Dynamics and Stability of Multiphase Flows $\ensuremath{\mathsf{HW}} 4$

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February 1, 2025

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${\bf Contents}$

1	Velocity Profile $U_{(z)}$ as a Funciton of z	2
2	ODE of Jet Radius $r_{(z)}$ 2.1 Formulate the ODE	
3	Numerical Solver	4
4	Solution	5
5	Compering The Results	5

List of Figures

1	The jet shape	5
2	R_1 as a function of z	5

1 Velocity Profile $U_{(z)}$ as a Function of z

Consider a liquid jet ejected from a circular orifice of radius a (point A). The jet has flux Q of fluid with density ρ that exits with velocity U_0 . The surface tension between the fluid and the air with pressure P_0 is σ .

By assuming laminar, inviscid, incpmressible, axisymmetric and steady flow, we get the Bernoulli equation, between point A and arbitrary point downstream B:

$$\frac{1}{2}\rho U_0^2 + \rho gz + P_A = \frac{1}{2}\rho U_{(z)}^2 + P_B \tag{1}$$

The Laplace pressure for a jet:

substituting into the Bernoulli equation:

$$\frac{1}{2}\rho U_0^2 + \rho gz + \mathcal{P}_0 + \frac{\sigma}{a} = \frac{1}{2}\rho U_{(z)}^2 + \mathcal{P}_0 + \sigma \left(\frac{1}{R_1} + \frac{1}{R_2}\right)$$
(3)

Isolating $U_{(z)}$ we get:

$$\frac{U_{(z)}}{U_0} = \sqrt{1 + \frac{2gz}{U_0^2} + \frac{2\sigma}{\rho U_0^2} \left(\frac{1}{a} - \frac{1}{R_1} - \frac{1}{R_2}\right)} \tag{4}$$

and in dimensionless form:

$$\frac{U_{(z)}}{U_0} = \sqrt{1 + \underbrace{\frac{2z}{Fr^2a}}_{\text{acceleration due to gravity}} + \underbrace{\frac{2}{We} \left(1 - \frac{a}{R_1} - \frac{a}{R_2}\right)}_{\text{due to surface tension}}} \tag{5}$$

where:

- $We = \frac{\rho U_0^2 a}{\sigma}$ is the Webber #
- $Fr = \frac{U_0^2}{ga}$ is the Froud #

Let's consider the curvature of the wall of the get parallel to the flow:

$$\underbrace{\kappa}_{\text{principal principal curvature}} = -\frac{R_{1zz}}{\left(1 + R_{1z}^2\right)^{\frac{3}{2}}} \underbrace{=}_{\text{linearization}} -R_{1zz} \tag{6}$$

The principal curvature equals one over the principal radius so:

$$\kappa = \frac{1}{R_2} = -\frac{\partial^2 R_1}{\partial z^2} \tag{7}$$

So, the velocity profile $U_{(z)}$ as a function of z is therefore:

$$\frac{U_{(z)}}{U_0} = \sqrt{1 + \frac{2z}{Fr^2a} + \frac{2}{We} \left(1 - \frac{a}{R_1} + a\frac{\partial^2 R_1}{\partial z^2}\right)}$$
(8)



2 ODE of Jet Radius $r_{(z)}$

2.1 Formulate the ODE

From conservation of flux:

$$Q = 2\pi \int_{0}^{R_{1}} U_{(z)} r dr = \pi R_{1}^{2} U_{(z)} \underbrace{=}_{z=0} \pi a^{2} U_{0}$$

$$Q_{(z)} = \pi U_{(z)} R_{1}^{2} \underbrace{=}_{\text{conservation of mass}} \pi a^{2} U_{0}$$

$$\underbrace{\frac{R_{1}}{a} = \sqrt{\frac{U_{0}}{U_{(z)}}}}_{\text{(9)}}$$

After substituting $U_{(z)}$:

$$R_{1} = \frac{a^{2}U_{0}}{\sqrt{U_{0}^{2} + 2gz + \frac{2\sigma}{\rho} \left(\frac{1}{a} - \frac{1}{R_{1}} + \frac{\partial^{2}R_{1}}{\partial z^{2}}\right)}}$$
(10)

2.2 Boundary Conditions

Assume BC:

1.
$$R_{1(z=0)} = a$$

$$2. \left. \frac{\partial R_1}{\partial z} \right|_{z=0} = 0$$

The first conditions is given.

The second condition is derived from the fact that at the beginning, the jet looks like a cylinder.



3 Numerical Solver

The selected method is the Runge-Kutta-Merson method. Isolating the second derivitive of R_1 :

$$\frac{a^4}{R_1^4} = 1 + \frac{2z}{Fr^2a} + \frac{2}{We} \left(1 - \frac{a}{R_1} + a \frac{\partial^2 R_1}{\partial z^2} \right)$$

$$\frac{We \cdot a^4}{2R_1^4} = \frac{We}{2} + \frac{Wez}{Fr^2a} + 1 - \frac{a}{R_1} + a \frac{\partial^2 R_1}{\partial z^2}$$

$$\frac{\partial^2 R_1}{\partial z^2} = \frac{We \cdot a^3}{2R_1^4} - \frac{We}{2a} - \frac{We \cdot z}{Fr^2a^2} - \frac{1}{a} + \frac{1}{R_1}$$
(11)

Defining the system of equations:

$$x = R_1 y = \frac{\partial R_1}{\partial z} (12)$$

$$\begin{cases}
\frac{\partial x}{\partial z} = g_{(z,x,y)} & g = y \\
\frac{\partial y}{\partial z} = f_{(z,x,y)} & f = \frac{\operatorname{We} \cdot a^3}{2x^4} - \frac{\operatorname{We}}{2a} - \frac{\operatorname{We} \cdot z}{\operatorname{Fr}^2 a^2} - \frac{1}{a} + \frac{1}{x}
\end{cases}$$
(13)

The Runge-Kutta-Merson method is defined as:

$$\begin{cases} x_{i+1} &= x_i + \frac{1}{6} (m_1 + 4m_4 + m_5) \\ y_{i+1} &= y_i + \frac{1}{6} (k_1 + 4k_4 + k_5) \end{cases}$$

$$\begin{cases} m_1 &= hg_{(z_i, x_i, y_i)} \\ k_1 &= hf_{(z_i, x_i, y_i)} \end{cases}$$

$$\begin{cases} m_2 &= hg_{(z_i + \frac{1}{3}h, x_i + \frac{1}{3}m_1, y_i + \frac{1}{3}k_1)} \\ k_2 &= hf_{(z_i + \frac{1}{3}h, x_i + \frac{1}{3}m_1, y_i + \frac{1}{3}k_1)} \end{cases}$$

$$\begin{cases} m_3 &= hg_{(z_i + \frac{1}{3}h, x_i + \frac{1}{6}(m_1 + m_2), y_i + \frac{1}{6}(k_1 + k_2))} \\ k_3 &= hf_{(z_i + \frac{1}{3}h, x_i + \frac{1}{6}(m_1 + m_2), y_i + \frac{1}{6}(k_1 + k_2))} \end{cases}$$

$$\begin{cases} m_4 &= hg_{(z_i + \frac{1}{2}h, x_i + \frac{1}{8}(m_1 + 3m_3), y_i + \frac{1}{8}(k_1 + 3k_3))} \\ k_4 &= hf_{(z_i + \frac{1}{2}h, x_i + \frac{1}{8}(m_1 + 3m_3), y_i + \frac{1}{8}(k_1 + 3k_3))} \end{cases}$$

$$\begin{cases} m_5 &= hg_{(z_i + h, x_i + \frac{1}{2}(m_1 - 3m_3 + 4m_4), y_i + \frac{1}{2}(k_1 - 3k_3 + 4k_4))} \\ k_5 &= hf_{(z_i + h, x_i + \frac{1}{2}(m_1 - 3m_3 + 4m_4), y_i + \frac{1}{2}(k_1 - 3k_3 + 4k_4))} \end{cases}$$



4 Solution

$$U_0 = 1 \left[\frac{\mathbf{m}}{\text{sec}} \right] \quad a = 0.05 \,[\mathbf{m}] \quad \rho = 998 \left[\frac{\mathrm{kg}}{\mathrm{m}^3} \right] \qquad \sigma = 0.0728 \left[\frac{\mathrm{N}}{\mathrm{m}} \right] \qquad g = 9.81 \left[\frac{\mathrm{m}}{\mathrm{sec}^2} \right]$$

$$h = 10^{-5} \,[m] \qquad \qquad z = [0, h, 2h, \dots, 5] \,[m]$$

$$(15)$$

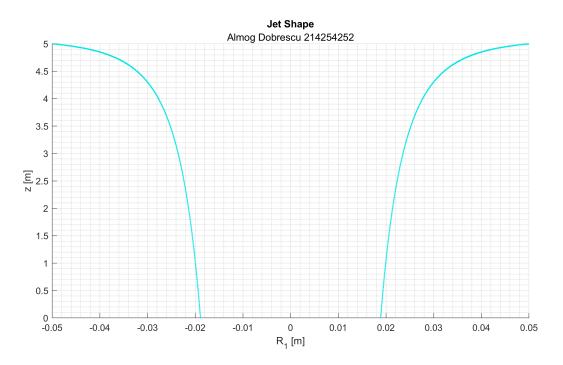


Figure 1: The jet shape

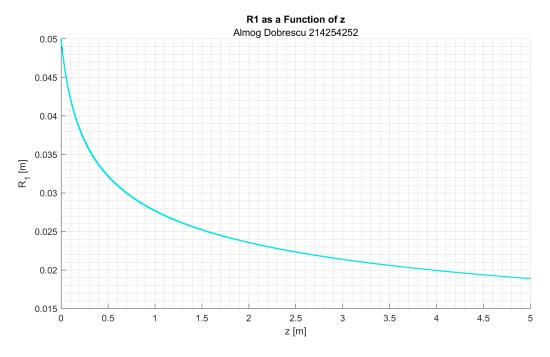


Figure 2: R_1 as a function of z

5 Compering The Results