Solutions to hw6 homework on Convex Optimization
https://web.stanford.edu/class/ee364a/homework.html

Andrei Keino

October 5, 2020

7.7

ML estimation of Poisson distributions. Suppose x_i, $i = 1, \ldots, n$ are independent random variables with Poisson distributions

$$\boldsymbol{prob}(x_i = k) = \frac{e^{-\mu_i}\mu_i^k}{k!}$$

with unknown means μ_i. The variables x_i represent number of times that one of n possible independent events occurs during a certain period. In emission tomography, for example, they might represent the number of photons emitted by n sources. We consider an experiment designed to determine the means μ_i. The experiment involves m detectors. If event i occurs, it is detected by detector j with probability p_{ij}. We assume the probabilities p_{ij} are given (with $pij \geq 0$, $\sum_{j=1}^{m} p_{ij} \leq 1$). The total number of events recorded by detector j is denoted y_j,

$$y_j = \sum_{i=1}^{m} y_{ij}, \quad i = 1, \ldots, m$$

Formulate the ML estimation problem of estimating the means μ_i based on observed values of y_j, $j = 1, \ldots, m$, as a convex optimization problem.

Hint: the variables y_{ij} have the Poisson distribution with means $p_{ij}\mu_i$, i.e

$$\boldsymbol{prob}(y_{ij} = k) = \frac{e^{-p_{ij}\mu_i}(p_{ij}\mu_i)^k}{k!}$$

The sum of n independent random variables with means $\lambda_1, \ldots \lambda_n$ has a Poisson distribution with mean $\lambda_1 + \cdots + \lambda_n$.

Solution:
It follows from the hint, that

$$\boldsymbol{prob}(y_j = k) = \frac{e^{-p_j^T\mu}(p_j^T\mu)^k}{k!}$$

1

Then likelihood of the set of $y_j = k_j$, $j = 1, \ldots, m$ is

$$L\{k_j\} = \prod_{j=1}^{m} \frac{e^{-p_j^T \mu}(p_j^T \mu)^{k_j}}{k_j!}$$

and the log likelihood is

$$LL\{k_j\} = \sum_{j=1}^{m}(-p_j^T \mu + k_j log(p_j^T \mu) - log(k_j!))$$

$LL\{k_j\}$ is a concave function of μ_j. So, the solution is:

$$\text{maximize } \sum_{j=1}^{m}(-p_j^T \mu + k_j log(p_j^T \mu))$$

$$\text{subject to } \mu \succeq 0$$

8.15

Minimum volume ellipsoid covering union of ellipsoids.
Formulate the following problem as a convex optimization problem. Find the minimum volume ellipsoid $\mathcal{E} = \{x \mid (x - x_0)^T A^{-1}(x - x_0) \le 1\}$ that contains K given ellipsoids

$$\mathcal{E}_i = \{x \mid (x)^T A_i x + 2b_i^T x + c_i \le 1\}$$

Hint. See appendix B.

Solution:

$\mathcal{E}$ contains $\mathcal{E}_i$ if
$\forall x: \ x^T A^{-1} x - 2x_0^T A^{-1} x + x_0^T A^{-1} x_0 - 1 \le 0$
$(x)^T A_i(x) + 2b_i^T x + c_i - 1 \le 0$
From the S - procedure in the appendix B we have it's true if there exists no such $\lambda \ge 0$ that:

$$\lambda_i \begin{bmatrix} A_i & b_i \\ b_i^T & c_i \end{bmatrix} - \begin{bmatrix} A^{-1} & -A^{-1}x_0 \\ -(A^{-1}x_0)^T & x_0 A^{-1} x_0 - 1 \end{bmatrix} \succeq 0$$

this is the same as:

$$\begin{bmatrix} \lambda_i A_i & \lambda_i b_i \\ \lambda_i b_i^T & \lambda_i c_i + 1 \end{bmatrix} - \begin{bmatrix} I \\ -x_0^T \end{bmatrix} A^{-1} \begin{bmatrix} I & -x_0^T \end{bmatrix} \succeq 0$$

So, we have SDP:

minimize $log(det(A^{-1}))$

subject to $\begin{bmatrix} \lambda_i A_i & \lambda_i b_i \\ \lambda_i b_i^T & \lambda_i c_i + 1 \end{bmatrix} - \begin{bmatrix} I \\ -x_0^T \end{bmatrix} A^{-1} \begin{bmatrix} I & -x_0^T \end{bmatrix} \succeq 0, \ i = 1, \ldots, K$

$\qquad\qquad \lambda_i \geq 0, \ i = 1, \ldots, K$

A13.2

Optimal spacecraft landing. We consider the problem of optimizing the thrust problem for a spacecraft to carry out a landing at a target position. The spacecraft dynamics are

$$m\ddot{p} = f - mge_3$$

where $m > 0$ is the spacecraft mass, $p(t) \in R^3$ is the spacecraft position with 0 the target landing position and $p_3(t)$ representing height, $f(t) \in R^3$ is the thrust force and g is the gravitational acceleration. (For simplicity we assume that the spacecraft mass is constant. This is not always a good assumption, since the mass decreases with fuel use. We will also ignore any atmospheric friction.) We must have $p(T^{td}) = 0$ and $\dot{p}(T^{td}) = 0$ where T^{dt} is the touchdown time. The spacecraft must remain in a region given by

$$p_3(t) \geq \alpha||p_1(t), p_2(t)||_2$$

where $\alpha > 0$ is a minimal given glide slope. The initial position $p(0)$ and velocity $\dot{p}(0)$ are given. The thrust force $f(t)$ is obtained from a single rocket engine on the spacecraft, with a given maximum thrust; an attitude control system rotates the spacecraft to achieve any desired direction of thrust. The thrust force is therefore characterized by the constraint $||f(t)||_2 \leq F^{max}$. fuel use rate is proportional to the thrust force magnitude, so the total fuel use is

$$\int_0^{T^{td}} \gamma||f(t)||_2 dt,$$

where $\gamma > 0$ is the fuel consumption coefficient. The thrust force is discretized in time, i.e., it is constant over consecutive time periods of length $h > 0$ with $f(t) = f_k$ for $t \in [(k-1)h, kh]$ for $k = 1, \ldots, K$, where $T^{td} = Kh$. Therefore we have:

$$v_{k+1} = v_k + (h/m)f_k - hge_3,$$
$$p_{k+1} = p_k + h/2(v_{k+1} + v_k)$$

For simplicity, we will impose the glide slope constraint only at the times $t = 0, h, 2h, \ldots, Kh$.

(a) Minimum fuel descent. Explain how to find the thrust profile $f_1, \ldots, f_k$ that minimizes fuel consumption, given the touchdown time $T^{td} = Kh$ and discretization time h.

(b) Minimum time descent. Explain how to find the thrust profile that minimizes the touchdown time, i.e., K, with h fixed and given. Your method can involve solving several convex optimization problems.

(c) Carry out the methods described in parts (a) and (b) above on the problem instance with data given in `spacecraft_landing_data.*`. Report the optimal total fuel consumption for part (a), and the minimum touchdown time for part (b). The data
files also contain plotting code (commented out) to help you visualize your solution. Use the code to plot the spacecraft trajectory and thrust pro
files you obtained for parts (a) and (b).

Solution:
(a) To find the minimum fuel consumption profile we should minimize the expression $\sum_{k=1}^{K} ||f_k||_2$. Then the problem is:

$$
\begin{aligned}
\text{minimize} \quad & \sum_{k=1}^{K} ||f_k||_2 \\
\text{subject to} \quad & v_{k+1} = v_k + (h/m)f_k - hge_3, \\
& p_{k+1} = p_k + h/2(v_{k+1} + v_k) \\
& p_1 = p(0) \\
& v_1 = \dot{p}(0) \\
& p_{K+1} = 0 \\
& v_{K+1} = 0 \\
& ||f_k||_2 \le F^{max}, \; p_3(t) \ge \alpha||p_1(t), p_2(t)||_2, \; k = 1, \ldots, K
\end{aligned}
$$

where $p_i, v_i, f_i, \; i = 1, \ldots, K$ are variables.

(b) We can solve a sequence of convex feasibility problems to find the minimum touchdown time. For each K we solve the problem

4

$$\begin{aligned}
\text{minimize} \quad & 0 \\
\text{subject to} \quad & v_{k+1} = v_k + (h/m)f_k - hge_3, \\
& p_{k+1} = p_k + h/2(v_{k+1} + v_k) \\
& p_1 = p(0) \\
& v_1 = \dot{p}(0) \\
& p_{K+1} = 0 \\
& v_{K+1} = 0 \\
& ||f_k||_2 \le F^{max}, \ p_3(t) \ge \alpha||p_1(t), p_2(t)||_2, \ k = 1, \ldots, K
\end{aligned}$$

where $p_i, v_i, f_i, \ i = 1, \ldots, K$ are variables.

If the problem is feasible, we reduce K, otherwise we increase K. We iterate until we

nd the smallest K for which a feasible trajectory can be found.

(c) The matlab code:

```
spacecraft_landing_data;
% solve part (a) (find minimum fuel trajectory)
cvx_solver sdpt3;
cvx_begin
variables p(3,K+1) v(3,K+1) f(3,K)
v(:,2:K+1) == v(:,1:K)+(h/m)*f-h*g*repmat([0;0;1],1,K);
p(:,2:K+1) == p(:,1:K)+(h/2)*(v(:,1:K)+v(:,2:K+1));
p(:,1) == p0; v(:,1) == v0;
p(:,K+1) == 0; v(:,K+1) == 0;
p(3,:) >= alpha*norms(p(1:2,:));
norms(f) <= Fmax;
minimize(sum(norms(f)))
cvx_end
min_fuel = cvx_optval*gamma*h;
p_minf = p; v_minf = v; f_minf = f;
% solve part (b) (find minimum K)
% we will use a linear search, but bisection is faster
Ki = K;
while(1)
cvx_begin
variables p(3,Ki+1) v(3,Ki+1) f(3,Ki)
v(:,2:Ki+1) == v(:,1:Ki)+(h/m)*f-h*g*repmat([0;0;1],1,Ki);

p(:,2:Ki+1) == p(:,1:Ki)+(h/2)*(v(:,1:Ki)+v(:,2:Ki+1));
p(:,1) == p0; v(:,1) == v0;
p(:,Ki+1) == 0; v(:,Ki+1) == 0;
p(3,:) >= alpha*norms(p(1:2,:));
```

```
norms(f) <= Fmax;
minimize(sum(norms(f)))
cvx_end
if(strcmp(cvx_status,'Infeasible') == 1)
Kmin = Ki+1;
break;
end
Ki = Ki-1;
p_mink = p; v_mink = v; f_mink = f;
end
% plot the glide cone
x = linspace(-40,55,30); y = linspace(0,55,30);
[X,Y] = meshgrid(x,y);
Z = alpha*sqrt(X.^2+Y.^2);
figure; colormap autumn; surf(X,Y,Z);
axis([-40,55,0,55,0,105]);
grid on; hold on;
% plot minimum fuel trajectory for part (a)
plot3(p_minf(1,:),p_minf(2,:),p_minf(3,:),'b','linewidth',1.5);
quiver3(p_minf(1,1:K),p_minf(2,1:K),p_minf(3,1:K),...
f_minf(1,:),f_minf(2,:),f_minf(3,:),0.3,'k','linewidth',1.5);
print('-depsc','spacecraft_landing_a.eps');
% plot minimum time trajectory for part (b)
figure; colormap autumn; surf(X,Y,Z);
axis([-40,55,0,55,0,105]); grid on; hold on;
plot3(p_mink(1,:),p_mink(2,:),p_mink(3,:),'b','linewidth',1.5);
quiver3(p_mink(1,1:Kmin),p_mink(2,1:Kmin),p_mink(3,1:Kmin),...
f_mink(1,:),f_mink(2,:),f_mink(3,:),0.3,'k','linewidth',1.5);
print('-depsc','spacecraft_landing_b.eps');
```

The optimal total fuel consumption is 193:0. The minimum touchdown time is K = 25.

A16.3

Utility versus latency trade-off in a network. We consider a network with m edges, labeled $1, \ldots m$, and n flows labeled $1, \ldots, n$. Each flow has an associated nonnegative flow rate f_j; each edge or link has an associated positive capacity c_i. Each flow passes over a fixed set of links (its route); the total traffic t_i on link i is the sum of flow rates over all flows that pass though the link i. The flow routes are described by a routing matrix $R \in \boldsymbol{R}^{m \times n}$, defined as:

$$R_{ij} = \begin{cases} 1 & \text{flow } j \text{ passes through link } i \\ 0 & \text{otherwise} \end{cases}$$

Thus, the vector fo link traffic, $t \in \boldsymbol{R}^m$ is given by $t = Rf$. The link capacity

6

constraint can be expressed as $Rf \preceq c$. The logarithmic network utility is defined as $U(f) = \sum_{j=1}^{n} log(f_j)$. The average queuing delay on link i is given by

$$d_i = \frac{1}{c_i - t_i}$$

multiplied by a constant that doesn't matter to us. We take $d_i = \infty$ for $t_i = c_i$. The delay or latency for flow j denoted l_j is the sum of link delays over all links that flow j passes through. We define the maximum flow latency as

$$L = max\{l_1, \dots l_n\}$$

We are given R and c, we are to choose f.

(a) How would you find the flow rates that maximize the utility U, ignoring flow latency? (In particular we allow $L = \infty$.) We'll refer to this maximum achievable utility as U^{max}.

(b) How would you find the flow rates that the maximum flow latency L, ignoring utility? (In particular we allow $U = -\infty$.) We'll refer to this minimum achievable latency as L^{min}.

(c) Explain how to find the optimal trade-off between utility U (which we want to maximize) and latency L (which we want to minimize).

(d) Find U^{max}, L^{min} and plot the optimal trade-off utility versus latency for the network with the data given in `net_util_data.m`, showing L^{min} and U^{max} on the same plot. Your plot should cover the range from $L = 1.1L^{min}$ to $L = 11L^{min}$. Plot U vertically using linear scale and L horizontally using log scale.

Note. For parts (a), (b), and (c), your answer can involve solving one or more convex optimization problems. But if there is a simpler solution, you should say so.

Solution:

(a) To maximize utility we solve the convex problem:

$$
\begin{array}{ll}
\text{maximize} & \sum_{j=1}^{n} log(f_j) \\
\text{subject to} & Rf \preceq c
\end{array}
$$

with variable f and the implicit constraint $f \succeq 0$.

(b) Link delay is monotonically increased with traffic and minimizes with zero traffic. Since flow latency is the sum of link delays, we can minimize the latency by setting the $f = 0$. With $f = 0$ we have $d_i = 1/c_i$. To sum these delays over the routes, we multiply d_i by R^T. Then $l = R^T(1/c_1, \ldots, 1/c_n)$, where l is a vector of flow latencies. So, we have:

$$L^{min} = max\{R^T(1/c_1, \ldots, 1/c_n)\}$$

where the $max\{R^T(1/c_1, \ldots, 1/c_n)\}$ is the maximum over the entries of the vector $R^T(1/c_1, \ldots, 1/c_n)$.

c Let $b_1, \ldots b_m$ denotes the rows of R. To find the optimal trade - off between U and L we solve the problem

$$
\begin{aligned}
& \text{maximize} && \sum_{j=1}^{n} log(f_j) \\
& \text{subject to} && Rf \preceq c \\
& && \sum_{i=1}^{m} \frac{R_{ij}}{c_i - b_i^T f} \leq L, \; j = 1, \ldots, n,
\end{aligned}
$$

for a range of values L. The variable here is f and there is an implicit constraint $f \succeq 0$. As $Rf \preceq c$, we should have $c_i - b_i^T f \geq 0$, so the constraints

$$\sum_{i=1}^{m} \frac{R_{ij}}{c_i - b_i^T f} \leq L$$

are convex (it can be easy verified considering the Hessian of $\frac{R_{ij}}{c_i - b_i^T f}$ with variable f. This is therefore a convex optimization problem.
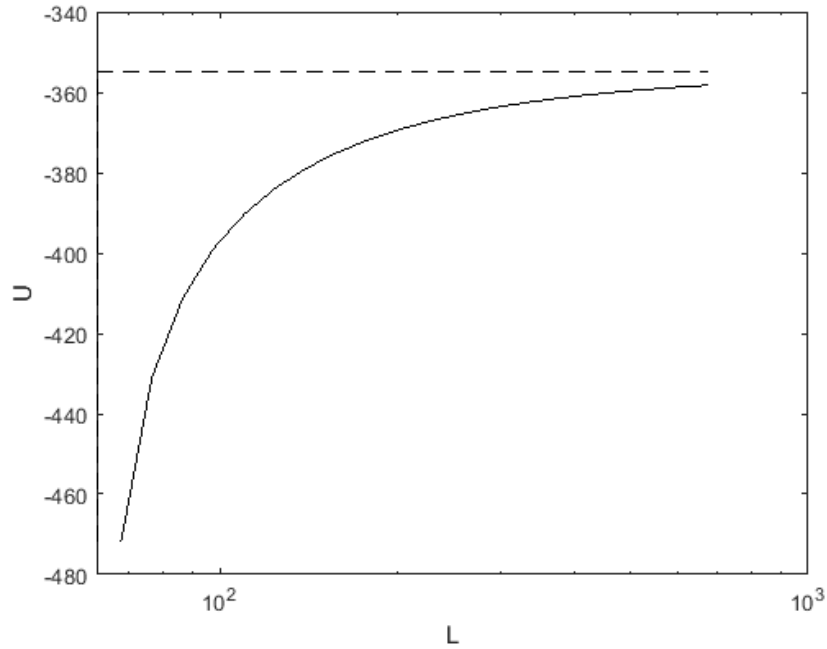
Figure 1: Optimal trade-off utility versus latency for the network.

The Matlab code:

```
% solution for network utility problem
clear all;
net_util_data;
% let's find max utility with no delay constraint
cvx_begin
variable f(n)
maximize geo_mean(f)
subject to
R*f <= c
f >= 0 % not needed; enforced by geomean domain
cvx_end
Umax=sum(log(f));

% let's find min latency with no utility constraint
% can be done analytically: just take f=0
Lmin = max(R'*(1./c));
% now let's do pareto curve
N = 20;
ds = 1.10*Lmin*logspace(0,1,N); % go from 10% above L
```

```
Uopt = [];
for d = ds
cvx_begin
variable f(n)
maximize geo_mean(f);
subject to
R'*inv_pos(c-R*f) <= d*ones(n,1)
f >= 0; % not needed; enforced by geomean domain
R*f <= c; % not needed; enforced by inv_pos domain
cvx_end
Uopt = [Uopt n*log(cvx_optval)];
end
semilogx(ds,Uopt,'k-',[Lmin,ds],[Umax,ones(1,N)*Umax],...
'k--',[1,1]*Lmin,[Uopt(1),Umax],'k--')
% axis([ds(1), ds(N), -480, -340]);
xlabel('L'); ylabel('U');
```

A18.2

Radiation treatment planning. In radiation treatment, radiation is delivered to a patient, with the goal of killing or damaging the cells in a tumor, while carrying out minimal damage to other tissue. The radiation is delivered in beams, each of which has a known pattern; the level of each beam can be adjusted. (In most cases multiple beams are delivered at the same time, in one 'shot', with the treatment organized as a sequence of 'shots'.) We let b_j denote the level of beam j, for $j = 1, \ldots, n$. These must satisfy $0 \leq b_j \leq B^{max}$, where B^{max} is the maximum possible beam level. The exposure area is divided into m voxels, labeled $i = 1, \ldots, m$. The dose d_i delivered to voxel i is linear in the beam levels, i. e., $d_j = \sum_{j=1}^{n} A_{ij} b_j$. Here $A \in R_+^{m \times n}$. is a (known) matrix that characterizes the beam patterns. We now describe a simple radiation treatment planning problem. A (known) subset of the voxels, $\mathcal{T} \subset 1, \ldots, m$ corresponds to the tumor or target region. We require that a minimum radiation dose D^{target} be administered to each tumor voxel, i.e., $d_i \geq D^{target}$ for $i \in \mathcal{T}$. For all other voxels, we would like to have $d_i \leq D^{other}$, where D^{other} is a desired maximum dose for non-target voxels. This is generally not feasible, so instead we settle for minimizing the penalty

$$E = \sum_{i \notin \mathcal{T}} ((d_i - D^{other})_+)^2$$

where $(\cdot)_+$ denotes the nonnegatie part. We can interpret E as the sum of the squares of the nontarget excess doses.

(a) Show that the treatment planning problem is convex. The optimization variable is $b \in R^n$; the problem data are A, D^{other}, D^{target}, $\mathcal{T}$, B^{max}.

(b) Solve the problem instance with data given in the
file `treatment_planning_data.m`. Here we have split the matrix A into
$Atarget$, which contains the rows corresponding to the target voxels, and $Aother$,
which contains the rows corresponding to other voxels. Give the optimal value.
Plot the dose histogram for the target voxels, and also for the other voxels.
Make a brief comment on what you see. Remark. The beam pattern matrix
in this problem instance is randomly generated, but similar results would be
obtained with realistic data.