



# Gateway Classes

**Semester -I & II****Common to All Branches****Fundamentals of Mech. Engg.(BME101/201)****Unit-1 : ONE SHOT- Introduction to Mechanics**

## Gateway Series for Engineering

- Topic Wise Entire Syllabus**
- Long - Short Questions Covered**
- AKTU PYQs Covered**
- DPP**
- Result Oriented Content**

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# Gateway Classes



## Fundamentals of Mech. Engg.(BME101/201)

### Unit-1

### Introduction to Mechanics

### Syllabus

**Force moment and couple, principle of transmissibility, Varignon's theorem. Resultant of force system- concurrent and non-concurrent coplanar forces, Types of supports (Hinge, Roller) and loads (Point, UDL, UVL), free body diagram, equilibrium equations and Support Reactions.**

**Normal and shear Stress, strain, Hooke's law, Poisson's ratio, elastic constants and their relationship, stress-strain diagram for ductile and brittle materials, factor of safety.**



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AKTU B.TECH I-YEAR FME



# FUND. OF MECHANICAL ENGG.

COMPLETE REVISION + NOTES

UNIT-1 MECHANICS

QUESTIONS तो यहीं से आएंगे !

LIVE

18 MARCH 9 PM

BY M S TOMER SIR





# AKTU : B.Tech. II SEMESTER

Sem-II

Session: 2023-24

Subjects	COMBO PACK
Maths-II	
PPS/FME	
Electrical / Electronics	
Physics / Chemistry	
Soft Skills / Environment	

Courses will be available 22 March 2024

# FUNDAMENTALS OF MECHANICAL ENGINEERING (FME)

## UNIT-1 : Introduction to Mechanics

AKTU : Syllabus

AKTU

Tht D x Nmm

- Force, moment and couple, principle of transmissibility, Varignon's theorem.
- Resultant of force system- concurrent and non-concurrent coplanar forces
- Types of supports (Hinge, Roller) and loads (Point, UDL, UVL), M
- Free body diagram, equilibrium equations and Support Reactions.
- Normal and shear Stress, strain, Hooke's law
- Poisson's ratio, elastic constants and their relationship
- Stress-strain diagram for ductile and brittle materials, factor of safety.

W/E/G/K

## Force

Force is an external agent which tends to change the speed or direction of a system.

A force on a rigid body may produce the following effects:

- Change its state of rest or motion

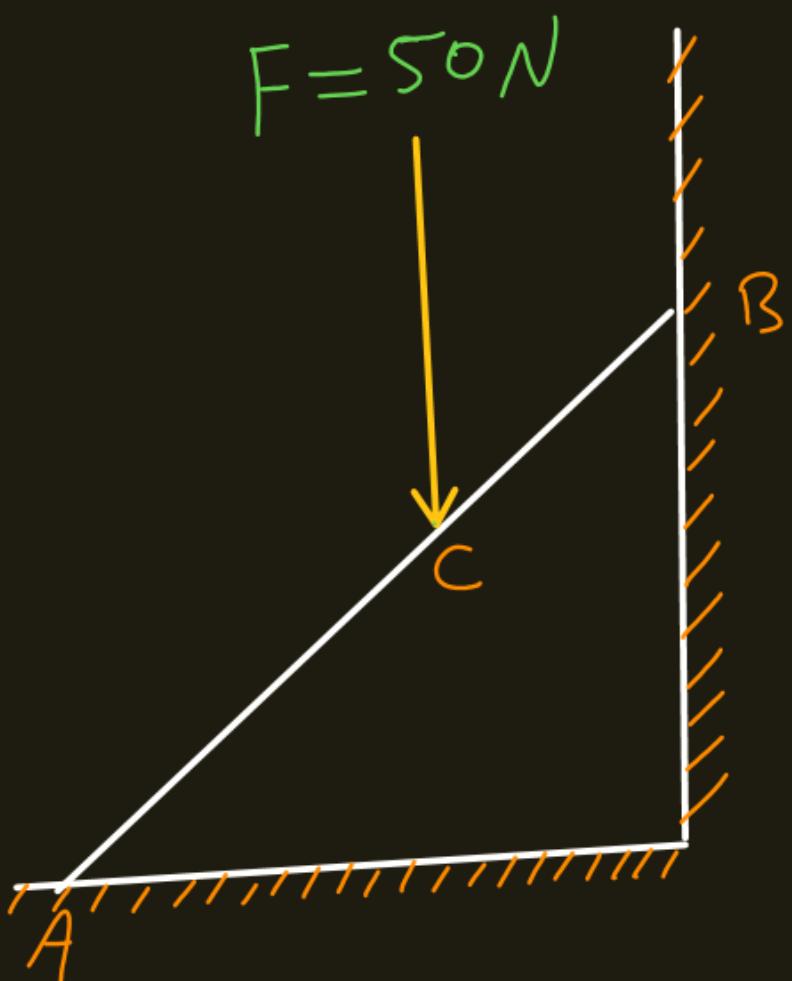
- Accelerate or retard its motion

- Change its shape and size

- Keep it in equilibrium.

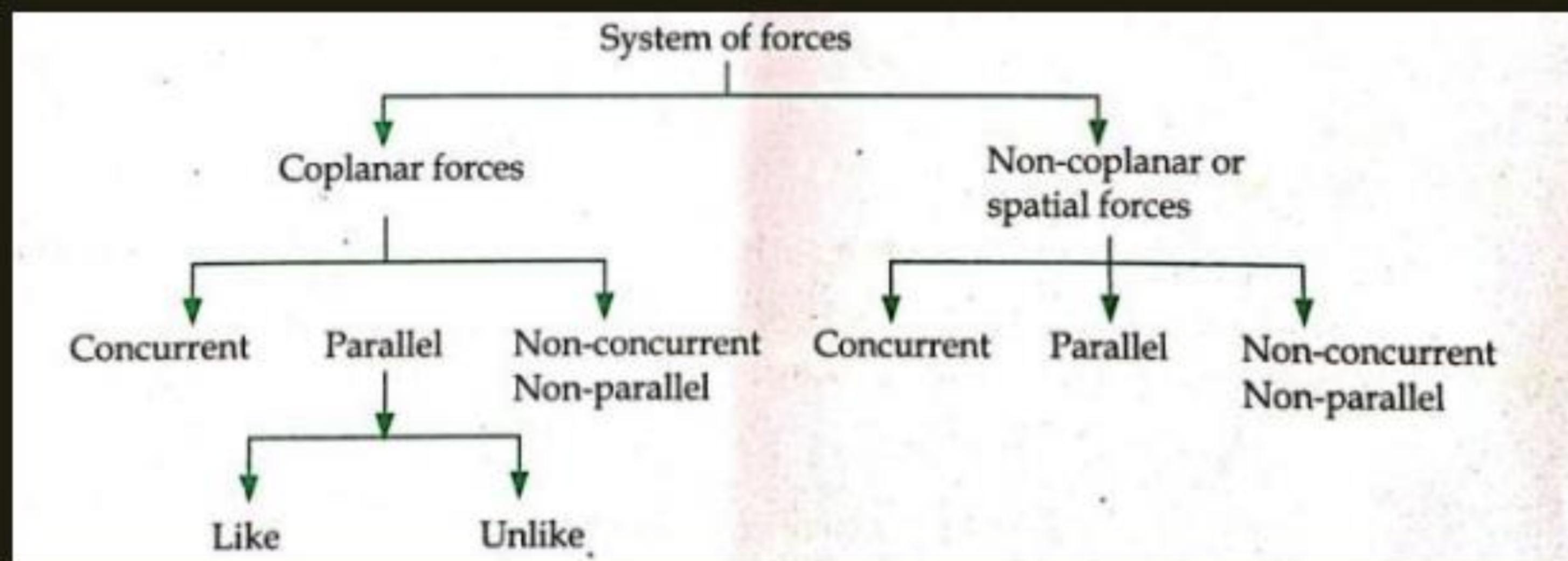
Force is a **vector** quantity determined completely by its

- Magnitude ✓  $50\text{N}$
- Point of application ✓ 'C'
- Line of action ✓ (Vertical)
- Direction ✓ (Downward)



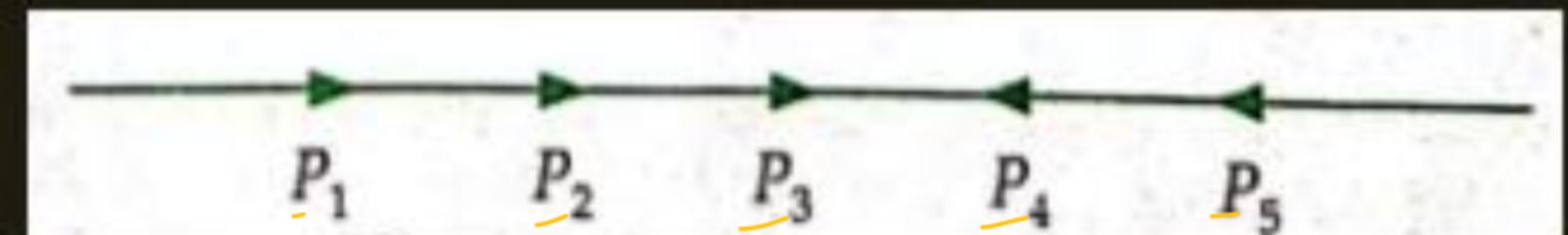
# Force System

When several forces of different magnitude and direction act upon a body, they constitute a force system.



## Force System

**Collinear forces** : The lines of action of all forces lie along the same straight line and in the same plane.

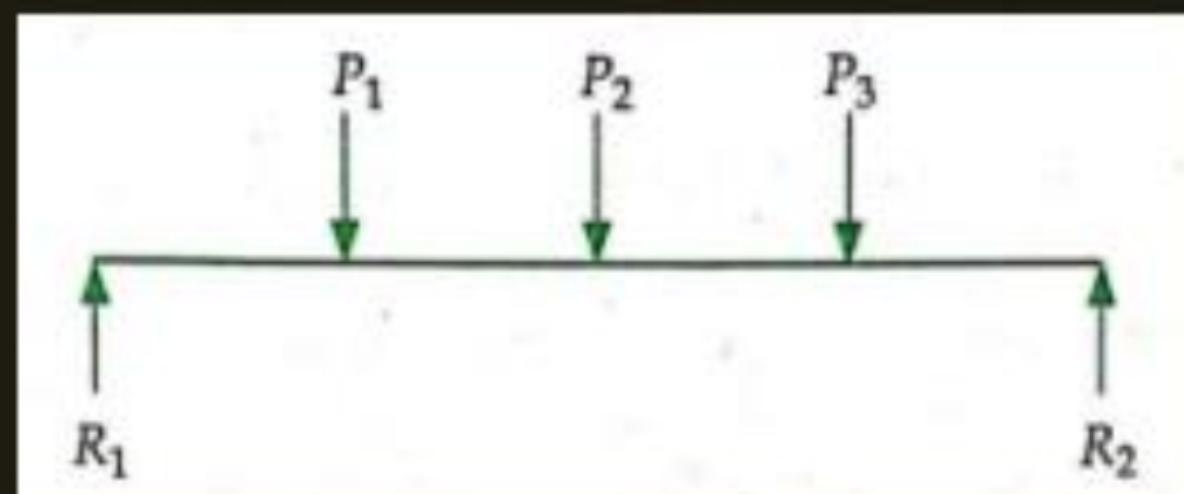


**Example:** Force on a rope in a tug of war.

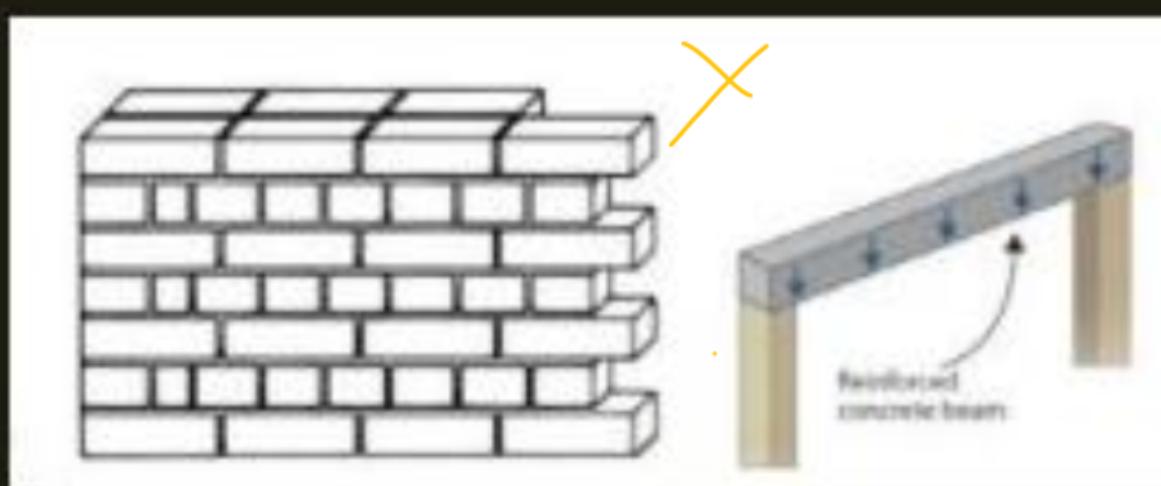


## Force System

**Coplanar parallel forces** : The lines of action of all forces are parallel to each other and lie in a single plane.



**Example:** System of vertical loads acting on a beam

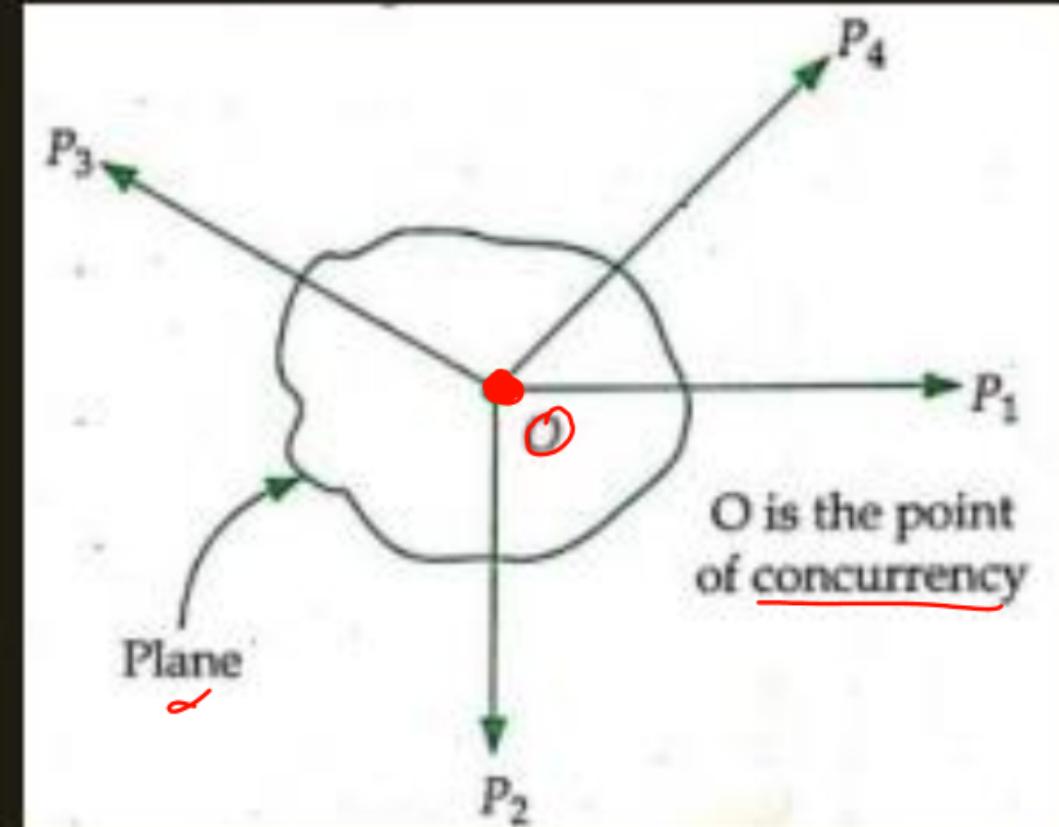


## Force System

Coplanar concurrent forces : All forces lie in the same plane, have different direction but their lines of action act at one point (pass through a single point).

The point where the lines of action of the forces meet is known as the point of concurrency of the system.

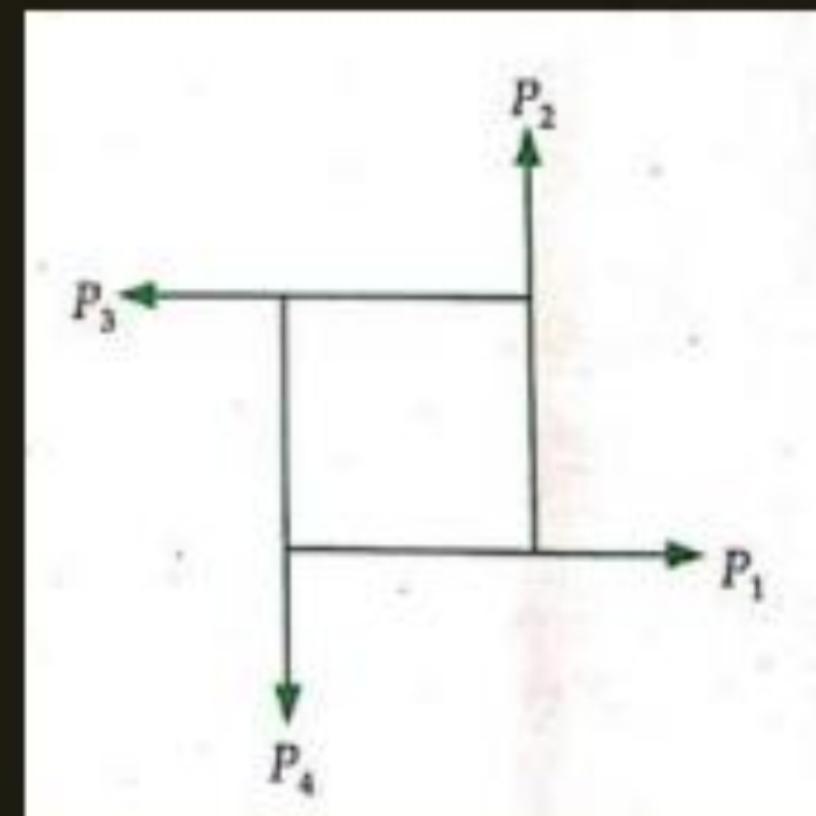
Example: Forces on a rod resting against a wall.



## Force System

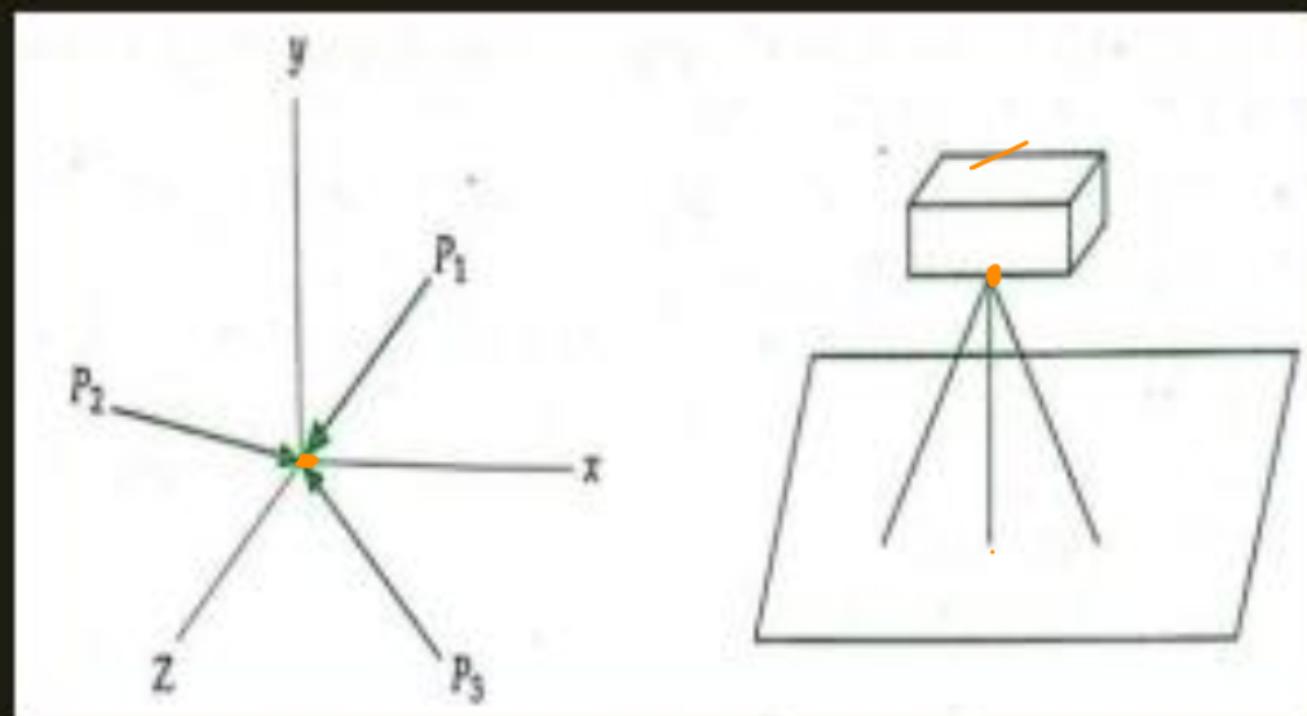
Coplanar non-concurrent forces : All forces lie in the same plane, but their lines of action do not pass through a single point.

**Example:** Forces on a ladder resting against a wall and a person standing on a rung which is not at its centre of gravity.



**Non-Coplanar concurrent forces** : All forces do not lie in the same plane, but their lines of action pass through a single point.

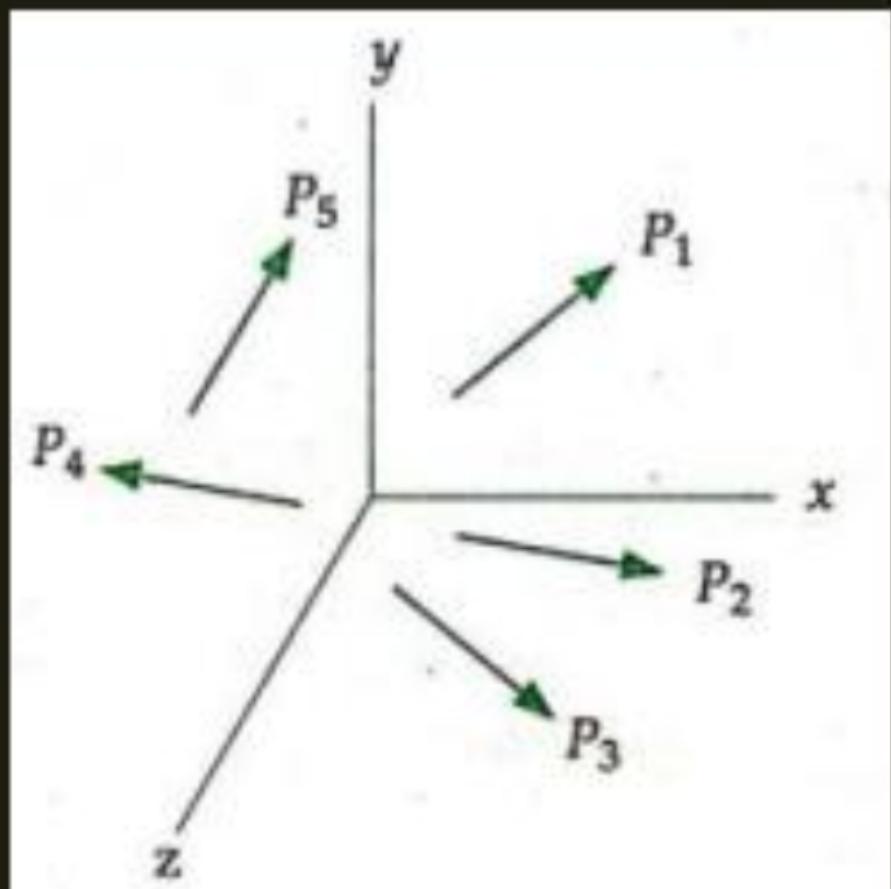
**Example: Forces on a tripod carrying a camera**



## Force System

**Non-Coplanar non-concurrent forces** : All forces do not lie in the same plane, and their lines of action do not meet at a single point.

**Example:** Forces acting on a moving bus.

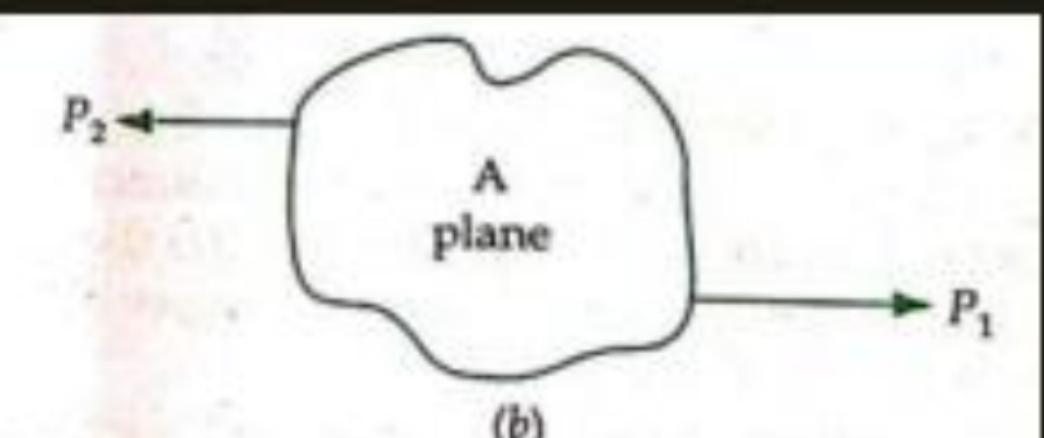
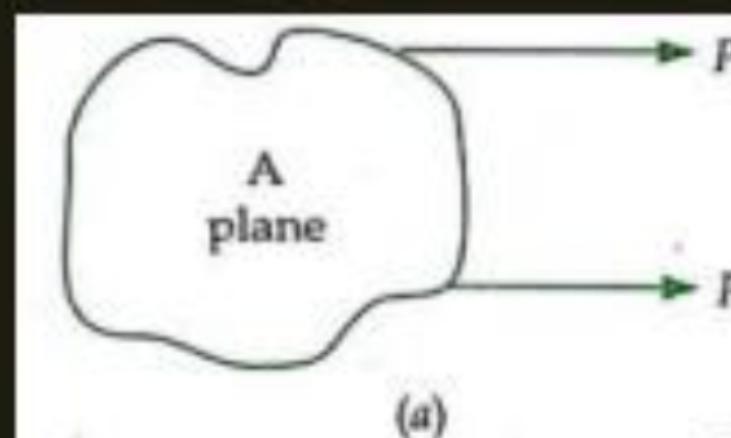


## Coplanar like and unlike parallel force system:

- Same plane
- Lines of action parallel to each other

### Further

- (i) The system is called coplanar **like parallel** force system if the forces are in the **same direction**.
- (ii) The system is called coplanar **unlike parallel** force system if the forces are in the **opposite direction**.



# Moment of Force

The moment of a force about a point is the product of magnitude of the force and the perpendicular distance from the line of action of the force from the same point.

$$M_O = F \times d$$



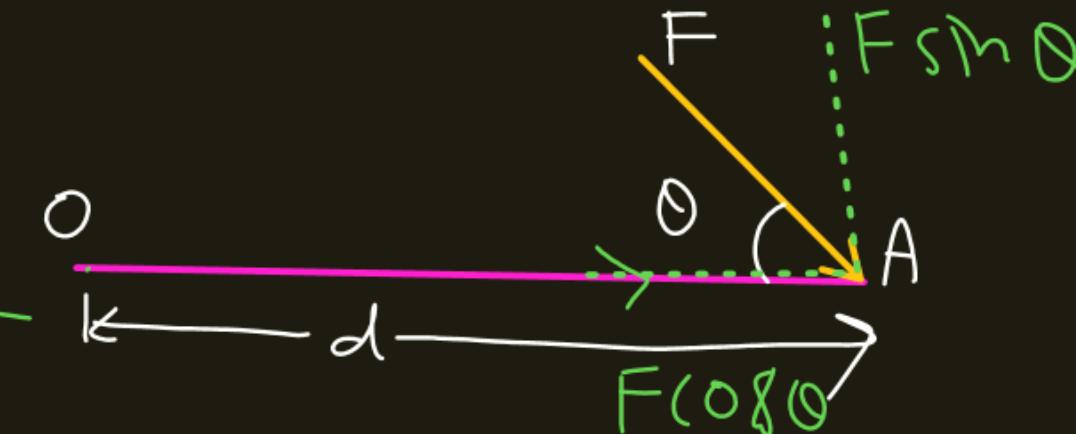
$$M_O = F \times O$$

$$\boxed{M_O = 0}$$



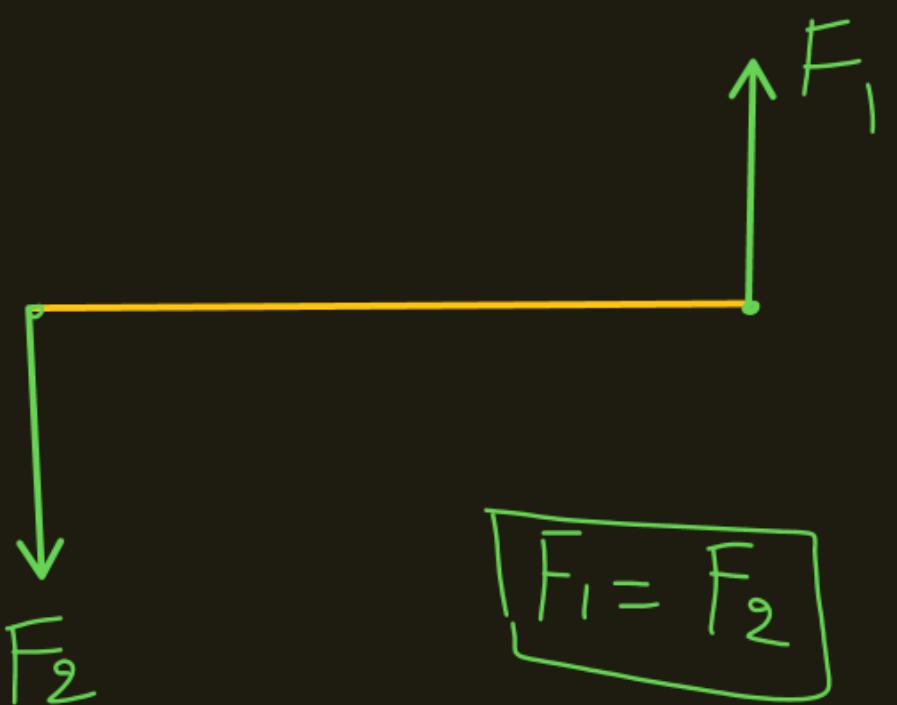
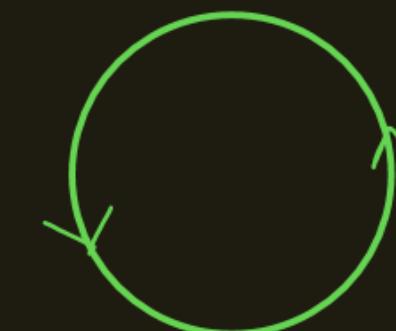
$$M_O = F \sin \theta \times d + F \omega l \theta \times O$$

$$\boxed{M_O = F d \sin \theta}$$



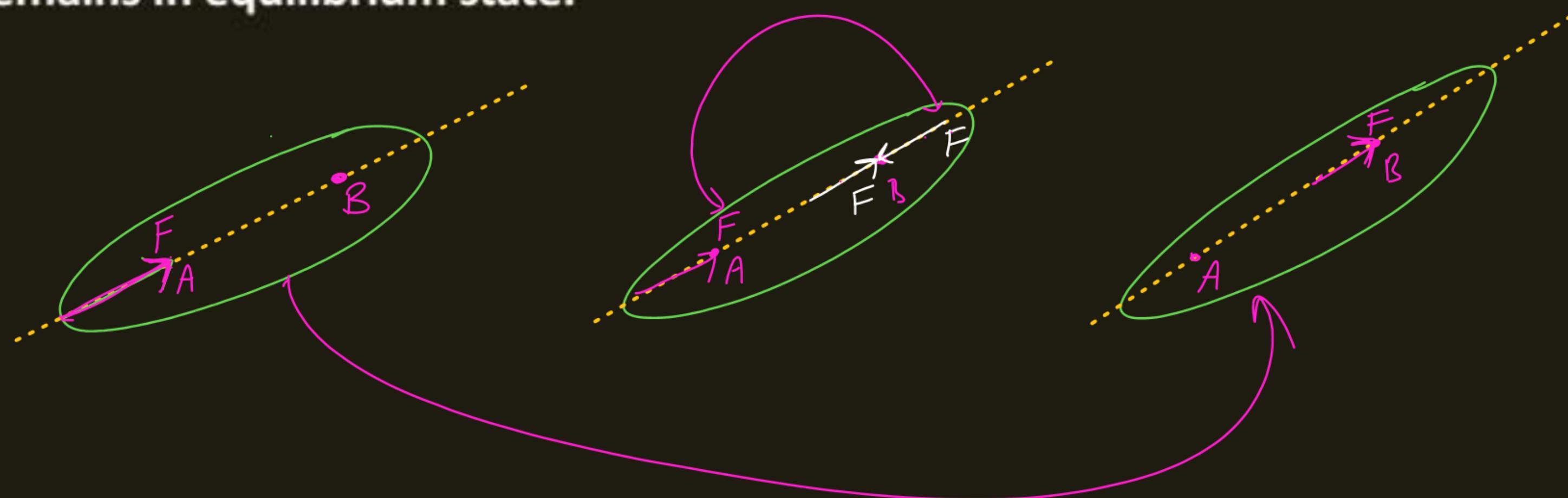
# Couple

When two equal and opposite forces are applied simultaneously at different points on a body, their resultant force is zero, but both these equal and opposite forces try to rotate the body in the same direction. These two forces formed a Couple.



## Principle of transmissibility of forces : AKTU-2022-23 (Odd Even)

It states that the point of application of a force may be shifted to any other point, along the line of action and in the same direction, body will remains in equilibrium state.



**FUNDAMENTALS OF MECHANICAL ENGINEERING by M S Tomer Sir**  
**AKTU PYQs**

**Topic : Complex Matrices**

**Q .1.1 Explain the principle of transmissibility of force. 2 marks AKTU-2022-23 (Odd Even)**

# Varignon's Theorem

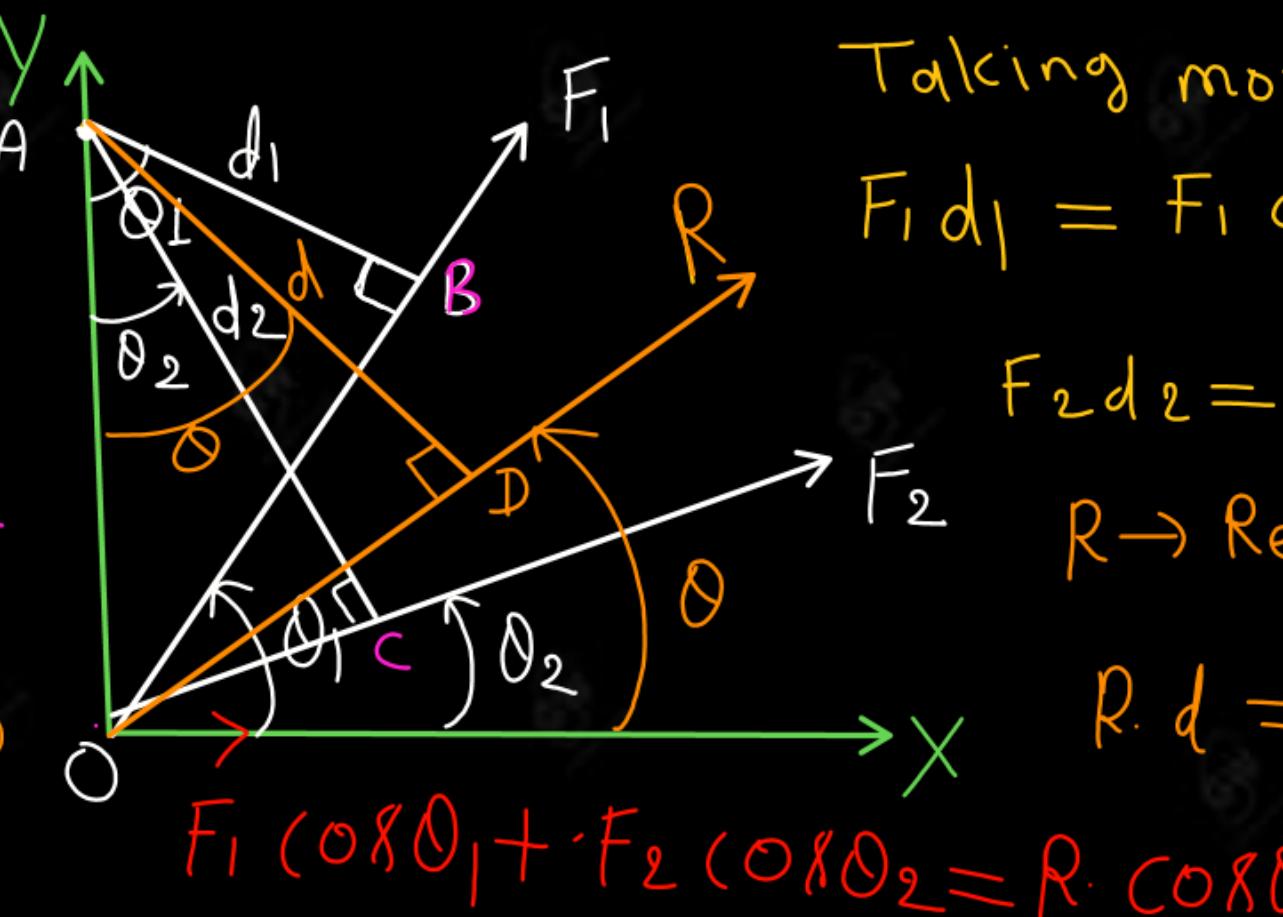
If many coplanar forces are acting on a body, then algebraic sum of moments of all the forces about a point in the plane of the forces is equal to the moment of their resultant about the same point.

**Proof:** Let us consider two forces  $F_1$  &  $F_2$  acting on a body as shown in fig

$$\triangle OBA \\ \cos\theta_1 = \frac{d_1}{OA} \Rightarrow d_1 = OA \cos\theta_1$$

$$\triangle OCA \\ \cos\theta_2 = \frac{d_2}{OA} \Rightarrow d_2 = OA \cos\theta_2$$

$$\triangle ODA \\ \cos\theta = \frac{d}{OA} \Rightarrow d = OA \cos\theta$$



Taking moment of  $F_1, F_2$  about 'A'

$$F_1 d_1 = F_1 OA \cos\theta_1 \dots \textcircled{I}$$

$$F_2 d_2 = F_2 OA \cos\theta_2 \dots \textcircled{II}$$

$R \rightarrow$  Resultant of  $F_1, F_2$

$$R \cdot d = R \cdot OA \cos\theta \dots \textcircled{III}$$

$$F_1 \cos\theta_1 + F_2 \cos\theta_2 = R \cos\theta$$

$$F_1 d_1 + F_2 d_2 = F_1 OA \cos \theta_1 + F_2 OA \cos \theta_2$$

$$F_1 d_1 + F_2 d_2 = OA (F_1 \cos \theta_1 + F_2 \cos \theta_2)$$

$$F_1 d_1 + F_2 d_2 = OA R \cos \theta - \text{--- (iv)}$$

from eqn (ii) & (iv)

$$\boxed{F_1 d_1 + F_2 d_2 = R \cdot d}$$

For 'n' no. of forces

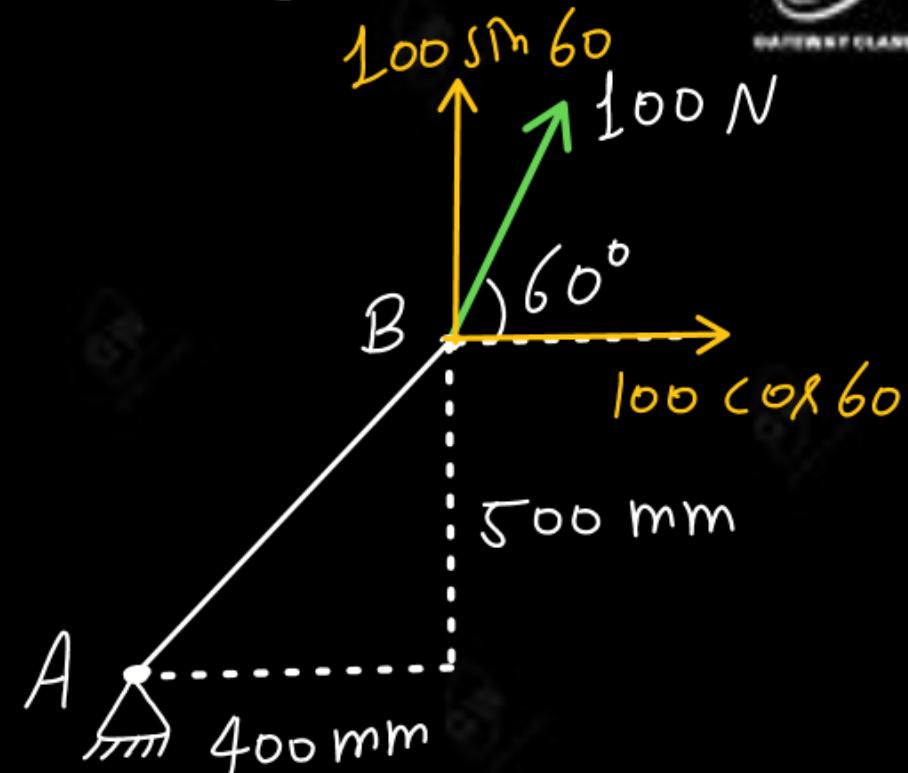
$$\boxed{F_1 d_1 + F_2 d_2 + F_3 d_3 + \dots + F_n d_n = R \cdot d}$$

Q.1 Find the moment of 100 N force acting at B about point A, as shown in fig. [AKTU] 2022-23

$$M_A = \frac{100 \cos 60 \times 500 - 100 \sin 60 \times 400}{}$$

$$M_A = -9641 \text{ N.mm}$$

$$M_A = 9641 \text{ N.mm} [A.C]$$



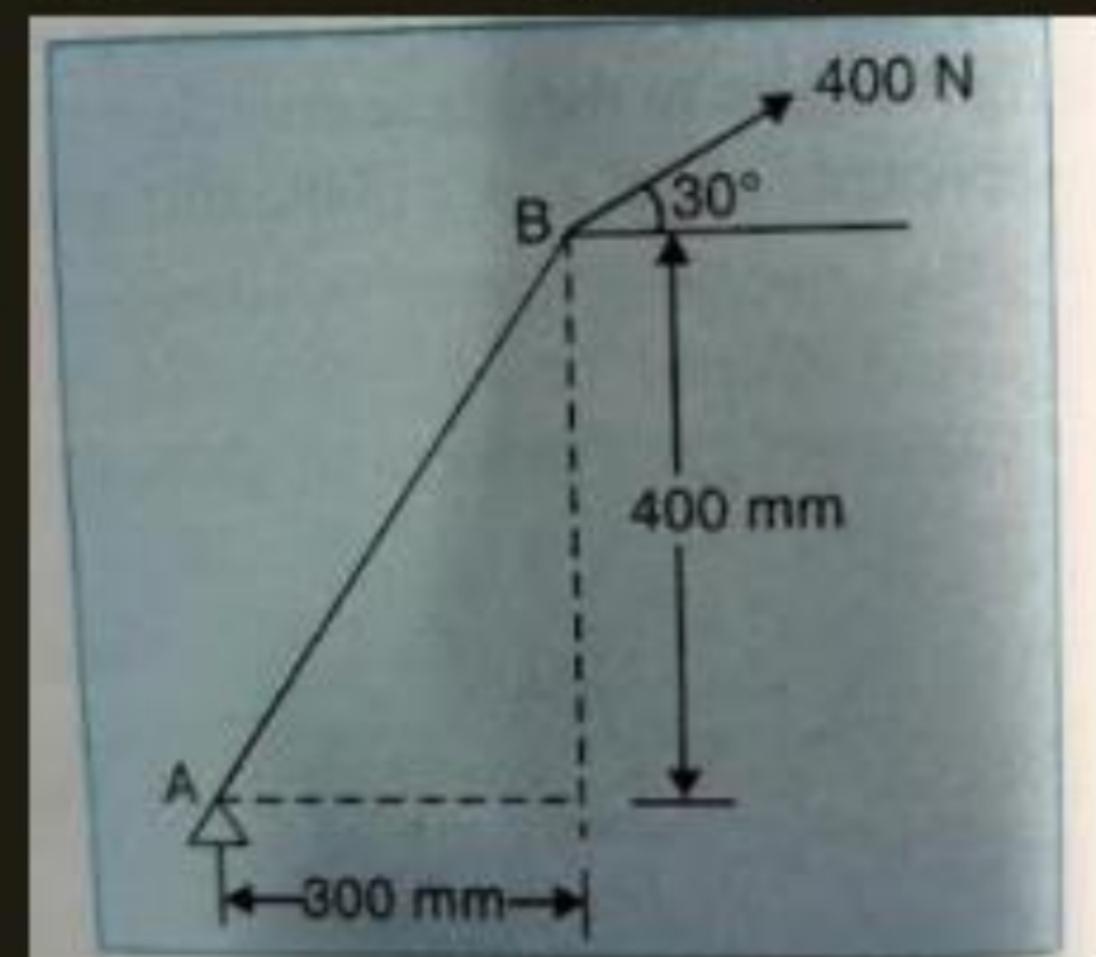
## FUNDAMENTALS OF MECHANICAL ENGINEERING

### DPP

#### Topic : Varignon's Theorem

**Q.1** Determine the moment of 400 N force acting at B in x-y plane about point A, as shown in fig.

**Ans. 78564 N-mm, Clockwise moment**



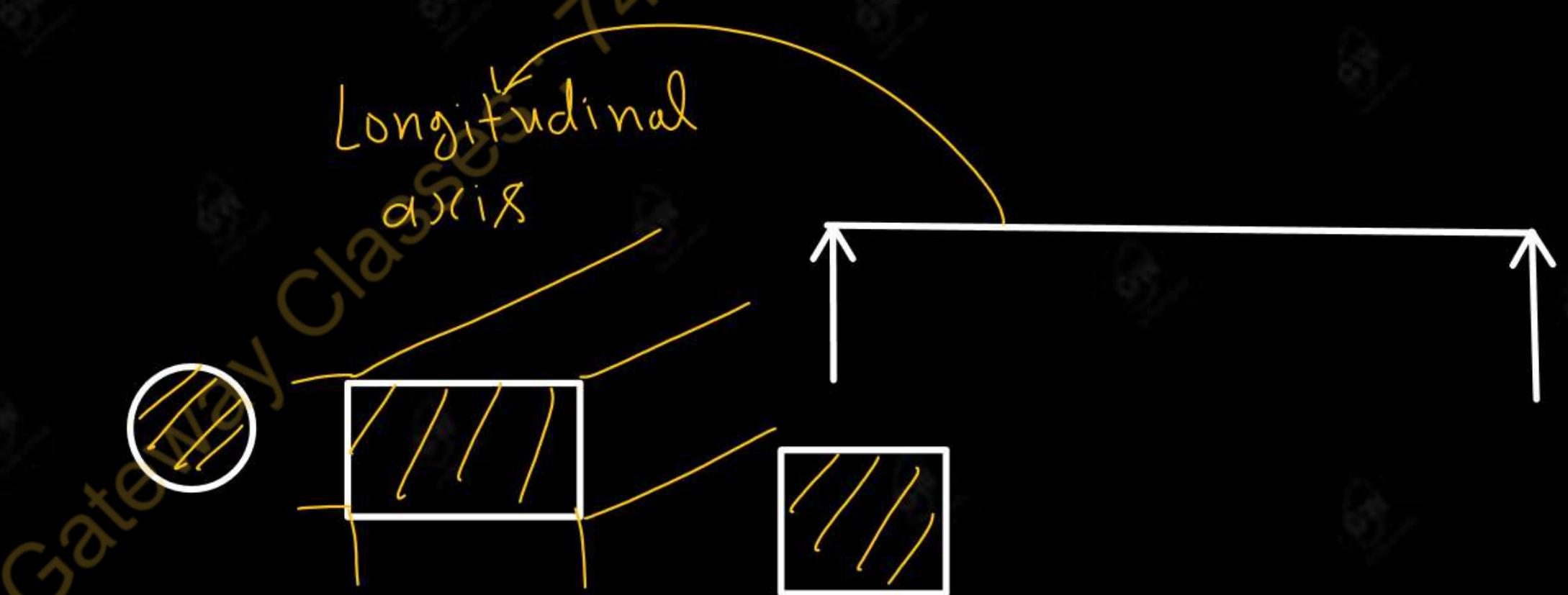
# Beam

Beam is defined as a structural member whose one dimension is very large as compare other lateral dimensions.

↳ Transverse (Lar)

or

Beam can be defined as a structural member which is subjected to shear forces.



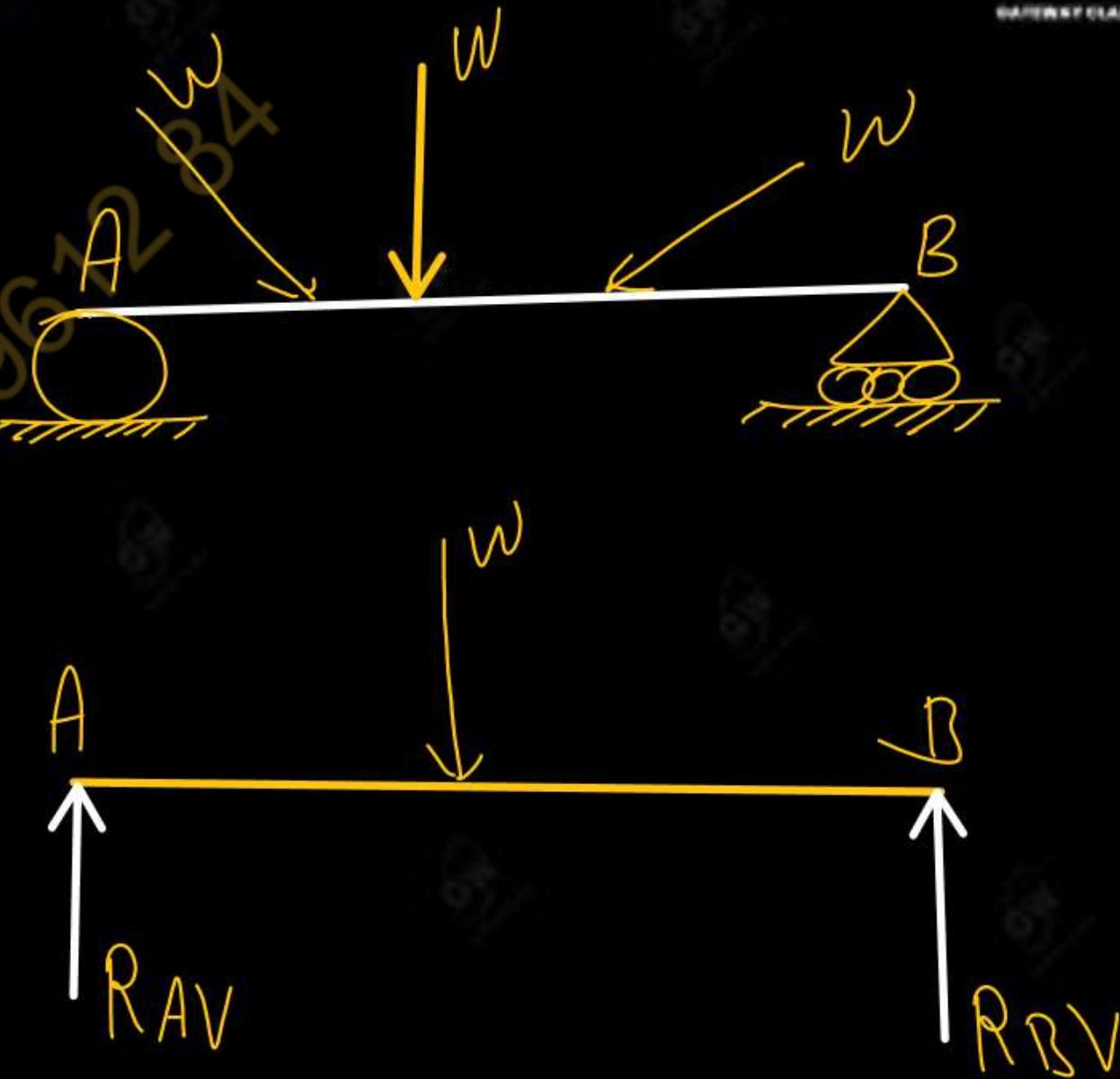
# Types of Supports

1. Roller support ✓
2. Hinge(Pin) Support ✓
3. Fixed (Built-in) Support ✓

Gateway Classes : 7455 9612 84

# 1. Roller support

- Axial motion is permitted ✓
- Rotation is permitted ✓
- Only vertical motion is restricted
- No of reaction = 1,  $R_V$

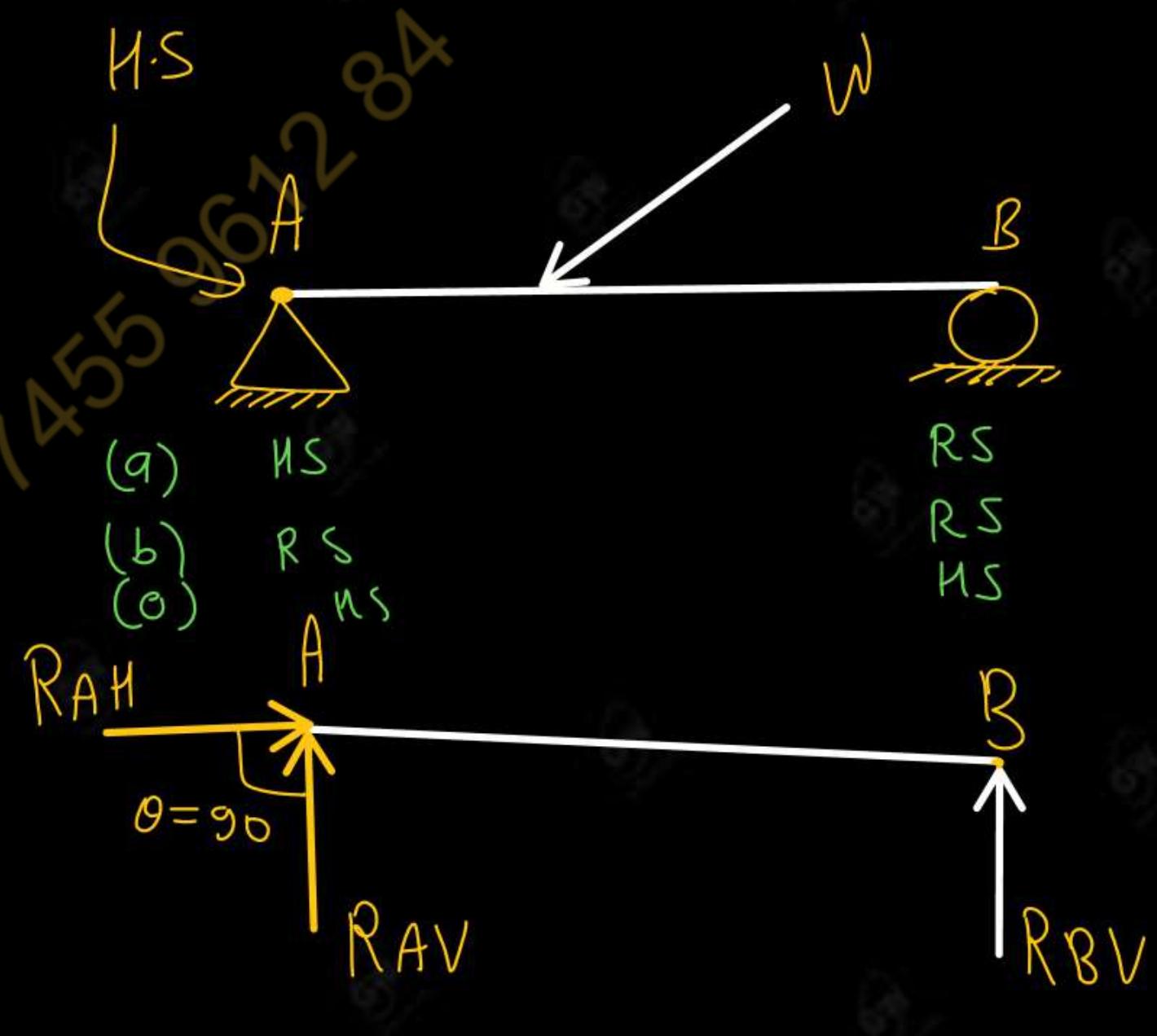


## 2. Pin(Hinged) support

- Vertical motion is restricted ✓
- Axial motion is restricted ✓
- Rotation is permitted
- No of reaction = 2,  $R_{AV}$  and  $R_{AH}$

$$R_A = \sqrt{(R_{AH})^2 + (R_{AV})^2}$$

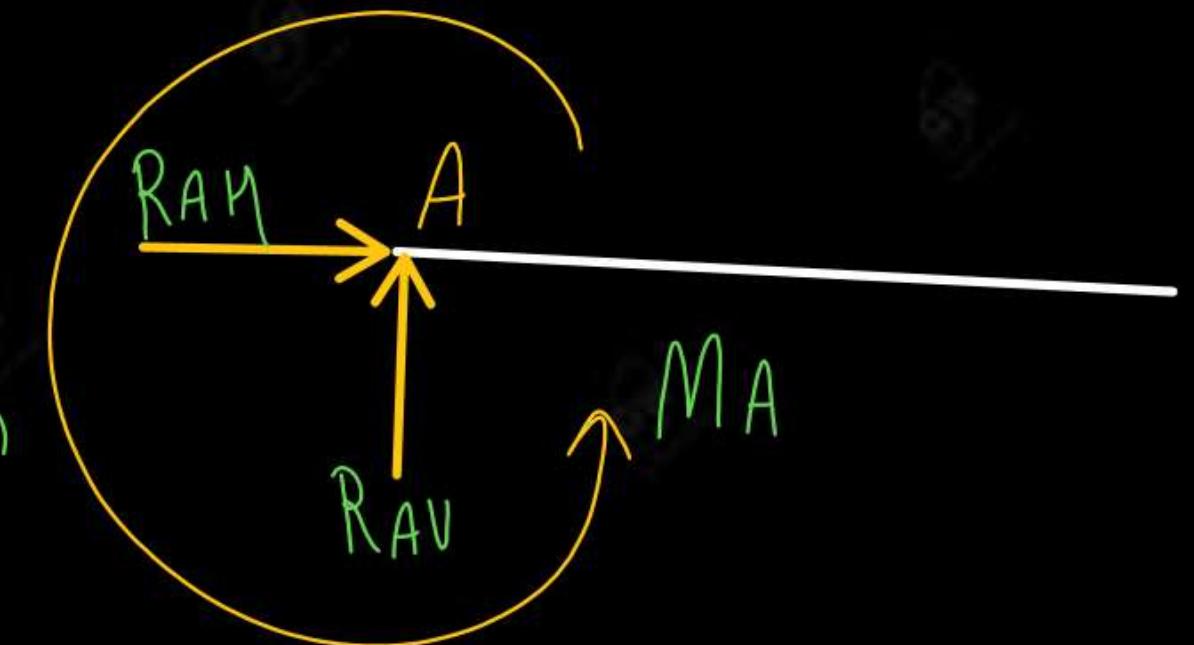
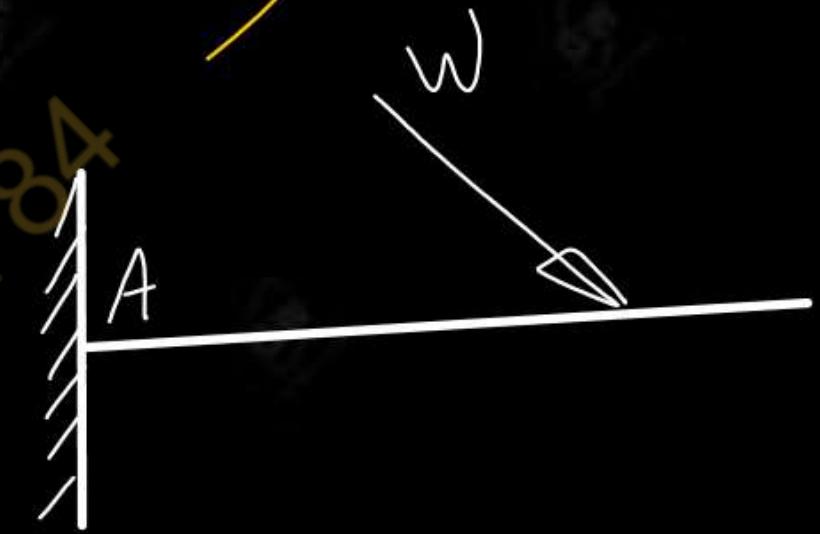
$$\theta = \tan^{-1} \left( \frac{R_{AV}}{R_{AH}} \right)$$



### 3. Fixed support (Built-in)

- Vertical motion is restricted. ✓
- Axial motion is restricted. ✓
- Rotation is restricted.
- No of reaction = 3,
- $R_{AV}$ ,  $R_{AH}$  and  $M_z$

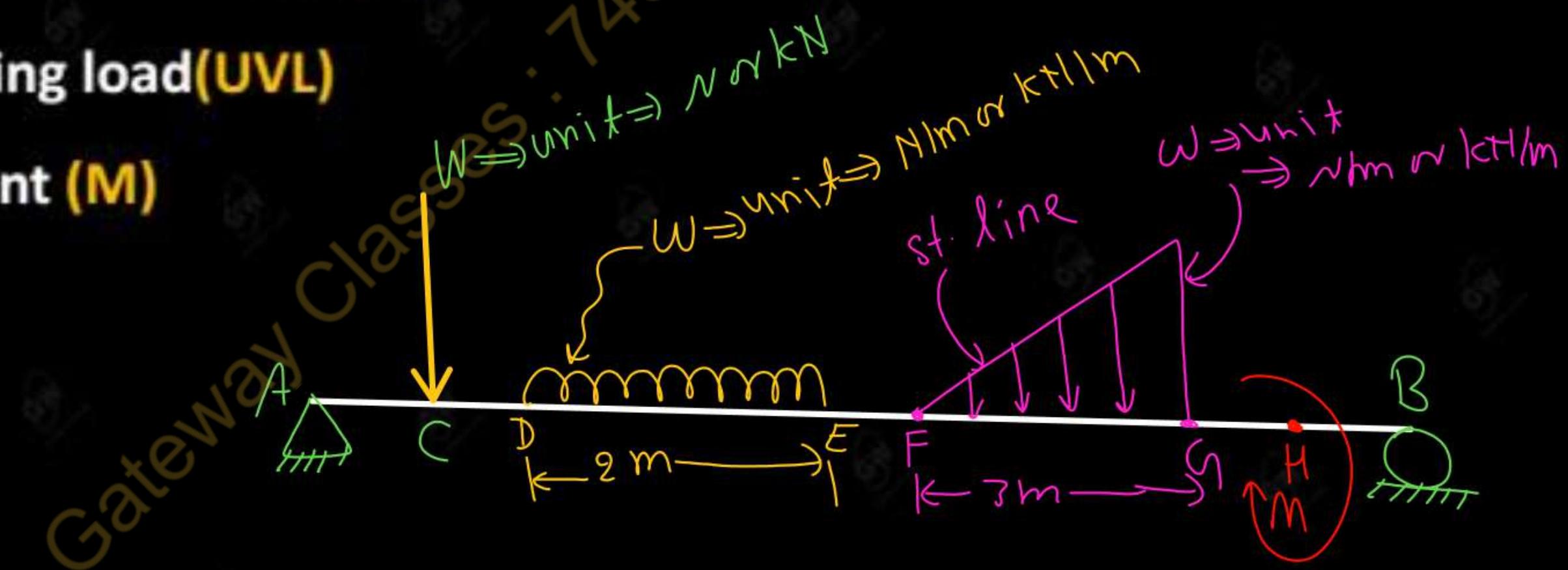
No. of Reactions = No. of motion  
Restricted



# Type of Load subjected to beam

Following are the types of load which are subjected to a beam

1. Concentrated load(**Point Load**)
2. Uniformly Distributed load(**UDL**)
3. Uniformly Varying load(**UVL**)
4. External Moment (**M**)



## 1. Concentrated load (Point load)

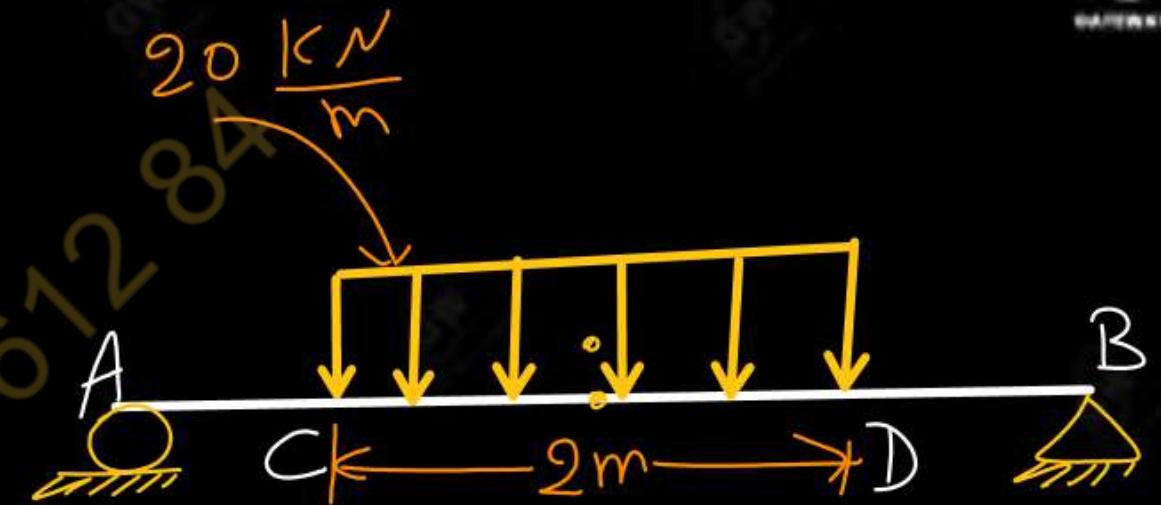
- This load acts at a point.
- It is represented by an arrow as shown in Fig.



Gateway Classes : 7455 9612 84

## 2. Uniformly Distributed load(UDL)

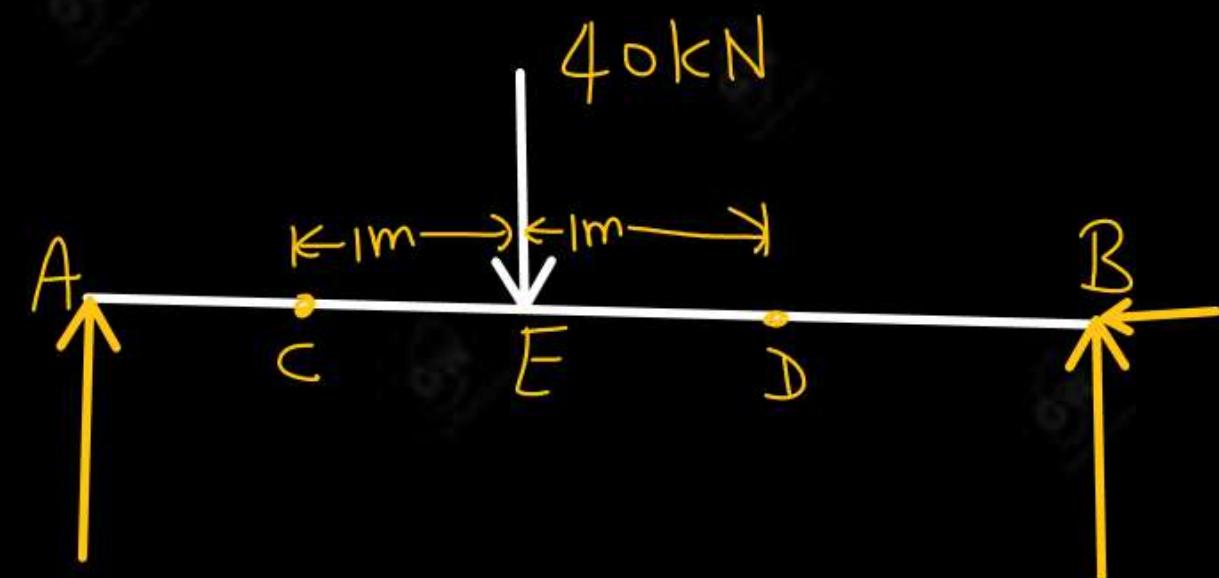
- Load acts over a certain length
- Intensity of load is uniform
- Total load = Area of plane fig (rectangle)
- Total load acts at middle of the loaded length.



$$\text{Total load} = 20 \frac{\text{kN}}{\text{m}} \times 2 \text{m}$$

$$\boxed{\text{Total load} = 40 \text{ kN}}$$

Point load



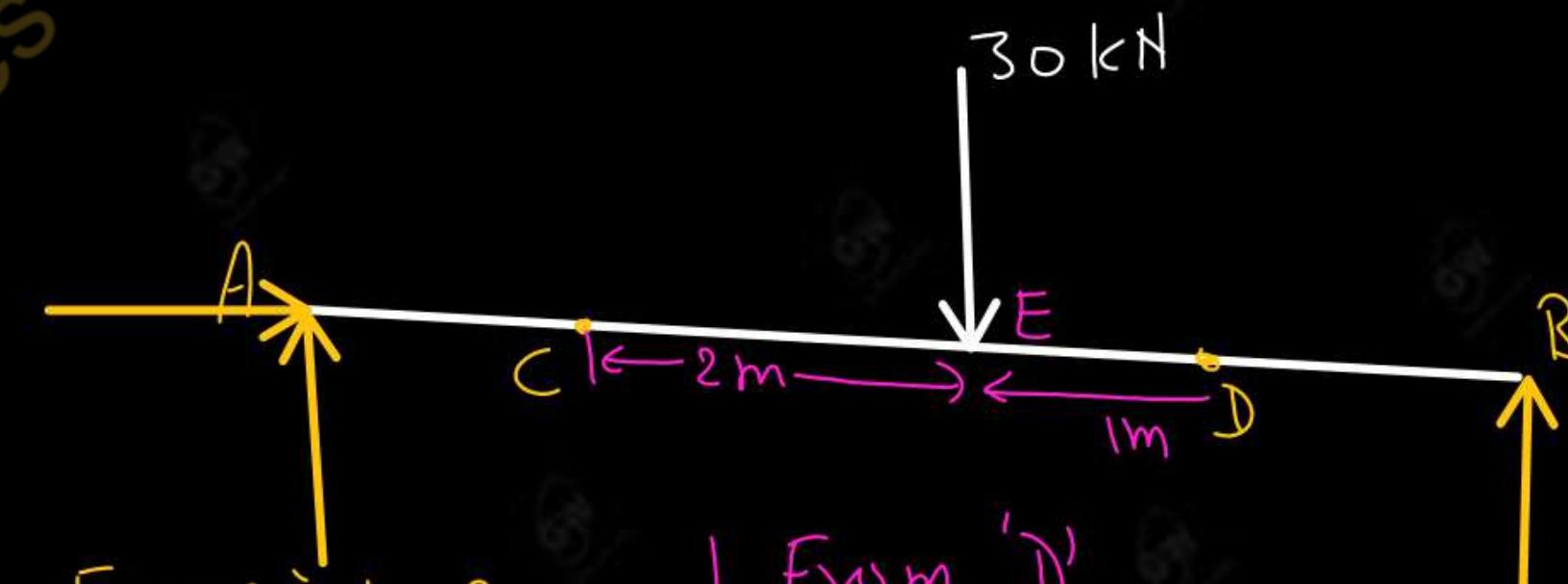
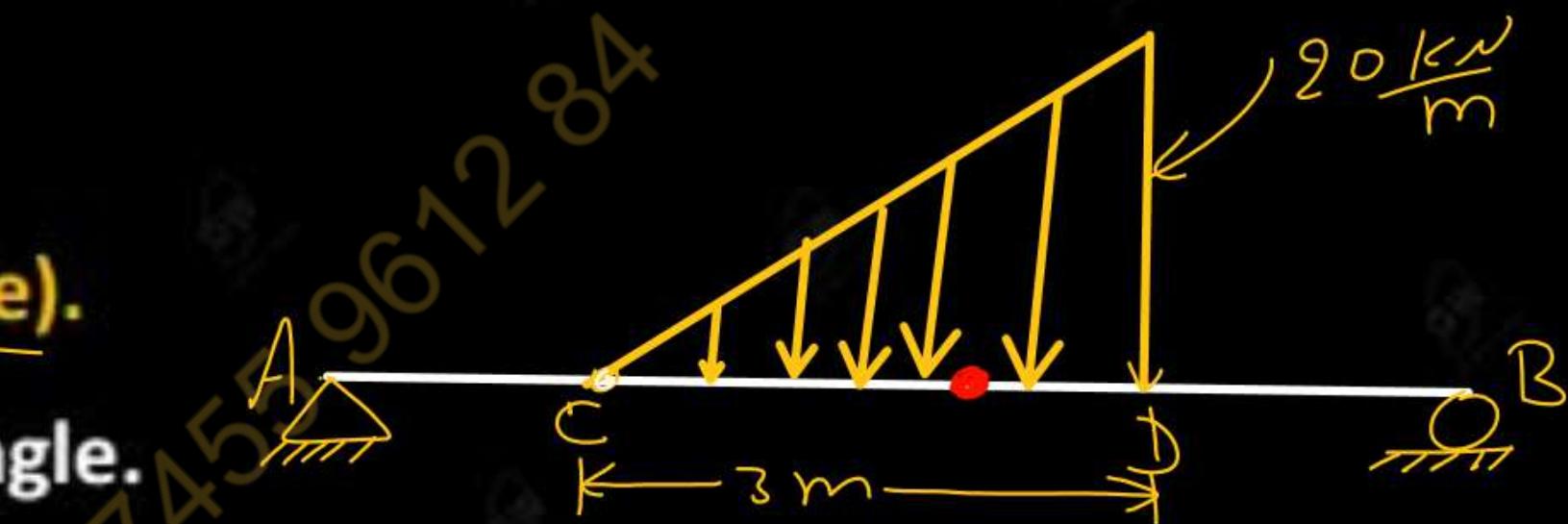
### **3. Uniformly Varying load(UVL)**

- The load varies Uniformly.
  - Total load = Area of plane fig. (Triangle).
  - Total load acts at Centroid of the triangle

$$\text{Total load} = \frac{1}{2} \times 3m \times 20 \frac{\text{kN}}{m}$$

Total load = 30 kN

## Point load



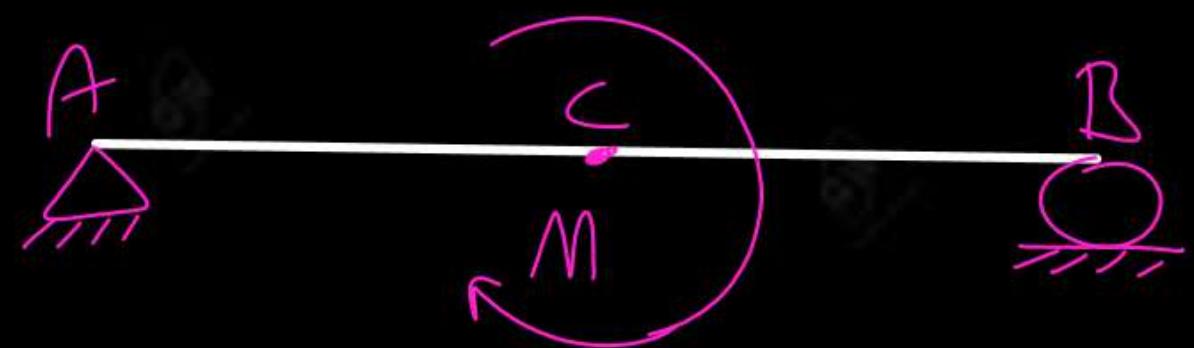
$$\text{From 'c' } = \frac{2}{3} \times 3m \\ = 2m$$

$$\text{From 'J') } \\ = \frac{1}{3} \times 3 \text{ m} \\ = 1 \text{ m}$$

## 4. Moment

- A beam may be subjected to external moment at certain points.
- Moment may be Clockwise or Anti-clockwise.

Anti-



**Q.1** Explain the types of supports in beam.

2 M

**Q.2** What are types of load subjected to beams ?

→ 45° ↗ 9 m 7 m ↘ 961284

Gateway Classes : 961284

# Equilibrium Equations

The body is said to be in equilibrium if the resultant of all the forces acting on it is zero.

## Coplanar Non-Concurrent Force System

Algebraic sum of forces in X-direction = 0 ,  $\sum F_x = 0$  — (I)

Algebraic sum of forces in Y-direction = 0 ,  $\sum F_y = 0$  — (II)

Algebraic sum of moment at any point = 0 ,  $\sum M = 0$  — (III)

# Stress

Stress :- "It is defined as internal resisting force per unit cross-sectional area".

$$\boxed{\sigma = \frac{R}{A}} \xrightarrow{* * *} \boxed{\sigma = \frac{P}{A} \rightarrow N} \rightarrow m^2$$

→ Stress

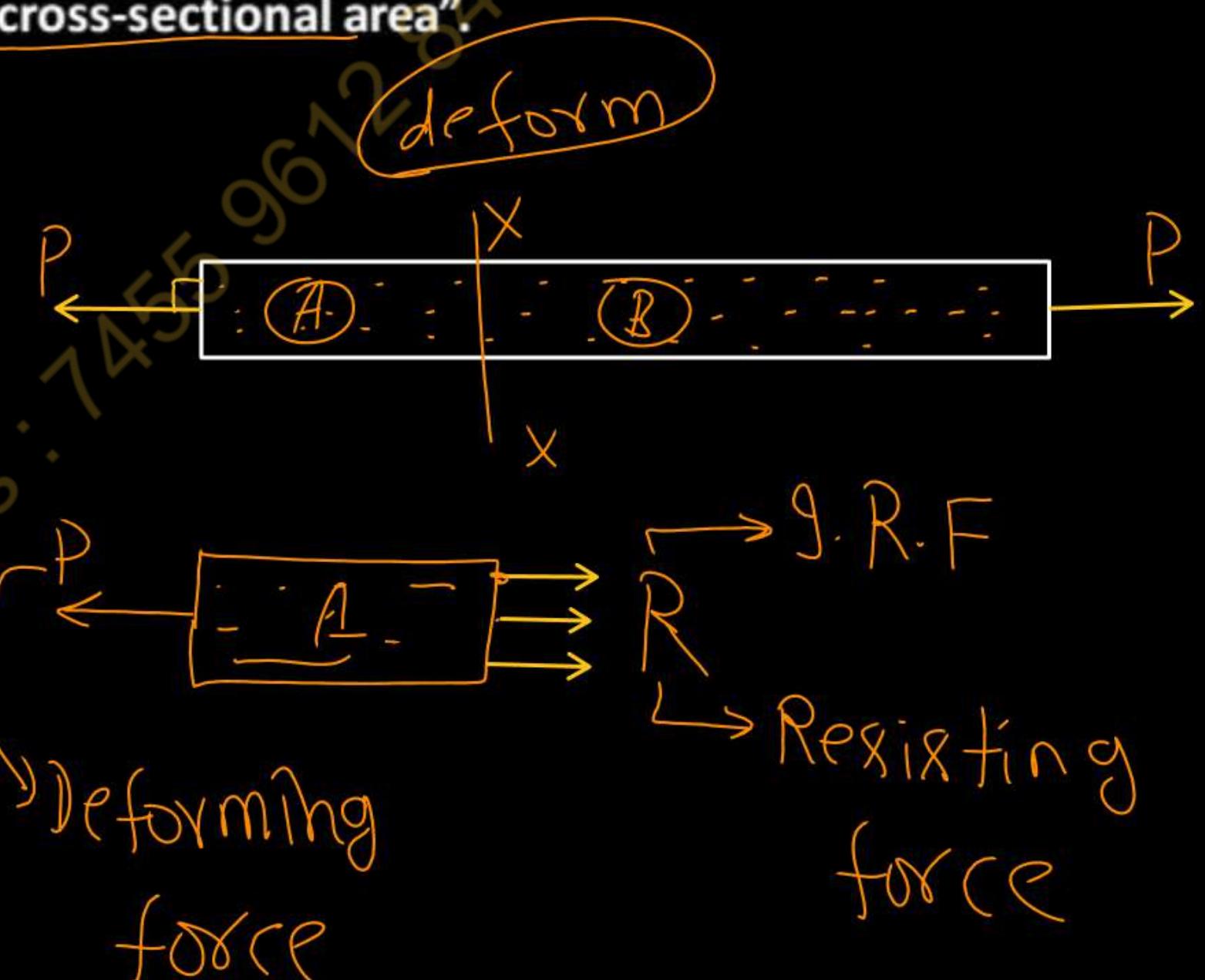
Unit of Stress

$$\rightarrow N/m^2$$

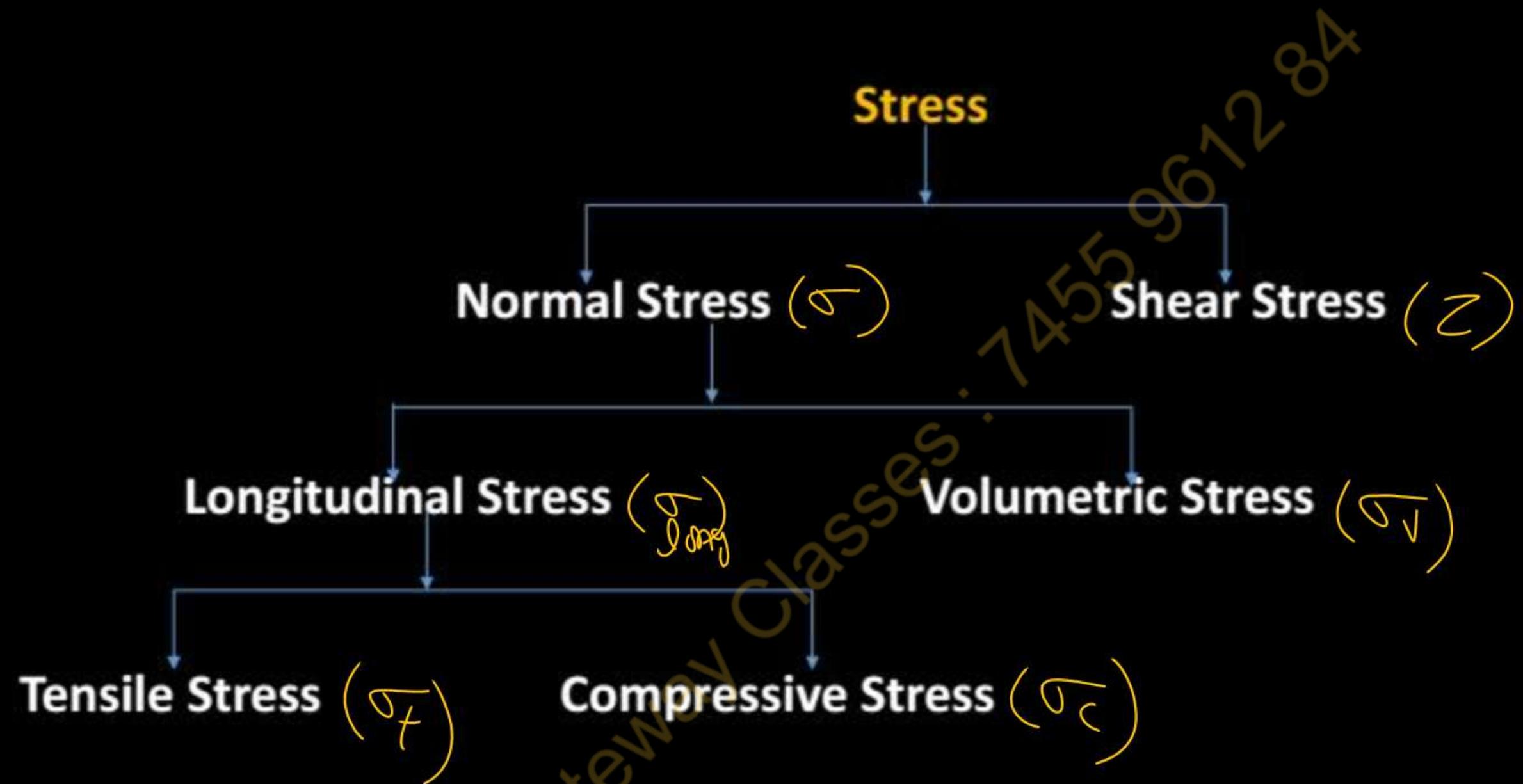
$$\rightarrow Pa \quad [1 Pa = 1 N/m^2]$$

$$kPa, MPa, GPa$$

$$| \quad | \quad | \\ 1 MPa = 1 N/mm^2$$



## Types of Stress



## Types of Stress

Lar  
T

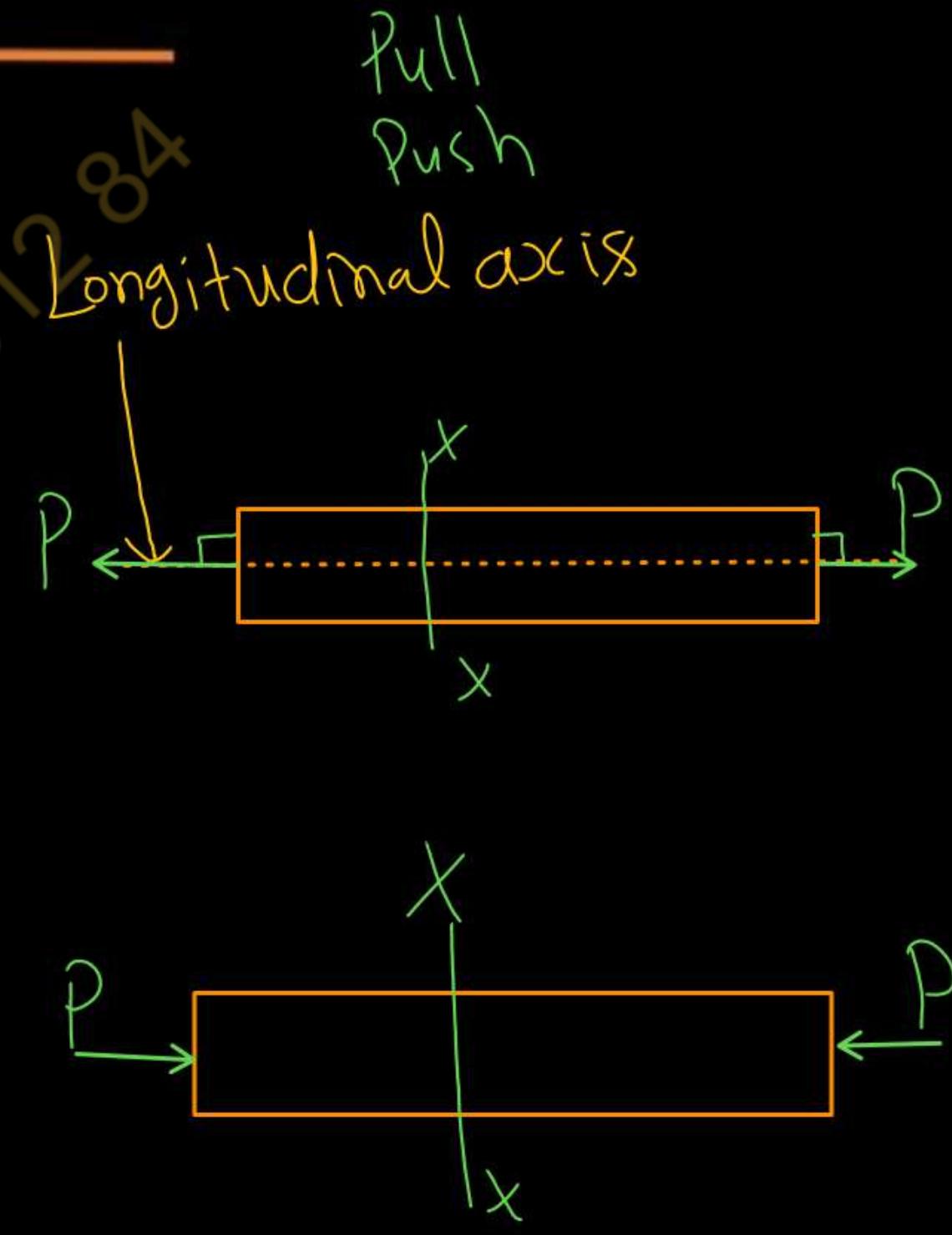
Normal Stress( $\sigma$ ):- Stress is said to be Normal stress when the direction of the deforming force is perpendicular to the cross-sectional area of the body.

➤ Tensile Stress( $\sigma_t$ ):- When a structural member is subjected to two equal and opposite tensile forces, the stress produced is called tensile stress. The tensile stress at any cross-section X-X is given as

$$\sigma_t = \frac{P}{A}$$

➤ Compressive Stress( $\sigma_c$ ):- When a structural member is subjected to two equal and opposite compressive forces, the stress produced is called compressive stress. The compressive stress at any cross-section X-X is given as

$$\sigma_c = \frac{P}{A}$$



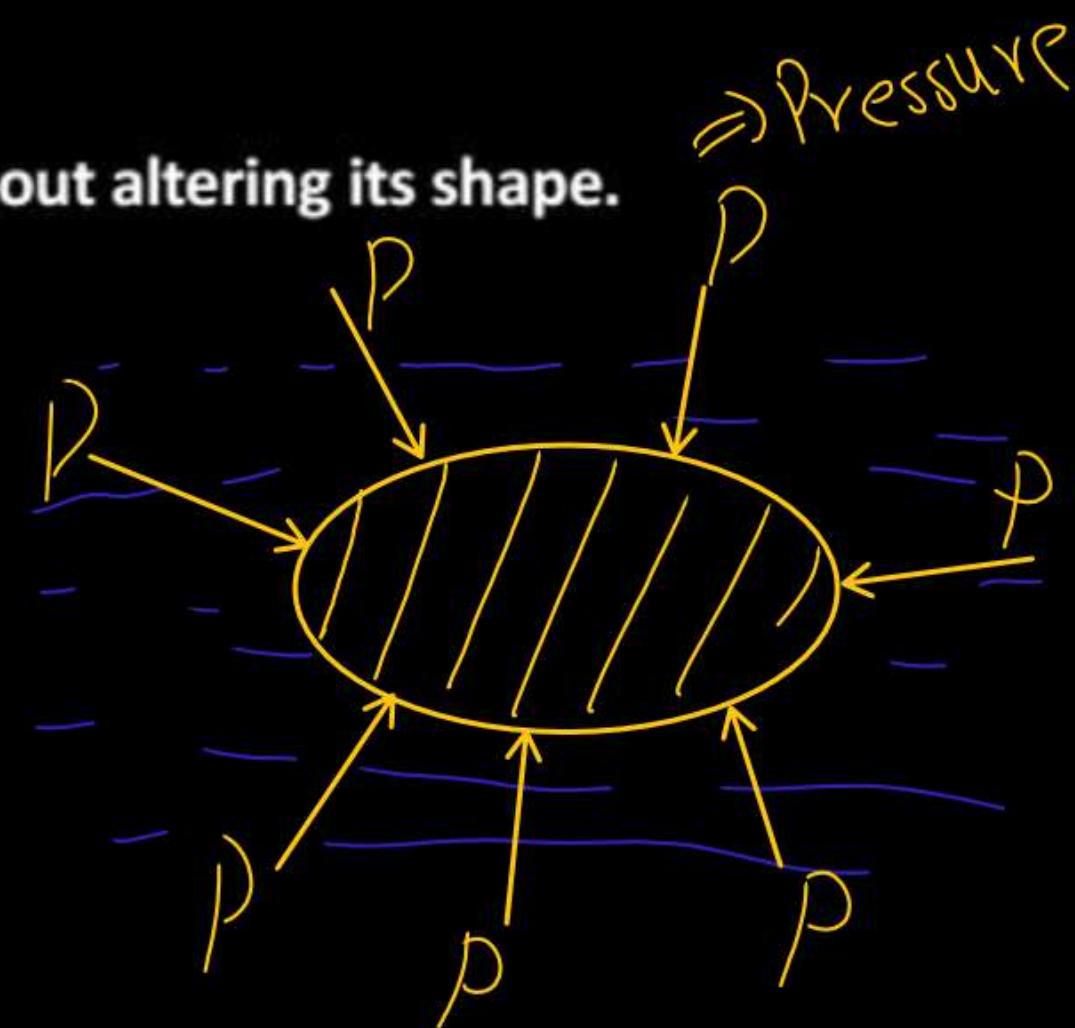
## Types of Stress

### ➤ Volumetric Stress ( $\sigma_v$ ):-

- Volumetric stress, also known as hydrostatic stress or bulk stress, is a type of stress that is associated with the uniform deformation of a material due to the application of an external force or pressure.
- Volumetric stress leads to changes in the volume of the material without altering its shape.
- Volumetric stress is typically expressed in terms of pressure.

$$\sigma_v = -P$$

→ Pressure  
N/m<sup>2</sup> (Pa)



## Types of Stress

Shear Stress ( $\tau$ ):- Stress produced by a force tangential to the surface of a body is known as shear stress.

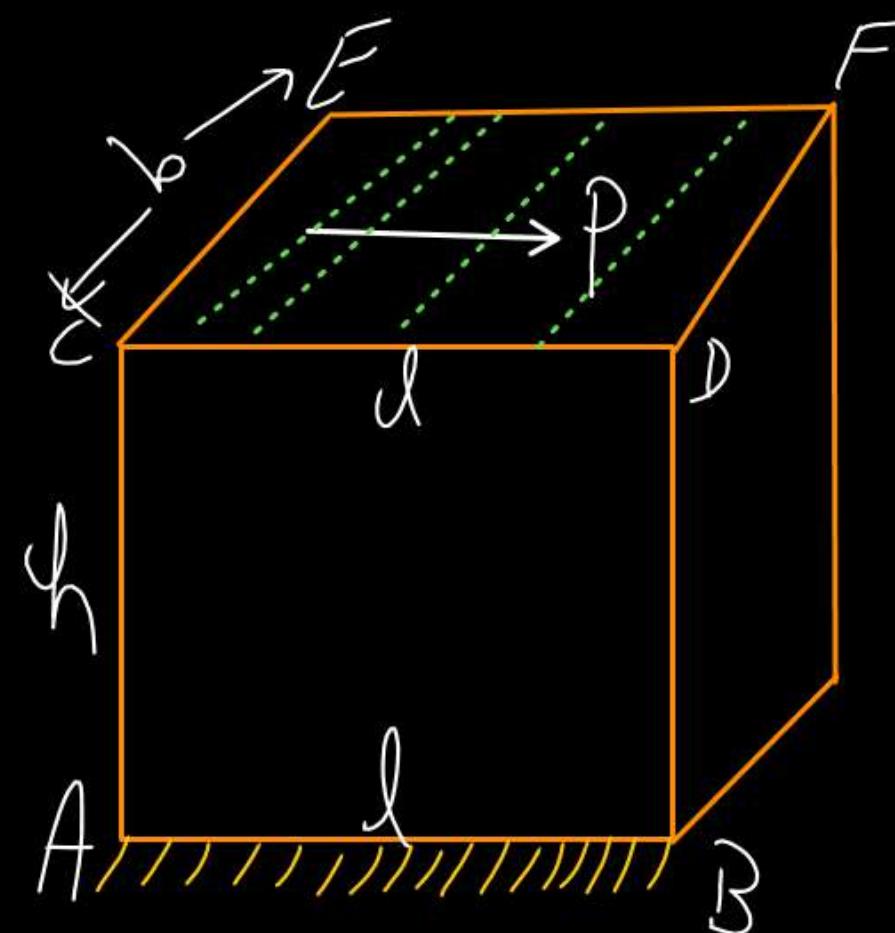
It is represented by  $\tau$ .

Consider a rectangular block ABCD fixed at the bottom plane and subjected to tangential force P at the upper plane.

Then

$$\text{Shear Stress, } \tau = \frac{\text{tangential force}}{\text{area of face DCFE}}$$

$$\tau = \frac{P}{bl}$$



## Strain

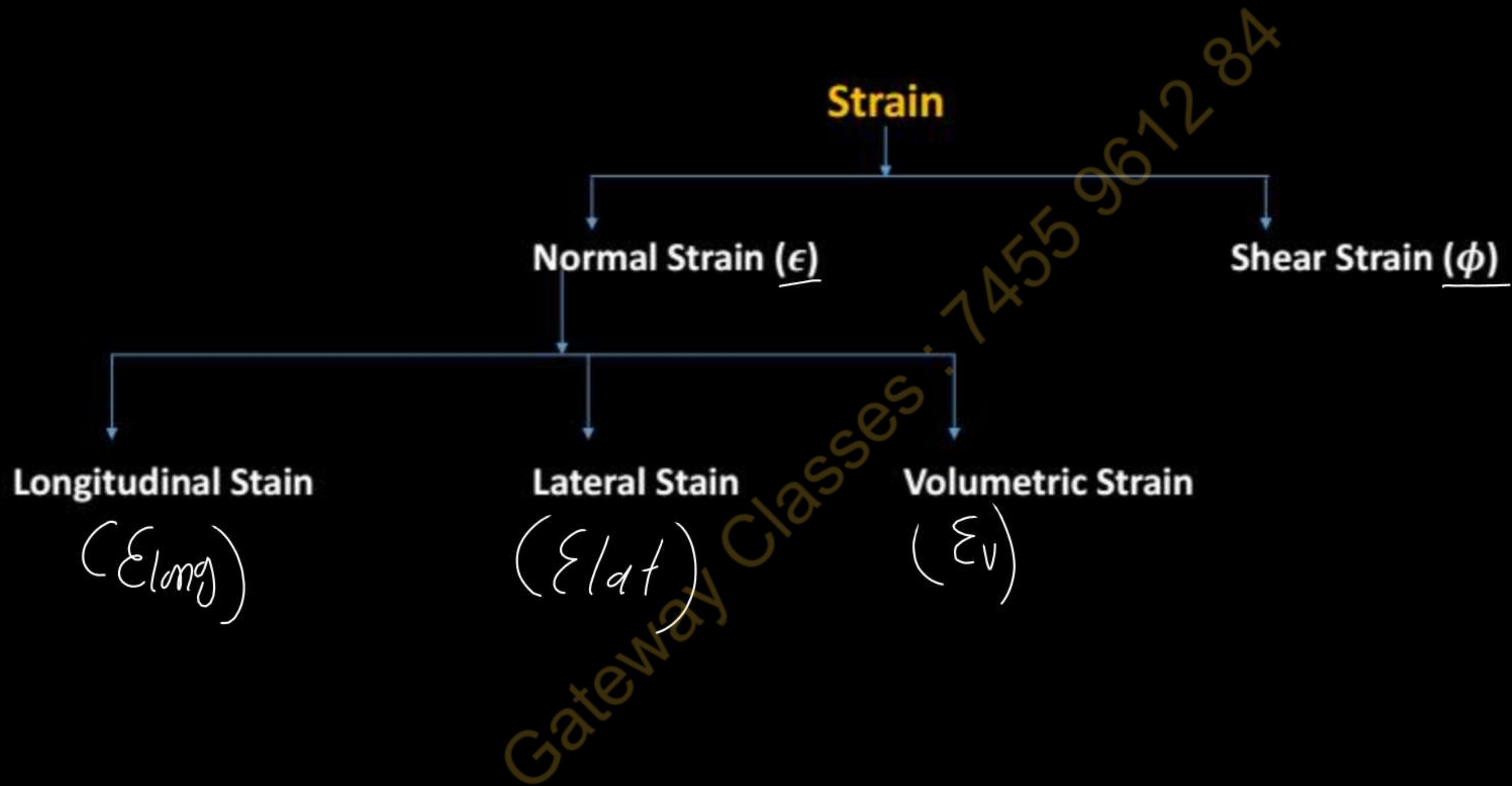
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- A body is said be strained when the relative position of the particles is changed due to loads acting on the member.
- This is deformation per unit original length is called strain.
- It can be defined as the ratio of change in dimension and original dimension.

$$\text{Strain} = \frac{\text{change in dimension}}{\text{original dimension}}$$

- Note: Strain is dimensionless quantity.

## Types of Strain



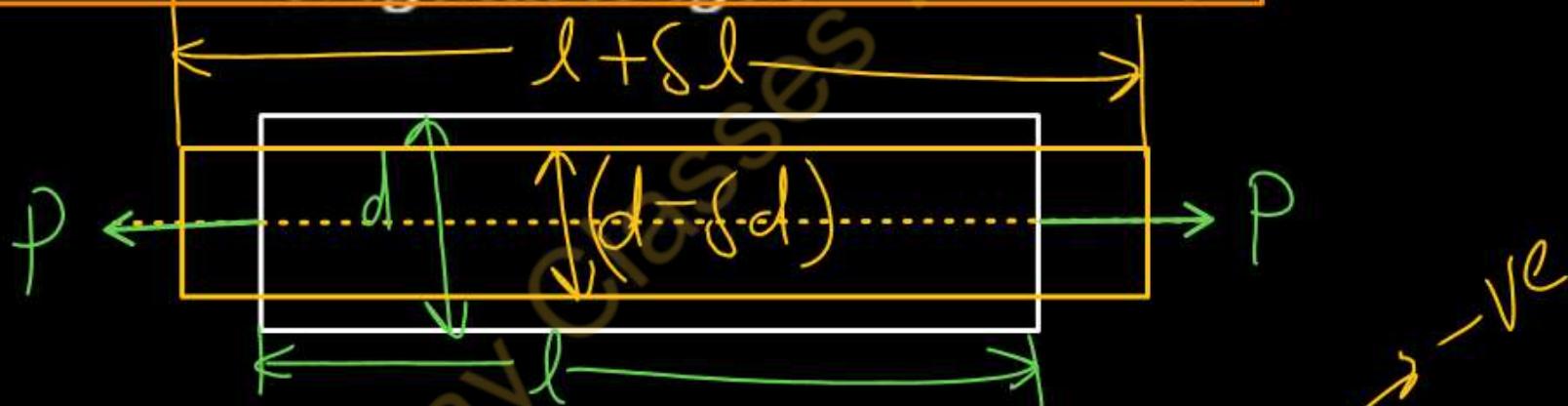
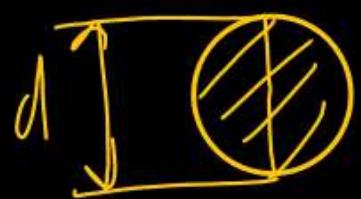
# Types of Strain

?

Longitudinal strain and Lateral strain:-

- Strain in the direction of applied load is called longitudinal strain or primary strain or linear strain.

$$\text{longitudinal strain} = \frac{\text{change in length}}{\text{original length}} \cdot \epsilon_{\text{long}} = \frac{\delta l}{l}$$



- Strain in the perpendicular direction of applied load is called lateral strain.

$$\text{lateral strain} = \frac{\text{change in diameter}}{\text{original diameter}}, \quad \epsilon_{\text{lat}} = \frac{\delta d}{d}$$

a   $A = a^2$

b   $A = a b$



$$A = \frac{\pi}{4} d^2$$

## Types of Strain

- **Shear Strain( $\phi$ ):-** Strain produced by a force, tangential to the surface of a body is known as Shear strain.
- Under the action of tangential force, the block ABCD gets distorted and takes the shape ABC'D' by deforming through an angle  $\phi$ .

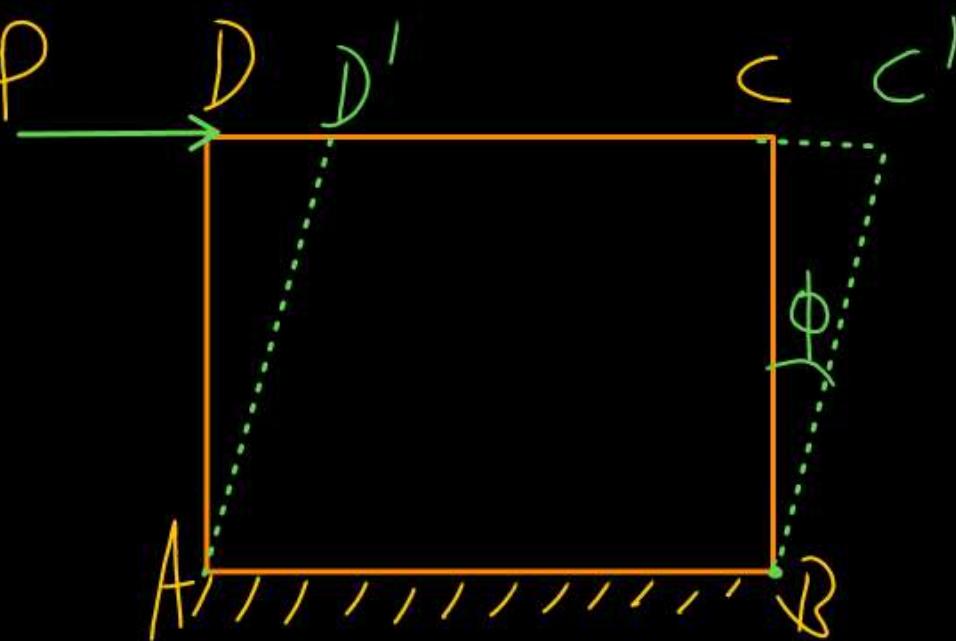
$\angle BCC'$

$$\tan \phi = \frac{cc'}{BC}$$

Because angle  $\phi$  is very small , then

$$\tan \phi = \phi = \frac{cc'}{BC}$$

The angular deformation  $\phi$  in radians represents the shear strain.



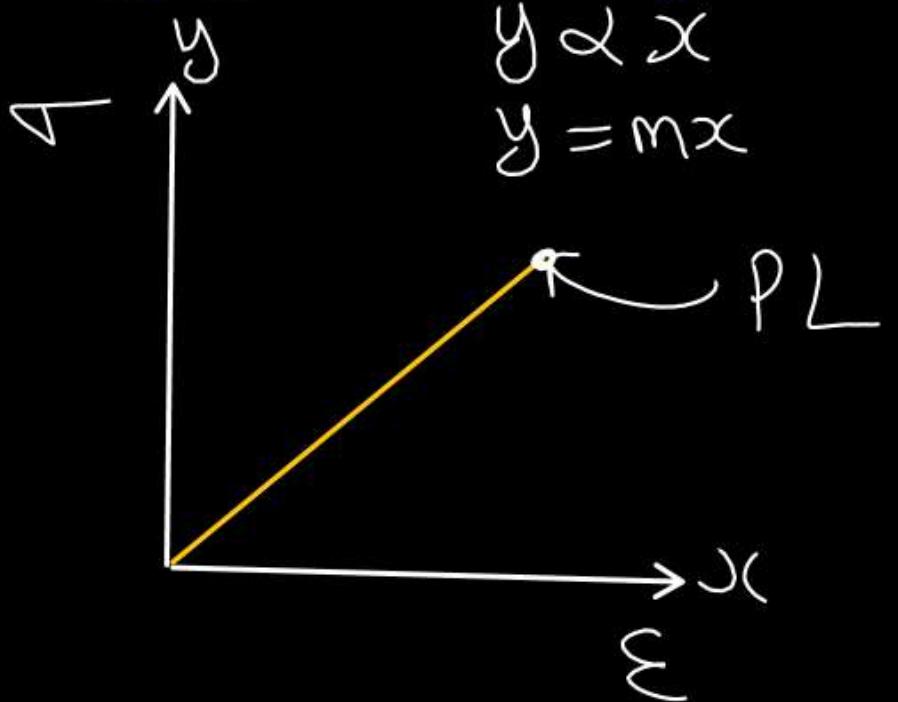
- **Hooke's Law:-** Hooke's law states that when a material is loaded within **proportional limit**, stress is directly proportional to strain.

Mathematically,

$$\sigma \propto \epsilon$$

$$\sigma = E \epsilon$$

$$E = \frac{\sigma}{\epsilon}$$



- Where the constant of proportionality  $E$  is called Young's modulus or modulus of elasticity.
- Young's modulus is the property of solid material.

## Elastic constants

**Elastic constants:-** For homogenous and isotropic material no. of elastic constants are 4 ( $\mu$ , E, G, K).

- (1) Poisson's Ratio ( $\mu$ ) (AKTU)
- (2) Modulus Of Elasticity or Young's Modulus (E)
- (3) Modulus Of Rigidity or Shear Modulus (G) (AKTU)
- (4) Bulk Modulus (K) (AKTU)

### Elastic constants

- (i) are used to determine strain theoretically
- (ii) are used obtain relationship between stress and strain

(1) **Poisson's Ratio( $\mu$ )**:- It is defined as the ratio of lateral strain to longitudinal strain.

$$\text{Poisson's ratio is } \mu = \frac{\text{Lateral strain}}{\text{Longitudinal strain}}$$

$$\Rightarrow \mu = \frac{\delta d / d}{\epsilon l / l}$$

- Poisson's ratio is dimensionless.
- The value of  $\mu$  lies between 0.25 to 0.33 for most of the engineering materials.

**(2) Modulus Of Elasticity or Young's Modulus(E):-** It is defined as the ratio of normal stress and normal strain, when material is loaded within elastic limit.

$$E = \frac{\sigma}{\epsilon} = \frac{\text{Normal Stress}}{\text{Normal Strain}}$$

- Unit of Young's modulus is same as unit of stress because strain is dimensionless quantity.
- N/m<sup>2</sup>, Pa, Kpa, Mpa, Gpa
- E is a property of the material.

(3) **Modulus Of Rigidity or Shear Modulus(G):-** It is defined as the ratio of shear stress and shear strain, when material is loaded within elastic limit.

$$G = \frac{\text{Shear stress } \tau}{\text{Shear strain } \phi}$$

- Unit of shear modulus is same as unit of stress because strain is dimensionless quantity.
- N/m<sup>2</sup>, Pa, Kpa, Mpa, Gpa
- G is a property of the material.

**(4) Bulk Modulus(K):-** It is defined as the ratio of volumetric stress and volumetric strain, when material is loaded within elastic limit.

$$K = \frac{\text{Volumetric Stress}}{\text{Volumetric Strain}} = \frac{\sigma_v}{\epsilon_v}$$

- Unit of Bulk modulus is same as unit of stress because strain is dimensionless quantity.
- N/m<sup>2</sup>, Pa, Kpa, Mpa, Gpa
- K is a property of the material.

## Elastic constants Derivations

$$\textcircled{I} \quad E = 3k(1-2\mu)$$

$$\textcircled{II} \quad E = 2G(1+\mu)$$

$$\textcircled{III} \quad E = \frac{9kG}{3k+G}$$

$$\mu \rightarrow \frac{1}{m}$$
  
$$G \rightarrow C$$

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# Elastic constants Derivations

$$K = \frac{\sigma_v}{\epsilon_v}$$

## Derivation-1 Relation between E, K and $\mu$

Let us consider a cubical block subjected to direct stress  $\sigma$  which act simultaneously along x, y, z directions.

Strain in x-direction

$$\epsilon_x = \frac{\sigma_x}{E} - \mu \frac{\sigma_y}{E} - \mu \frac{\sigma_z}{E}$$

$$\epsilon_{yx} = \frac{\sigma_x}{E} - \mu \frac{\sigma_y}{E} - \mu \frac{\sigma_z}{E}$$

$$\epsilon_{xz} = \frac{\sigma_x}{E} [1 - 2\mu] \quad \dots \dots \dots \textcircled{1}$$

Strain in y-direction

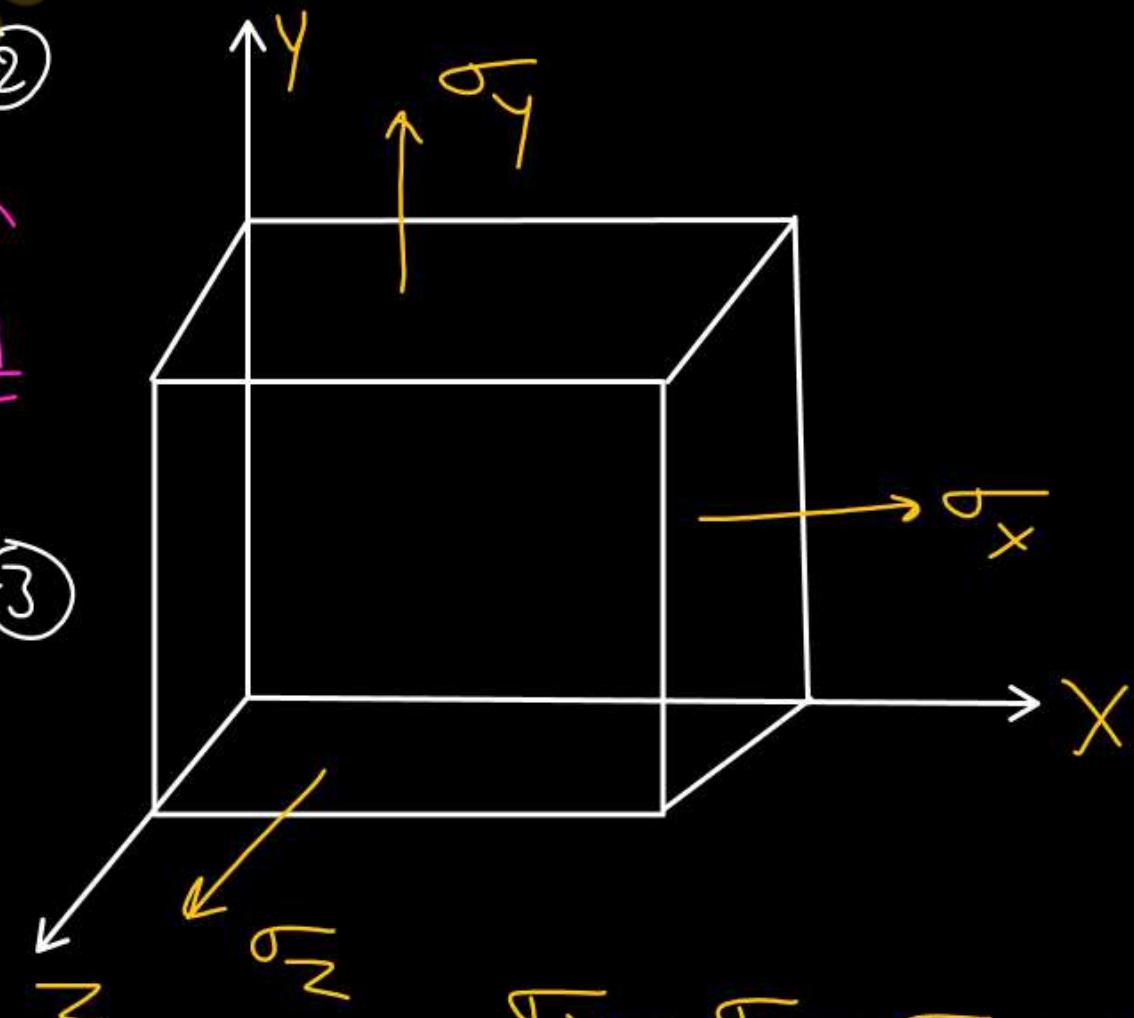
$$\epsilon_y = \frac{\sigma_y}{E} - \mu \frac{\sigma_x}{E} - \mu \frac{\sigma_z}{E}$$

$$\epsilon_y = \frac{\sigma}{E} [1 - 2\mu] \quad \dots \dots \textcircled{2}$$

Strain in z-direction

$$\epsilon_z = \frac{\sigma_z}{E} - \mu \frac{\sigma_x}{E} - \mu \frac{\sigma_y}{E}$$

$$\epsilon_z = \frac{\sigma}{E} [1 - 2\mu] \quad \dots \dots \textcircled{3}$$



$$\sigma_x = \sigma_y = \sigma_z = \sigma$$

Volumetric strain

$$\varepsilon_v = \varepsilon_x + \varepsilon_y + \varepsilon_z$$

$$\varepsilon_v = \frac{\sigma}{E}(1-\nu) + \frac{\sigma}{E}(1-\nu) + \frac{\sigma}{E}(1-\nu)$$

$$\boxed{\varepsilon_v = \frac{3\sigma}{E}(1-\nu)}$$

We know that

Bulk modulus,  $K = \frac{\sigma_v}{\varepsilon_v} = \frac{\sigma_v}{\dot{\varepsilon}_v}$

$$K = \frac{\sigma}{\frac{3\sigma}{E}(1-\nu)}$$

$$\boxed{E = 3K(1-\nu)}$$

## Derivation-2 : Relation between $E$ , $G$ and $\mu$

$$A C = A E$$

Let us consider a cubical block ABCD subjected to shear force at the top face and fixed at the bottom face.

The block experiences the following effect due to shear force F

1. Shear stress is induced at the face DC and AB.
2. The block distort to a new configuration ABC'D'
3. The diagonal AC elongate (Tension) and diagonal BD shorten (Compression)

$$\Delta E C' C$$

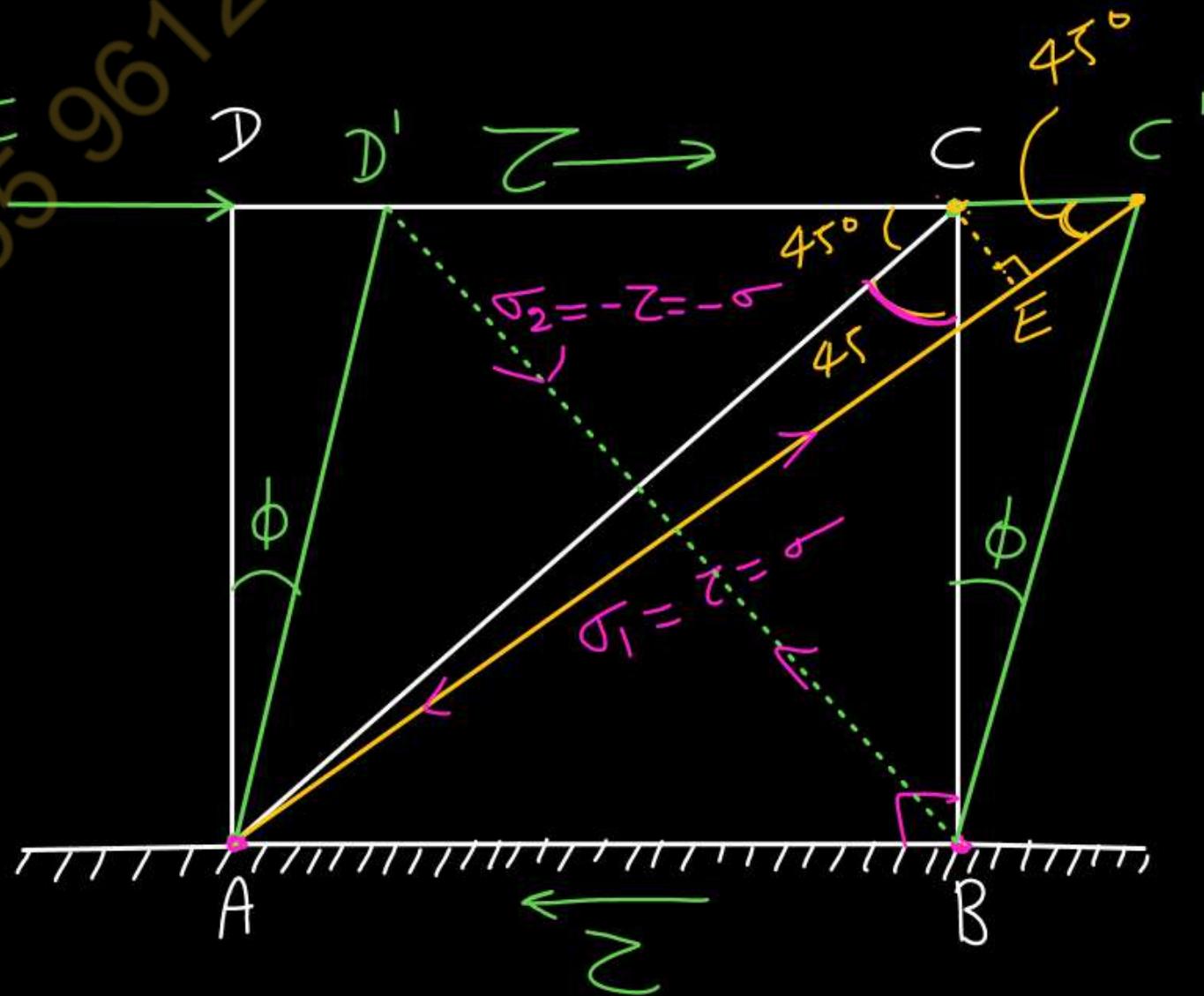
$$\cos 45^\circ = \frac{E C'}{C C'}$$

$$E C' = C C' \cos 45^\circ$$

$$\Delta A B C$$

$$\cos 45^\circ = B C / A C$$

$$A C = B C / \cos 45^\circ$$



$$\triangle BCC'$$

$$\tan \phi = \frac{CC'}{BC}$$

since  $\phi$  is very-very small

$$\tan \phi = \phi = \frac{CC'}{BC} \quad \text{--- (1)}$$

Draw  $CE \perp AC'$

linear strain in diagonal  $AC$

$$\epsilon = \frac{AC' - AE}{AC} = \frac{EC'}{AC}$$

$$\epsilon = \frac{CC' \cos 45}{BC} \frac{\cos 45}{\cos 45}$$

$$\epsilon = \frac{CC'}{BC} \cos^2 45$$

$$\epsilon = \frac{CC'}{BC} \left(\frac{1}{\sqrt{2}}\right)^2 = \frac{CC'}{2BC}$$

$$\epsilon = \frac{CC'}{2BC} \quad \text{--- (II)}$$

from eqn (1) & (I)

$$\epsilon = \frac{\phi}{2}$$

we know

$$\gamma = \frac{\epsilon}{\phi}$$

$$\Rightarrow \phi = \frac{\gamma}{\gamma}$$

$$\epsilon = \frac{\gamma}{2\gamma} \quad \text{--- (III)}$$

In case of pure shear

$$\sigma = \tau \sin \theta \quad [\theta = 45]$$

$$\tau = \tau \sin 90^\circ$$

$$\boxed{\sigma = \tau} \quad \text{***}$$

strain in diagonal  $AC$

$$\epsilon = \frac{\sigma}{2G} \quad \text{--- (IV)}$$

strain in diagonal  $AC$   
also can be written

$$\epsilon = \frac{\sigma_1}{E} - \mu \frac{\sigma_2}{E}$$

$$= \frac{\sigma}{E} - \mu \left(-\frac{\sigma}{E}\right)$$

$$\epsilon = \frac{\sigma}{E} + \mu \frac{\sigma}{E}$$

$$\epsilon = \frac{\sigma}{E} (1+\mu) \quad \text{--- (V)}$$

from  $\epsilon_1^n \textcircled{iv} \& \textcircled{v}$

$$\frac{\sigma}{\sigma_n} = \frac{\sigma}{E}(1+\mu)$$

$$E = \sigma_n(1+\mu)$$

## Derivation-3 : Relation between E, G and K

we know that,

$$E = 2G(1+\mu) \dots \dots \dots \text{(i)}$$

$$E = 3K(1-2\mu) \dots \dots \dots \text{(ii)}$$

eliminate  $\mu$  from eqn (i) and (ii)

$$E = 2G(1+\mu)$$

$$1 + \mu = \frac{E}{2G}$$

$$\mu = \frac{E}{2G} - 1$$

Put value of  $\mu$  in eqn (ii)

$$E = 3K \left( 1 - \frac{E}{G} + 2 \right)$$

$$E = 3K \left( 3 - \frac{E}{G} \right)$$

$$E = 9K - \frac{3KE}{G}$$

$$E + \frac{3KE}{G} = 9K$$

$$E \left( 1 + \frac{3K}{G} \right) = 9K$$

$$E = \frac{9KG}{3K+G}$$

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$$\frac{1}{m}, E, C, K$$
$$\downarrow \quad \downarrow \quad \downarrow \quad \downarrow$$
$$m \quad E \quad C \quad K$$

①  $E = 2G(1+m)$

②  $E = 3K(1-2m)$

③  $E = \frac{9Kb}{3K+b}$

$$E = 2G(1+m) = 3K(1-2m) = \frac{9Kh}{3K+h}$$

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## Stress-strain diagram for ductile and brittle materials

- A stress-strain diagram is a graphical representation of the relationship between stress and strain in a material.
- It provides valuable information about a material's mechanical properties and behavior under various loading conditions.
- Ductile and brittle materials have different stress-strain behaviors, and their diagrams look distinct.

Ductile Material: [Mild steel]

- Ductile materials are capable of undergoing substantial plastic deformation before failure.

## Stress-strain diagram for ductile materials (mild steel)

The stress-strain curve for a ductile material have the following characteristics:

- ❖ **Proportional limit(A):**
  - Upto this limit, stress is a linear function of strain and material obeys Hook's law.
  - O-A is a straight line and its slope represents the value of modulus of elasticity.  
$$\sigma \propto \epsilon$$
$$\sigma = E\epsilon \Rightarrow E = \frac{\sigma}{\epsilon}$$
- ❖ **Elastic limit(B):**
  - It represents maximum stress upto which material is still able to regain its original shape and size after removal of load i.e. upto this point deformation is recoverable.

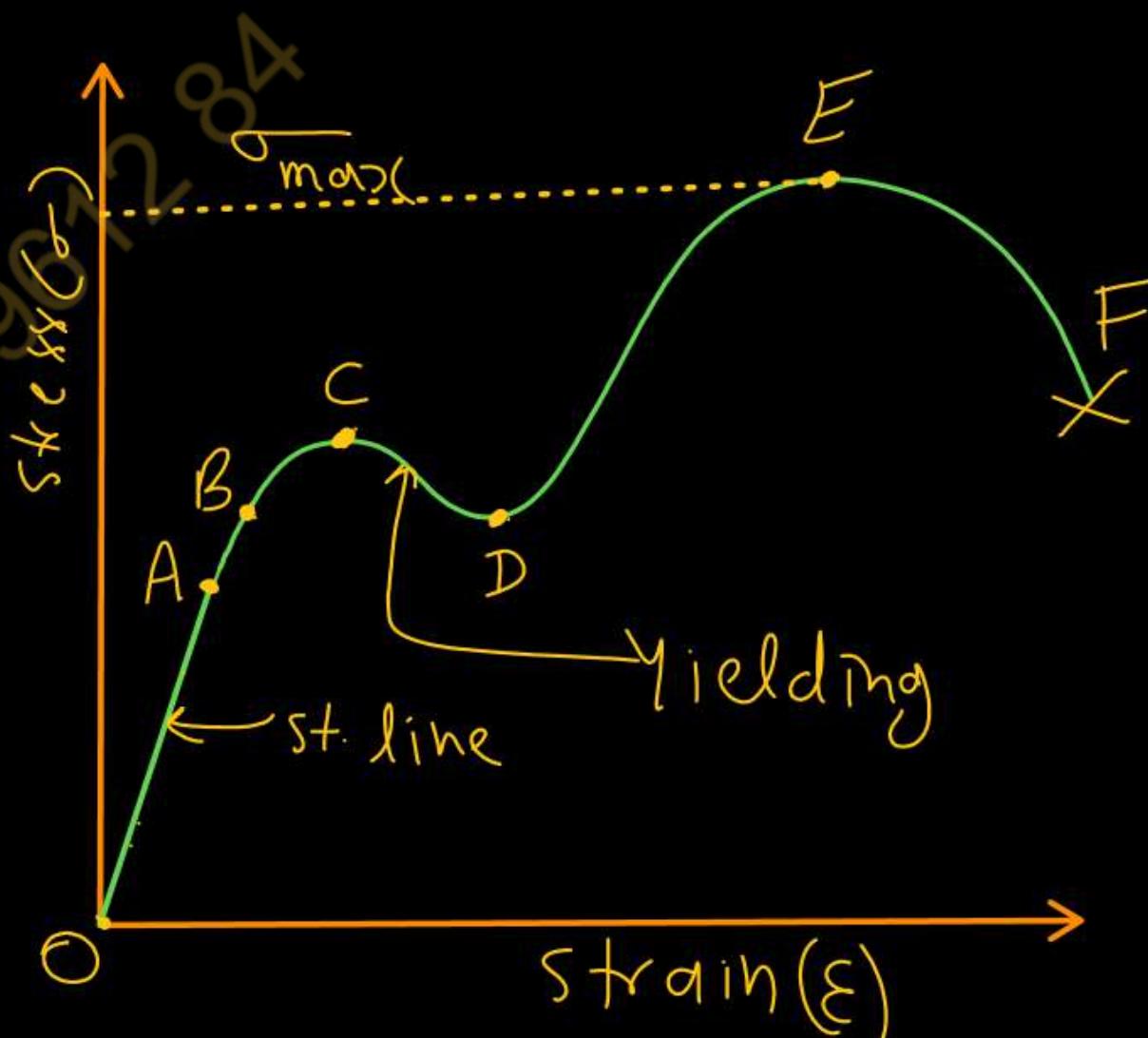


fig. Stress-strain diagram  
for ductile materials (mild steel)

- ❖ **Upper yield point(C) and Lower yield point(D):**
  - Beyond elastic limit, the material shows considerable strain even though there is no increase in load.
  - Deformation is not fully recoverable i.e. the behaviour of material is inelastic.
  - This phenomenon from C to D is called yielding.
  
- ❖ **Ultimate stress point(E):**
  - After yielding has taken place, the material becomes hardened and increase in load is required to take the material to its maximum stress at point E.
  - Point E represents the maximum stress of this curve and this point is known as ultimate stress point.
  
- ❖ **Breaking point(F):**
  - In the portion EF, there is falling off the load(stress) from the maximum until fracture takes place at F.
  - The point F is known as fracture or breaking point and corresponding stress is called the breaking stress.

## Stress – Strain diagram for brittle material (i.e. Cast Iron)

- For brittle materials, like cast Iron, no appreciable deformation is obtained and the failure occurs without yielding.
- Ductile materials give indication before failure.
- Brittle material fails suddenly with out any indication.

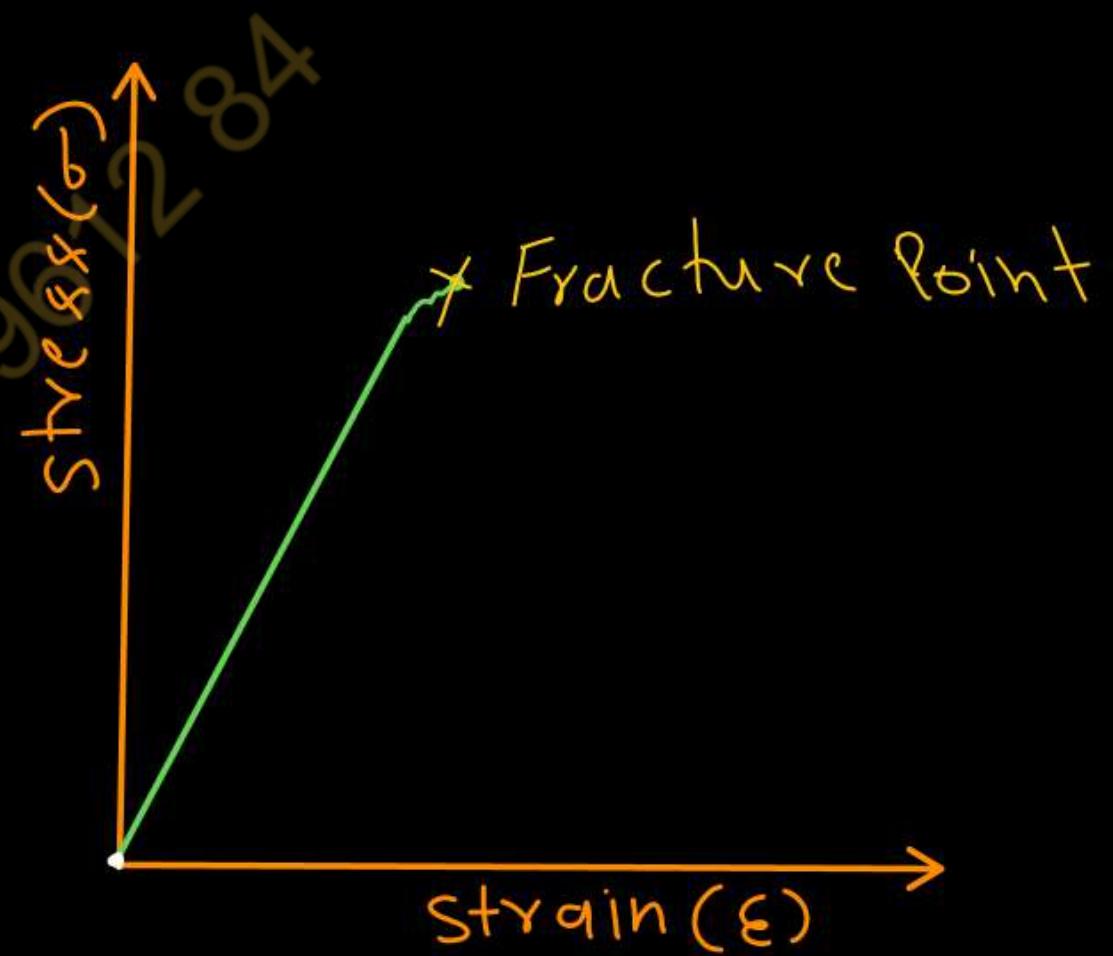


Fig. Stress – Strain diagram for brittle material (i.e. Cast Iron)

FOS is defined as Ultimate stress to working stress.

$$\text{FOS} = \frac{\text{Ultimate stress}}{\text{Working stress}} = \frac{\text{Ultimate stress}}{\text{Allowable stress}}$$

- For safe design FOS must be greater than unity.  $\text{FOS} > 1$
- How much FOS is required depends on-
  - Materials Properties , Design, Loading Conditions , Uncertainties, Applications etc.
- For example, civil engineering structures like bridges and buildings may have a factor of safety around 1.5 to 2, while aerospace and nuclear engineering applications may require factors of safety that are much higher, often exceeding 3 or 4.
- As the FOS increases, the cost of the product also increases therefore cost is also a considerable parameter.

$$\text{FOS} \uparrow \Rightarrow \text{Cost} \uparrow$$

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