

Running SU2

POINTWISE® AND SU2 JOINT WORKSHOP
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DEFINITION
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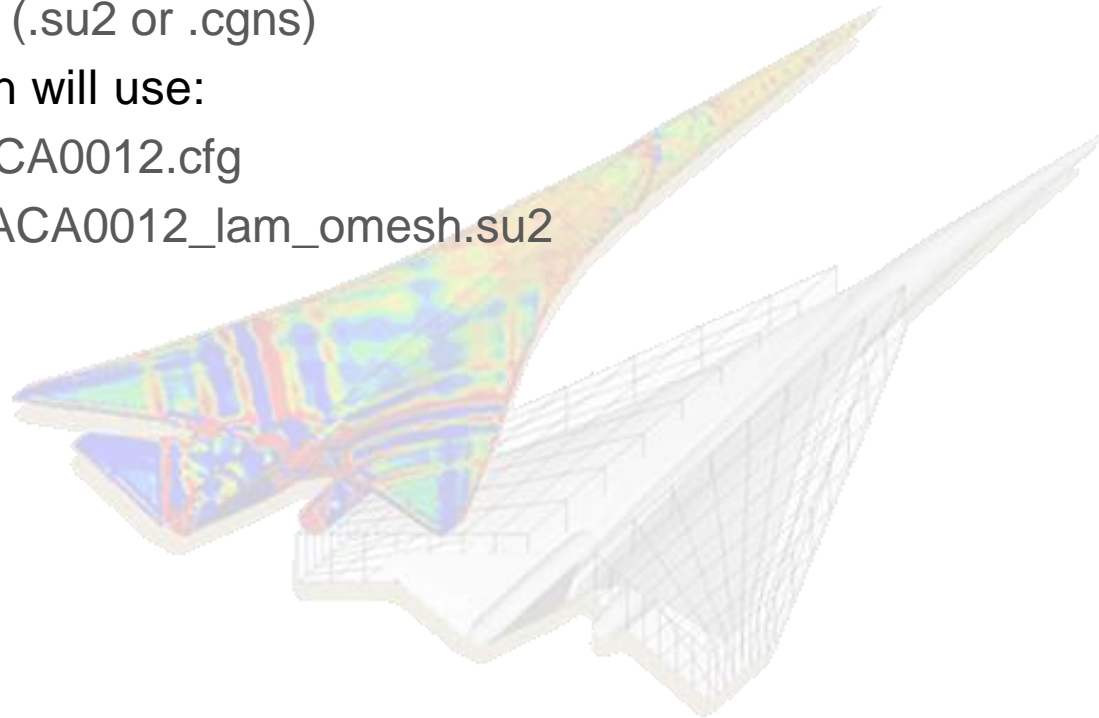
Running Simulations with SU2

What do I need to run simulations with SU2?

- Configuration file (.cfg)
- Mesh file (.su2 or .cgns)

This session will use:

- Lam_NACA0012.cfg
- Mesh_NACA0012_lam_omesh.su2



Quick Start Tutorial

NACA 0012 Airfoil

Transonic, Euler flow

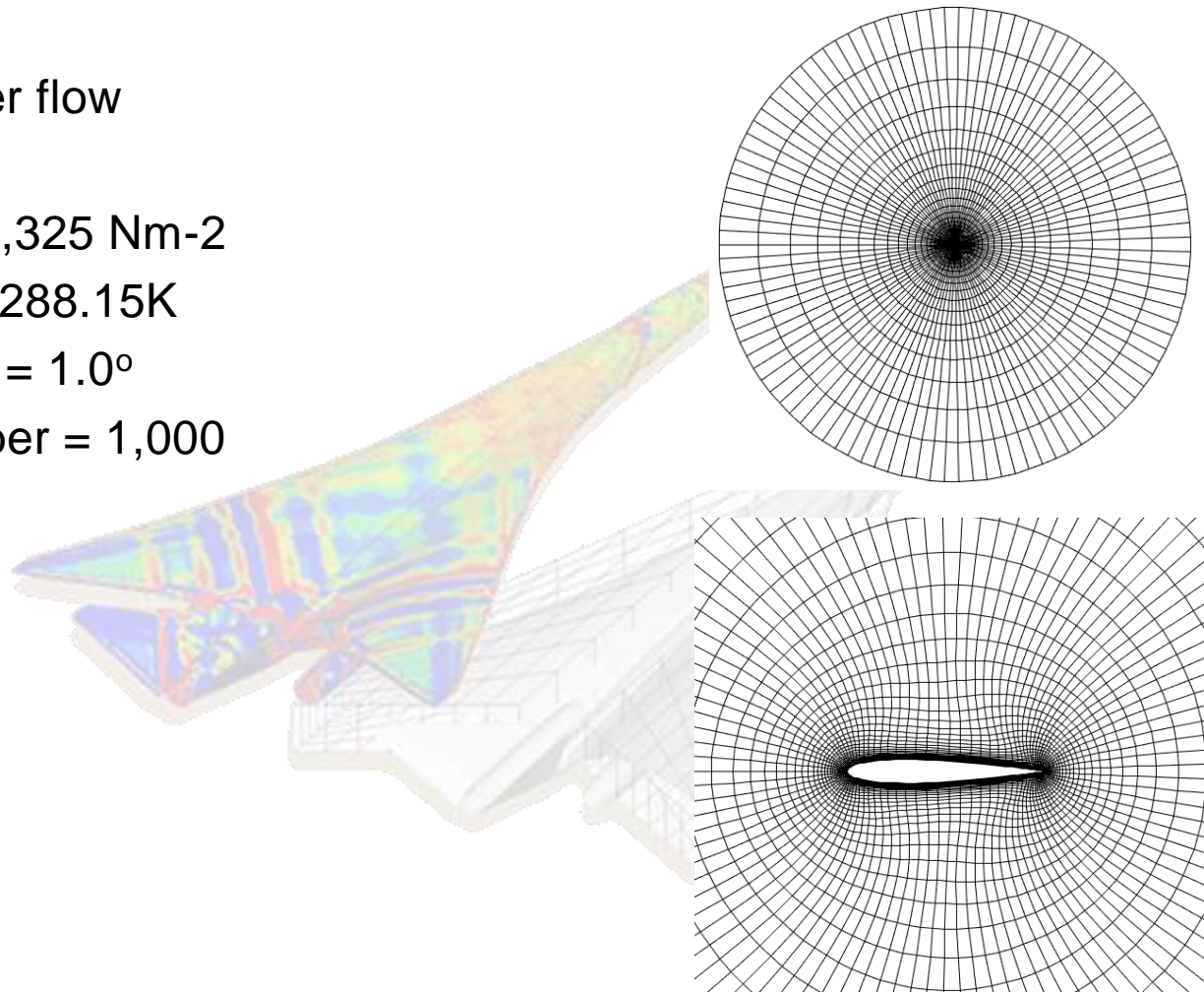
Mach No. = 0.5

Pressure = 101,325 Nm⁻²

Temperature = 288.15K

Angle of attack = 1.0°

Reynolds number = 1,000



Flow Solution

Config options:

```
PHYSICAL_PROBLEM= NAVIER_STOKES
%
MATH_PROBLEM= DIRECT
%
MACH_NUMBER= 0.5
%
AoA= 1.00
%
REYNOLDS_NUMBER=1000.0
%
FREESTREAM_TEMPERATURE= 288.15
%
MESH_FILENAME= mesh_NACA0012_lam_omesh.su2
```

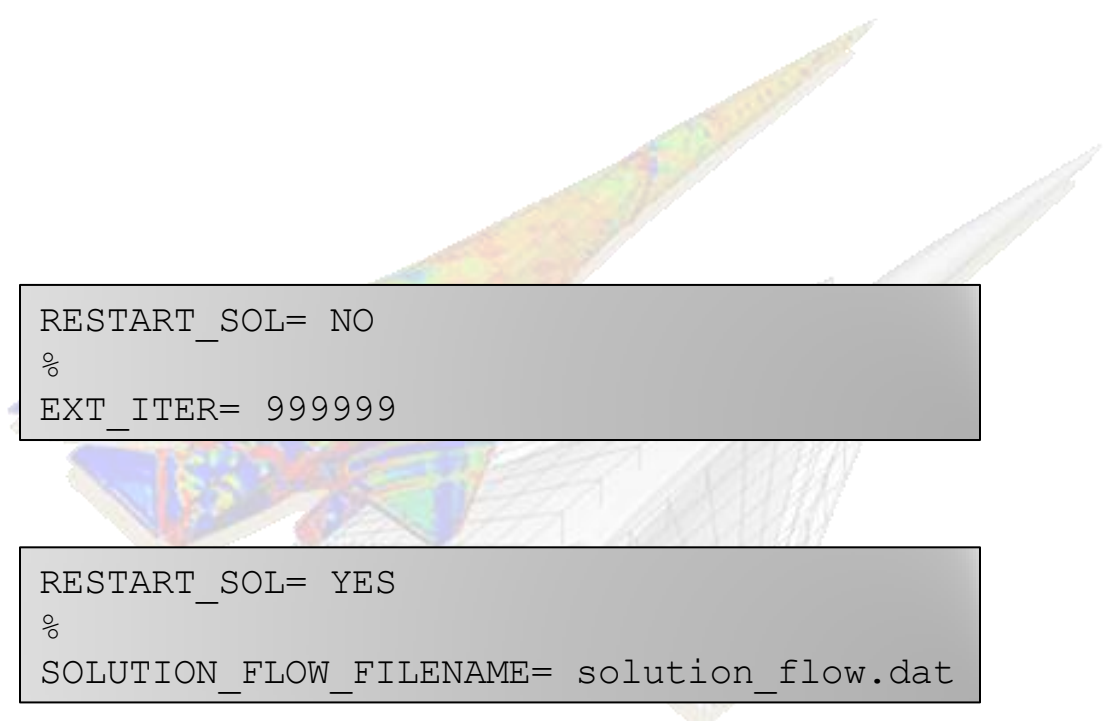
Most parameters have default values

The order of config options is not important

Restart

Simulations can be restarted from partially converged results

Config options:



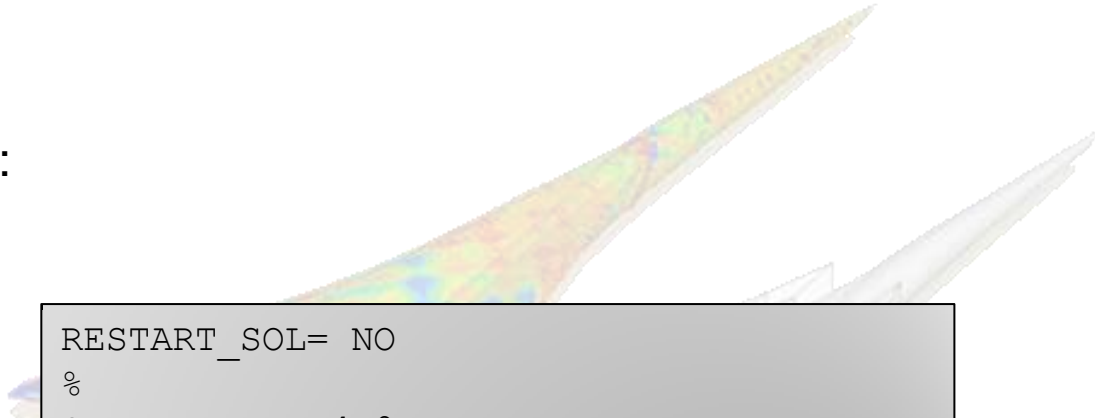
```
RESTART_SOL= NO
%
EXT_ITER= 999999
```

```
RESTART_SOL= YES
%
SOLUTION_FLOW_FILENAME= solution_flow.dat
```

Solver Parameters

Among the many options in the config file, various parameters exist to modify the solution method

Config options:

A background image showing a 3D visualization of flow around an airfoil. The flow is represented by a color gradient from blue to yellow, indicating pressure or velocity variations. The airfoil is a NACA 0012 type, shown in a perspective view.

```
RESTART_SOL= NO
%
CFL_NUMBER= 4.0
%
CFL_RAMP= ( 1.1, 10, 10.0 )
%
CONV_NUM_METHOD_FLOW= ROE
%
SPATIAL_ORDER_FLOW= 2ND_ORDER
```

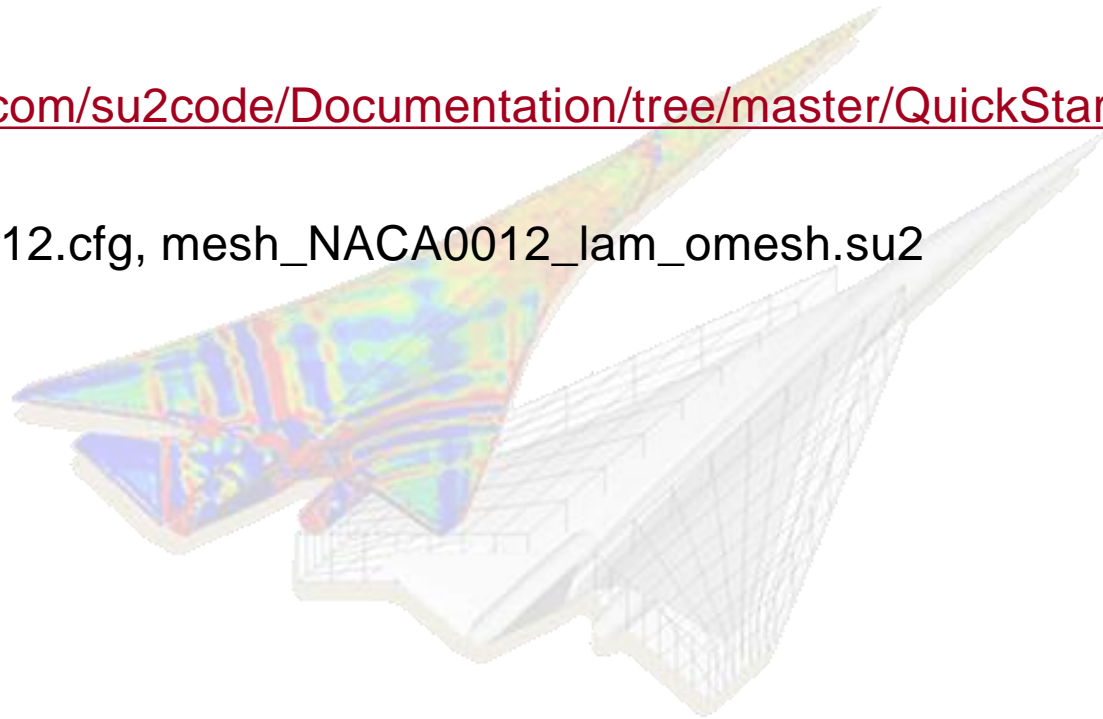
Interactive

Direct Solution demonstration

Materials:

https://github.com/su2code/Documentation/tree/master/QuickStart_NS/

Lam_NACA0012.cfg, mesh_NACA0012_lam_omesh.su2



Adjoint Solution

Sensitivity of a functional to changes in the flow

- e.g., How does changing the airfoil shape affect drag?

Additional required file:

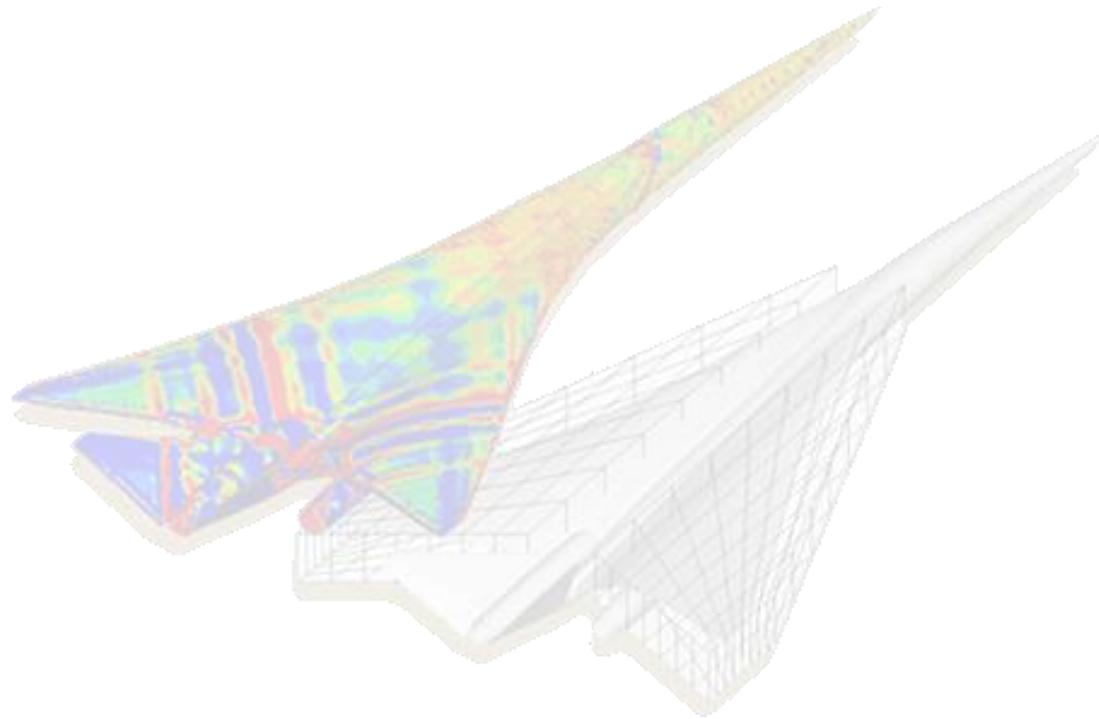
- Converged flow solution

Config options:

```
MATH_PROBLEM= ADJOINT
%
RESTART_SOL= NO
%
OBJECTIVE_FUNCTION= DRAG
%
SOLUTION_FLOW_FILENAME= solution_flow.dat
```

Interactive

Adjoint Solution Demonstration



Additional Resources

Online documentation

- <http://su2.stanford.edu>

Online tutorials

- su2.stanford.edu > Training
- also accessible via su2.stanford.edu > Guides > User's Tutorials

TestCases directory

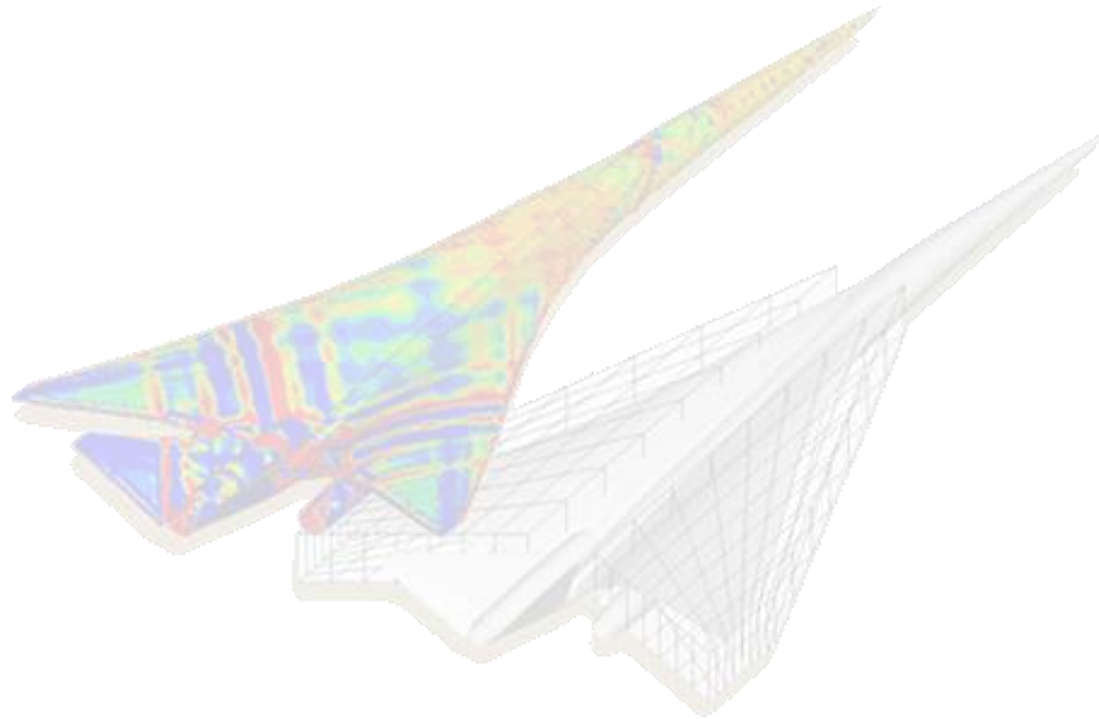
- github.com/su2code/TestCases/

CFD Online forum

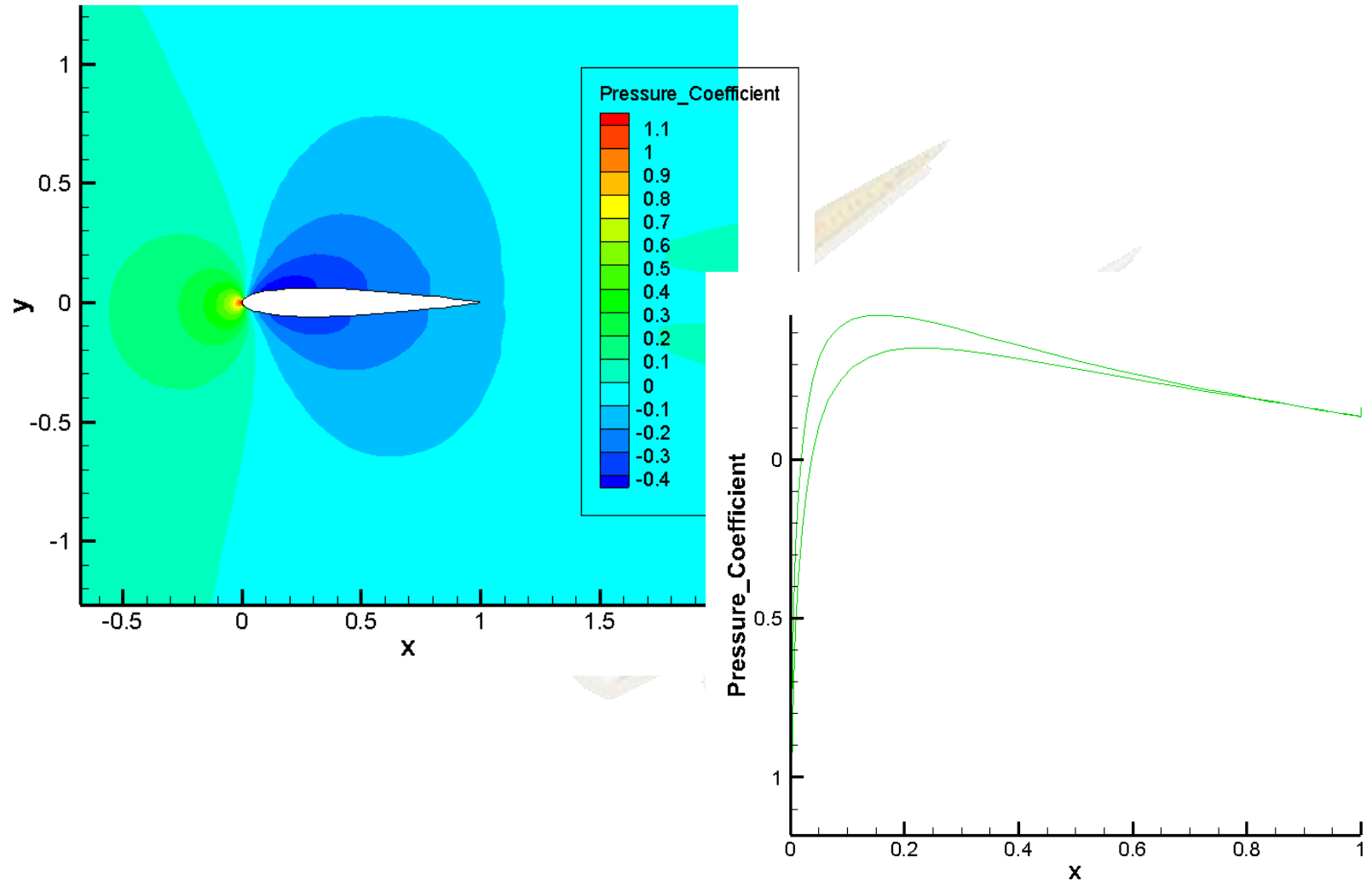
- <http://www.cfd-online.com/Forums/su2/>



Questions?



Results: Direct Solution



Results: Adjoint Solution

