

$$\underline{k_{m-n}:$$

$$M \cos(-m_n) = 0.0294 \cos(-4.29158^\circ) = 0.029318$$

$$\frac{A}{\cos(-m_n)} = \frac{7.86442}{\cos(-4.29158^\circ)} = 7.88631$$

$$\underline{k_{m-n} = 1}$$

$$\underline{k_{f,i}:$$

$$\frac{l_f'}{b} = \frac{11.5625}{72.5} = 0.159483$$

$$\frac{A}{\cos(-m_n)} = 7.88631$$

$$\underline{k_v = 0.9875}$$

$$\underline{\left(\frac{C_{\alpha\beta}}{C_L}\right)_A i}$$

$$A = 7.86442$$

$$\lambda = 0.5445$$

$$\underline{\left(\frac{C_{\alpha\beta}}{C_L}\right)_A = -0.0004}$$

lateral stability (w/o)

$$P = 2 (deg)$$

$$\left(\frac{C_{\beta}}{P} \right);$$

$$A = 7.86442$$

$$\lambda = 0.5445$$

$$\lambda_{cr} = -4.29158 (deg)$$

$$\left(\frac{C_{\beta}}{P} \right) = -0.0002375$$

$$k_{mp};$$

$$MC_{\beta}(\lambda_{cr}) = 0.029318$$

$$\frac{A}{C_{\beta}(\lambda_{cr})} = 7.88631$$

$$k_{mp} = 1$$

$$\left(\frac{\Delta C_{\beta}}{P} \right) = -0.0005 \sqrt{A} \left(\frac{d}{b} \right) = -0.0005 \sqrt{7.86442} \left(\frac{5.719}{72.5} \right) = -0.000633$$

$$\left(\frac{\Delta C_{\beta}}{P} \right) = -0.000633$$

$$\left(\Delta C_{\beta} \right)_{z_w} = \frac{1.2 \sqrt{A}}{57.3} \left(\frac{z_w}{b} \right) \left(\frac{z_d}{b} \right) = \frac{1.2 \sqrt{7.864}}{57.3} \left(\frac{-3.8125}{72.5} \right) \left(\frac{2 \cdot 5.719}{72.5} \right) = 0.000487$$

$$\left(\Delta C_{\beta} \right)_{z_w} = 0.000487$$

Lateral stability (wing)

$$(C_{\beta})_{L,P} = 0.8988 \left[(-0.0005)(1)(0.9875) + (-0.0004) \right] \\ + 2 \left[(-0.0002375)(1) + (-0.000633) \right] \\ + (-0.000487)$$

$$(C_{\beta})_{L,P} = -0.00303 / \text{deg}$$

Wing contribution to
Lateral stability

M#	$(\frac{1}{\text{deg}})$ $(C_{\beta})_{L,P}$
0.0294	-0.003031
0.04408	-0.002585
0.05877	-0.002429
0.07347	-0.002357
0.08816	-0.002317

$$(C_{lp})_v = -k(L_{2v}) \left(1 + \frac{\partial \sigma}{\partial \beta}\right) \eta_v \bar{V}_v (z_v \cos(\alpha) - d_v \sin(\alpha))$$

$k(L_{2v}) \left(1 + \frac{\partial \sigma}{\partial \beta}\right) \eta_v \bar{V}_v$ was found while calculating the directional stability.

Alpha at different velocities was found using AVL.

M#	(deg) α	(Y_{rod}) $(C_{lp})_v$
0.0294	19.426	-0.025215
0.04408	5.63196	-0.080742
0.05877	1.1573	-0.097956
0.07347	-0.88866	-0.10569
0.08816	-1.99734	-0.109885

Lateral Stability (wing + v-tail)

$$(C_{lP})_{w,p,v} = (C_{lP})_{w,p} + (C_{lP})_v$$

(1/rd)

M#	$(C_{lP})_{w,p,v}$
0.0294	-0.198879
0.04408	-0.228852
0.05877	-0.237127
0.07347	-0.240736
0.08816	-0.242639