
Table of Contents

.....	1
Housekeeping	2
1) Sample S, C _L , rho, and V 10,000 times	2
2) Calculate lift in kilonewtons for each of the 10,000 samplings/simulations.	2
3) Calculate the best estimate and error for lift	2
4) Plot a histogram (use the "histogram" command) of L with 30 bins.	2
Bonus 1) Calculate drag in kilonewtons	3
Bonus 2) Make a scatterplot of Lift vs Drag.	3

%%
%%

% CODE CHALLENGE 2 - Monte Carlo Analysis

%

% The purpose of this challenge is to perform a Monte-Carlo analysis
on the

% lift generated by an aircraft. The aircraft has the following
characteristics:

% Wing surface area, S = 80 m²

% Lift coefficient, C_L = 0.90 +/- 0.03

%

% And is flying under the following conditions

% Air density, rho = 0.653 kg/m³

% Airspeed, V = 100 +/- 10 m/s

%

%

%

% To complete the challenge, execute the following steps:

% 1) Sample S, C_L, rho, and V 10,000 times.

% 2) Calculate lift in kilonewtons for each of the 10,000 samplings/
simulations.

% 3) Calculate the best estimate and error for lift and report it to
the

% command window using appropriate significant figures.

% 4) Plot a histogram of L.

% Bonus 1) Calculate drag in kilonewtons for each of the 10,000
samplings/simulations.

% Bonus 2) Make a scatterplot of Lift vs Drag.

%

% NOTE: DO NOT change any variable names already present in the code.

%

% Upload your team's script to Canvas to complete the challenge.

%

% NAME YOUR FILE AS Challenge2_Sec{section number}_Group{group
breakout #}.m

% ***Section numbers are 1 or 2***

% EX File Name: Challenge2_Sec1_Group15.m

%

%

```
% 1) Lila Monprode
% 2) Jordi Brownlow
% 3) Andrew Logue
% 4) Julian Claxton
% 5) Joshua Quarderer
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
```

Housekeeping

(Please don't "clear all" or "clearvars", it makes grading difficult)

```
close all    % Close all open figure windows
clc         % Clear the command window
```

1) Sample S, C_L, rho, and V 10,000 times

(i.e. the S variable should contain 10000 samples of the wing surface area)

```
N = 1e04;
S = 80*ones(N,1); % [m^2]
C_L = .9+(.06*(rand(N,1)))-.03;
rho = 0.653*ones(N,1); % [kg/m^3]
V = 100+(20*(rand(N,1)))-10; % [m/s]
```

2) Calculate lift in kilonewtons for each of the 10,000 samplings/simulations.

Given that the equation for lift is: $L = 0.5 * \rho * V^2 * C_L * S$ (Newtons)

```
L = (0.5 .* rho .* V.^2 .* C_L .* S)/1000; % [kN]
```

3) Calculate the best estimate and error for lift

Report it to the command window using appropriate significant figures.

```
L_best = mean(L);
L_err  = std(L);

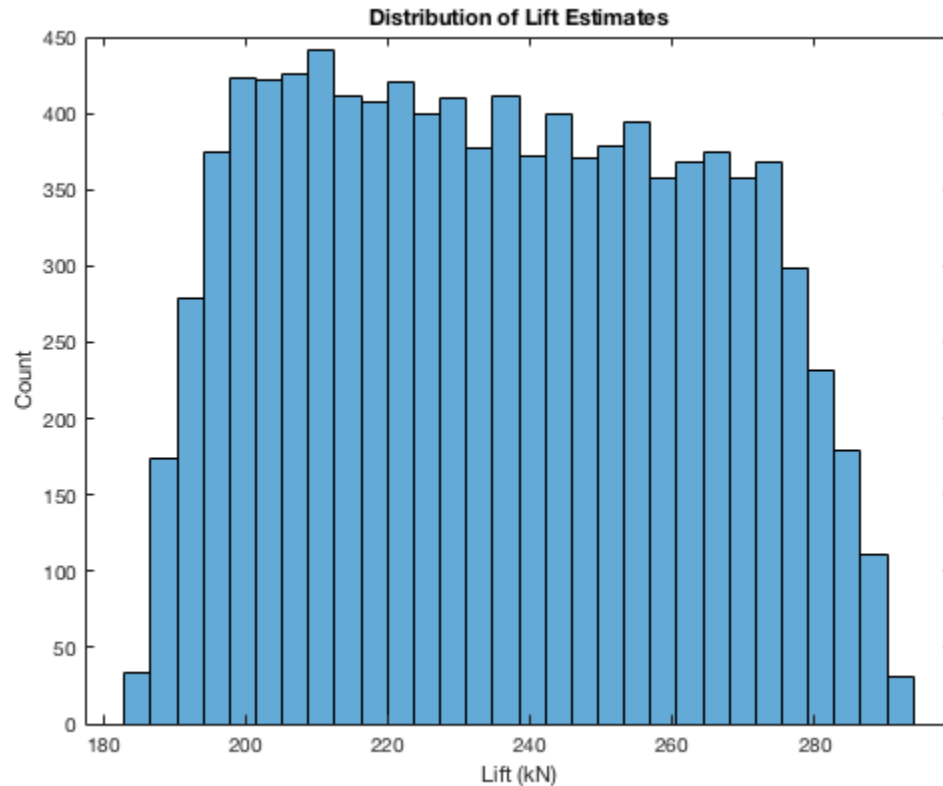
fprintf("Best Estimate of Lift: %f kN\n",L_best);
fprintf("Error for Lift Estimate: %f\n", L_err);

Best Estimate of Lift: 235.727177 kN
Error for Lift Estimate: 27.509089
```

4) Plot a histogram (use the "histogram" command) of L with 30 bins.

Add annotations and labels for style points!

```
figure();
histogram(L,30)
title("Distribution of Lift Estimates")
xlabel("Lift (kN)")
ylabel("Count")
```



Bonus 1) Calculate drag in kilonewtons

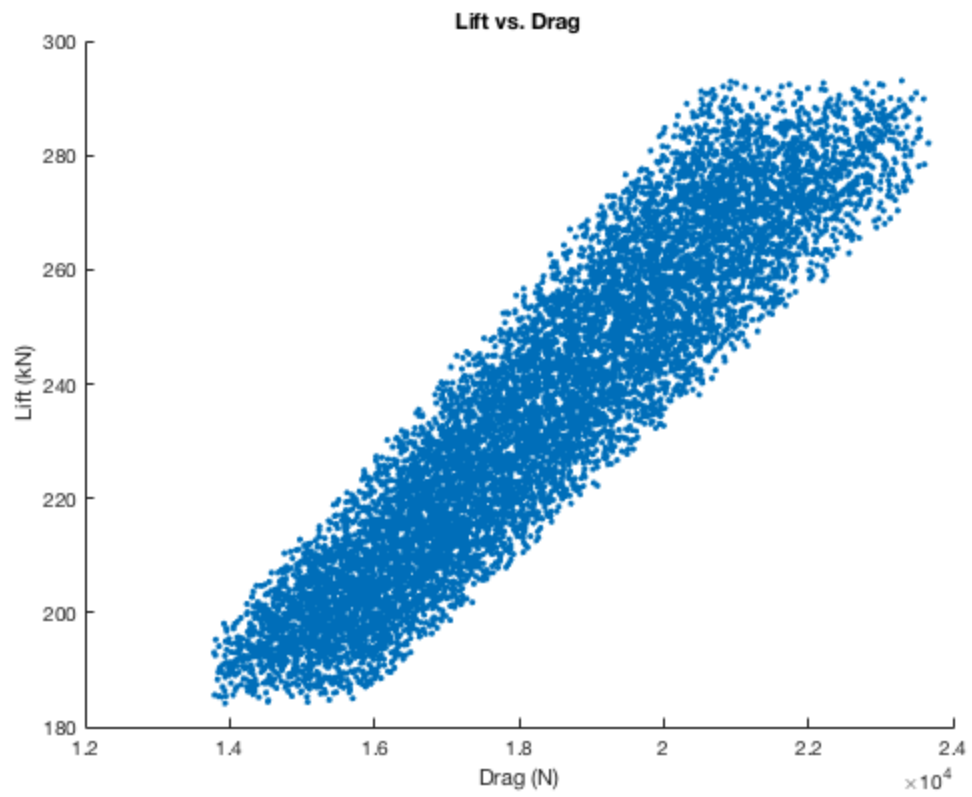
For each of the 10,000 samplings/simulations, given that the equation for drag is: $D = 0.5 * \rho * V^2 * C_D * S$ (Newtons) and that $C_D = 0.070 \pm 0.005$

```
C_D = .070+(.01*(rand(N,1)))-.005;
D = 0.5 * rho .* V.^2 .* C_D .* S;
```

Bonus 2) Make a scatterplot of Lift vs Drag.

Think about the following (no work to do): - Why do you think the points are spread into an ellipse and not a circle? - What is the significance of the general trend/slope of the data? - How could this sort of analysis be useful when dealing with more complicated systems and equations?

```
figure();
scatter(D,L,1, '.')
title("Lift vs. Drag")
xlabel("Drag (N)")
ylabel("Lift (kN)")
```



Published with MATLAB® R2019a