representation sensor position significance each application position was already discussed figure 4 schematic representation of sensors positions for temperature and moisture assessment b schematic representation sensor position engine oil pressure compressor pressure b fuel pump pressure for fuel regulation c hydraulics braking system air data measurement in pitot tubes e environmental air conditioning and pressurization materials 2023 16 1433 8 of 29 among different sensor technologies optical fiber based sensors are growing research field in the sensor community due to their advantages such as compactness electro magnetic fields immunity passive operation multiplexing capabilities chemical stability and biocompatibility 32 as mentioned before for these reasons they are used for applications in different areas such as industry 33 shm 34 biochemical 35 and medicine 36 a high versatility is found in of since many approaches were employed throughout the year where the intensity variation 37 fluorescence absorbance 38 long period gratings 39 fbgs 14 40 non uniform gratings 41 nonlinear effects 42 specklegrams 43 interferometer 44 and surface plasmon resonance 45 sensors are generally employed as an important advantage of optical fibers sensing approaches related to their material feature which include a flexibility compactness and chemical stability such sensors are able of being embedded in rigid and flexible structures 46 as well as the integration in different dyes 47 and dopants 48 the embedment or integration of different materials in of is especially important in humidity sensors development using silica optical fiber since such an optical material is not intrinsically sensitive to humidity moisture absorption 49 an interferometer based approach for humidity assessment uses a mach zehnder interferometer mzi fabricated from a taper where a composite film composed of graphene oxide and pva is coated on the optical fiber sensor 50 the applied coating is sensitive to relative humidity variation in which there is a variation as a function of the environmental humidity which leads to the possibility of humidity sensing due to mzi transmitted spectrum variation as a function of the in another interferometer sensor for humidity measurement fabry perot interferometer fpi proposed 51 case interferometer cavity is obtained on the tip of the fiber where a thin film 168 nm 3.5 thickness additional humidity sensitive film 1621 nm thickness made sio₂ which is enclosed with another thin film with 168 nm thickness thus the 2.3.5.3.5 films are used as reflective surfaces to create the fpi cavity where the sio₂ film presents 2 variations in the refractive index as a function of the relative humidity where such variations lead to a wavelength shift on the fpi reflected spectrum similarly the use of extrinsic fpi for humidity measurement can also be achieved using optical adhesive or polymer films that present swelling with the moisture absorption which also lead to spectral variations of the fpi 52 considering the fbgs sensors developments in humidity assessment the use of silica optical fibers is also related to the optical fiber coating with different humidity sensitive material 53 to that extent the fbg can be coated with polymer films with humidity sensitivity where the film swelling due to the moisture absorption leads to an increase in the strain on the fiber which leads to a Bragg wavelength shift in the fbg in this case use of pva composite was investigated in 54 for the humidity sensitivity using the aforementioned principle it is also worth noting that the cladding removal of the fbg using chemical processes such as etching 55 results in a sensitivity of the fbg with the external air

which lead to the possibility of using thin films with variation as a function of the relative humidity for FBG based humidity moisture assessment approach
 in contrast with the humidity sensing principles using silica optical fibers the use of polymer optical fibers (POFs) provides advantageous features in some applications due to their inherent sensitivity to humidity 56
 for this reason intensity variation based sensors for humidity assessment were evaluated using the polymer swelling and material features dependency as a function of the humidity as the sensing principle 57 in addition
 advances in polymer processing and major breakthroughs in FBG inscription led to the development of FBGs in POFs these so called POFBGs 58 in such developments the use of FBGs in polymethyl methacrylate (PMMA) POFs results in a sensor intrinsically sensitive to relative humidity and moisture where there is no need for coating the optical fiber with sensitive materials 59
 it is also important to mention that etching treatments as well as the diameter reduction of POFs can lead to the improvement on the response time of these sensors in which the real time moisture and humidity assessment can be achieved 60 materials 2023 16 1433 9 of 29 furthermore heat treatment applied POFs obtain insensitivity to temperature variations as well as hysteresis reduction in order to extend the performance of FBG based humidity sensor 61 moreover flexibility of POFs fabrication resulted in the possibility of developing POFs with different transparent polymers 62 63 for this reason POFBGs with tailored properties were developed to achieve high humidity sensitivity using intrinsic POFBG sensors 49
 another critical parameter environmental sensing temperature sensor mandatory for such assessment if the aircraft applications are considered there is a high range temperature sensor temperature 1000 °C regions close to the engines and thermal equipment and temperatures below 0 °C for structural analysis for in flight conditions 64 in both scenarios high and low temperatures the POFs were already employed where the stability of material properties at low temperatures already shows the possibility of using such fiber even in cryogenic applications 65
 considering the low temperatures obtained on in flight conditions conventional optical fiber interferometers 66 distributed temperature sensing 67 and FBGs 68 can be used using their intrinsic sensitivity to temperature variations along the fiber in addition
 embedment of the optical fibers in different materials with high thermal expansion coefficient can extend the temperature sensor performance especially in terms of sensitivity and resolution 69
 if high temperature applications are concerned applications with temperatures close to the glass transition temperature or the processing temperatures of glass material silica optical fibers need the evaluation of the sensors due to the variations in silica material feature temperature 70 extent application sensor based
 fluorescence was proposed using the optical fiber coating in different photoluminescent material such as yttrium aluminum garnet (YAG) sapphire and MgAl due to their 2–4 resistance to high temperatures 71
 the fluorescence intensity ratio is commonly used application photoluminescent material present fluorescence function of the temperature at a specific wavelength 72 in this approach a ratio between intensity wavelength fluorescence occurs intensity reference wavelength without the fluorescence is obtained and analyzed as a function of the temperature 73
 another important breakthrough in high temperature development is micromachining especially using the femtosecond (fs) laser for the fabrication of micro structured devices in optical fibers such as interferometers 74 these devices can be used in high temperatures but below the silica optical fiber processing temperatures in addition the encapsulation of different material well use air cavity extend temperature application range of such sensors device 75 in this approach there is the application of sapphire wafers for the micro cavity development 76 thus the use of sapphire optical fiber generally used sensor device high temperature assessment due to temperature resistance that makes them suitable for temperature applications in a range higher than 1000 °C since the melting point of such materials is around 2045 °C 71
 the temperature sensor applications using optical fiber based approaches are generally related to FBG sensors due to their inherent sensitivity to temperature variations general
 the FBGs are inscribed using UV lasers with holographic interferometric phase mask techniques 77 which typically result in type I gratings that operate in temperatures below around 450 °C since high temperatures lead to the erasing of the grating 78 address this issue in high temperature operations

the direct inscription using fs lasers result in the possibility of using such sensors in temperatures close to the ones of the material processing 79 in addition the annealing treatments in silica optical fibers for grating regeneration lead to changes in the fb g that enable its applications in temperatures higher than 1000 c due to the changes in the optical fiber material and grating structures 78 another straightforward approach for fb g based high temperature sensing is the use of sapphire optical fibers due to their high temperature resistance to that extent fs lasers were used in the fb g inscription in sapphire fibers using direct inscription 80 and phase materials 2023 16 1433 10 of 29 mask 81 inscription method result temperature sensor able to withstand temperatures higher than 1500 c 3 2 pressure sensing pressure assessment is critical in different fields for structural condition monitoring environmental assessment control units and health monitoring therefore pressure assessment is used in applications ranging from industrial measurements 82 to medicine 83 and biomechanics 84 just to name a few in aerospace pressure sensing can influence the predictive maintenance and optimization of its costs 85 structural health monitoring of aerospace and aeronautics assets 86 fuel economy and even in the flight navigation 87 the pressure measurements in engines stages of aircrafts are an important field of investigation where the pressure assessment in turbine airfoils as well as the effect of position dynamic variation aerodynamic structure obtained 64 addition the pressure assessment in the cabin and cockpit is related to the safety of the crew members and early detection of components malfunction 23 figure 4 b presents the schematic representation of the regions and components of aircrafts for pressure sensors application where it is also important to mention that some of these regions are subjected to high temperatures up to 800 c and such issues also increase the demands of pressure sensors for high temperature operation 87 considering the applications and physical properties related to the pressure sensing turbulence important phenomenon unsolved feature quantitative prediction of structures under turbulent flow 87 in this case there are variations in the pressures and velocities which need to be dynamically and precisely measured since they result in the possibility of obtaining temporal and spatial variations related to the reynolds number in turbulence 23 despite the large dimensions of aircrafts the aerodynamics of their components generally result in a microfluidics study since there is the necessity of thin boundary layers investigation to evaluate critical effects such as flow separation and friction drag 87 the pressure assessment in such small layers is important on the assessment of such effects for proper design and mitigation the optical fiber based sensors for pressure sensing generally employ the advantages small dimension flexibility multiplexing capability device embedment different structure considering variety geometry configurations 62 in this context the assessment of mechanical parameters e g pressure force and displacement usually requires the integration of the optical fiber sensor in different structure which include cantilevers 88 diaphragm 89 or platforms 90 in general diaphragm embedded structure lead to a compact sensor device with the possibility of customizing the sensor performance using different diaphragm materials geometry assembly methods 91 in the diaphragm configuration there are two major geometric assemblies for the optical fiber integration in such scenarios where the diaphragm can be positioned on the tip of the optical fiber 92 perpendicular configuration or along the optical fiber parallel configuration 93 considering the case with the diaphragm on the tip of the optical fiber the advantages are the possibility of a miniaturized sensor development and the possibility of high resolution measurement where the diaphragm has the same dimensions as the cross sectional area of the optical fiber 94 in this configuration the intensity variation based sensors and interferometer especially the fpi are used with micromachined flexible diaphragms positioned in a hollow structure 91 in the case of fpi sensors using the diaphragm on the tip of the optical fiber there is an extrinsic fpi formed in the region between the optical fiber tip which is used as a reflector and the tip of the diaphragm that can also include reflective surface 95 case sensor sensitivity related diaphragm mechanical properties since the spectral variations on the fpi are due to the cavity length variation diaphragm deformation thus use flexible diaphragm elastomers or other materials lead to a highly sensitive device as discussed in 96 make suitable measurement

small pressure variation addition the materials 2023 16 1433 11 of 29
 possibility of increasing the dynamic range as well as the possibility of measuring higher pressure
 is achieved by optimizing the diaphragm material properties which make it suitable for gas pressure sensing 92
 despite the difficulties and higher demands on the fabrication tolerance of such small diaphragms
 the configuration using diaphragm on the
 tip of the optical fiber also inhibits the multiplexing capability of the OFS using a single fiber cable
 since there is only one diaphragm at each fiber 91 in another configuration for optical fiber pressure sensors
 the diaphragm positioned along the optical fiber enables the development of diaphragm
 embedded OFS based on intensity variation 97 interferometer 98 and FBGs 99 due to larger dimensions of
 the diaphragm in this configuration the fabrication tolerance is smaller when compared diaphragm tip
 optical fiber multiplexing capability favorable since many diaphragms can be employed along the optical fiber
 100 it is also important to mention that such a configuration leads to the possibility of positioning the
 optical fiber in different regions of the diaphragm considering different planes and related
 to the position on different thickness 101 as well as the transverse area positioning fiber edge diaphragm 102
 reason possible use this approach on the development of distributed systems for density profiling 103
 pressure mapping 104 the operation principle of such sensors is based on the diaphragm strain due to pressure
 applied sensor assembly transmitted optical fiber leading to spectral variations in the sensors 82
 to that extent not only the diaphragm properties are important on the sensors performance
 but also the optical fiber mechanical properties
 since the sensor is based on the strain transmitted to the optical fiber thus
 the use of OFS generally leads to higher sensitivity and resolution in the pressure sensor due to their lower Young's
 modulus when compared with conventional silica optical fibers 99
 it is worth mentioning that such pressure sensors irrespective of the configuration
 are generally sensitive to temperature variations not only due to inherent temperature sensitivity of the OFS
 but also the thermal expansion and mechanical properties variations of the diaphragms 105 in addition
 there is a temperature variation on aircraft applications as mentioned above
 which increases the demand for temperature insensitivity on the OFS pressure sensing application reason
 different temperature compensation techniques have been proposed throughout the literature 106
 such techniques include the use of a temperature sensor without pressure sensitivity to obtain a temperature ref-
 erence system which is compared with the results of the pressure sensor using the direct
 difference between both sensor signals considering their sensitivities this approach can
 be used with different OFS approaches including FBGs uniform and non uniform 107 different interferometers
 108 and intensity variation based sensors 109 furthermore
 the use of mechanical structures for the development of temperature insensitive pressure sensor
 which include the application of a metallic sheet in the diaphragm region for the
 positioning of the temperature compensation reference system 110 it is also possible to
 position two FBGs in the same diaphragm for the simultaneous assessment of pressure and temperature
 where the sensor system is characterized as a function of the pressure
 and temperature prior to its application in a real scenario in which the temperature and
 pressure are simultaneously varied 111
 the intrinsic or extrinsic interferometric cavities along an optical channel generate an interferometric sensor 51
 112 interferometric sensors with practical applications include FPI sensor low coherent interferometric
 sensor called SOFI interferometric sensor 113 and FPI sensor may have a resolution as high as 0.15 $\mu\epsilon$
 a strain measurement range 1000 $\mu\epsilon$ may be expanded 5000 $\mu\epsilon$ capacity function
 temperatures ranging from 40 °C to 250 °C FPI sensors are extremely compact ranging
 in length from 1 mm to 20 mm and can be incorporated in certain structural components
 without incurring any weight penalty or negative impacts however its slow multiplexing
 capacity is a disadvantage materials 2023 16 1433 12 of 29 stated SOFI interferometric sensor successful
 low coherent interferometric sensor structural health monitoring SHM
 successfully placed in hundreds of structures including bridges building oil pipes tunnel
 SOFI interferometric sensors are long gauge sensors in contrast to FPI sensors measurement range beginning
 0.25 extending 10 even 100 m with a micrometer level resolution and temperature insensitivity
 high precision and stability however they can only measure elongations and contractions at low speeds 0.1 Hz

1 Hz and are unable to detect impact damage in aircraft structures

there are three types of distributed fiber optic sensors Rayleigh based optical time domain reflectometry (OTDR), Raman based optical time domain reflectometry (ROTDR) and Brillouin based optical time domain reflectometry (BOTDR). OTDR is the first generation of distributed fiber optic sensors employing Rayleigh scattering to reflect the attenuation profiles of long distance optical fiber networks [114]. An optical pulse is introduced into an optical fiber link and the power of the Rayleigh backscattered light is measured by a photodetector as the light pulse propagates along the fiber link. This measurement is typically used to determine fiber loss and break locations as well as to evaluate splices and connectors in recent years. ROTDR and BOTDR have been utilized for distributed sensing applications. Their operation methods rely on the nonlinearities of optical fibers which generate additional spectral components. These additional spectral components are impacted by environmental conditions external to the system. Consequently, changes in external measurement can be determined by evaluating the spectral content appropriately. ROTDR is based on the Raman scattering phenomenon which generates both anti-Stokes and Stokes components [115]. As the fiber connection itself is the sensor, the intensity ratio between these two components can provide temperature information at any point along the fiber link. Since the amplitude of the Stokes components is independent of temperature, ROTDR can only measure temperature with a temperature resolution of 0.2 °C and not strain. As a spatial resolution of 1 m, the sensing distance of ROTDR is typically restricted to around 8 km. In BOTDR, light is partially scattered back based on Brillouin scattering phenomenon [116]. BOTDR can monitor both temperature and strain since the frequency of the scattered light is independent of the temperature and strain applied to the fiber link. The basic BOTDR measurement distance is 30 km and can be expanded to 200 km. The resolution ranges from 1 to 4 m for the purpose of increasing the service life of aging airplanes. The SHM of damaged aircraft panels fixed with bonded patches has garnered considerable interest using FBG sensor position form fatigue crack disbond front identified aircraft panels fixed with double sided bonded patches [117, 118]. The specifications, sensor performance and other technical information are shown in Table 1. Table 1: Performance evaluation of selected fiber optic sensor technologies for aircraft monitoring. SOF sensors [112]. Interferometric OTDR [114], ROTDR [115], BOTDR [116], FBG sensors [117], sensor [113]. Point sensor type: point, long gauge, distributed, distributed, distributed, semi distributed, temperature, temperature, main sensing, strain, deformation, fiber loss, temperature, strain, parameter, rotation, strain, break, temperature, strain, rotation, force, location, pressure, pressure, quasi parallel, parallel, multiplexing, time, time, distributed, distributed, distributed, wavelength, division, division, division. Materials [2023, 16, 1433, 13 of 29]. Table 1: Cont. SOF sensors [112], interferometric OTDR [114], ROTDR [115], BOTDR [116], FBG sensors [117], sensor [113]. Depending on the: depending on the, depending on the, measurement point, 1, 1, range and, range and, range and, 10, 50, in online, resolution, resolution, resolution, typical resolution, 0, 15, 1, n, n, 20, 1, strain, μ strain, temperature, c, 0, 1, n, 0, 1, 0, 2, 0, 1, capability for large wavelength shift, yes, yes, detection, 10 nm, spatial resolution, 0, 1, 0, 1, 1, 10, 1, 1, 0, 1, capability of fast response for acoustic signal, yes, yes, detection, 100 kHz, linearity in response, infinite, infinite, accurate, high, long gauge, sensing, sensing, advantage, sensitivity, high spatial, wide applications, point, point, high, resolution, accurate, resolution, fiber, fiber, inherent, integrated, integrated, WDM, encoding, low speed, detection, temperature, disadvantage, single point, 10, limitation, cross sensitivity, cross sensitivity, high cost, 4, sensors for navigation, inertial navigation systems, INS are important systems for aircraft navigation that can also apply to personal navigation, car navigation, and unmanned aerial vehicles [119]. The continuous advances in autonomous vehicles and navigation systems placed demands: compactness, well, precision, navigation system [120], demands lead to developments in fiber optic gyroscopes, fog, and general MEMS in development, system, small, error, position, attitude, due, reduction of the sensors uncertainties and nonlinear signal processing approaches [121]. In this context, calibration methods and automatic error corrections increase the accuracy and general performance of INS as another common approach for the development of reliable navigation systems. The INS can be integrated with the global positioning system (GPS) where the GPS enables the calibration and reduction of bias in INS [119]. In addition,

the fusion between the INS and GPS enables the tracking performance of the GPS [122]. Such improvement is achieved using error calibration techniques based on feedforward or feedback methods. Moreover, the integration can be achieved by means of fusing only the GPS for the position and velocity calculations, whereas the navigation filter estimates the position, velocity, and attitude from the INS and the GPS. Position and velocity data are used as the reference for calibration of the INS [119]. Figure 5 presents a general schematic about inertial navigation and global positioning systems on an aircraft. One of the sensors for the navigation system is the displacement sensor, where conventional linear variable differential transformers (LVDT) and rotary variable differential transformer (RVDT) are employed [23]. Displacement sensor capable of operating in high temperature ranges from below 0 °C to higher than 200 °C and of the possibility of positioning in different regions of the aircraft, but at the cost of complex signal processing, to that extent, the development of optical fiber based displacement sensors can address some of the issues of the electronic sensors, such as magnetic field immunity and potential applications in high temperature conditions [123].

Materials [2023, 16, 1433, 14 of 29]. Figure 5. Inertial navigation system and global positioning of an aircraft. Among the sensors for the inertial navigation systems, the continuous improvements in gyroscope technology, rotation measurement, critical evolution, navigation system [124]. Case rotation rate indicates variation in heading and attitude of the system. Moreover, the accelerometers are also important in the navigation system for the assessment of acceleration amplitude and direction. If the gyroscopes are considered, such devices are positioned on a frame or mechanical structure to obtain the angular velocity of the rotating structure [125]. A typical classification of gyroscope includes mechanical gyroscope, optical gyroscope, MEMS gyroscope [125]. However, we focus on the optical gyroscopes in this review. The development of the optical gyroscopes is motivated by sequential breakthroughs in optoelectronic technology [126].

include widespread optical fiber technology, fiber angular velocity sensor using optical fibers, where the Sagnac effect is used for the assessment of the rotational rate by considering the advantages of high sensitivity, temperature and pressure resistance, in conjunction with the well-known electromagnetic field insensitivity [127]. For these reasons, fiber gyroscopes are widely used in general navigation systems, especially in aviation [128] and aerospace [129]. Application: another approach for fiber gyroscopes is based on the resonant structures, resulting in the so-called ring fiber gyroscopes [124]. In the last years, the use of novel optical fiber structures, e.g., the use of photonic crystal fibers (PCFs), has accelerated in order to obtain structural birefringence sensing structure [130]. Hollow core PCF used for fiber gyroscopes even result in smaller cross-sensitivity function, temperature, conjunction, lower backscatter and nonlinear effect, which increases the polarization noise controllability of the sensing structure [131]. For these reasons, the use of optical components in the coupling hollow core PCF resonator structure proposed [131], resulting in high temperature stability of the device. In addition, a high finesse can be obtained in the resonant cavity using a PCF coupled with a single mode fiber, resulting in a hybrid PCF resonator structure, where such a structure resulted in a gyro bias stability of 0.5 [132]. Also worth noting, use hollow core PCF gyroscope prototype reduces the polarization crosstalk of the structure [133]. Furthermore, the use of hollow core anti-resonant fiber (ARF) structure resulted in the extended performance of the fiber gyroscopes with the possibility of using a frequency differential ring structure to obtain a stability as high as 0.05 h [134].

optical fiber based accelerometers have even higher developments and applications since used in navigation system, also structural health monitoring [2023, 16, 1433, 15 of 29]. Monitoring [135] and physiological parameters monitoring [136]. To that extent, the optical fiber based accelerometers were developed using different approaches, as thoroughly discussed in [137]. The simplest approach for optical fiber accelerometers is the use of the intensity variation principle, where the transmitted optical power variation is analyzed as a function of the acceleration [138]. In this case, the sensor can operate in the coupling principle, where a fiber is positioned in a light source, whereas another fiber is positioned in a photodetector [139]. If this configuration is used, one of the fibers is isolated from the vibration/acceleration variation, and the other end is connected to the system with proof mass for the vibration transmission [140].

Some important drawbacks of this approach are that the sensor is sensitive to environmental variations and presents low precision due to the high sensitivity to misalignments. As higher precision for optical fiber based accelerometers

the interferometric based accelerometer developed using cavity generally based fpi 137 general cavity based accelerometers use a movable structure connected to the cavity in which the cavity length varies as a function of the acceleration in this case a sub nanometer potential resolution achieved using configuration 141 application micro optical electro mechanical systems moems resulted in the novel configurations for the optical accelerometer that can further increase the accelerometer performance 142 in this approach there is a microscale proof mass and a silicon frame to create the accelerometer structure where the displacement and acceleration are obtained from the spectral features variation of the cavity with the possibility of tuning the accelerometer parameters such as proof mass and stiffness to achieve high resonant frequencies and small noise in another moems accelerometer the bioinspired shape was proposed in 143 where a battery was used as proof mass and the sensors were embedded in a transparent web like structure for movement analysis and highly sensitive displacement measurements one of the most common optical fiber based approaches for accelerometers development is the integration of fbgs in mechanical structures 144 in this context many different approaches using single 145 and double 146 cantilever as well as diaphragm 147 and flexible hinge 148 structure were proposed this approach is based on the strain produced in the fbg due to the inertial displacement of the proof mass under acceleration 149 for this reason it is possible to develop 2 and 3 axis 150 accelerometers using different assembly conditions and using the multiplexing capabilities of the fbgs thus it is possible to develop fbg based accelerometers for multipoint measurement which play an important role not only in navigation systems but also in structural monitoring in aircrafts furthermore such devices are able of measuring tilt angles by means of embedding in mechanical structures and using different types of fbgs such as the tilted fbgs 151 can measure such parameters with the additional advantages of self referencing signals using the spectral features of the gratings such gratings based devices can also include additional information for the navigation systems and enable novel data fusion approaches for ever higher accuracy and reliability of in flight data 5 pilot vital health monitoring there are over one million active pilots in the world with over two thirds residing in the united states the average age of pilots is increasing across all aviation industries due to aging pilot over the age of 50 may be more prone to accidents than pilots in their 30 in addition to mechanical and system failures certain examinations have shown that pilot error is a major cause of accidents as a result the flight safety foundation identified cognitive health as a crucial component in pilot safety subsequently risk classification and user experience outcomes validate an immersive virtual reality vr cognitive screening and intervention tool for elderly pilots 152 this vr simulation tool serves as the basis for this cognitive evaluation since it provides a portable cost effective and dependable approach for evaluating pilot cognition 153 today aircraft engine flight control system equipped number sensing unit determining pilot normal health condition 154 157 recent materials 2023 16 1433 16 of 29 variation of the infectious severe acute respiratory syndrome coronavirus 2 sars cov 2 cause coronavirus disease 2019 covid 19 pneumonia afflicted entire world it mostly affects the aviation industry and poses a severe risk to pilots numerous problems are observed in pilots infected with covid 19 including neurologic issues during the acute phase following recovery and even after immunization covid 19 patients reported variety neurologic complication including encephalitis encephalopathy stroke headache loss of smell and taste dizziness seizure refractory status epilepticus myelitis myopathy acute disseminated encephalomyelitis leukoencephalopathy kawasakis syndrome guillain barré syndrome and neuroleptic malignant syndrome 158 in a recent case study a 43 year old helicopter pilot was sent to the emergency department due to influenza like symptoms after receiving treatment for covid 19 recovered and returned to flying during flight pilot experienced moments of dizziness and unconsciousness although the co pilots salvaged the helicopter it has been determined that robust sensory elements are required to identify the pilots present health condition during flight operations 158 monitoring the pilot's vital signs is essential for ensuring the safety of an aircraft and its occupants in this instance a cockpit equipped with a health monitoring system provides a means for both the occupant and the ground station to be aware of the pilot's normal health state throughout

flight multiple type sensor pulse oximetry sensor
 are positioned at a preset area of the pilot seat that is within reach of the occupant
 the sensor for a pulse oximeter is put at a preset stable point for the insertion of a finger
 which offers a satisfactory response to the sensor while the individual is seated the sensor linked
 communication link via control box aid data processing
 presents the data in an appropriate format on the cockpit multifunctional display system throughout the flight
 the pilot can enter his or her finger into the sensing unit to monitor
 his or her pulse oximetry and view the parameters on the multifunctional device similarly republic singapore air
 force rsaf adopted search rescue sar heli medevac
 based services in 1971 for the singapore flight information region in which
 the flight doctor and medic are supplied for medical care on board the helicopter 159
 researchers from the united states air force are investigating methods for determining the health of pilots in high
 performance aircraft they intend to develop a technology
 that monitors and notifies pilot of potential problems with physiological states that affect
 pilot performance during flight such as a sudden drop in oxygen concentration in this work
 emphasis must be placed primarily on integrating and experimenting with existing sensor technologies
 as well as doing the evaluation researchers are interested in inte
 grating sensors and monitoring many parameters such as flight environment monitoring sensor fusion
 monitoring of pilot vital signs and respiratory function data storage and processing onboard analytics
 and pilot alerts to develop integrated sensors multiple hardware components are necessary
 the employed hardware is small lightweight self
 powered enough to be easily integrated into the cockpit of an aircraft this type of
 hardware system consists of smartphones tablet computers and embedded computing
 that facilitate wireless networking such as bluetooth low energy the in line sensor tech
 nologies were developed for measuring the air quality and supply the aircraft life support system
 the air quality consists of the appropriate level of carbon dioxide oxygen flow pressure similar impurity
 similarly pilot vital health monitoring system incor
 porates the sensor system to provide direct or indirect evidence of pilot blood or tissue oxygenation
 blood circulation peripheral capillary oxygen saturation spo2 respiration rate
 and estimated core temperature u air force scientists also developed a unique
 cockpit sensor for the measurement of the air any difficulty in the abilities function and alert
 to the pilot to behave properly during flight in high performance aircraft researchers are
 trying to develop and showcase existing sensor technologies to help keep pilots functioning feeling good
 cockpit looking self contained self powered sensor hardware that is small enough to fit into airplane cockpits
 materials 2023 16 1433 17 of 29 the international civil aviation organization icao
 defines fatigue as a physiological
 state of reduced mental or physical performance capability caused by sleep deprivation protracted awake
 circadian phase workload mental physical activity impairs pilot attentiveness ability perform safety
 related operation correctly 156 a proper pilot monitoring system in an aircraft consists of the plurality sensor
 arranged monitor several health parameter pilot instance
 aircraft state monitoring system is used to monitor the flight situation data of the aircraft
 and the analysis system is used to determine the incapacitation level of the pilot for health parameter thereafter
 human machine interface utilized to interface with one or more
 processor to offer an interface between the pilot and pilot monitoring system
 human machine interface help to notify the pilot based on the determined incapacitation level
 as illustrated in figure 6a g the system hardware design includes a controller box printed circuit board pcb
 textile electrode and electrode button on the surface of human skin 153 figure 6 seat belt point
 care system in an airplane illustration of four developed chest band layered hardware structure b
 the produced chest band device c difference in aspiration
 between normal breathing and sleep apnea produced by airflow photographs of a human person
 wearing a chest belt e a flow chart of the monitoring system operation including hardware and software f
 neural network used classifier system network red node represents the input layer
 the blue nodes represent the hidden layers and the gray nodes represent the output layer g

diagram of an airline point care monitoring system in the user client
 bluetooth data transmission to the mobile terminal for hybrid physiological signal detection adapted 160 6
 novel nanomaterials for aerospace applications due to their unique properties such as high aspect ratio
 large surface area tailorable electrical and thermal conductivities high anisotropy
 and distinctive optical properties the synthesis and characterization of nanomaterials 100 nm size
 have been the primary focus of academic private and government organizations for the past two decades 161
 these characteristics lead to the development of lightweight high strength multipurpose structure
 efficient energy storage systems strong radiation shielding and cutting edge optical fiber sensors 162
 trying to align the reason for biomolecular research based on nonmaterial assisted
 optical fiber sensors help the reader in understanding the need for sensor development for materials 2023 16
 1433 18 of 29 aeronautical applications one of the most severe human diseases diabetes can be caused
 by high blood glucose levels 163 diabetes is characterized by elevated blood sugar levels
 that result in consequences such as renal failure and diabetic retinopathy 164 165 india diabetes diagnosis
 the food industry biotechnology and aerospace glucose sensors play a significant role in these works
 gold nanoparticles graphene oxide and m-xenon materials were primarily used to develop optical fiber sensors for glucose biomolecule detection acute myocardial
 infarction (AMI) emerged leading cause death global scale work cerium oxide nanoparticles used develop
 optical fiber sensor for CTN detection 166 within 1.5 to 3 h of symptom onset acute myocardial infarction cause
 irreversible myocardial damage blood troponin release important biomarker AMI cardiac troponin (cTnI)
 creatinine byproduct human muscle metabolism and an important clinical indicator of diabetes kidney disease
 kidney failure and muscle wasting 167 168 to lower the incidence of renal disease
 early detection and prevention of elevated creatinine levels are crucial these sensor probes feature graphene
 oxide gold nanoparticles molybdenum disulfide nanoparticles niobium carbide m-xenon materials
 cholesterol is an essential part of the human body created liver also part healthy diet precursor
 extremely vital biological substance abnormal level of cholesterol in the blood can lead
 to excessive blood pressure heart disease coronary artery disease cerebral thrombosis atherosclerosis
 and a number of other common conditions 169 170 these optical fiber
 sensor probes are developed utilizing gold nanoparticles silver nanoparticles and zinc oxide nanoparticles
 alanine aminotransferase (ALT) is a serum aminotransferase whose activity must be determined to diagnose liver
 disease study 171 molybdenum disulfide nanoparticles cerium oxide nanoparticles combined
 increase the performance of an optical sensor when certain metabolic conditions such as obesity high fat diabetes
 and other symptoms are present in the body clinical trials and laboratory
 tests have demonstrated that ALT levels may be somewhat raised in addition alanine
 transaminase activity leads to somewhat increased alanine transaminase levels in a range of muscle disorders
 including viral hepatitis and muscular dystrophy uric acid (UA)
 an extremely significant component in human serum and urine 172 the concentration of UA in blood increased
 lack exercise poor diet improper drug 173 excess uric acid produces solid urate crystals in the body leading
 to catastrophic conditions such as gout and kidney stones 174 similarly wearable sensors are ground
 breaking health monitoring gadgets that enable continuous monitoring physical biological characteristic
 recent paper 175 detailed the development of innovative optical sensors for wearable vital health monitoring
 device detailed discussion substrate sensor platform biofluids utilized for the detection of target molecules
 in this way low cost wearable technologies could improve the quality of health monitoring systems and permit continuous and early disease diagnosis in aerospace applications in addition
 and not less importantly it is obvious that advanced materials such as
 thermal protection systems based on carbon carbon composites and turbine blades based on metal alloys
 play a significant role in aircraft applications multiple industries have
 begun to develop improved aircraft materials single crystal nickel based turbine blades
 and aluminum bulk alloys are examples of applications for nanomaterials research and
 development evolved and broadened its application to encompass among others radiation protection
 thermal protection structural nanostructures space propulsion electronics sensor numerous applications
 carbon nanostructures nanotubes CNTs graphene and inorganic nanomaterials are used
 silica and metal oxides subsequently customized nanoparticles enabled significant improvement
 structural non structural components of virtually all space and aviation systems it primarily provides a

reduction in weight maintenance of mechanical strength increased adaptability multi component monitoring storage and transmission efficient power production greater radiation protection and long term life support for exploration the advances in the synthesis and materials 2023 16 1433 19 of 29 characterization of carbon nanostructures cns and carbon nanotubes nanosheets cnt nanosheets produced carbon fiber provide new opportunity using multi functional features of these materials in aviation components several scientific studies suggested that cnt based coatings film and sheets facilitate electrostatic dissipation and electromagnetic interference shielding with low effort 176 there are vacuums severe temperatures space debris micrometeoroid and significant variance owing sunspot activity space environment criterion determine the design and construction of aircraft systems and spacecrafts primarily system performance degrades when exposed to atomic oxygen the special materials used in aircraft applications can withstand high temperatures and reduce the erosion yield factor 177 due to high phase temperatures aeronautical structures require materials with high strength and stiffness for mechanical aspects 178 this can be accomplished by the use of composites containing a blend of graphene and epoxy resin as a curing agent to produce desired result aeronautical application boost base material strength and temperature resistance hence enhancing the material tribological behavior thus improved composites contribute to the development of lightweight aircraft constructions with corrosion and fatigue resistance 179 carbon glass fiber composite also utilized production sensor mounted on aircrafts shm 180 prior to that in 1940 Bower was the first person to employ the nanocomposite technology 180 typically nanocomposite consists of nanoparticles embedded in a reinforcing metal matrix material composed of aluminum and boron 181 this technology introduces the qualities of robustness portability and affordability compared copolymer nanocomposites greater pyroelectric coefficient due low density strong damage resistance material also offer lengthy service life the behavior of tribology in aeronautical constructions is determined by severe temperature in these nanocomposites the function of aluminum particles is to minimize the resistance to wear and friction nanocomposite materials such as alumina and polytetrafluoroethylene can be utilized to optimize performance existing advanced structure carbon nanotube cnt carbon nano bead multi walled carbon nano tube mwcnt carbon nanorods diamond like carbon carbon nanofibers and carbon nanocones are present in the nanocomposite materials 182 the performance of these materials exceeds that of conventional material mwcnts cnts graphene oxide and polymer clay nanocomposites are the most important nanocomposites for aerospace industries 183 184 as shown in figure 7 figure 7 properties of nanomaterials for aerospace applications adapted from 183 materials 2023 16 1433 20 of 29 similarly molybdenum disilicide nanoparticles combined with an aluminum matrix play a significant role in aerospace and provide wear resistance to prevent degradation 185 through the use of nanocomposites based on high strength titanium based materials possible achieve superior aerospace quality nanocomposites graphene oxide and titanium nanopowders can be employed as the matrix system reinforcement in order to offer a high hardness which is often used in a number of aerospace structural component 186 using laser sintering the graphene may be simply and rapidly dispersed throughout the matrix due to the development of multifunctional materials space exploration has accelerated in recent years using polymer nanocomposites with cnts sheet reinforcement vibration damping factor decreased mwcnts distinctive electrical mechanical and thermal qualities that are advantageous for aeronautical applications 187 as shown in figure 8 figure 8 applications of nanomaterials in the aircraft industry these nanocomposites also enable the aerospace industry to withstand sub zero and high temperature enabling resist severe condition lower earth orbit outer space 188 cnt widely used aerospace application result significant property similarly advancement Al-Cr-Al-Cu-Al coated advanced structures boost heat and adhesive resistance 189 nanocomposites provide the optimal technology for aircraft applications due to the intertwined nature sustainable impact constant generational advancement incorporating nanocomposites into complex aeronautical designs waste creation during the manufacturing process can be reduced this also facilitates the creation of low maintenance

lightweight structures 190 materials 2023 16 1433 21 of 29 7 future perspectives well known enlarged use
 composite structure aeronautical
 dust started over the past few decades with newer aircrafts made largely of composite
 material and produced many new fruitful contributions to the progress of the sea area
 for composite aeronautical and aerospace structures applying shm in the aeronautical
 and aerospace industries have not yet fully matured and their applications to composite
 structures are particularly challenging in terms of a realistic aircraft maintenance scenario requires full
 damage diagnostic four shm level mean damage detection damage localization
 damage type identification and damage severity presently the scientific industrial and end
 user communities generally view optical sensors as a
 technology with the highest potential for the continuous real time monitoring of aircraft structures and
 not less importantly the tracking of some critical biochemical parameters as well as destructive contaminants
 the optical sensors which can include optical fiber based sensors have also found ap
 plications in feeding back real time measurements of weight distribution reliable aviation fuel gauging sensors
 or even water content detection and online monitoring in aviation fuel distributed optical sensor able test
 structural integrity the wings and fuselage following the four shm levels mentioned as well as judge the
 performance engine icing wing loading landing gear importantly the cockpit environment
 not only in commercial aircrafts but also in military one 191 193 in this way
 with the continuous development of autopilot systems and flight assistance systems
 pilot performance and air safety from the cockpit temperature humidity pressure
 and also detecting biological contaminants aboard aircrafts for example need to be improved significantly
 pilot behavior recognition based on multi modality fusion technology 194
 using physiological features acquired online as well as critical parameters like fatigue hypo hyperoxia
 surges of acute stress level or sudden loss in blood flow 161 175 195 197 of pilots is crucial to obtain in real
 time data and process them via an integrated smart tool additional nanofilms nanofibers nanoparticles
 demonstrate variety advanced capability including good thermal electrical property safer coating
 cleaning corrosion resistance and potential toxicity facilities in a variety of disciplines aviation component
 surface coating also safeguard aircraft component from severe hazards in addition
 it improves the unique advantages and performance characteristic
 compared to the standard metals and composites used in the production of many types of aircraft components
 the incorporation of nanomaterial structures and nanomaterial
 based devices into the manufacturing of aircraft helps their maintenance and repair
 consequently reducing the operational costs and it is quite important to consider
 them not only for aircraft components improvements but also to progress in terms of the
 physiological features of the cabin crew pilot and passengers 8 conclusion
 the building blocks of future aircrafts both commercial and military are in place
 allowing for the next phase of aerospace development to attain significantly higher levels
 of sustainability in the aviation industry while reducing its carbon footprint with the help
 of lightweight technologies such as soft increasing public concern about climate change
 coupled with government support of green projects should foster innovation in this field
 and optical technology can be a viable alternative for the continued expansion of such a vital industry
 the employment of soft and innovative nanophotonic materials in aircrafts
 to enhance the safety of the aviation sector as well as the safety of pilots and passengers is strongly encouraged
 as described in the article the development of this industry will
 continue concurrently with the development of technologies fed by the optical domain
 due to their distinct features materials 2023 16 1433 22 of 29 author contributions conceptualization c
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since many approaches were employed throughout the: 0.0041210929631779365
since many diaphragms can be employed along the optical fiber: 0.0041210929631779365
since such an optical material is not intrinsically sensitive to humidity: 0.0041210929631779365
since the amplitude of the Stokes components is independent of temperature: 0.0041210929631779365
since the frequency of the scattered light: 0.0041210929631779365
since the melting point of such materials is around 2045: 0.0041210929631779365
since there is only one diaphragm at each fiber: 0.0041210929631779365
since there is the necessity: 0.0041210929631779365
since the sensor is based on the strain transmitted to the optical fiber: 0.0041210929631779365
since the spectral variations on the fiber are due to the cavity length: 0.0041210929631779365
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whereitisalsoimportantto mention that some of these regions are subjected: 0.0041210929631779365
wheresuchaconcept has been brought under huge affirmation: 0.0041210929631779365
wheresuchastructure resulted in a gyro bias stability of 0: 0.0041210929631779365
wheresuch variations lead to: 0.0041210929631779365
wherethe aeronautical industry needs to align the progress: 0.0041210929631779365
wherethe air data system is responsible to provide the flight: 0.0041210929631779365
wherethedamage may be present inside: 0.0041210929631779365
wherethedesign can be performed while also considering: 0.0041210929631779365
wherethediaphragm can be: 0.0041210929631779365
wherethediaphragm has the same dimensions as the cross: 0.0041210929631779365
wherethedisplacement and acceleration are obtained from the spectral features: 0.0041210929631779365
wherethe environmental or local conditions can be used to: 0.0041210929631779365
wherethe films swelling due to the moisture absorption lead to an increase in: 0.0041210929631779365
wherethe gps enables the calibration and reduction of bias in ins: 0.0041210929631779365
wherethe intensity variation: 0.0041210929631779365
wherethemost capable options are acoustic: 0.0041210929631779365
wherethe pressure assessment in turbine airfoils: 0.0041210929631779365

wherethereisnoneedforcoatingtheopticalfiberwith: 0.0041210929631779365
wherethesagnaceffectisusedforthe: 0.0041210929631779365
wherethesensorsystemischaracterizedasafunctionofthepressure: 0.0041210929631779365
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whichleadstoabragg wavelength shift in the fb: 0.0041210929631779365
whichleadstothe possibility of humidity sensing: 0.0041210929631779365
whichleadstothe possibility of using thin films with variation as a function: 0.0041210929631779365
whichmakeit: 0.0041210929631779365
whichmeans: 0.0041210929631779365
whichneedtobedynamicallyandpreciselymeasured: 0.0041210929631779365
whichoffersasatisfactoryresponse to the sensor while the individual is seated: 0.0041210929631779365
whichplayan: 0.0041210929631779365
which typically result in type I gratings that operate in temperatures: 0.0041210929631779365
while reducing its carbon footprint: 0.0041210929631779365
while this serious change also takes: 0.0041210929631779365
white: 0.0041210929631779365
wide applications: 0.0041210929631779365
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wing shape: 0.0041210929631779365
with adequate physical and spatial: 0.0041210929631779365
with a history of coronavirus disease 2019: 0.0041210929631779365
with a micrometer: 0.0041210929631779365
with bonded patches using fiber Bragg gratings sensors: 0.0041210929631779365
with corrosion and fatigue resistance: 0.0041210929631779365
with distributed fb optical sensors: 0.0041210929631779365
with high thermal expansion coef: 0.0041210929631779365
within 1: 0.0041210929631779365
with newer aircrafts such as Boeing 787 and Airbus A350 made: 0.0041210929631779365
without incurring any weight penalty or negative impacts: 0.0041210929631779365
without the fluorescence: 0.0041210929631779365
with over two: 0.0041210929631779365

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