Design of Shell and Tube Heat Exchanger Semester 5 PBL Heat and Mass Transfer



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Problem Statement

Following are the input physical parameters:

	Fluid	T _i (°C)	T _o (°C)	Mass Flow Rates (kg/s)
Shell Side	Kerosene	200	94	5.52
Tube Side	Crude Oil	38	77	18.8

Shell diameter = 1.5 m

Tube diameter = 0.05 m

Baffle Spacing = 0.5 m

In the design problem you are expected to:

- 1. Calculate the heat transfer
- 2. Calculate overall heat coefficient
- 3. Calculate the area
- 4. Incorporate the fouling factor of a typical STHX with the given conditions
- 5. Find out appropriate number of tubes, length and thickness of the tubes
- 6. Calculate the pressure drop
- 7. Comment on the results
- 8. Propose improvement measures
- 9. Prepare a small report with all the design considerations, calculations and results.

Note: Use appropriate values of the missing or required parameters in the problem.

Properties of Fluids

Physical properties of 42° API Kerosene:

Kerosene	Inlet	Mean	Outlet	Units
Temperature	200	147	94	°C
Specific Heat	2.72	2.47	2.26	kJ/kg°C
Thermal Conductivity	0.130	0.132	0.135	W/m°C
Density	690	730	770	kg/m ³
Viscosity	0.22	0.43	0.8	mNs/m²

Physical Properties of 34° API Light Crude Oil

Crude Oil	Inlet	Mean	Outlet	Units
Temperature	38	57.5	77	°C
Specific Heat	1.93	1.97	2.01	kJ/kg°C
Thermal Conductivity	0.133	0.134	0.135	W/mºC
Density	800	820	840	kg/m ³
Viscosity	2.4	3.2	4.3	mNs/m ²

Assumptions

- 1 Shell and 2, 4, 6 etc (any multiple of 2), tube passes
 - Number of tubes assumed = 6
- Tube Outer Diameter = 10% of Tube Inner Diameter

Fluid Properties

- Thermal Conductivity of Copper Tubes (K) = 385 W/m°C
- $R_{fi} = 0.00035$
- $R_{fo} = 0.00020$
- Specific Gravity of Kerosene (Sk) = 0.785
- Specific Gravity of Crude Oil (Sc) = 0.850

Calculations

I shell and L, Y, 6 etc tube passes of hence,

hence,

n = 26 tubes

Tube!

m = pAnne 820 x T x0.05° x b

= 1,946 ms -1

Pe = NDf = 1.946 x 0.05 x 210

8.2 x 10 -3

= 24938.125 = 0.24983 x 10

(Re > 10000, hence Purbalent
Flow)

Pr = M CP = 8.2 × 10-3 × 1.97 × 10 3 = 0.184

NU2 ND 2 0.023 Re0.8 PIDM

1(0.08) 20.028 (0.249 x105)0.8 (47.04)0.4

11-82 h 2 946.5 W/mt.K

Shell:

hased on type of fluids.

m70.0 = 10 Q = A; U; Nm 1445. L5 x103 = Ai x 525.70 x 8004 Ai = 84.84 m2 Q = A000 A ?m 1448.25 ×103 = A0 × 482.77 × 80.04 Ao 2 37,40 m2 ": Taking Ao as it is highest area.

Ao = NT Pol 37.40 = 67 x 0.06 xL (200 000) ml + L= 198.42m (length of all tubes) 198.42 = 198.42 = 88.07 m 05.757 = 10 Me 6 (length of 1 tube)

Philiceness of tube = Deo - Dei 0.06 - 0.05 2000.0 2001M 210 MM

30.0x 0001 30.0 x 0001

```
Pube Pressure:
                         SLEU DEMINE!
b = 0.4137 Re-0.2885
 2 0.4137 (14988.125)-0.185
Re = 4m (hp/nt) (1000 - 700) × 700× 700
      TOHH
24988.125 = 4(18.0) (np/6)
          x(0.01)(8.2 x10-3)
np21 ~ 2 (even) 1 20 x = 30
42 m (np/nt) = 18.8 (1/6)
             2 8191.59
AP<sub>2</sub> proled disco
    = 0.030L1 x L x 198.42 x 3191.592 28.8 x10-8 Pa
      7. 5 x 1012 x 0.05 x 0.850 x1
 SP8 21.884 ×10-13 (Lnp-1.5) 4 | Sc
     = 1.884 ×10-18 (2(2) -1.5) (3191.59)2 0.850
     2 3.996 ×10-5 Pa 20 × 3 × (14 11) = 91
 DP 2 DP, + DP,
    = (3.8 ×10-3) + (3.996 ×10-1)
    2 3.84 x10-3 Pa Tall y &
```

Shell Pressure!

$$V = \frac{m}{14} = \frac{1.01}{730 \times 0.07} = 0.0133 \text{ m/n}$$

References

$$\dot{Q} = \dot{m}_c C_{pc} (T_{c, \text{ out}} - T_{c, \text{ in}})$$
 (13-9)

and

$$\dot{Q} = \dot{m}_h C_{ph} (T_{h, \text{in}} - T_{h, \text{out}})$$
 (13-10)

where the subscripts c and h stand for cold and hot fluids, respectively, and

 \dot{m}_{c} , $\dot{m}_{h} = {
m mass}$ flow rates C_{pc} , $C_{ph} = {
m specific}$ heats $T_{c, {
m out}}$, $T_{h, {
m out}} = {
m outlet}$ temperatures $T_{c, {
m in}}$, $T_{h, {
m in}} = {
m inlet}$ temperatures

$$\dot{Q} = UA_s \, \Delta T_{\rm lm} \tag{13-24}$$

where

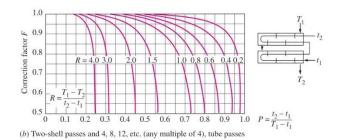
$$\Delta T_{\rm lm} = \frac{\Delta T_1 - \Delta T_2}{\ln{(\Delta T_1/\Delta T_2)}} \tag{13-25}$$

$$\Delta T_{\rm lm} = F \, \Delta T_{\rm lm, \, CF} \tag{13-26}$$

$$P = \frac{t_2 - t_1}{T_1 - t_1} \tag{13-27}$$

and

$$R = \frac{T_1 - T_2}{t_2 - t_1} = \frac{(\dot{m}C_p)_{\text{tube side}}}{(\dot{m}C_p)_{\text{shell side}}}$$
(13-28)



$$\frac{1}{U\!A_s} = \frac{1}{U_i A_i} = \frac{1}{U_o A_o} = R = \frac{1}{h_i A_i} + \frac{R_{f,i}}{A_i} + \frac{\ln{(D_o/D_i)}}{2\pi kL} + \frac{R_{f,o}}{A_o} + \frac{1}{h_o A_o} \quad \text{(13-8)}$$

Tube Pressure Formulae

Process Heat Transfer Principles and Applications Book • 2007

https://www.sciencedirect.com/book/9780123735881/process-heat-transfer

$$\Delta P_f = \frac{f \, n_p L G^2}{7.50 \times 10^{12} D_i s \phi} \tag{5.1}$$

where f = Darcy friction factor (dimensionless) L = tube length (ft) $G = \text{mass flux (lbm/h · ft}^2)$

 $D_i = \text{tube ID (ft)}$

 b_i = tube ID (II) s = fluid specific gravity (dimensionless) ϕ = viscosity correction factor (dimensionless) = $(\mu/\mu_w)^{0.14}$ for turbulent flow = $(\mu/\mu_w)^{0.25}$ for laminar flow

$$f = 0.4137 Re^{-0.2585} (5.2)$$

$$\Delta P_r = 1.334 \times 10^{-13} \alpha_r G^2 / s \tag{5.3}$$

Table 5.1 Number of Velocity Heads Allocated for Minor Losses on Tube Side

Flow regime	Regular tubes	U-tubes
Turbulent	$2n_p - 1.5$	$1.6 n_p - 1.5$
Laminar, $Re \ge 500$	$3.25 n_p - 1.5$	$2.38 n_p - 1.5$

Shell Pressure Formulae

https://www.enggcyclopedia.com/2019/05/shell-tube-heat-exchanger-pressuredrop/#pressure_drop_calculation

Effective Area = Ae = Ds × Bs × (P-Dt) / P where, Ds = shell diameter Bs = Baffle spacing P = pitch (distance between center axes of two adjacent tubes) Dt = Tube diameter $V = Mass \ flow/(\rho \times Ae \times 3600)$ $f_k = 1.79 \times (\rho \times V \times D/\mu)^{-1.9}$ $De = 4 \times (P^2 - (\pi \ Dt^2/4)) / \pi \ Dt$ $\Delta P = (\ (N+1) \times f_k \times Ds \times \rho V^2) / (\ 2 \times De)$ where, N = number of baffles