

Model for the solar system using ordinary differential equations

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December 22, 2019

Abstract

This article constructs a model for simulating the solar system. The Forward Euler and velocity Verlet methods will be used to solve the ordinary differential equations that describes the system. Taking an object oriented approach to implementation of the code makes for an more affordable task of expanding the system.

I. INTRODUCTION

This article focus on how to solve coupled ordinary differential equations with Euler Forward and velocity Verlet. With these two we will simulate the solar system and look further on the stability of the two different methods. Further the aim is to develop a structured code which is object oriented for an easier way of creating the desired system we want to simulate. The system modelled is only affected by the gravitational force between the planets. For this we use Newton's law of gravitation given by:

$$F_G = \frac{GM_{\odot}M_{\text{Earth}}}{r^2} \quad (1)$$

II. THEORY

i. Escape velocity

It is interesting looking at what the velocity of earth has to be for it to leave its orbit around the sun. This happens when the potential energy equals the kinetic energy. This leads to the equation;

$$0.5M_1v_{\text{escape}}^2 = G \frac{M_1M_2}{r} \quad (2)$$

$$v_{\text{escape}} = \sqrt{\frac{2GM}{r}} \quad (3)$$

For a system consisting of just the sun and Earth the Earth's escape velocity will be:

$$v_{\text{escape}} = \sqrt{\frac{2(4\pi^2(AU^3/\text{year}^2))}{1(AU)}} = 2\pi\sqrt{2}(AU/\text{year}) \quad (4)$$

III. ALGORITHMS

The methods for integrating the system has it offspring from Taylor expansion of functions.

$$f(x+h) = f(x) + h \frac{df}{dx}(x) + \frac{1}{2!} \frac{d^2f}{dx^2}(x) + ..$$

One of the methods (Euler Forward) make use of the two first terms in the Taylor expansion while Velocity Verlet uses three terms.

i. Euler Forward algorithm

Euler Forward then defines the approximation for next value to be

$$x_{i+1}^{\vec{}} = \vec{x}_i + h\vec{v}_i \quad (5)$$

and

$$v_{i+1}^{\vec{}} = \vec{v}_i + h\vec{a}_i \quad (6)$$

```
for i= 0,1,.., n-1 do
  find a from forces
  then compute velocity and position
   $x_{i+1}^{\vec{}} = \vec{x}_i + h\vec{v}_i$ 
   $v_{i+1}^{\vec{}} = \vec{v}_i + h\vec{a}_i$ 
```

*<https://github.com/AndreasFagerheim/Project-4>

end for

By using this algorithm the calculation consist of 4 FLOPS (2 in each calculation of next velocity and position). The error is $O(h^2)$ for both x_{i+1} and \vec{x}_{i+1} . Euler Forward can therefore be said to trade of its low cost in nedd of power for calculations (4 FLOPS) for a lower precision.

ii. Verlet method

Velocity Verlet make use of three terms, as earlier stated;

$$x_{i+1} = x_i + hx_i^{(1)} + \frac{h^2}{2}x_i^{(2)} + O(h^3) \quad (7)$$

and

$$v_{i+1} = v_i + hv_i^{(1)} + \frac{h^2}{2}v_i^{(2)} + O(h^3) \quad (8)$$

Here we know all values except the second derivative of the velocity. By Taylor expansion of the first derivative of velocity:

$$v_{i+1}^{(1)} = v_i^{(1)} + hv_i^{(2)} + O(h^2)$$

$$hv_i^{(2)} \approx v_{i+1}^{(1)} - v_i^{(1)}$$

Using this and we can rewrite equations 7 and 8 containing only known values;

$$x_{i+1} = x_i + hv_i + \frac{h^2}{2}v_i^{(1)} + O(h^3) \quad (9)$$

and

$$v_{i+1} = v_i + \frac{h}{2} \left(v_{i+1}^{(1)} + v_i^{(1)} \right) + O(h^3) \quad (10)$$

Due to $v_{i+1}^{(1)}$ being dependent on x_{i+1} calculating position at updated time (t_{i+1}) is necessary for calculating the new velocity. WE also know that $v_i^{(1)} = a_i$. In pseudo code this will look something like the figure below.

```

for i= 0,1,..., n-1 do
  find  $a$  from forces
  then compute velocity and position
   $\vec{x}_{i+1} = \vec{x}_i + h\vec{v}_i + \frac{h^2}{2}\vec{a}_i$ 
   $\vec{v}_{i+1} = \vec{v}_i + \frac{h}{2}(\vec{a}_{i+1} + \vec{a}_i)$ 
end for

```

This algorithm has 9 FLOPS in its calculation (5 FLOPS for position and 4 FLOPS for velocity). The strenght of the method is instead its local error witch is in order $O(h^3)$.

iii. Object oriented

The code is structured with with classes after a straightforward approach. To do this there is implemented three different classes to make up our system. The first class is the class Univers which sets up the system we want to simulate with its bodies and natural laws they have to obey. This urges the class Legemer (Bodies) which holds the attributes of bodies. Last is the class Intregrand wich implements to the two different methods for integration.

IV. RESULTS

Figure 1 and **2** show plots of the precision of the algorithms as a function of Δt . The plot is the position over 1 year for all of the different Δt . For Euler the precision could say to be low and a loss of stability to occur when Δt is not small enough. The velocity Verlot seems to achieve great precision and stability for $\Delta t = 0.01$ and is superior to Euler in this aspect as expected.

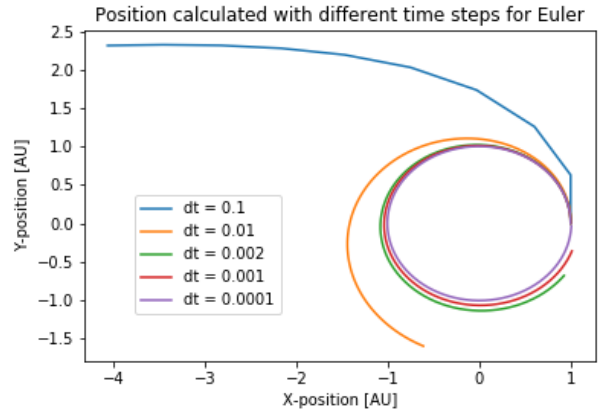


Figure 1: Stability plot of the Euler Forward method as a function of Δt .

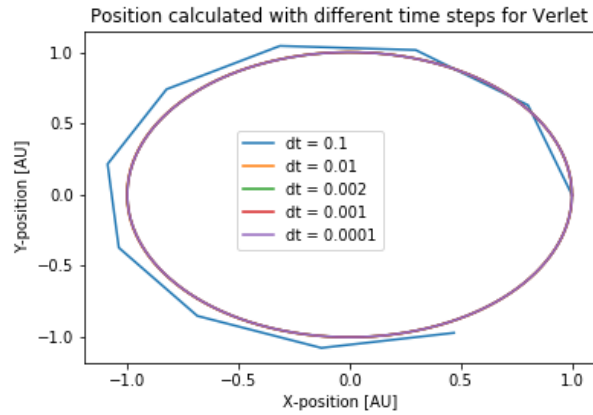


Figure 2: Stability plot of the velocity Verlet method as a function of Δt .

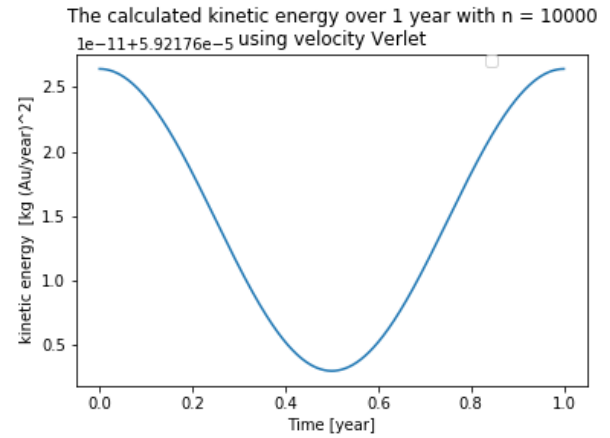


Figure 4: The kinetic energy for the system over a year calculated using Velocity Verlet method with 10000 iterations

i. Energy

The energy is relatively conserved taking in to account the distance is in AU and speed in AU/year. Figure 3-6 show that the changes are minimal through the year.

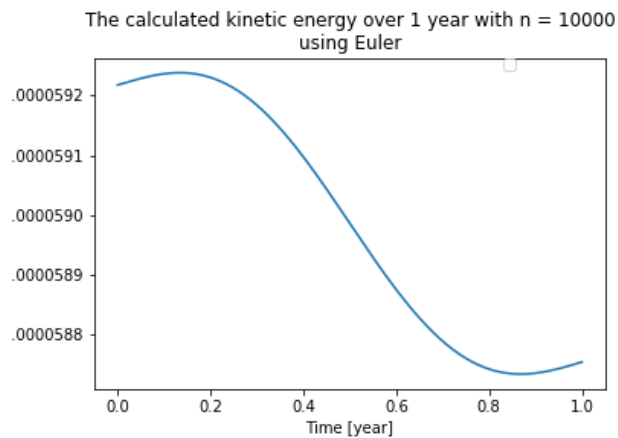


Figure 3: The kinetic energy for the system over a year calculated using Euler Forward method with 10000 iterations

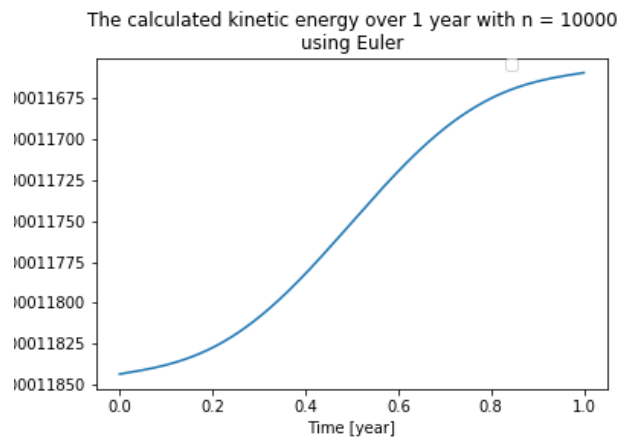
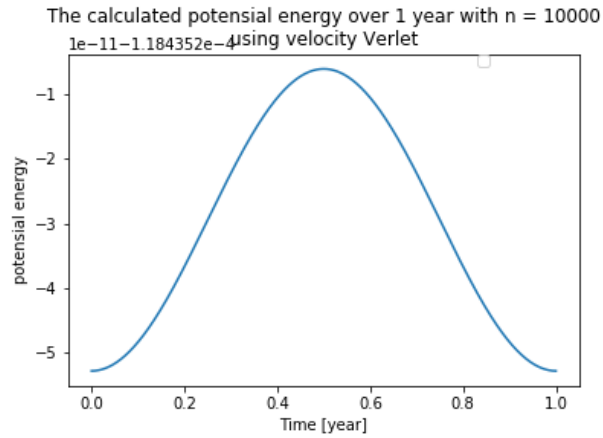


Figure 5: The potential energy for the system over a year calculated using Euler Forward method with 10000 iterations



[Hjorth-Jensen] Hjort-Jensen, M.
<https://github.com/CompPhysics/ComputationalPhysics>

Figure 6: The potential energy for the system over a year calculated using Velocity Verlet method with 10000 iterations

ii. Escape velocity for earth

The escape velocity for earth orbiting the sun was earlier found to be $v_e = 2\pi\sqrt{2}(AU/year)$ and looking at Figure 7 this is close to $V = 2.82\pi$ which also seem to escape the orbit of the sun.

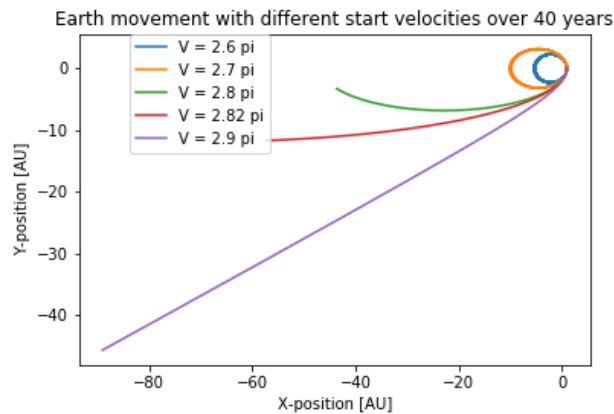


Figure 7: Plot of Earth orbit with different start velocities (V). The plot is over a period of 40 years.

V. CONCLUSION

REFERENCES

[Hjorth-Jensen, 2015] Hjort-Jensen, M. (2015). Computational Physics.