MAE/ECE 5320 Mechatronics

2025 Spring semester

Lecture 01

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Course introduction

- 1. Syllabus
- 2. Lab sessions: in person (7 sessions) or remote (do it yourself)
- A) Experiment stations (ENLAB 112), or
- B) Option: buy an Arduino package (~\$100); will post the link on Arduino.

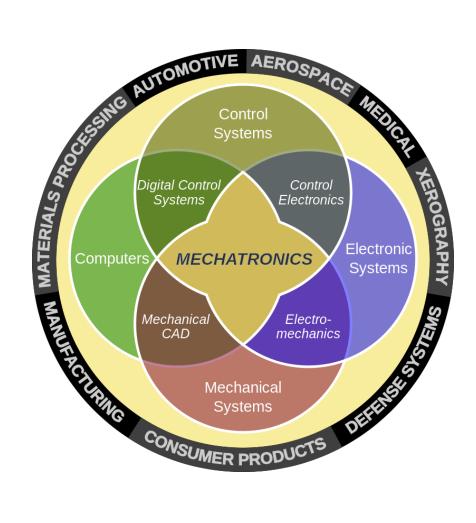
- 3. Final project: a team with 1-3 members
- A) The last 3-4 weeks are reserved to work on final project.

What is covered in this class?

Modeling physical systems
 Hardware components, sensors, actuators, Arduino (micro-controller)
 Basic signal processing, digitalization
 Control (PID) tuning
 Programming on Arduino using Matlab/Simulink
 Final project

Mechatronic System Overview

Mechatronic = mechanical + electronic



Mechatronic system is a multidisciplinary engineering field, including

Dynamic systems and Controls

(MAE/ECE-5310)

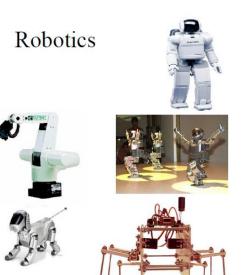
- Systems and signals (ECE-3620-3640)
- Mechanical systems (ENGR-2030 Dynamics) (MAE-5300 Vibrations)
- Digital system (ECE-3700)
- Computer engineering

Examples



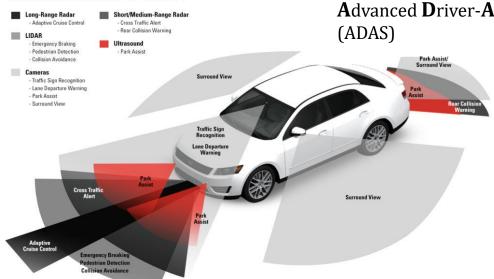








ADAS: THE CIRCLE OF SAFETY



Advanced **D**river-**A**ssistance **S**ystems



Example 1: self-balancing two-wheel robot





Controller/microchip: Arduino MKR 1000 with WiFi



Motor: GB37 12V 1:4 gear DC motor with hall effect encoder with a resolution of 1320



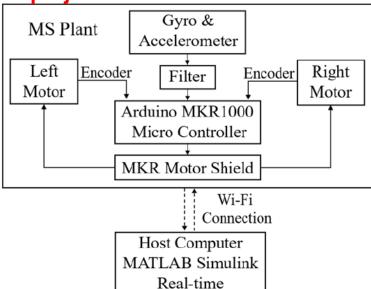
Gyro: MPU 6050 three-axle accelerometer and gyro chip

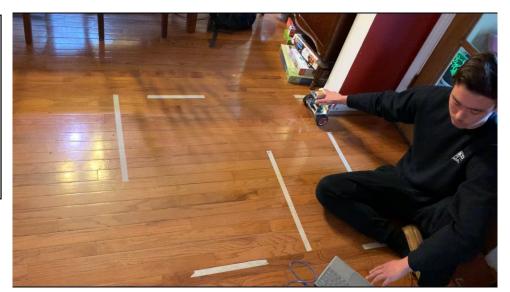


Motor Carrier: 12V PWM DC motor driver with 2 encoder reading channels



Accessories: Wheels, screws, motor mounts, controller mounts, and cart body





Example 2: VTOL aircraft

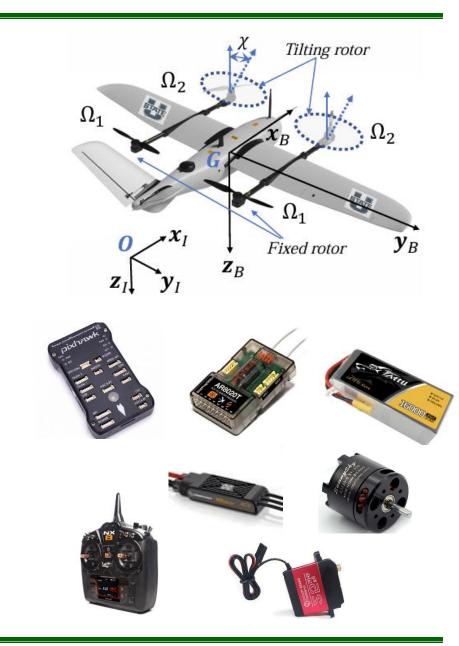


Hardware available for final project !!!

https://usu.box.com/s/ir5e4axf0wmopk2ilb86e 0a5l3hm53su

https://usu.box.com/s/uragp991q64ctqhcv4hvkp1rhgs80kjb

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Example 3:Soft robotic arm

Stepper Motor-Arduino Uno-Stepper Motor. Driver Soft Robotic Arm Stickers Webcam-

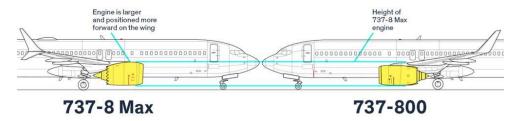
Hardware available for final project !!!

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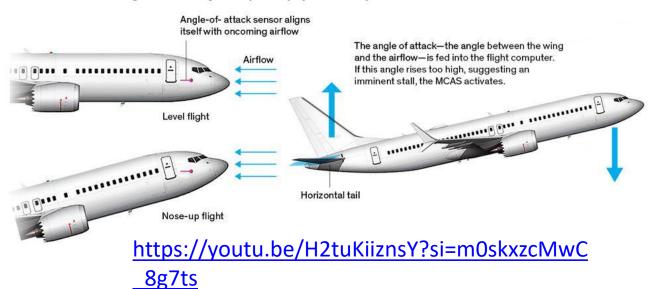
Example 4: How a disaster happens



Ethiopian Airlines Flight ET302, a Boeing 737 Max airliner that crashed on 11 March in Bishoftu, Ethiopia, killing all 157

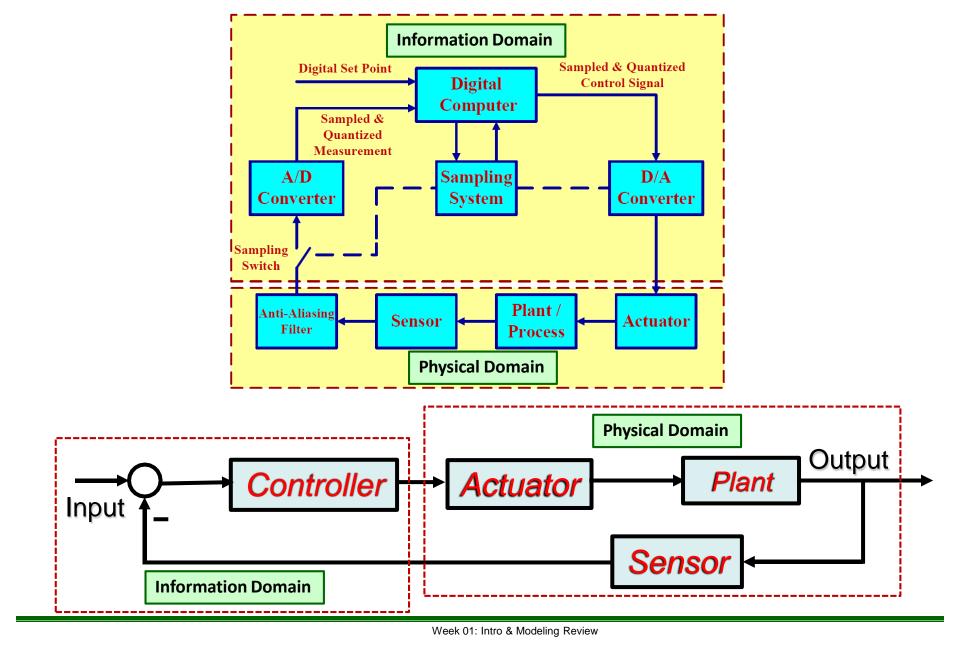


How the new Max flight-control system (MCAS) operates to prevent a stall

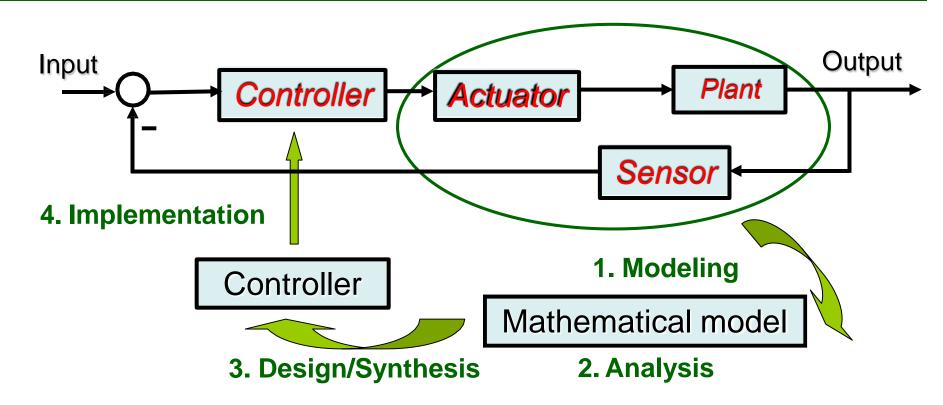


Read more: https://spectrum.ieee.org/how-the-boeing-737-max-disaster-looks-to-a-software-developer

Information and Physical Domains



Modeling – Part of Control Design Process



develop mathematical models for

- Electrical systems
- Mechanical systems
- Electro-mechanical system

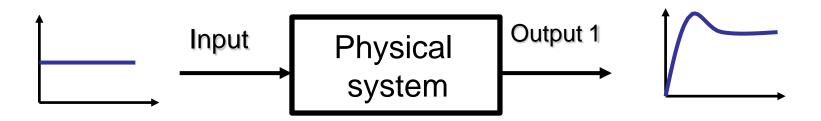
Electrical Systems:

Kirchhoff's voltage & current laws

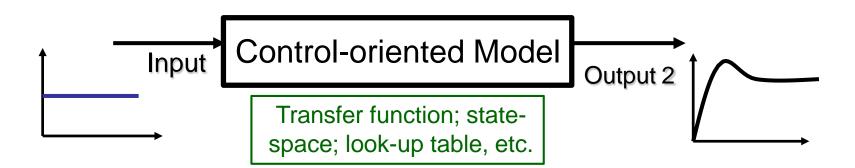
Mechanical systems:

Newton's laws

Modeling Approaches

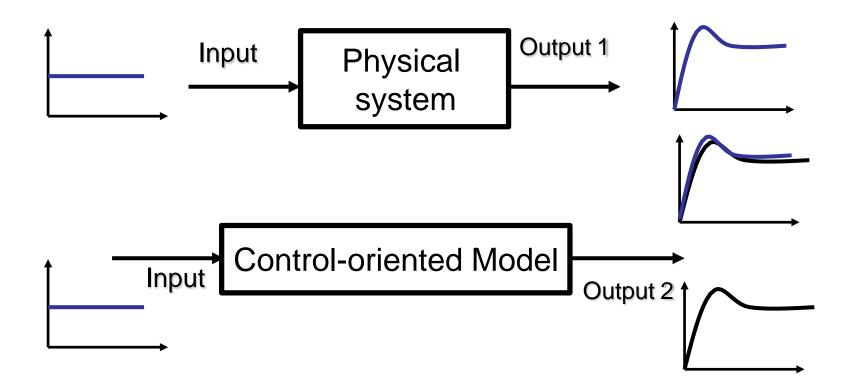


Physics-based model	Hybrid approach	Black box approach
Kirchhoff's Newton's Law Newton-Euler equations Partial Differential Equations Ordinary Differential Equations	System parameter identification with known structure	System identification to get System order & system parameter Neural-network AI (Artificial Intelligence) Machine learning



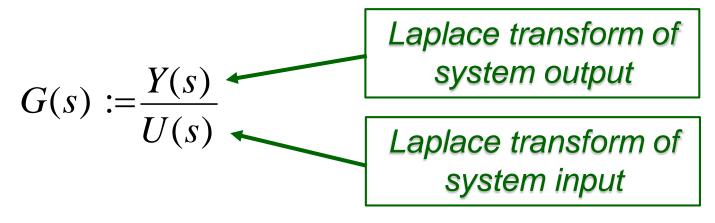
Modeling Discussion

- A control-oriented model is used for analysis, controller design and simulation validation
- No model is exact, but some are useful
- A good model is simple but captures the essential dynamics



Modeling – Transfer Function

A transfer function is defined by its Laplace transform

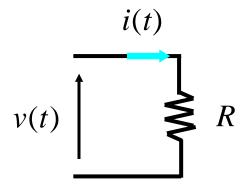


Time-domain
$$u(t)$$
 $g(t)$ $y(t)$ Frequency-domain $U(s)$ $G(s)$ $Y(s)$

Assumption: zero initial condition

Modeling – Electrical Elements

Resistance



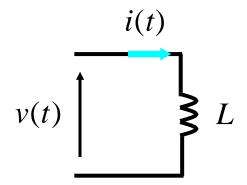
$$v(t) = R i(t)$$



Laplace transform

$$\frac{V(s)}{I(s)} = R$$

Inductance



$$v(t) = L \frac{di(t)}{dt}$$

$$(i(0) = 0)$$

$$\frac{V(s)}{I(s)} = sL$$

Capacitance

$$v(t) = \bigcap_{t \in \mathcal{C}} C$$

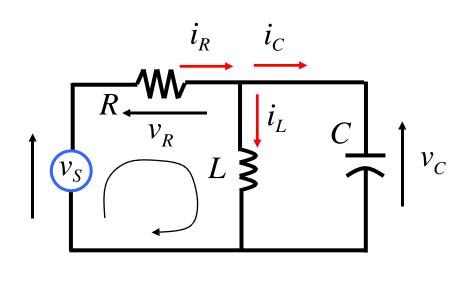
$$v(t) = L \frac{di(t)}{dt} \quad v(t) = v(0) + \frac{1}{C} \int_{0}^{t} i(t)dt$$

$$(v(0) = 0)$$

$$\frac{V(s)}{I(s)} = \frac{1}{sC}$$

Modeling – Kirchhoff's Voltage & Current Laws

- The algebraic sum of voltage drops around any loop is zero.
- The algebraic sum of currents into any junction is zero.



Zero sum voltage:

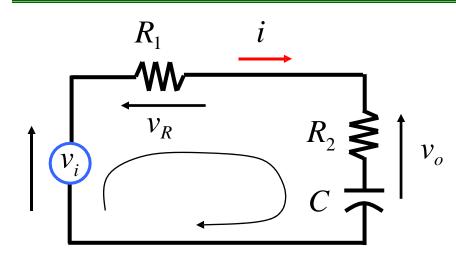
$$-v_S + v_R + v_C = 0$$

$$-v_S + v_R + v_L = 0$$

Zero junction current:

$$i_R - i_C - i_L = 0$$

Modeling – Electrical Example 1



Kirchhoff voltage law (zero initial conditions):

$$R_{2} \begin{cases} \uparrow \\ C \end{cases} v_{o} \quad v_{i} = (R_{1} + R_{2})i(t) + \frac{1}{C} \int_{0}^{t} i(\tau)d\tau$$

$$v_{o} = R_{2}i(t) + \frac{1}{C} \int_{0}^{t} i(\tau)d\tau$$

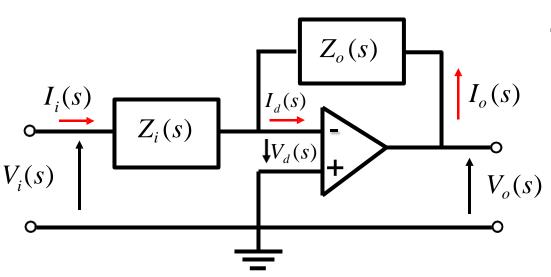
Laplace Transformation

$$V_i(s) = (R_1 + R_2)I(s) + \frac{1}{sC}I(s)$$
 $V_o(s) = R_2I(s) + \frac{1}{sC}I(s)$

Transfer Function

$$G(s) = \frac{V_o(s)}{V_i(s)} = \frac{R_2 I(s) + \frac{1}{sC} I(s)}{(R_1 + R_2)I(s) + \frac{1}{sC} I(s)} = \frac{1 + R_2 Cs}{1 + (R_1 + R_2)Cs}$$

Modeling – Electrical Example 2 (OP Amp)



OP Amp Assumptions:

 $I_o(s)$ Infinity gain: $V_d(s) = 0$

Infinity input impedance:

$$I_d(s) = 0$$

$$V_i(s) = Z_i(s)I_i(s)$$

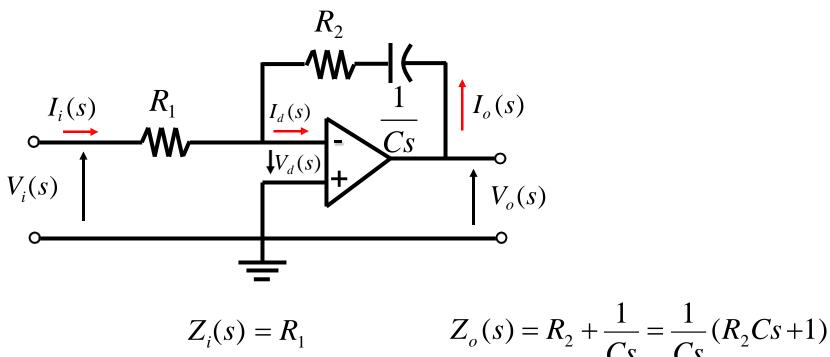
$$V_{i}(s) = Z_{i}(s)I_{i}(s)$$
 $V_{o}(s) = Z_{o}(s)I_{o}(s)$ $I_{i}(s) = -I_{o}(s)$

$$I_i(s) = -I_o(s)$$

Transfer Function

$$G(s) = \frac{V_o(s)}{V_i(s)} = \frac{Z_o(s)I_o(s)}{Z_i(s)I_i(s)} = -\frac{Z_o(s)}{Z_i(s)}$$

Modeling – Electrical Example 3

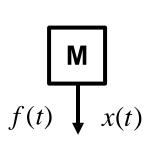


Transfer Function

$$G(s) = \frac{V_o(s)}{V_i(s)} = -\frac{Z_o(s)}{Z_i(s)} = -\frac{R_2Cs+1}{R_1Cs}$$

Modeling – Translational Mechanism

Mass

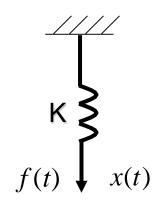


$$f(t) = M\ddot{x}(t)$$
$$x(0) = 0$$

$$\dot{x}(0)=0$$

$$F(s) = Ms^2X(s)$$

Spring

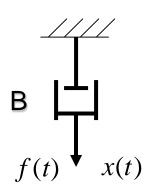


$$f(t) = K x(t)$$



$$F(s) = KX(s)$$

Damper



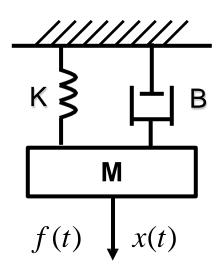
$$f(t) = B \dot{x}(t)$$



$$x(0) = 0$$

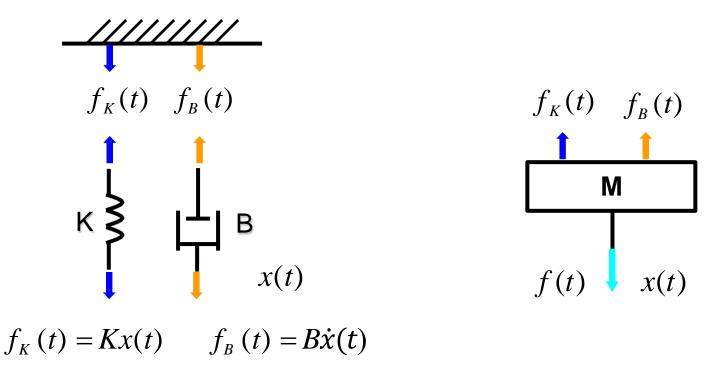
$$F(s) = BsX(s)$$

Modeling – Spring-Mass-Damper Systems



$$M\ddot{x}(t) + Kx(t) + B\dot{x}(t) = f(t)$$

Modeling – Free Body Diagram



Note: x(t) represents the displacement change for spring resting position

Using Newton's Law : $F(t) = M\ddot{x}(t)$

$$M\ddot{x}(t) = f(t) - f_K(t) - f_B(t) = f(t) - Kx(t) - B\dot{x}(t)$$

$$M\ddot{x}(t) + Kx(t) + B\dot{x}(t) = f(t)$$

Modeling - Rotational Mechanism

Moment of inertia

torque $\tau(t)$ $\theta(t)$

rotation angle

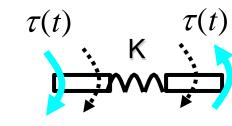
$$\tau(t) = J\ddot{\theta}(t)$$

$$\theta(0) = 0$$

$$\dot{\theta}(0) = 0$$

$$T(s) = Js^2\Theta(s)$$

Rotational spring



$$\theta(t) = \theta_1(t) - \theta_2(t)$$

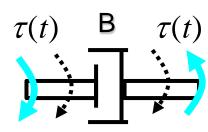
$\tau(t) = K\theta(t)$



$$T(s) = K\Theta(s)$$

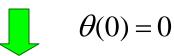
Friction

Friction of air/fluid resistance



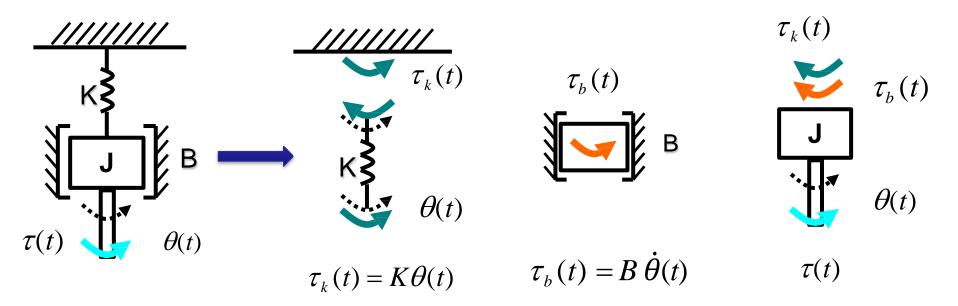
$$\theta(t) = \theta_1(t) - \theta_2(t)$$

$$\tau(t) = B \dot{\theta}(t)$$



$$T(s) = Bs\Theta(s)$$

Modeling - Torsional Pendulum System (Ex. 2.12)

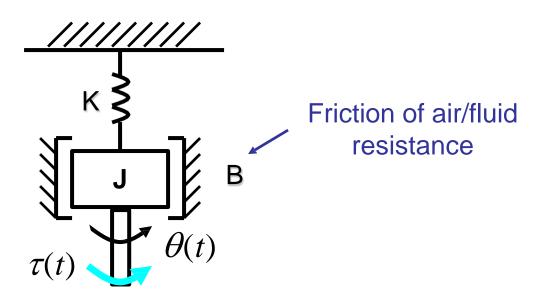


Newton's law

$$J\ddot{\theta}(t) = \tau(t) - \tau_k(t) - \tau_b(t) = \tau(t) - K\theta(t) - B\dot{\theta}(t)$$

$$J\ddot{\theta}(t) + B\dot{\theta}(t) + K\theta(t) = \tau(t)$$

Modeling - Torsional Pendulum System (cont'd)



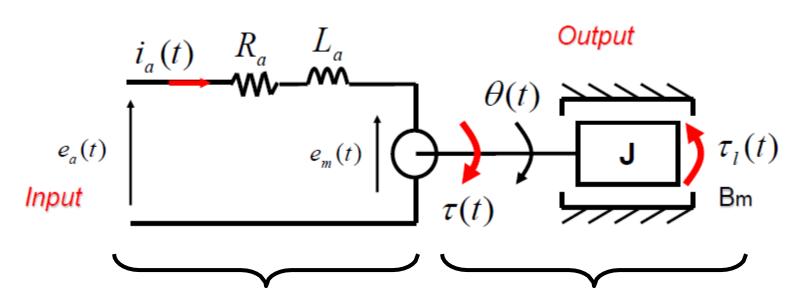
Equation of Motion

$$J\ddot{\theta}(t) + B\dot{\theta}(t) + K\theta(t) = \tau(t)$$

By Laplace transform (with zero ICs),

$$G(s) = \frac{\Theta(s)}{T(s)} = \frac{1}{Js^2 + Bs + K}$$
 (2nd order system)

Modeling – Model of DC Motor



Armature circuit

 e_a : applied voltage

 i_a : armature current

*e*_m back EMF voltage

(counter-electromotive force/voltage)

Mechanical load

 θ : angular position

 ω : angular velocity

J: rotor inertia

B: viscous fricton

Modeling - Model of DC Motor in Time Domain

Armature circuit

$$e_a(t) = R_a i_a(t) + L_a \frac{di_a(t)}{dt} + e_m(t)$$

- Motor torque
- Back EMF

$$\tau(t) = K_{\tau} i_a(t)$$

$$e_m(t) = K_m \omega(t)$$

Mechanical load

$$J\ddot{\theta}(t) = \tau(t) - B\dot{\theta}(t) - \tau_l(t)$$

Load torque

Rigid connection between mechanical/electrical parts

$$\omega(t) = \dot{\theta}(t)$$

Modeling - Model of DC Motor in "s" Domain

Armature circuit

$$I_a(s) = \frac{1}{R_a + L_a s} (E_a(s) - E_m(s))$$

- Connection between mechanical/electrical parts
 - Motor torque

$$T(s) = K_{\tau}I_{\alpha}(s)$$

Back EMF

$$E_m(s) = K_m \Omega(s)$$

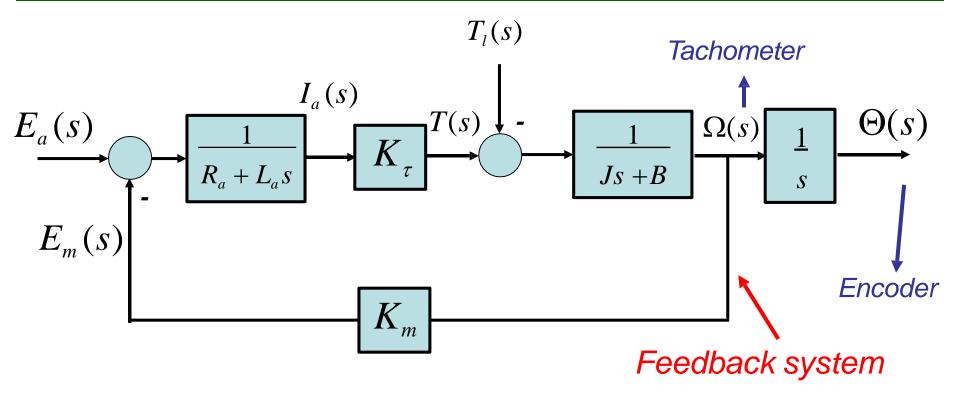
Mechanical load

$$\Omega(s) = \frac{1}{Js + B} (T(s) - T_{L}(s))$$

Angular position (zero initial condition)

$$\Omega(s) = s\Theta(s)$$

Modeling – DC Motor Block Diagram



$$\Omega(s) = \frac{K_{\tau}}{(L_a s + R_a)(J s + B)} (E_a(s) - K_m \Omega(s)), \quad T_l(s) = 0$$

$$\Omega(s) = \frac{1}{(Js + B)} (-T_l(s) - \frac{K_m K_{\tau}}{L_a s + R_a} \Omega(s)), \quad E_a(s) = 0$$

Modeling – DC Motor Transfer Function

$$\frac{\Omega(s)}{E_{a}(s)} = \frac{\frac{K_{\tau}}{(L_{a}s + R_{a})(Js + B)}}{1 + \frac{K_{\tau}K_{m}}{(L_{a}s + R_{a})(Js + B)}} = \frac{K_{\tau}}{(L_{a}s + R_{a})(Js + B) + K_{\tau}K_{m}} =: G_{1}(s)$$

$$\frac{\Omega(s)}{T_{1}(s)} = \frac{1}{1 + \frac{K_{\tau}K_{m}}{(L_{a}s + R_{a})(Js + B)}} = \frac{-(L_{a}s + R_{a})}{(L_{a}s + R_{a})(Js + B) + K_{\tau}K_{m}} =: G_{2}(s)$$

$$\Omega(s) = G_1(s)E_a(s) + G_2(s)T_l(s)$$

$$\Theta(s) = \frac{1}{s}\Omega(s) = \frac{1}{s}(G_1(s)E_a(s) + G_2(s)T_l(s))$$

Modeling – DC Motor Transfer Function (cont'd)

Note: In many cases $L_a << R_a$. Then, an approximated TF is obtained by setting $L_a = 0$.

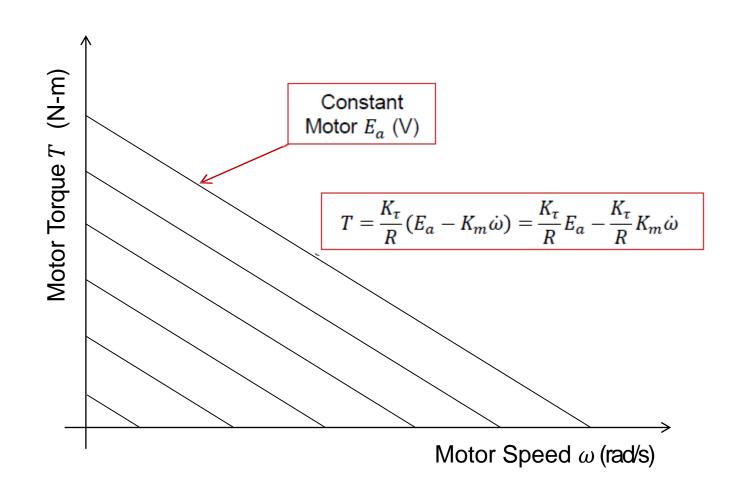
$$\frac{\Omega(s)}{E_{a}(s)} = \frac{K_{\tau}}{(L_{a}s + R_{a})(Js + B) + K_{\tau}K_{m}} \approx \frac{K_{\tau}}{R_{a}(Js + B) + K_{\tau}K_{m}}$$

$$=: \frac{K}{Ts + 1} \left(K := \frac{K_{\tau}}{R_{a}B + K_{m}K_{\tau}}, T = \frac{R_{a}J}{R_{a}B + K_{m}K_{\tau}} \right)$$

$$\frac{\Omega(s)}{T_{l}(s)} = \frac{-(L_{a}s + R_{a})}{(L_{a}s + R_{a})(Js + B) + K_{m}K_{\tau}} \approx \frac{-R_{a}}{R_{a}(Js + B) + K_{m}K_{\tau}}$$

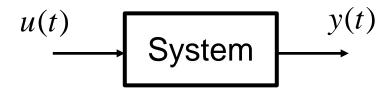
$$\frac{\Theta(s)}{E_a(s)} = \frac{K}{s(Ts+1)}$$

Modeling – DC Motor SS Characteristics



Modeling – What Is a Linear System?

A linear system satisfies Principle of Superposition

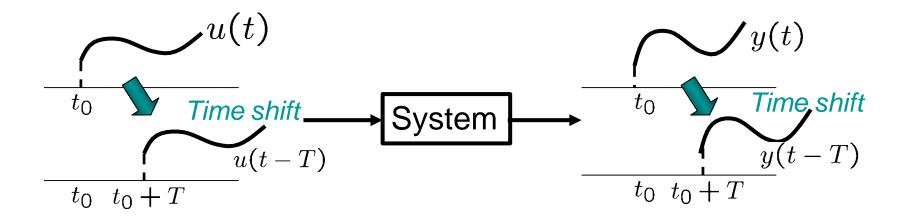


$$\begin{aligned} u_1(t) & \to y_1(t) \\ u_2(t) & \to y_2(t) \end{aligned} \Rightarrow \alpha_1 u_1(t) + \alpha_2 u_2(t) \to \alpha_1 y_1(t) + \alpha_2 y_2(t) \\ & \forall \alpha_1, \alpha_2 \in \Re$$

A nonlinear system does not satisfy the principle of superposition.

Modeling – Time Invariant and Varying Systems

- A system is called time-invariant (time-varying) if system parameters do not (do) change in time.
- Example: vehicle/rocket $M \ddot{x}(t) = f(t)$ and $M(t)\ddot{x}(t) = f(t)$
- For time-invariant systems:



This course deals with time-invariant systems.

Modeling – Why Linear System?

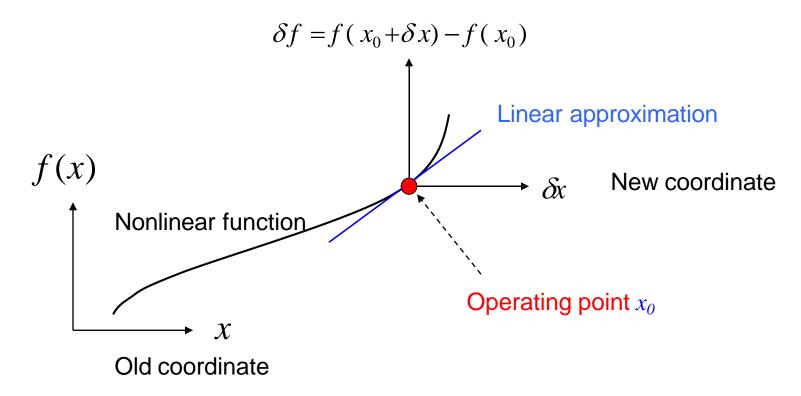
- Easy to understand and obtain solutions
- Linear ordinary differential equations (ODEs),
 - Homogeneous solution and particular solution
 - Transient solution and steady-state solution
 - Solution caused by initial values and forced solution
- Add many simple solutions to get more complex ones (using superposition!)
- Easy to check the Stability of stationary states (Laplace Transform)

Modeling – Why Linearization

- Actual physical systems are inherently nonlinear. (linear system rarely exists!)
- TF models are only for Linear Time-Invariant (LTI) systems.
- Many control analysis/design techniques are available only for linear systems.
- Nonlinear systems are difficult to deal with mathematically.
- Often, we linearize nonlinear systems before analysis and design. How?

Modeling – How to Linearize It?

- Nonlinearity can be approximated by a linear function for small deviations δx around an operational point x_0
- Use a Taylor series expansion at x₀

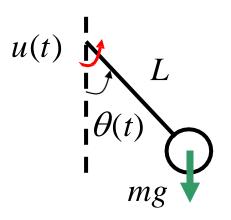


Modeling – Summary: 6 Steps

- Step 1: identify the system model's input r(t) and ouput y(t) With initial conditions $\dot{r}_0 = 0, \dot{y}_0 = 0, \ddot{r}_0 = 0$
- Step 2: express model in the form $f(r, \dot{r}, ..., y, \dot{y}, ...) = 0$
- Step 3: define equilibrium operation point r_0 , y_0 and set all derivatives to zeros at the equilibrium points
- Step 4: perform Taylor expansion about (r_0, y_0) and only retain 1st order derivative
- Step 5: change variables to derivatives $(\tilde{r}=r-r_0,\dot{\tilde{r}}=\dot{r}-\dot{r}_0,...,\tilde{y}=y-y_0,\dot{\tilde{y}}=\dot{y}-\dot{y}_0)$
- Step 6: rewrite the function in step 5 into a standard ODE

Modeling - Pendulum Linearization

Motion of the pendulum



$$mL^2\ddot{\theta}(t) + mgL\sin\theta(t) = u(t)$$

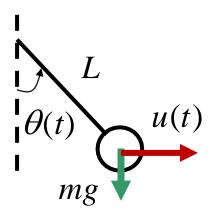
$$\frac{\ddot{\theta}(t) + \frac{g\sin\theta(t)}{L} - \frac{u(t)}{mL^2}}{\int_{f(\theta,\ddot{\theta},u)} dt} = 0$$

- Linearize it at $\theta_0 = \pi$
- Find u_0 $\ddot{\pi} + \frac{g \sin \pi}{L} \frac{u_0}{mL^2} = 0 \rightarrow u_0 = 0$
- New coordinates:

$$\theta = \theta_0 + \delta\theta = \pi + \delta\theta$$
$$u = u_0 + \delta u = 0 + \delta u$$

Modeling - Pendulum Linearization

Motion of the pendulum



$$mL^2\ddot{\theta}(t) + mgL\sin\theta(t) = u(t)$$

$$\frac{\ddot{\theta}(t) + \frac{g\sin\theta(t)}{L} - \frac{u(t)}{mL^2}}{\int_{f(\theta,\ddot{\theta},u)} dt} = 0$$

- Linearize it at $\theta_0 = \pi$
- Find u_0 $\ddot{\pi} + \frac{g \sin \pi}{L} \frac{u_0}{mL^2} = 0 \rightarrow u_0 = 0$
- New coordinates:

$$\theta = \theta_0 + \delta\theta = \pi + \delta\theta$$
$$u = u_0 + \delta u = 0 + \delta u$$

Modeling – 6 Steps (Example 1)

System ODE:

$$mL^2\ddot{\theta}(t) + mgL\sin\theta(t) = u(t)$$

Step One (input/output):

System input : u(t), system output : $\theta(t)$

Step Two (f(...)=0):

$$\underbrace{\ddot{\theta}(t) + \frac{g\sin\theta(t)}{L} - \frac{u(t)}{mL^2}}_{f(\theta,\ddot{\theta},u)} = 0$$

Step Three (θ_0, u_0) :

Let
$$\theta_0 = \pi$$
, and $\dot{\theta}_0 = \ddot{\theta}_0 = 0$,
$$\ddot{\theta} + \frac{g \sin \pi}{I} - \frac{u_0}{mI^2} = 0 \rightarrow u_0 = 0$$

Modeling – 6 Step Example 1 (cont'd)

Step Four (Taylor expansion):

$$f(\theta, \ddot{\theta}, u) \cong \underbrace{f(\theta_0, \ddot{\theta}_0, u_0)}_{0} + \frac{\partial f}{\partial \theta} \Big|_{\substack{\theta = \theta_0 \\ \ddot{\theta} = \dot{\theta}_0 \\ u = u_0}} (\theta - \theta_0) + \frac{\partial f}{\partial \ddot{\theta}} \Big|_{\substack{\theta = \theta_0 \\ \ddot{\theta} = \ddot{\theta}_0 \\ u = u_0}} (\ddot{\theta} - \ddot{\theta}_0) + \frac{\partial f}{\partial u} \Big|_{\substack{\theta = \theta_0 \\ \ddot{\theta} = \ddot{\theta}_0 \\ u = u_0}} (u - u_0)$$

$$= \frac{g}{L} \cos \theta \Big|_{\theta = \pi} (\theta - \pi) + 1 \cdot (\ddot{\theta} - 0) - \frac{1}{mL^2} (u - 0)$$

$$= -\frac{g}{L} (\theta - \pi) + (\ddot{\theta} - 0) - \frac{1}{mL^2} (u - 0)$$

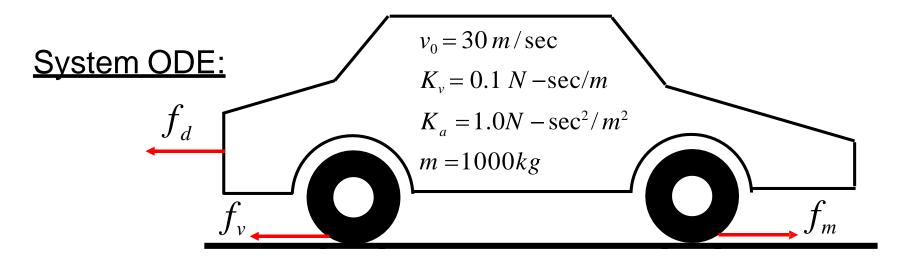
Step Five $(\delta\theta_0, \delta u_0)$:

$$f(\theta, \ddot{\theta}, u) \cong \delta \ddot{\theta} - \frac{g}{L} \delta \theta - \frac{1}{mL^2} \delta u = 0$$

Step Six (rewrite):

$$\delta \ddot{\theta} - \frac{g}{L} \delta \theta = \frac{1}{mL^2} \delta u \Rightarrow \frac{\Delta \Theta(s)}{\Delta U(s)} = \frac{1}{mL^2 s^2 - mLg}$$

Modeling – 6 Steps (Example 2)



Tractive force input : $f_m = f_m(t)$

Viscous friction force: $f_v = f_v(t) = K_v v$

Aerodynamic drag force: $f_d = f_d(t) = K_a v^2$

$$m\dot{v} = f_m - f_d - f_v = f_m - K_v v - K_a v^2$$

 $m\dot{v} + K_v v + K_a v^2 = f_m$

Modeling – 6 Step Example 2 (cont'd-1)

Vehicle System:

$$m\dot{v} + K_v v + K_a v^2 = f_m$$

Step One (input/output):

System input (tractive force): $f_m(t)$, system output (velocity): v(t)

Step Two (f(...)=0):

$$\underbrace{m\dot{v} + K_v v + K_a v^2 - f_m}_{f(f_m, v, \dot{v})} = 0$$

Step Three (v_0, \dot{v}_0, u_0) :

Let
$$v_0 = 30$$
 and $\dot{v}_0 = 0$, and find f_{m0}

$$f_{m0} = K_v v_0 + K_a v_0^2 = 0.1 \times 30 + 1 \times 30^2 = 903N$$

Modeling – 6 Step Example 2 (cont'd-2)

Step Four (Taylor expansion):

$$f(f_{m}, v, \dot{v}) \cong \underbrace{f(f_{m0}, v_{0}, \dot{v}_{0})}_{0} + \underbrace{\frac{\partial f}{\partial f_{m}}}_{\substack{f_{m} = f_{m0} \\ v = v_{0} \\ \dot{v} = \dot{v}_{0}}}^{f_{m} = f_{m0}} (f_{m} - f_{m0}) + \underbrace{\frac{\partial f}{\partial v}}_{\substack{f_{m} = f_{m0} \\ v = v_{0} \\ \dot{v} = \dot{v}_{0}}}^{f_{m} = f_{m0}} (v - v_{0}) + \underbrace{\frac{\partial f}{\partial v}}_{\substack{f_{m} = f_{m0} \\ v = v_{0} \\ \dot{v} = \dot{v}_{0}}}^{f_{m} = f_{m0}} (\dot{v} - \dot{v}_{0})$$

$$= 0 - 1(f_{m} - f_{m0}) + (K_{v} + 2K_{a}v_{0}) \cdot (v - v_{0}) + m(\dot{v} - \dot{v}_{0}) = 0$$

Step Five
$$(\widetilde{f}_m = f_m - f_{m0}, \widetilde{v} = v - v_0, \widetilde{\dot{v}} = \dot{v} - \dot{v}_0)$$
:

$$\underbrace{m}_{1000} \widetilde{\dot{v}} + (\underbrace{K_v + 2K_a v_0}_{0.1 + 2 \times 1 \times 30 = 60.1}) \cdot \widetilde{v} - \widetilde{f}_m = 0$$

Step Six (rewrite):

$$1000\widetilde{\dot{v}} + 60.1\widetilde{v} = \widetilde{f}_m \Longrightarrow \frac{V(s)}{F_m(s)} = \frac{1}{1000s + 60.1}$$