# Technion - Israel Institute of Technology Faculty of Aerospace Engineering Numerical Methods in Transonic Flows Exercise no. 0

#### **Problem Definition**

Consider the following differential equation:

$$\frac{\partial u}{\partial t} = \mu \frac{\partial^2 u}{\partial y^2}$$

$$\mu = 1.0$$

The equation is obtained by neglecting the convective term in Burgers equation.

## Solution Method

Solve the equation using the following implicit/explicit method:

$$\left(I - \frac{\alpha \Delta t \mu}{\Delta y^2} \delta_{yy}\right) \Delta u_i^n = \frac{\mu \Delta t}{\Delta y^2} \delta_{yy} u_i^n$$

where  $\alpha=0$  corresponds to an explicit method,  $\alpha=0.5$  corresponds to the Crank-Nicolson scheme and  $\alpha=1$  corresponds to a fully implicit (Euler Implicit) scheme. You may use 11 grid points.

## **Boundary and Initial Conditions**

Use the following boundary conditions:

$$u\left(y=0\right) = 0.0$$

$$u\left(y=1\right) \ = \ 1.0$$

and the following initial conditions:

$$u\left(t=0\right) = 1.0$$

### Case Studies

Use the program to solve for the following cases:

- 1.  $\alpha = 0$ , choose  $\Delta t$  appropriately and then verify the stability limit. Examine the stability and convergence behavior as you vary the time step.
- 2.  $\alpha = 0.5$ , choose a small to moderate  $\Delta t$  and a very large  $\Delta t$ . Explain the results.
- 3.  $\alpha = 1$ ,  $\Delta t$  should be very large.

## **Short Report Preparation Instructions**

The report should include the following:

- 1. A short description of the mathematical problem.
- 2. A short description of the numerical scheme and how it is developed.
- 3. Stability analysis of the scheme.
- 4. Results (plots, NOT pages with meaningless numbers).
- 5. Discussion and conclusions.
- 6. The computer program.

In any event it has to be a SHORT report.