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```
function []= robustControl(theta10,theta20,dtheta10, dtheta20,theta1f, theta2f,dtheta1f,dtheta2f,tf)
```

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### testing algorithm with a set of initial and final states.

---

```
clc
clear all
close all
%link1 initial and final position and velocities.
theta10=-0.6;
dtheta10 =0;
theta1f = 0.9;
dtheta1f=0;

%time
tf= 60;

%link2 initial and final position and velocities.
theta20=-0.8;
dtheta20= 0.1;
theta2f = 0.6;
dtheta2f=0;

% the nominal model parameter:
m1 =10; m2=5; l1=1; l2=1; r1=0.5; r2 =.5; I1=10/12; I2=5/12; % parameters in the paper.
% the nominal parameter vector b0 is
b0 = [ m1* r1^2 + m2*l1^2 + I1; m2*r2^2 + I2; m2*l1*r2];
```

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### Trajectory planning block

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Initial condition (TODO: CHANGE DIFFERENT INITIAL AND FINAL STATES)

```
x0=[-0.6,-0.8,0,0.1];
x0e = [-0.7,0.5,-0.2,0]; % an error in the initial state.
xf=[0.9,0.6, 0, 0];
% The parameter for planned joint trajectory 1 and 2.
global a1 a2 % two polynomial trajectory for the robot joint
nofigure=1;
% Traj generation.
a1 = planarArmTraj(theta10,dtheta10, theta1f, dtheta1f,tf, nofigure);
a2 = planarArmTraj(theta20,dtheta20, theta2f, dtheta2f,tf, nofigure);

torque=[];
options = odeset('RelTol',1e-4,'AbsTol',[1e-4, 1e-4, 1e-4, 1e-4]);
```

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### TODO: IMPLEMENT THE CONTROLLER

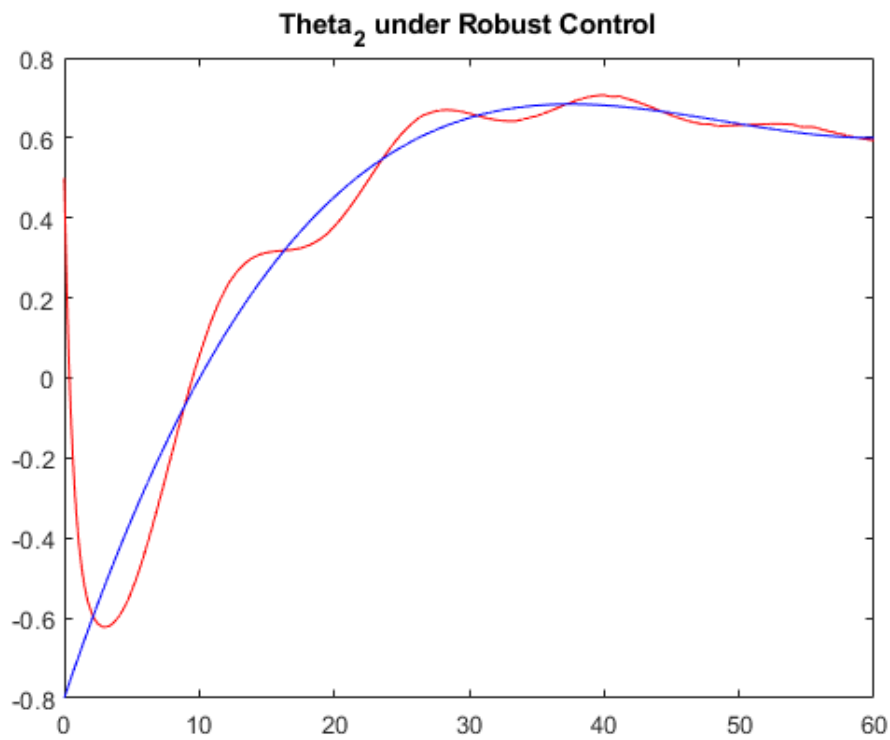
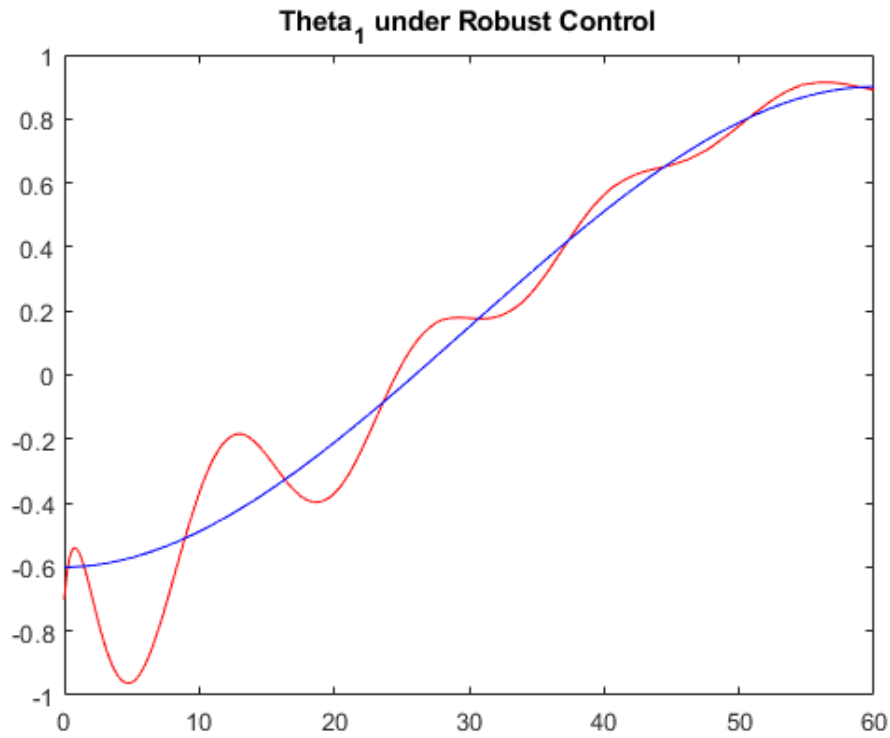
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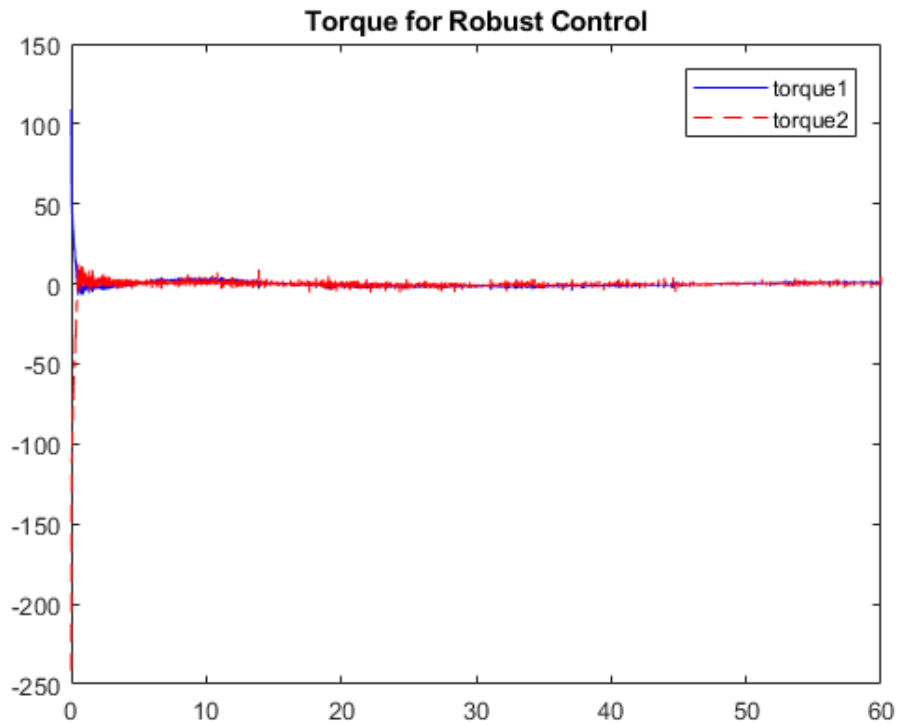
```
[T,X] = ode45(@(t,x)planarArmODERobust(t,x),[0 tf],x0e,options);
figure('Name','theta1');
plot(T, X(:,1),'r-');
hold on
plot(T, a1(1)+a1(2)*T+ a1(3)*T.^2+a1(4)*T.^3,'b-');
```

```
title('Theta_1 under Robust Control');
figure('Name','theta2');
plot(T, X(:,2), 'r-');
hold on
plot(T, a2(1)+a2(2)*T+ a2(3)*T.^2+a2(4)*T.^3, 'b-');
title('Theta_2 under Robust Control');

figure('Name', 'I/p- Robust control')
plot(T, torque(1,1:size(T,1)), 'b-');
hold on
plot(T, torque(2,1:size(T,1)), 'r--');
legend('torque1', 'torque2')
title('Torque for Robust Control');
torque = [];
```

---





**TODO: IMPLEMENT THE CONTROLLER TO AVOID CHATTERING.**

```
[T,X] = ode45(@(t,x)planarArmODERobustApprx(t,x),[0 tf],x0e,options);

figure('Name','theta1');
plot(T, X(:,1),'r-');
hold on
plot(T, a1(1)+a1(2)*T+ a1(3)*T.^2+a1(4)*T.^3,'b-');
title('Theta_1 under Robust Control w/o chattering');
figure('Name','theta2');
plot(T, X(:,2),'r-');
hold on
plot(T, a2(1)+a2(2)*T+ a2(3)*T.^2+a2(4)*T.^3, 'b-');
title('Theta_2 under Robust Control w/o chattering');

figure('Name','I/p- Robust control w/o chattering')
plot(T, torque(1,1:size(T,1)), 'b-');
hold on
plot(T, torque(2,1:size(T,1)), 'r--');
legend('torque1', 'torque2')
title('Torque for Robust Control without chattering');
torque = [];

% Robust control with chattering
function [dx ] = planarArmODERobust(t,x)
    %Todo: Select your feedback gain matrix Kp and Kd.

    % Compute the desired state and their time derivatives from planned
    % trajectory.
    vec_t = [1; t; t^2; t^3]; % cubic polynomials
    theta_d= [a1'*vec_t; a2'*vec_t];
    %ref = [ref,theta_d];
    % compute the velocity and acceleration in both theta 1 and theta2.
    a1_vel = [a1(2), 2*a1(3), 3*a1(4), 0];
    a1_acc = [2*a1(3), 6*a1(4),0,0 ];
    a2_vel = [a2(2), 2*a2(3), 3*a2(4), 0];
    a2_acc = [2*a2(3), 6*a2(4),0,0 ];
    dtheta_d =[a1_vel*vec_t; a2_vel* vec_t];
    ddtheta_d =[a1_acc*vec_t; a2_acc* vec_t];
    theta= x(1:2,1);
    dtheta= x(3:4,1);
```

```

%the true model
m2t = m2+ 10*rand(1);% m1 true value is in [m1, m1+epsilon_m1] and epsilon_m1 a random number in [0,10];
r2t = r2 + 0.5*rand(1);
I2t = I2 + (15/12)*rand(1);

a = I1+ I2t +m1*r1^2+ m2t*(l1^2+ r2t^2);
b = m2t*l1*r2t;
d = I2t+ m2t*r2t^2;

%lower bound
a_l = I1+I2+m1*r1^2+ m2*(l1^2+ r2^2);
b_l = m2*l1*r2;
d_l = I2 + m2*r2^2;

%Upper bound
m2t_u = m2+ 10;
r2t_u = r2 + 0.5;
I2t_u = I2 + (15/12);

a_u = I1+I2t_u+m1*r1^2+ m2t_u*(l1^2+ r2t_u^2);
b_u = m2t_u*l1*r2t_u;
d_u = I2t_u+ m2t_u*r2t_u^2;

% lower bound M
M_l = [a_l + 2*b_l*cos(x(2)), d_l + b_l*cos(x(2)); d_l + b_l*cos(x(2)), d_l];
%Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];

% upper bound M
M_u = [a_u + 2*b_u*cos(x(2)), d_u + b_u*cos(x(2)); d_u + b_u*cos(x(2)),d_u];
%Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];

% Control law M and C
global M_bar C_bar
M_bar = inv((M_u + M_l)/2);
d_bar = M_bar(2,2);
b_bar = (M_bar(2,1) - d_bar)*sec(x(2));
C_bar = [-b_bar*sin(x(2))*x(4), -b_bar*sin(x(2))*(x(3)+x(4)); b_bar*sin(x(2))*x(3),0];

% actual dynamic model of the system is characterized by M and C
global Mmat Cmat
Mmat = [a+2*b*cos(x(2)), d+b*cos(x(2)); d+b*cos(x(2)), d];
Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];
invM = inv(Mmat);
invMC = invM*Cmat;

%TODO: compute the robust controller
tau = Rob_ctrl(theta_d, dtheta_d, ddtheta_d, theta, dtheta);
torque = [torque, tau];

%TODO: update the system state, compute dx
dx=zeros(4,1);
dx(1) = x(3);
dx(2) = x(4);
dx(3:4) = -invMC* x(3:4) +invM*tau; % because ddot theta = -M^{-1}(C \dot Theta) + M^{-1} tau
end

% Robust control without chattering
function [dx] = planarArmODERobustApprx(t,x)
%Todo: Select your feedback gain matrix Kp and Kd.

% Compute the desired state and their time derivatives from planned
% trajectory.
vec_t = [1; t; t^2; t^3]; % cubic polynomials
theta_d= [a1'*vec_t; a2'*vec_t];
%ref = [ref,theta_d];
% compute the velocity and acceleration in both theta 1 and theta2.
a1_vel = [a1(2), 2*a1(3), 3*a1(4), 0];
a1_acc = [2*a1(3), 6*a1(4),0,0 ];
a2_vel = [a2(2), 2*a2(3), 3*a2(4), 0];

```

```

a2_acc = [2*a2(3), 6*a2(4),0,0 ];
dtheta_d =[a1_vel*vec_t; a2_vel* vec_t];
ddtheta_d =[a1_acc*vec_t; a2_acc* vec_t];
theta= x(1:2,1);
dtheta= x(3:4,1);

%the true model
m2t = m2+ 10*rand(1);% m1 true value is in [m1, m1+epsilon_m1] and epsilon_m1 a random number in [0,10];
r2t = r2 + 0.5*rand(1);
I2t = I2 + (15/12)*rand(1);

a = I1+I2+m1*r1^2+ m2t*(l1^2+ r2t^2);
b = m2t*l1*r2t;
d = I2t+ m2t*r2t^2;

%lower bound
a_l = I1+I2+m1*r1^2+ m2*(l1^2+ r2^2);
b_l = m2*l1*r2;
d_l = I2 + m2*r2^2;

%Upper bound
m2t_u = m2+ 10;
r2t_u = r2 + 0.5;
I2t_u = I2 + (15/12);

a_u = I1+I2t_u+m1*r1^2+ m2t_u*(l1^2+ r2t_u^2);
b_u = m2t_u*l1*r2t_u;
d_u = I2t_u+ m2t_u*r2t_u^2;

% lower bound M
M_l = [a_l + 2*b_l*cos(x(2)), d_l + b_l*cos(x(2)); d_l + b_l*cos(x(2)), d_l];
% Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];

% upper bound M
M_u = [a_u + 2*b_u*cos(x(2)), d_u + b_u*cos(x(2)); d_u + b_u*cos(x(2)),d_u];
% Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];

% Control law M and C
global M_bar C_bar
M_bar = inv((M_u + M_l)/2);
d_bar = M_bar(2,2);
b_bar = (M_bar(2,1) - d_bar)*sec(x(2));
C_bar = [-b_bar*sin(x(2))*x(4), -b_bar*sin(x(2))*(x(3)+x(4)); b_bar*sin(x(2))*x(3),0];

% actual dynamic model of the system is characterized by M and C
global Mmat Cmat
Mmat = [a+2*b*cos(x(2)), d+b*cos(x(2)); d+b*cos(x(2)), d];
Cmat = [-b*sin(x(2))*x(4), -b*sin(x(2))*(x(3)+x(4)); b*sin(x(2))*x(3),0];
invM = inv(Mmat);
invMC = invM*Cmat;

%TODO: compute the robust controller
taur = Rob_ctrlapprx(theta_d, dtheta_d, ddtheta_d, theta, dtheta);
torque = [torque, taur];

%TODO: update the system state, compute dx
dx=zeros(4,1);
dx(1) = x(3);
dx(2) = x(4);
dx(3:4) = -invMC*x(3:4) +invM*taur; % because ddot theta = -M^{-1}(C \dot Theta) + M^{-1} tau
end

% torque function for Robust Control
function tau = Rob_ctrl(theta_d, dtheta_d, ddtheta_d, theta, dtheta);
global M_bar C_bar Mmat Cmat
Kp = 300*eye(2);
Kv = 200*eye(2);
e = theta_d - theta;
de = dtheta_d - dtheta;
y1 = 0.01;
y2 = 0.02;

```

```

y3 = 1;
P = eye(2);
B = [0;1];
%for link 1
x_err1 = [e(1,1) ; de(1,1)];
p1 = y1*norm(x_err1) + y2*(norm(x_err1)^2) + y3;
w1 = transpose(x_err1)*P*B;
if w1 == 0
    v1=0;
else
    v1 = -((w1*p1)/ norm(w1));
end

%for link 2
x_err2 = [e(2,1) ; de(2,1)];
p2 = y1*norm(x_err2) + y2*(norm(x_err2)^2) + y3;
w2 = transpose(x_err2)*P*B;
E = norm((inv(Mmat)*M_bar) - eye(2));
eta = norm(((inv(Mmat)*M_bar) - eye(2)) + inv(Mmat)*((C_bar-Cmat)*dtheta));
if w2 ==0
    v2=0;
else
    v2 = -((w2*p2)/ norm(w2));
end
% Added input
v = [v1;v2];

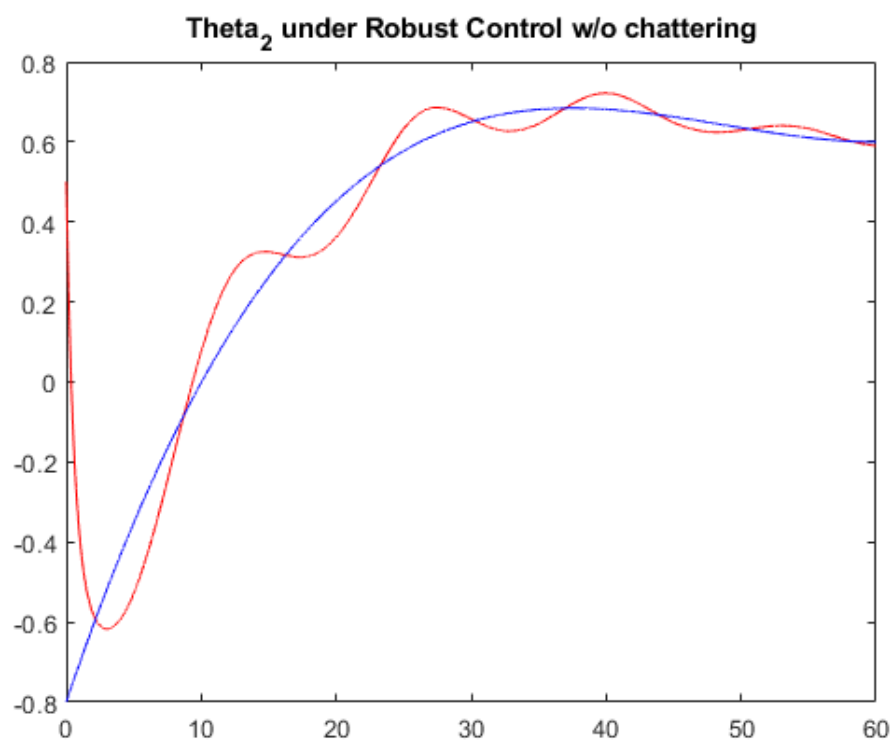
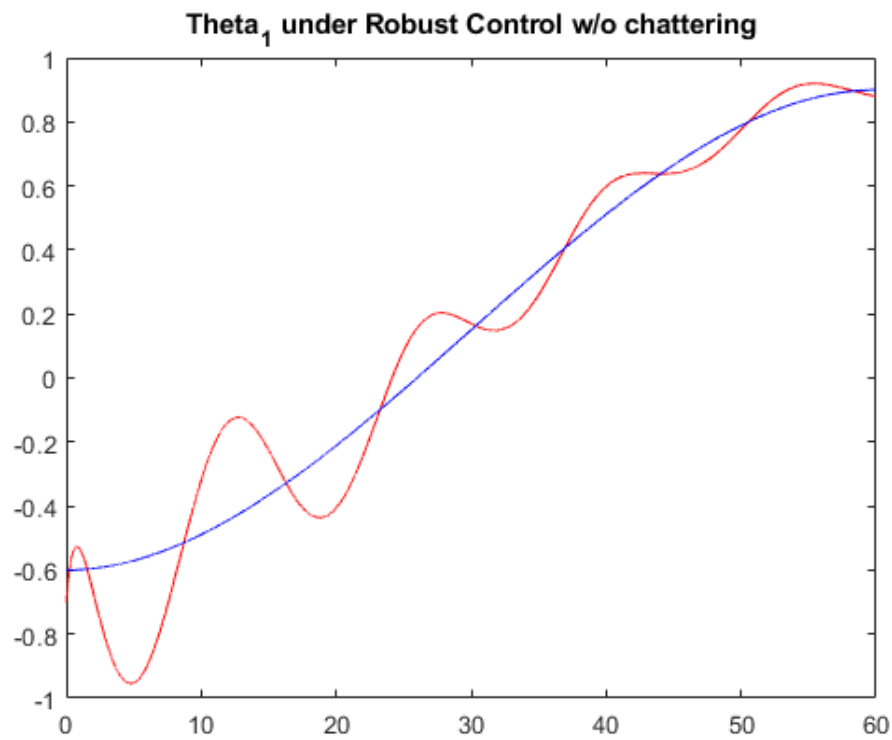
aq = (ddtheta_d + Kp*e + Kv*de + v);
tau = M_bar*aq + C_bar*dtheta ;
end
% torque function for Robust Control without Chattering
function taur = Rob_ctrlapprx(theta_d, dtheta_d, ddtheta_d, theta, dtheta);
global M_bar C_bar Mmat Cmat
Kp = 300*eye(2);
Kv = 200*eye(2);
e = theta_d - theta;
de = dtheta_d -dtheta;
y1 = 0.01;
y2 = 0.02;
y3 = 1;
P = eye(2);
B = [0;1];
%for link 1
x_err1 = [e(1,1) ; de(1,1)];
p1 = y1*norm(x_err1) + y2*(norm(x_err1)^2) + y3;
w1 = transpose(B)*P*(x_err1);
eps1 = 0.01;
if w1 <= eps1
    v1= -((w1*p1)/eps1);
else
    v1 = -((w1*p1)/ norm(w1));
end

%for link 2
x_err2 = [e(2,1) ; de(2,1)];
p2 = y1*norm(x_err2) + y2*(norm(x_err2)^2) + y3;
w2 = transpose(B)*P*(x_err2);
E = norm((inv(Mmat)*M_bar) - eye(2));
eta = norm(((inv(Mmat)*M_bar) - eye(2)) + inv(Mmat)*((C_bar-Cmat)*dtheta));
eps2 = 0.01;
if w2 <= eps2
    v2= -((w2*p2)/eps2);
else
    v2 = -((w2*p2)/ norm(w2));
end
% Added input
v = [v1;v2];

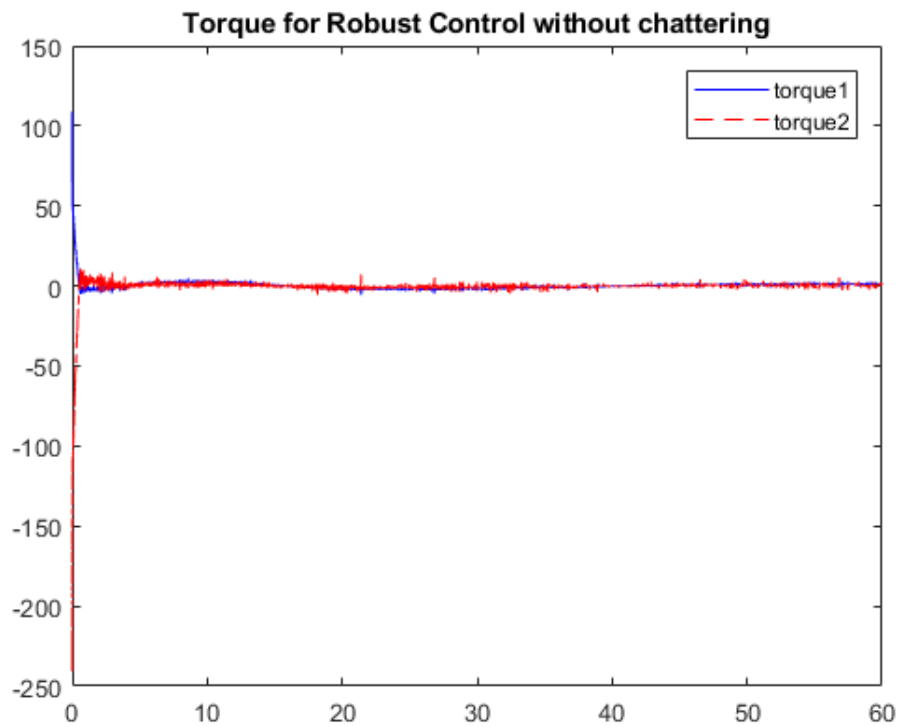
aq = (ddtheta_d + Kp*e + Kv*de + v);

```

```
taur = M_bar*aq + C_bar*dtheta ;  
end
```







In Robust control without chattering, the model performed slightly better in the sense that the sudden high spikes were reduced. It basically eliminated the discontinuity problem of previous case.

#### Trajectory planning using polynomial functions.

```
function [a] = planarArmTraj(theta10,dtheta10, theta1f, dtheta1f,tf, nofigure)
% Input: Initial and final position and velocities, planning horizon [0,tf]
% nofigure=1 then do not output the planned trajectory.
% Cubic polynomial trajectory.

% formulate the linear equation and solve.
M= [1 0 0 0;
    0 1 0 0;
    1 tf tf^2 tf^3;
    0 1 2*tf 3*tf^2];
b=[theta10; dtheta10;theta1f; dtheta1f];
a=M\b;
t=0:0.01:tf;

if nofigure==1
    return
else

figure('Name','Position (degree)');
plot(t,a(1)+a(2)*t+ a(3)*t.^2+a(4)*t.^3,'LineWidth',3);
title('Position (degree)')
grid

figure('Name','Velocity (degree/s)');
plot(t,a(2)+ 2*a(3)*t +3*a(4)*t.^2,'LineWidth',3);
title('Velocity (degree/s)')
grid

figure('Name','Acceleration (degree/s^2)');
plot(t, 2*a(3) +6*a(4)*t,'LineWidth',3);
title('Acceleration (degree/s^2)')
grid
end
end
```

```
end
```

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