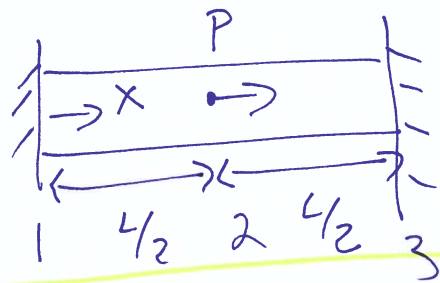


FEM 1D

Example-0 Recap



E, A, L given

Find displacement

$$\begin{aligned}\Pi = & \frac{1}{2} \int_{x_1}^{x_2} E \left(\overset{\textcircled{1}}{N}_1' u_1 + \overset{\textcircled{1}}{N}_2' u_2 \right)^2 A dx + \\ & \frac{1}{2} \int_{x_2}^{x_3} E \left(\overset{\textcircled{2}}{N}_1' u_2 + \overset{\textcircled{2}}{N}_2' u_3 \right)^2 A dx - P u_2 - R_1 u_1 \\ & - R_3 u_3\end{aligned}$$

$$\Pi = \Pi^{\textcircled{1}} + \Pi^{\textcircled{2}}$$

Minimize P.E.

$$\begin{bmatrix} \frac{2\pi}{2u_1} \\ \frac{2\pi}{2u_2} \\ \frac{2\pi}{2u_3} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} \frac{2\pi^{(1)}}{2u_1} + \frac{2\pi^{(2)}}{2u_1} \\ \frac{2\pi^{(1)}}{2u_2} + \frac{2\pi^{(2)}}{2u_2} \\ \frac{2\pi^{(1)}}{2u_3} + \frac{2\pi^{(2)}}{2u_3} \end{bmatrix}$$

elements →

of unknowns ↓

element ① does not have node 3

Assembly process

F - element 1

$$\frac{2\pi^{(1)}}{2u_1} = \int_{x_1}^{x_2} EA (N_1^{(1)'} u_1 + N_2^{(1)'} u_2) N_1^{(1)'} dx - R_1$$

$$\frac{2\pi^{(1)}}{2u_2} = \int_{x_1}^{x_2} EA (N_1^{(1)'} u_1 + N_2^{(1)'} u_2) N_2^{(1)'} dx - P$$

[K] local stiffness matrix local force vector

$$\begin{bmatrix} \frac{2\pi^{(1)}}{2u_1} \\ \frac{2\pi^{(1)}}{2u_2} \end{bmatrix} = \int_{x_1}^{x_2} EA dx \begin{bmatrix} N_1^{(1),2} & N_2^{(1),2} \\ N_1^{(1),1} & N_2^{(1),1} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} - \begin{bmatrix} R_1 \\ P \end{bmatrix}$$

$$K = \int_{x_1}^{x_2} EA \begin{bmatrix} N_1'^2 & N_1'N_2' \\ N_1'N_2' & N_2'^2 \end{bmatrix} dx \quad \text{for each element}$$

$$N_1 = \frac{x_2 - x}{x_2 - x_1} \quad \left\{ \begin{array}{l} \text{for} \\ \text{each} \\ \text{element} \end{array} \right.$$

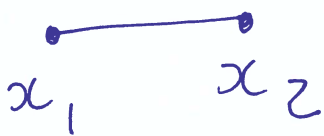
$$N_1' = \frac{\partial N_1}{\partial x}$$

$$N_2 = \frac{x - x_1}{x_2 - x_1}$$

$$= \frac{-1}{x_2 - x_1} = \frac{-1}{l}$$

$$N_2' = \frac{+1}{x_2 - x_1} = \frac{1}{l}$$

$x_2 - x_1 = \text{length of the element}$



$$K = \int_{x_1}^{x_2} EA \begin{bmatrix} \frac{1}{l^2} & -\frac{1}{l^2} \\ -\frac{1}{l^2} & \frac{1}{l^2} \end{bmatrix} dx$$

$$= \frac{EA}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} (x_2 - x_1) \rightarrow l$$

$$K = \frac{EA}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

← used it on the first H.W.

length of element j

$$K_{11} = \frac{EA}{l} \cdot 1$$

code

Going back for element 2.

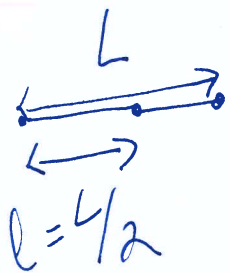
$$\begin{bmatrix} \frac{2\pi}{2u_2} \\ \frac{2\pi}{2u_3} \end{bmatrix} = \int_{x_2}^{x_3} EA \begin{bmatrix} N_1' & N_1' N_2' \\ N_1' N_2' & N_2' \end{bmatrix} dx \begin{bmatrix} u_2 \\ u_3 \end{bmatrix} - \begin{bmatrix} 0 \\ R_3 \end{bmatrix}$$

Plug these into $\begin{bmatrix} \frac{2\pi}{2u_1} \\ \frac{2\pi}{2u_2} \\ \frac{2\pi}{2u_3} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$

- just provide local stiffness and local force. The code will do the assembly

only external forces here

$$\begin{bmatrix} \frac{2\pi}{2u_1} \\ \frac{2\pi}{2u_2} \\ \frac{2\pi}{2u_3} \end{bmatrix} = \frac{2\pi}{2u_1} \begin{bmatrix} K_{11} & K_{12} & 0 \\ K_{21} & K_{22} + K_{11} & K_{12} \\ 0 & K_{21} & K_{22} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} - \begin{bmatrix} R_1 \\ P \\ R_3 \end{bmatrix}$$



$$= \frac{EA}{l} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1+1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} - \begin{bmatrix} R_1 \\ P \\ R_3 \end{bmatrix}$$

Now you can delete rows/columns

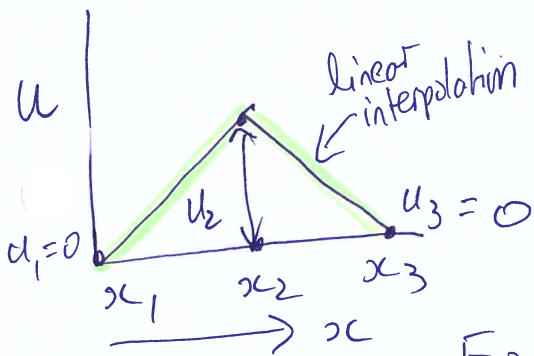
$$\frac{EA}{l} \begin{bmatrix} \cancel{1} & \cancel{-1} & \cancel{0} \\ -1 & 1+1 & -1 \\ \cancel{0} & \cancel{-1} & \cancel{1} \end{bmatrix} \begin{bmatrix} \cancel{u_1} \\ u_2 \\ \cancel{u_3} \end{bmatrix} - \begin{bmatrix} \cancel{R_1} \\ P \\ \cancel{R_3} \end{bmatrix} = \begin{bmatrix} \cancel{0} \\ 0 \\ \cancel{0} \end{bmatrix}$$

Now you can apply B.C's and delete rows & columns - This is condensation

$$\frac{EA}{l} 2 u_2 - P = 0$$

$$u_2 = \frac{Pl}{2EA} = \frac{PL}{4EA}$$

Interpolate: you can find solution anywhere
- This is an advantage over truss method, where you only know displacements at the ends



For 100 elements?

code

$$\text{local stiffness} = \frac{EA}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

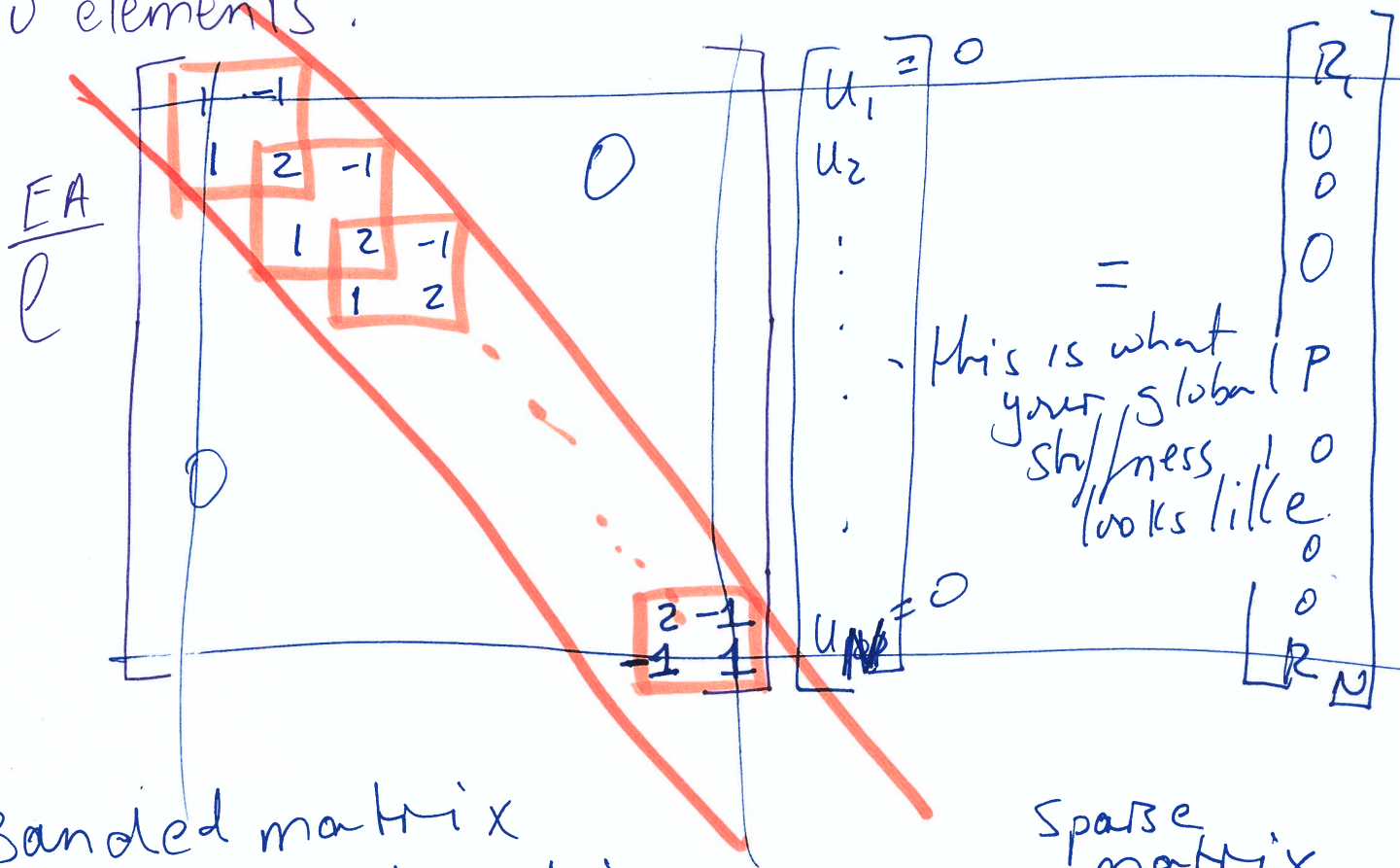
$$\text{local force} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \leftarrow \text{always kept as zeros}$$

except if you have body forces

+ $\begin{bmatrix} \text{Thermal or body force} \end{bmatrix}$

* Add external point forces at the end to the global force vector

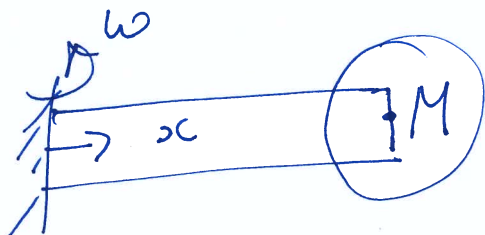
What does your stiffness matrix look like for 100 elements?



Sparse matrix
(don't need to store the whole matrix)

* The local stiffness remains the same. Don't need to write it over and over.

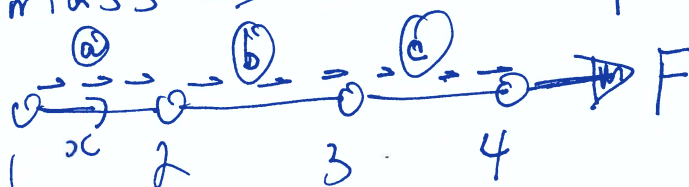
H.W.



Assume 1D

Point mass \Rightarrow external point force

centrifugal force getting bigger



$$b = \frac{m}{V} x \omega^2 = \rho \omega^2 x$$

\downarrow body force (N/m^3) \downarrow density

$$\int \bar{u}^T b dV$$

The whole P.F.

$$\Pi = \frac{1}{2} \int_0^L EA u'^2 dx - \int \rho \omega^2 x u A dx$$

$$- (M \omega^2 L) u(L)$$

FEM code

local stiffness (same as previous)

$$= \frac{EA}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

length
of element $\rightarrow l$

The local force is different for this problem.

The point forces come at the end.

Let's consider a generic element (not the first or last)