



# Lecture 10 State Variable Model

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# Introduction



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Recall the spring-mass-damper system

$$M \frac{d^2 y(t)}{dt^2} + b \frac{dy(t)}{dt} + ky(t) = r(t),$$

To obtain the time response  $y(t)$

$$M \left( s^2 Y(s) - sy(0^-) - \frac{dy}{dt}(0^-) \right) + b(sY(s) - y(0^-)) + kY(s) = R(s).$$

When

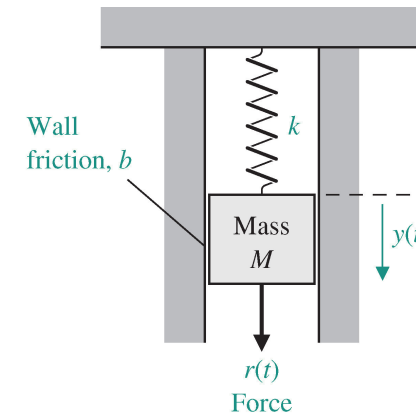
$$r(t) = 0, \quad \text{and} \quad y(0^-) = y_0, \quad \text{and} \quad \left. \frac{dy}{dt} \right|_{t=0^-} = 0,$$

We have

$$Ms^2 Y(s) - Msy_0 + bsY(s) - by_0 + kY(s) = 0.$$

Solving for  $Y(s)$  yields

$$Y(s) = \frac{(Ms + b)y_0}{Ms^2 + bs + k} = \frac{p(s)}{q(s)}. \quad \longrightarrow \quad \textbf{Transfer Function}$$



Limited to:  
Linear  
Time-invariant  
SISO



# Introduction

utilizing a set of ordinary differential equations in a convenient matrix-vector form.

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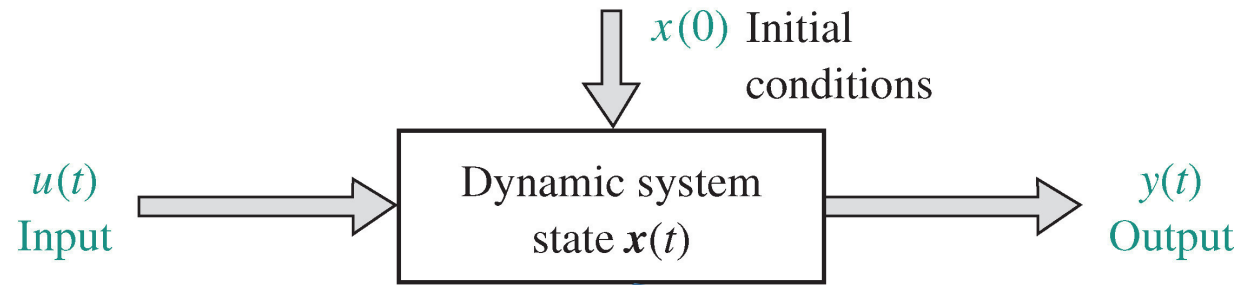
In this chapter, we consider system modeling using *time-domain methods*.

- *Straight forward*
- *LTI SISO models, can be represented via state variable models. Powerful mathematical concepts from *linear algebra* and matrix-vector analysis, as well as effective computational tools, can be utilized.*
- *Readily extended to *nonlinear, time-varying, and multiple input– output* systems.*
- *Computer works in time-domain as well!*

For example, the mass of an *airplane* varies as a function of time as the fuel is expended during flight.

## Outcomes :

- Understand state variables, state differential equations, and output equations.
- Recognize that state variable models can describe the dynamic behavior of physical systems and can be represented by *block diagrams*
- Know how to obtain the *transfer function model from a state variable model*, and vice versa.
- Be aware of *solution methods for state variable models* and the role of the state transition matrix in obtaining the time responses.



*the state of a system is described in terms of a set of state variables*  
 $x(t) = (x_1(t), x_2(t), \dots, x_n(t))$

Again, consider the spring-mass-damper system

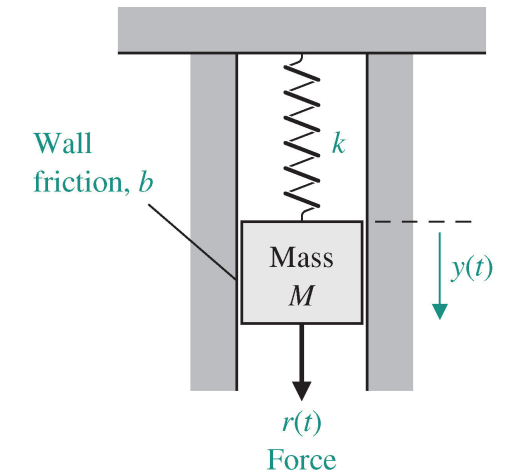
*Define a set of state variables*

$$\mathbf{x}(t) = (x_1(t), x_2(t)),$$

$$x_1(t) = y(t) \quad \text{and} \quad x_2(t) = \frac{dy(t)}{dt}.$$

*The differential equation describes the behavior of the system*

$$M \frac{d^2 y(t)}{dt^2} + b \frac{dy(t)}{dt} + k y(t) = u(t).$$



*A set of state **variables** **sufficient** to describe this system includes : the position and the velocity of the mass.*



# State Space Model



*Substitute the state variables as already defined and obtain*

$$M \frac{dx_2(t)}{dt} + bx_2(t) + kx_1(t) = u(t).$$

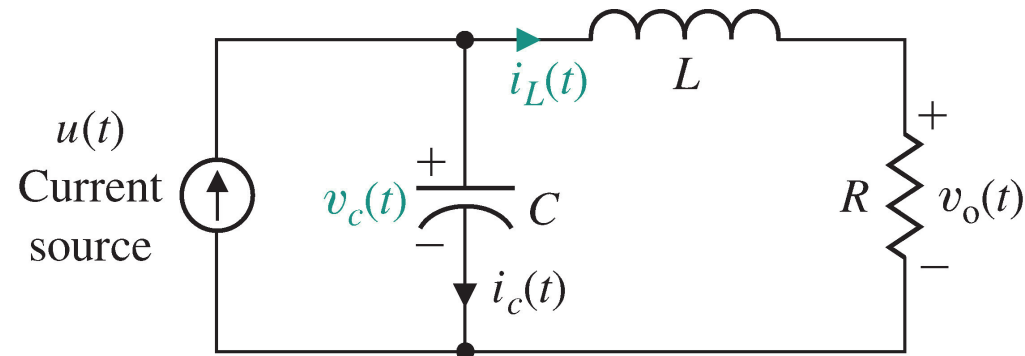
*Further, write as the set of two first-order differential equations*

$$\frac{dx_1(t)}{dt} = x_2(t)$$

$$\frac{dx_2(t)}{dt} = \frac{-b}{M}x_2(t) - \frac{k}{M}x_1(t) + \frac{1}{M}u(t).$$

**State Space Model**

*Another example*





# State Space Model



*Define a set of state variables*

$$\mathbf{x}(t) = (x_1(t), x_2(t)).$$

where  $x_1(t)$  is the capacitor voltage  $v_c(t)$  and  $x_2(t)$  is the inductor current  $i_L(t)$

*Utilizing Kirchhoff's current law at the junction, we obtain*

$$i_c(t) = C \frac{dv_c(t)}{dt} = +u(t) - i_L(t).$$

$$L \frac{di_L(t)}{dt} = -Ri_L(t) + v_c(t).$$



## State Space Model

$$\frac{dx_1(t)}{dt} = -\frac{1}{C}x_2(t) + \frac{1}{C}u(t),$$

$$\frac{dx_2(t)}{dt} = +\frac{1}{L}x_1(t) - \frac{R}{L}x_2(t).$$

*The output of this system is represented by*

$$v_o(t) = Ri_L(t).$$



$$y_1(t) = v_o(t) = Rx_2(t).$$

- The engineer's interest is primarily in physical systems, where the variables typically are *voltages, currents, velocities, positions, pressures, temperatures*, and similar physical variables.
- The state variables that describe a system *are not a unique set*





# State Differential Equations



*general form*

$$\frac{d}{dt} \begin{pmatrix} x_1(t) \\ x_2(t) \\ \vdots \\ x_n(t) \end{pmatrix} = \begin{bmatrix} a_{11} & a_{12} & \cdots & a_{1n} \\ a_{21} & a_{22} & \cdots & a_{2n} \\ \vdots & & \ddots & \vdots \\ a_{n1} & a_{n2} & \cdots & a_{nn} \end{bmatrix} \underbrace{\begin{pmatrix} x_1(t) \\ x_2(t) \\ \vdots \\ x_n(t) \end{pmatrix}}_{\text{state vector}} + \begin{bmatrix} b_{11} & \cdots & b_{1m} \\ \vdots & & \vdots \\ b_{n1} & \cdots & b_{nm} \end{bmatrix} \underbrace{\begin{pmatrix} u_1(t) \\ \vdots \\ u_m(t) \end{pmatrix}}_{\text{inputs}}.$$

*compact notation of the state differential equation as*

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t).$$

*output equation*

$$\mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) + \mathbf{D}\mathbf{u}(t),$$

if we have n states, m inputs, r outputs, what's the dimension of A, B, C, D?



# State Differential Equations



*RLC*

$$\frac{dx_1(t)}{dt} = -\frac{1}{C}x_2(t) + \frac{1}{C}u(t),$$

$$\frac{dx_2(t)}{dt} = +\frac{1}{L}x_1(t) - \frac{R}{L}x_2(t).$$

$$y_1(t) = v_o(t) = Rx_2(t).$$

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} 0 & -\frac{1}{C} \\ \frac{1}{L} & -\frac{R}{L} \end{bmatrix} \mathbf{x}(t) + \begin{bmatrix} \frac{1}{C} \\ 0 \end{bmatrix} u(t)$$

$$y(t) = [0 \quad R]\mathbf{x}(t).$$

When  $R = 3$ ,  $L = 1$ , and  $C = 1/2$ , we have

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} 0 & -2 \\ 1 & -3 \end{bmatrix} \mathbf{x}(t) + \begin{bmatrix} 2 \\ 0 \end{bmatrix} u(t)$$

$$y(t) = [0 \quad 3]\mathbf{x}(t).$$

$y(t) = ?$





# Solution of Differential Equations



Consider the first-order differential equation

$$\dot{x}(t) = ax(t) + bu(t),$$

Take the Laplace transform

$$sX(s) - x(0) = aX(s) + bU(s);$$

therefore,

$$X(s) = \frac{x(0)}{s - a} + \frac{b}{s - a}U(s).$$

The inverse Laplace transform

$$x(t) = e^{at}x(0) + \int_0^t e^{+a(t-\tau)}bu(\tau)d\tau.$$

We expect the solution of the general state differential equation *to be similar and to be of exponential form.*

The *matrix exponential* function is defined by in a similar Taylor series form

$$e^{\mathbf{A}t} = \exp(\mathbf{A}t) = \mathbf{I} + \mathbf{A}t + \frac{\mathbf{A}^2t^2}{2!} + \cdots + \frac{\mathbf{A}^kt^k}{k!} + \cdots,$$



# Solution of Differential Equations



*matrix exponentials is a DEFINITION*

$$X(t) = e^{tA} = \sum_{i=0}^{\infty} \frac{1}{i!} [tA]^i .$$

- We have  $X(0) = I$ .
- The time derivative of  $X$  satisfies  $\dot{X}(t) = A \cdot X(t)$ .
- $X(t)$  commutes with  $A$ , i.e.,  $AX(t) = X(t)A$ .
- If  $A \cdot B = B \cdot A$ , then  $e^{A+B} = e^A \cdot e^B$ .
- **But in general  $e^{A+B} \neq e^A \cdot e^B$  !!!**
- $X(t_1) \cdot X(t_2) = X(t_1 + t_2)$  for all  $t_1, t_2 \in \mathbb{R}$ .
- The function  $X(t) = e^{tA}$  is invertible,  $X(t)^{-1} = e^{-tA}$ .



# Solution of Differential Equations



*Conclusion:* The solution of the state differential equation

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

is found to be

$$\mathbf{x}(t) = \exp(\mathbf{A}t)\mathbf{x}(0) + \int_0^t \exp[\mathbf{A}(t - \tau)]\mathbf{B}\mathbf{u}(\tau)d\tau.$$

*Proof:*

- *Uniqueness of Solutions*

If we have two solutions  $x_1, x_2 : \mathbb{R} \rightarrow \mathbb{R}^{n_x}$ , then  $y = x_1 - x_2$  satisfies

$$\dot{y}(t) = Ay(t) \quad \text{with} \quad y(0) = 0.$$

The auxiliary function  $v(t) = e^{-At}y(t)$  satisfies

$$\dot{v}(t) = -Ae^{-At}y(t) + e^{-At}Ay(t) = -Ae^{-At}y(t) + Ae^{-At}y(t) = 0$$

$$v(0) = 0,$$

$$\implies v(t) = y(t) = 0 \implies x_1 = x_2.$$



# Solution of Differential Equations



*Conclusion:* The solution of the state differential equation

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t)$$

is found to be

$$\mathbf{x}(t) = \exp(\mathbf{A}t)\mathbf{x}(0) + \int_0^t \exp[\mathbf{A}(t - \tau)]\mathbf{B}\mathbf{u}(\tau)d\tau.$$

*Proof:*

- Verify the ODE

*Generalized Leibniz integral rule.*

$$\frac{d}{dt} \int_{a(t)}^{b(t)} g(t, \tau) d\tau = g(t, b(t)) \dot{b}(t) - g(t, a(t)) \dot{a}(t) + \int_{a(t)}^{b(t)} g_t(t, \tau) d\tau .$$

$$\begin{aligned} \dot{x}(t) &= \mathbf{A} e^{\mathbf{A}t} x(0) + e^{\mathbf{A}(t-t)} \mathbf{B} u(t) + \int_0^t \mathbf{A} e^{\mathbf{A}(t-\tau)} \mathbf{B} u(\tau) d\tau \\ &= \mathbf{A} \left[ e^{\mathbf{A}t} x(0) + \int_0^t e^{\mathbf{A}(t-\tau)} \mathbf{B} u(\tau) d\tau \right] + \mathbf{B} u(t) \\ &= \mathbf{A} x(t) + \mathbf{B} u(t) \end{aligned}$$



# Solution of Differential Equations



*Specially, the solution of **an unforced system***

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t)$$

*is found to be*

$$\mathbf{x}(t) = \exp(\mathbf{A}t)\mathbf{x}(0)$$

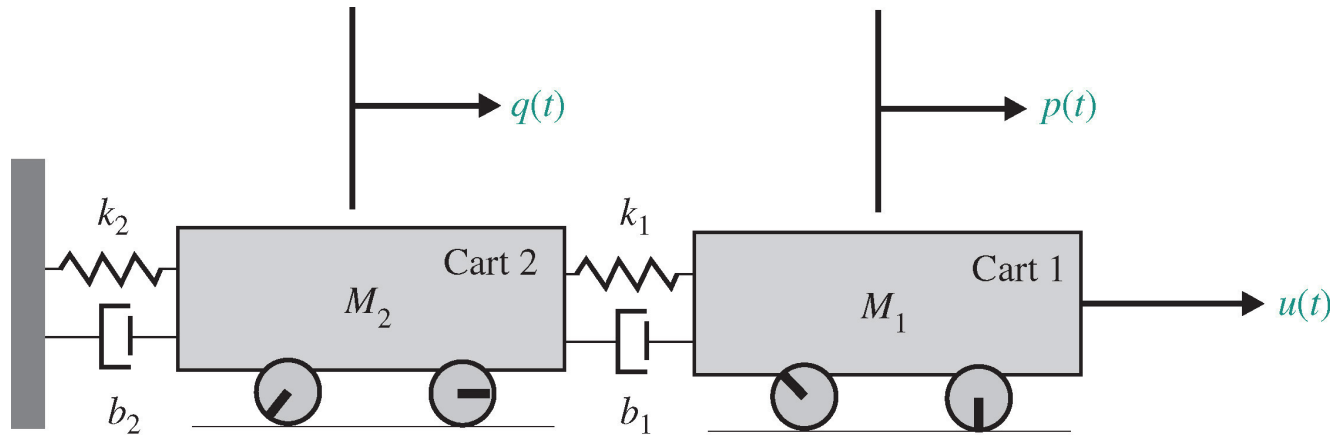
*The matrix exponential function describes the unforced response of the system and is called **the fundamental or state transition matrix**  $\Phi(t, 0)$ .*

*Thus, the general solution can be written as*

$$\mathbf{x}(t) = \Phi(t)\mathbf{x}(0) + \int_0^t \Phi(t - \tau)\mathbf{B}\mathbf{u}(\tau)d\tau.$$

**NOTE, up to now, we are talking about LTI system, for nonlinear or time-varying, there is NO nice general solution form.**

*Example: Two rolling carts*



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$M_1, M_2 =$  mass of carts

$p(t), q(t) =$  position of carts

$u(t) =$  external force acting on system

$k_1, k_2 =$  spring constants

$b_1, b_2 =$  damping coefficients

*We assume that the carts have negligible rolling friction*

*we use Newton's second law*

$$M_1 \ddot{p}(t) + b_1 \dot{p}(t) + k_1 p(t) = u(t) + k_1 q(t) + b_1 \dot{q}(t),$$

$$M_2 \ddot{q}(t) + (k_1 + k_2) q(t) + (b_1 + b_2) \dot{q}(t) = k_1 p(t) + b_1 \dot{p}(t).$$

by defining  $x_1(t) = p(t),$

$$x_2(t) = q(t).$$

$$x_3(t) = \dot{x}_1(t) = \dot{p}(t),$$

$$x_4(t) = \dot{x}_2(t) = \dot{q}(t).$$

*Choose the position difference between Cart1 and Cart2 as the output. Write the state space model of the system in a compact form, i.e. identify the matrices  $A, B, C, D$*



# Block Diagram and Canonical Form



*There are several block diagrams that could represent the same transfer function.*

*Let us initially consider the fourth-order transfer function*

$$G(s) = \frac{Y(s)}{U(s)} = \frac{b_0}{s^4 + a_3s^3 + a_2s^2 + a_1s + a_0}$$

*Rearranging the terms in above Equation and taking the inverse Laplace transform yields*

$$\frac{d^4(y(t)/b_0)}{dt^4} + a_3\frac{d^3(y(t)/b_0)}{dt^3} + a_2\frac{d^2(y(t)/b_0)}{dt^2} + a_1\frac{d(y(t)/b_0)}{dt} + a_0(y(t)/b_0) = u(t).$$

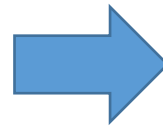
*Define the four state variables as follows:*

$$x_1(t) = y(t)/b_0$$

$$x_2(t) = \dot{x}_1(t) = \dot{y}(t)/b_0$$

$$x_3(t) = \dot{x}_2(t) = \ddot{y}(t)/b_0$$

$$x_4(t) = \dot{x}_3(t) = \dddot{y}(t)/b_0.$$



$$\dot{x}_1(t) = x_2(t),$$

$$\dot{x}_2(t) = x_3(t),$$

$$\dot{x}_3(t) = x_4(t),$$

$$\dot{x}_4(t) = -a_0x_1(t) - a_1x_2(t) - a_2x_3(t) - a_3x_4(t) + u(t);$$

$$y(t) = b_0x_1(t).$$





# Block Diagram and Canonical Form



$$G(s) = \frac{Y(s)}{U(s)} = \frac{b_0}{s^4 + a_3s^3 + a_2s^2 + a_1s + a_0}$$

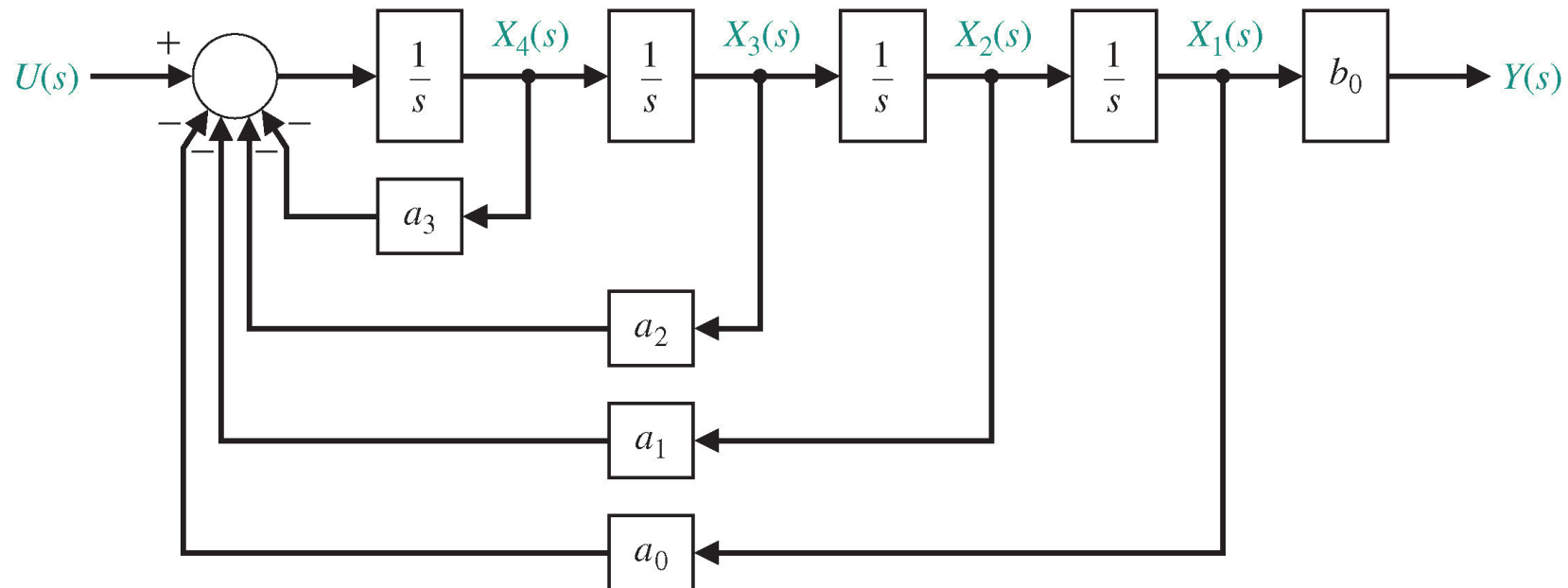
$$\dot{x}_1(t) = x_2(t),$$

$$\dot{x}_2(t) = x_3(t),$$

$$\dot{x}_3(t) = x_4(t),$$

$$\dot{x}_4(t) = -a_0x_1(t) - a_1x_2(t) - a_2x_3(t) - a_3x_4(t) + u(t);$$

$$y(t) = b_0x_1(t).$$





# Block Diagram and Canonical Form



Now consider when the numerator is a polynomial in  $s$ , so that we have

$$G(s) = \frac{Y(s)}{U(s)} = \frac{b_3s^3 + b_2s^2 + b_1s + b_0}{s^4 + a_3s^3 + a_2s^2 + a_1s + a_0} \frac{Z(s)}{Z(s)}.$$

*the intermediate variable*

Equating the numerator and denominator polynomials yields

$$Y(s) = [b_3s^3 + b_2s^2 + b_1s + b_0]Z(s)$$

$$U(s) = [s^4 + a_3s^3 + a_2s^2 + a_1s + a_0]Z(s).$$

$$y(t) = b_3 \frac{d^3z(t)}{dt^3} + b_2 \frac{d^2z(t)}{dt^2} + b_1 \frac{dz(t)}{dt} + b_0 z(t)$$

$$u(t) = \frac{d^4z(t)}{dt^4} + a_3 \frac{d^3z(t)}{dt^3} + a_2 \frac{d^2z(t)}{dt^2} + a_1 \frac{dz(t)}{dt} + a_0 z(t).$$

Define the four state variables as follows:

$$x_1(t) = z(t)$$

$$x_2(t) = \dot{x}_1(t) = \dot{z}(t)$$

$$x_3(t) = \dot{x}_2(t) = \ddot{z}(t)$$

$$x_4(t) = \dot{x}_3(t) = \dddot{z}(t).$$

$$\dot{x}_1(t) = x_2(t),$$

$$\dot{x}_2(t) = x_3(t),$$

$$\dot{x}_3(t) = x_4(t),$$

$$\dot{x}_4(t) = -a_0x_1(t) - a_1x_2(t) - a_2x_3(t) - a_3x_4(t) + u(t),$$

$$y(t) = b_0x_1(t) + b_1x_2(t) + b_2x_3(t) + b_3x_4(t).$$

*In matrix form, we can represent the system*

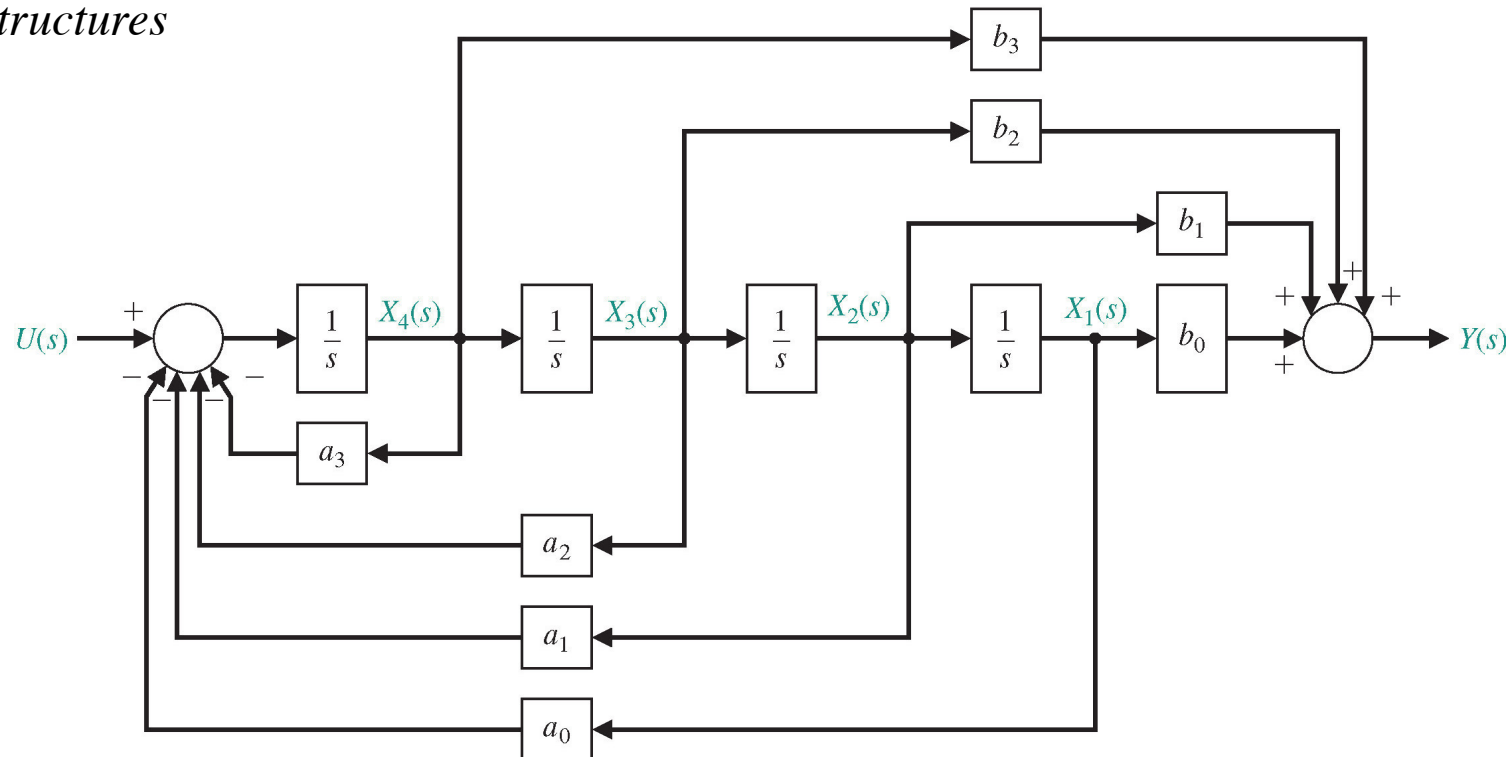
$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}u(t),$$

Controllable Canonical Form

$$\frac{d}{dt} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{pmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -a_0 & -a_1 & -a_2 & -a_3 \end{bmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{pmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1 \end{bmatrix} u(t).$$

$$y(t) = \mathbf{C}\mathbf{x}(t) = [b_0 \quad b_1 \quad b_2 \quad b_3] \begin{pmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{pmatrix}.$$

*The graphical structures*





# Block Diagram and Canonical Form



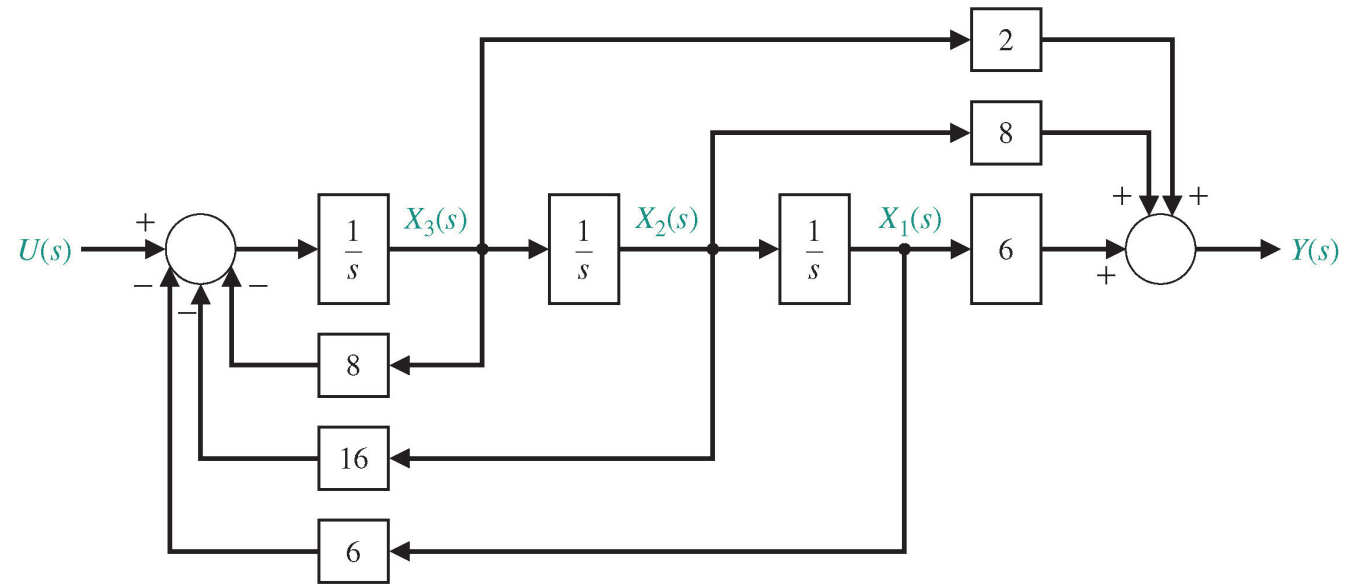
*Example:*

$$T(s) = \frac{Y(s)}{U(s)} = \frac{2s^2 + 8s + 6}{s^3 + 8s^2 + 16s + 6}.$$

*Write down the state space model and corresponding block diagram*

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -6 & -16 & -8 \end{bmatrix} \mathbf{x}(t) + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u(t),$$

$$y(t) = [6 \quad 8 \quad 2] \mathbf{x}(t).$$



A second form of the model we need to consider is the decoupled response modes.

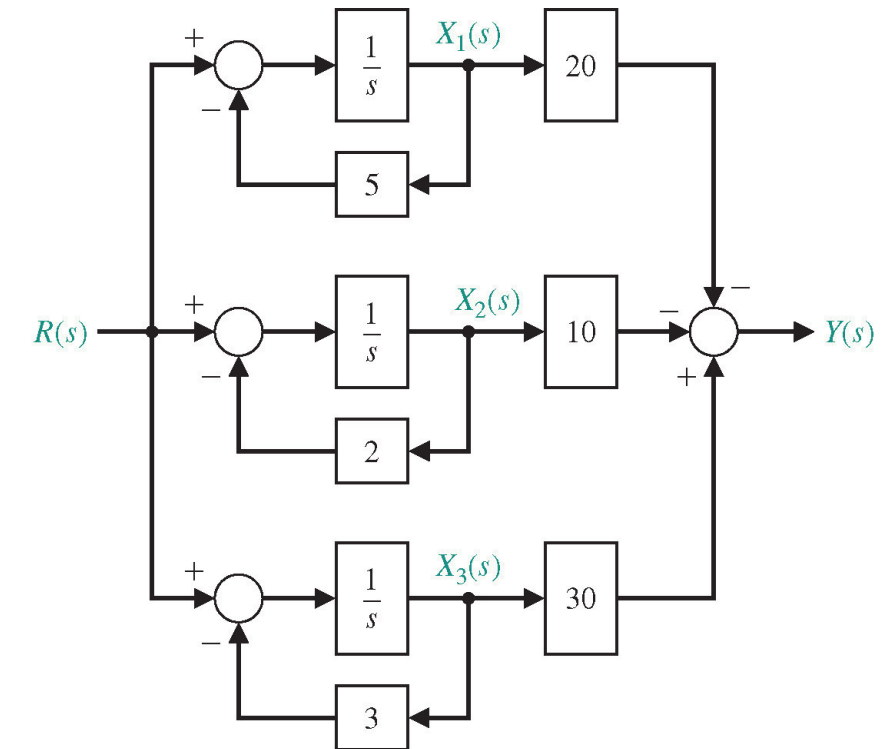
$$\frac{Y(s)}{R(s)} = T(s) = \frac{k_1}{s+5} + \frac{k_2}{s+2} + \frac{k_3}{s+3},$$

where we find that  $k_1 = -20$ ,  $k_2 = -10$ , and  $k_3 = 30$ .

The state variable matrix differential equation and block diagram are

$$\dot{\mathbf{x}}(t) = \begin{bmatrix} -5 & 0 & 0 \\ 0 & -2 & 0 \\ 0 & 0 & -3 \end{bmatrix} \mathbf{x}(t) + \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix} r(t)$$

$$y(t) = [-20 \quad -10 \quad 30] \mathbf{x}(t).$$



this format is often called the diagonal canonical form.



# Block Diagram and Canonical Form



The *state space model is NOT unique* in the sense that

Any invertible linear matrix transformation is represented by  $z = Mx$  can transform the  $x$ -vector into the  $z$ -vector by means of the  $M$  matrix.

$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx + Du\end{aligned}$$



$$sX(s) = AX(s) + BU(s)$$

$$Y(s) = CX(s) + DU(s)$$



$$X(s) = (sI - A)^{-1}BU(s)$$

$$Y(s) = [C(sI - A)^{-1}B + D]U(s)$$



$$G(s) = \frac{Y(s)}{U(s)} = C(sI - A)^{-1}B + D$$

$$z = Mx$$



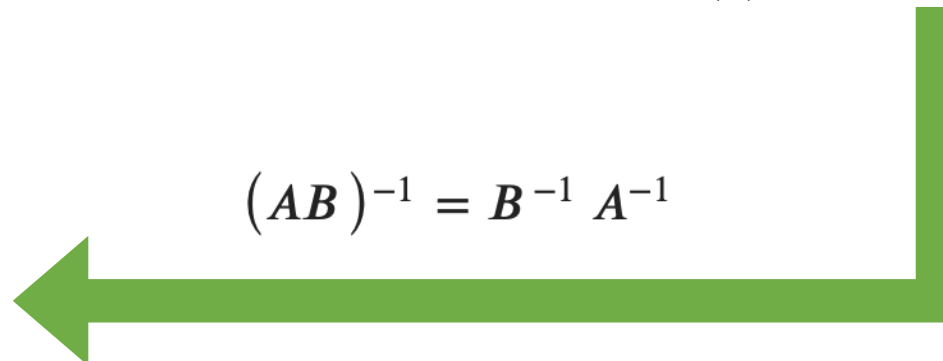
$$\begin{aligned}\dot{z} &= M\dot{x} \\ &= MAM^{-1}z + MBu \\ y &= CM^{-1}z + Du\end{aligned}$$

The Laplace transforms



$$G(s) = \frac{Y(s)}{U(s)} = CM^{-1}(sI - MAM^{-1})^{-1}MB + D$$

$$(AB)^{-1} = B^{-1}A^{-1}$$

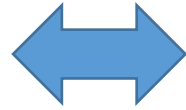




# Block Diagram and Canonical Form



$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx + Du\end{aligned}$$



$$G(s) = \frac{Y(s)}{U(s)} = C(sI - A)^{-1}B + D = \frac{k_p N(s)}{M(s)}$$

*This indicates that*

The location of poles of the system depends on the dynamic matrix  $A$ ,  
to be more specific, the eigenvalues of  $A$ .

*Naturally, a new stability criterion arise*

The system described by *the state space mode*  $(A, B, C, D)$  is said to be  
*Hurwitz or stable if all eigenvalues of  $A$  have negative real parts.*





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# THANKS!

