SERVICE BULLETIN REVISION TRANSMITTAL SHEET

AIRBUS INDUSTRIE
CUSTOMER SERVICES DIRECTORATE
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31707 BLAGNAC CEDEX FRANCE
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ATA SYSTEM: 21

TITLE: AIR CONDITIONING - AVIONICS EQUIPMENT VENTILATION - ADD RESISTOR ON AEVC

INLET SKIN AIR VALVE CONTROL LINE.

MODIFICATION No.: 30934P6961

This page transmits Revision No. 01 of Service Bulletin No. A320-21-1129.

ADDITIONAL WORK

No additional work is required by this revision for aircraft modified by any previous issue.

REASON

Revision No. 01 issued to add reference to TFU 21.26.00.020 and modify the effectivity.

CHANGES

SUMMARY:

- EFFECTIVITY
 - . Operators updated and NOTE modified.
- REFERENCES/REPERCUSSIONS
 - . Reference to TFU 21.26.00.020 added.

PLANNING INFORMATION:

- EFFECTIVITY
 - . Effectivity updated and NOTE modified.
- REFERENCES
 - . Reference to AWM 20-60-00 becomes ESPM 20-30-00.

ACCOMPLISHMENT INSTRUCTIONS:

- GENERAL
- Standard Practices

5 DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 1 of 2

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SERVICE BULLETIN REVISION TRANSMITTAL SHEET

. Reference to AWM 20-60-00 becomes ESPM 20-30-00.

NOTE: Lay-out changes due to computerization of this revision are not identified.

FILING INSTRUCTIONS

This Service Bulletin has been generated electronically and is reissued as a complete document. Replace the complete document.

Put this Revision Transmittal Sheet in front of the Service Bulletin.

HISTORY OF PREVIOUS REVISIONS

No previous revisions.

REVISION SEQUENCE

ORIGINAL: Apr 03/01

REVISION No.: 01 - Mar 06/02

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

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This summary is for information only and is not approved for modification of the aircraft

ATA SYSTEM: 21

TITLE: AIR CONDITIONING - AVIONICS EQUIPMENT VENTILATION - ADD RESISTOR ON AEVC

INLET SKIN AIR VALVE CONTROL LINE.

MODIFICATION No.: 30934P6961

REASON/DESCRIPTION/OPERATIONAL CONSEQUENCES

The skin air inlet valve closure command in the Avionics Equipment Ventilation Computer (AEVC) can be aborted by a voltage spike related to the valve internal relay.

Some electrical characteristics of wiring and the normal spectral reponse of the valve lead to an unstable oscillation in the command circuit and cross over the threshold of the current limitation protection of the AEVC.

In order to reduce the occurence of aborted closure command, this Service Bulletin proposes addition of a resistor on the closure command wiring.

By an amplitude modulation rejection and an uncoupling between the valve, the wiring and the computer, the resistor shifts the oscillating point to an other which has shown through experience to be out of AEVC range.

EVALUATION TABLE								
COMPLIANCE	Desirable	CANCELS INSPECTION SB	No					
POTENTIAL AD	No	A/C OPERATION AFFECTED	No					
RELIABILITY AFFECTED	No	PAX COMFORT AFFECTED	No					
COST SAVING	No	ETOPS AFFECTED	No					
STRUCTURAL LIFE EXTN	No	VENDOR SB INVOLVED	No					
KIT PRICE (USD) A01R02	260							

EFFECTIVITY

This Service Bulletin is applicable to these operators :

5 DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 1 of 4

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SERVICE BULLETIN SUMMARY

53Z AAA AAR ACA ADR AEF AES AFR AIH AJM ALK AMC AMM AMU ANA AUA AWE AWQ AZA BAW BES BMA C3J C9G CES CIB CJG CMM CNW CSC CSN CTN CYP DCS DLH EDW EEZ EIN EWG F2I FIN G2X G4I GBL GFA HDA HVN IAC IBE IWD JBU JKK JMC KAC KZW LAJ LAN LBT LTU LXR MEA MON MSR MXA NWA O2X OHY OYC PAL PIC QAF QTR RJA RYN S2A SAB SEU SHJ SHK SLK SSV SWR SYR TAI TAM TAP TAR TAS TNA TWJ UAL USA VIR VLE WGT

NOTE: This modification is embodied before delivery on A319 aircraft MSN 1371, 1373, 1375, 1395, 1401, 1414, 1429, 1444, 1445, 1463, 1466, 1479 and subsequent, on A320 aircraft MSN 1367, 1407, 1405, 1409, 1418, 1424, 1430, 1432, 1435, 1439, 1452 and subsequent, on A321 aircraft MSN 1356, 1408, 1417, 1431, 1433, 1455, 1462, 1465, 1476, 1481 and subsequent.

CONCURRENT REQUIREMENTS

None

REFERENCES/REPERCUSSIONS

TFU : 21.26.00.020

OEB : None
AOT : None
SIL : None
LIFE LIMIT : None
LINE MAINTENANCE AFFECTED : No
OTHERS : None

NATURE OF THE WORK

AIRCRAFT : YES EQUIPMENT : NO HARD : NO SOFT : NO

OBRM: NO

MANPOWER

Kit 211129A01R02

TOTAL MANHOURS 4.0

ELAPSED TIME (HOURS) 4.0

MATERIAL INFORMATION

AIRCRAFT DATA

Kit 211129A01R02

Resistor assy, bundle, placard set.

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 2

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SERVICE BULLETIN SUMMARY

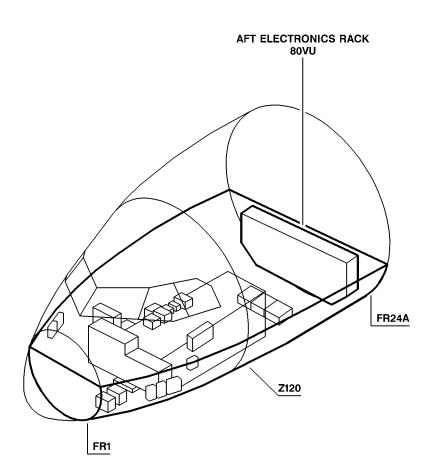
<u>APPENDICES</u>

None

DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No.: 01 - Mar 06/02 Page: 3

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DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

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TITLE: AIR CONDITIONING - AVIONICS EQUIPMENT VENTILATION - ADD RESISTOR ON AEVC

INLET SKIN AIR VALVE CONTROL LINE.

MODIFICATION No.: 30934P6961

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1. PLANNING INFORMATION

A. <u>EFFECTIVITY</u>

(1) Models

319-111 319-112 319-113 319-114 319-115 319-131 319-132 319-133 320-111 320-211 320-212 320-214 320-231 320-232 320-233 321-111 321-112 321-131 321-211 321-231

- (2) Aircraft
 - (a) Effectivity by MSN

This Service Bulletin is applicable to aircraft MSN:

0002-0008 0010-0014 0016-0078 0080-0104 0106-0373 0375-0432 0434-0480 0482-0535 0537-0707 0709-1277 1279-1355 1357-1366 1368-1370 1372 1374 1376-1394 1396-1400 1402-1404 1406 1410-1413 1415 1416 1419-1423 1425-1428 1434 1436-1438 1440-1443 1446-1451 1453 1456 1458 1460 1468 1471 1472 1474 1477

NOTE: This modification is embodied before delivery on A319 aircraft MSN 1371, 1373, 1375, 1395, 1401, 1414, 1429, 1444, 1445, 1463, 1466, 1479 and subsequent, on A320 aircraft MSN 1367, 1407, 1405, 1409, 1418, 1424, 1430, 1432, 1435, 1439, 1452 and subsequent, on A321 aircraft MSN 1356, 1408, 1417

5 DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 1 of 20

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SERVICE BULLETIN

, 1431 , 1433 , 1455 , 1462 , 1465 , 1476 , 1481 and subsequent.

(b) Effectivity by Operator

The Operator/MSN relationship is provided for information only and is correct at the time of issue in accordance with the information available to AIRBUS. Any future changes resulting from transfer of an aircraft from one operator to another will not be reflected in this list unless the Service Bulletin is revised for another reason.

OPERATOR	MSN									
53Z	1468									
AAA									0140 0662	
AAR	0771	0855	1060	1293						
ACA	0154 0248 0302 0350 0639 0697	0159 0253 0305 0359 0649 0711 0769	0174 0254 0309 0378 0656 0719	0175 0255 0310 0384 0670 0721	0183 0265 0311 0403 0672 0728	0210 0277 0324 0404 0682 0732	0231 0279 0330 0426 0688 0736	0232 0283 0333 0546 0691 0740	0149 0233 0284 0341 0572 0693 0742 0829	0242 0290 0342 0634 0695 0752
ADR	0043	0113	0114							
AEF					0661 1421		0787	0792	0954	0990
AES	0739	0743	0839	0892						
AFR	0019 0101 0145 0214 0244 0498 0637 0938 1169	0020 0102 0155 0215 0270 0509 0644 0956	0021 0108 0156 0220 0278 0521 0647 0985 1190	0033 0115 0184 0226 0285 0529 0660 0998	0036 0128 0186 0227 0286 0544 0674 1000	0044 0129 0187 0228 0287 0598 0675 1020	0061 0130 0188 0235 0337 0600 0684 1025	0062 0131 0203 0236 0352 0608 0761 1036	0014 0063 0133 0204 0237 0377 0618 0777 1133 1344	0100 0144 0211 0239 0491 0625 0796 1151
AIH		0164 0808				0193	0362	0363	0424	0444
AJM		0528	0624	0626	0628	0630	0666	0775	1213	1390
ALK	0406									
AMC	0112									
AMM					1017					
AMU	0550	0557	0573	0582	0620	0631	0805	0908		

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

SERVICE BULLETIN

	OPERATOR	MSN									
	ANA	0300 0549	0139 0328 0554 1227	0365	0383	0413	0482	0501	0507	0531	0534
	AUA		0570 1387		0768	0776	0797	0920	0935	1137	1150
	AWE	0081 0448 0866 1064 1196	0053 0082 0455 0889 1075 1223 1419	0091 0471 0924 1079	0092 0527 0931 1083	0098 0543 0953 1088	0099 0584 1003 1110	0238 0762 1043 1111	0271 0770 1049 1116	0315 0803 1050 1166	0317 0856 1056 1178
I	AWQ	0542									
	AZA	0526	0477 0532 0951	0576	0583	0586	0593	0599	0765	0819	
	BAW	1082	0008 1115 1258	1118	1142	1193	1197	1222	1225	1232	1236
	BES	1256									
	BMA		0810 1177				0968	0974	1014	1045	1153
	C3J	0772	0799	0828	0883	0914	0925				
1	C9G	0460									
	CES		1028 1357				1093	1108	1285	1303	1312
1	CIB	0774 1416	0793 1437	0809	0884	0894	0905	0957	0971	1381	1402
	CJG	0707	0724	0876	0921	1296					
1	CMM	0132	0397	0409							
	CNW		0754 1316		0854	0897	0909	0939	0967	0984	0986
	CSC	0540	0551	0556	0878	0915	0919	1013			
	CSN		0698 0859								
	CTN	0258	0767	0833	1009	1029	1237	1252			
	CYP	0028	0035	0037	0038	0180	0256	0295	0316	0414	0430
	DCS	1053									

DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

SERVICE BULLETIN

	OPERATOR	MSN									
	DLH	0110 0200 0346 0502 0616 0717	0111 0201 0382 0505 0623 0723	0116 0202 0401 0518 0627 0729	0117 0209 0412 0560 0636 0738	0078 0135 0216 0458 0563 0652 0744 1260	0137 0217 0468 0564 0689 0853	0147 0218 0473 0567 0692 0860	0161 0267 0474 0595 0694 0875	0162 0268 0484 0609 0699	0172 0269 0493 0610 0700
•	EDW	0942	0947	1026							
I	EEZ	1448									
	EIN	0815	0841	0926	0991	1023	1094	1242	1394	1443	
	EWG	0646	0654	0794	1016	1172					
1	F2I	1054	1086	1364							
	FIN	0941	0961	1073	1107	1185	1241	1309	1352		
1	G2X	0249									
	G4I	1002	1157								
	GBL	1215	1428								
	GFA	0313 0497		0345	0375	0419	0421	0438	0445	0459	0466
	HDA	0633	0756	0784	0812	0816	0877	0930	0993	1024	1253
	HVN	0590 0648		0594	0597	0601	0605	0607	0611	0617	0619
	IAC	0074 0327	0075	0080 0396	0089 0398	0049 0090 0416	0095	0096	0097	0247	0257
	IBE	0223 0323 1087	0224 0356 1099	0240 0992 1101	0241 0994 1119	0158 0246 1021 1154 1292	0264 1027 1180	0266 1047 1200	0274 1059 1208	0303 1063 1221	0312 1067 1229
	IWD	0225	0308	0314	0879	0888					
	JBU	1123 1398		1173	1235	1240	1257	1270	1280	1302	1327
	JKK	1276	1333	1349	1366						
1	JMC	0354	0357	0443	0716						
	KAC	0181	0182	0195							
I	KZW	0641	0651	0679							
	LAJ	0376	0386	0480	1194						
	LAN	1304	1332	1351	1355						
	LBT	1121									
	LTU	0530	1385	1441							

DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

SERVICE BULLETIN

	OPERATOR	MSN									
1	LXR	0191	0395								
•	MEA		0663	0668	0676	0760					
I	MON	0379	0389	0391	0392	0446	0983	1015	1202		
	MSR	0165 0725	0166	0178	0194	0198	0351	0366	0680	0687	0715
	MXA	0320	0259 0321 0428	0332							
	NWA	0125 0262 0318 0380 0766 0907 1058 1164	0032 0152 0263 0319 0381 0778 0911 1062 1167 1453	0153 0272 0329 0387 0786 0923 1071 1186	0160 0273 0339 0388 0801 0962 1078	0171 0281 0340 0399 0807 0964 1090	0192 0282 0355 0400 0818 0981 1091	0197 0297 0358 0408 0830 0988 1126	0206 0298 0360 0410 0832 0996 1129	0208 0306 0367 0417 0846 1011 1131	0213 0307 0372 0418 0903 1037 1149
	02X	0304									
I	OHY	0364									
	OYC		0222		0299	0301	0348	0349			
_	PAL		0745	0753							
<u> </u>	PIC	0604									
!	QAF	0927		0042	0077						
l	QTR RJA		0932 0088			0560					
	RYN		0394				0725				
1	S2A	1413		0411	0714	0730	0733				
İ	SAB	0970	0995 1283					1102	1124	1145	1160
•	SEU	0653	0657	0737	0749	0973					
	SHJ	0910									
	SHK	0322	0326	0344	0478						
	SLK	0872	0899	0969	1074	1098	1228	1422			
	SSV		0338 0671			0437	0447	0476	0496	0525	0579
	SWR	0559 0612	0519 0562 0621 0713	0566 0629	0574 0635	0577 0642	0578 0643	0585 0664	0588 0673	0596 0681	0603 0701
	SYR	0886	0918	1032	1076	1085	1117				

DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

SERVICE BULLETIN

OPERATOR	MSN
TAI	0425 0461 0558 0561 0733 0741 0747 0789 0874 0902 0912 0916 1007 1066 1113 1140 1159 1300 1334 1339 1353 1374 1400
TAM	0243 0250 0251 0334 0335 0440 0453 0758 0976 1010 1092 1096 1103 1139 1143 1158 1246 1251 1368 1376
TAP	0185 0234 0750 0755 0763 0790 0821 0837 0906 0917 0933 0945 0960 0979 0982 1034 1100 1106 1120 1127 1165 1181 1206 1231 1307 1399
TAR	0119 0123 0124 0205 0370 0390 0402 0511 0869 0880 0958 0975 1187
TAS	0371 0814 0937 1372
TNA	0441 0538 0555 0602 0606 0731 0746 0791 0822
TWJ	0913
UAL	0435 0439 0442 0450 0452 0454 0456 0457 0462 0463 0464 0465 0470 0472 0475 0479 0483 0485 0487 0489 0500 0503 0504 0506 0508 0510 0512 0523 0539 0568 0571 0587 0589 0592 0613 0638 0655 0678 0683 0686 0690 0702 0748 0751 0759 0780 0783 0788 0798 0804 0820 0824 0825 0826 0834 0836 0842 0843 0847 0850 0851 0857 0858 0862 0865 0867 0871 0873 0882 0893 0898 0944 0948 0952 0955 0965 0980 0989 1001 1022 1031 1104 1105 1128 1146 1163 1192 1211 1243 1248 1266 1272 1282 1290 1291 1321 1341 1343 1359 1363 1420 1426 1427 1460 1474 1477
USA	0844 0861 0863 0868 0885 0890 0896 0904 0922 0929 0936 0946 0949 0972 0997 1019 1033 1038 1040 1044 1046 1051 1052 1055 1061 1065 1069 1077 1084 1089 1095 1097 1109 1112 1114 1122 1134 1135 1136 1141 1147 1148 1155 1170 1171 1182 1203 1205 1209 1210 1224 1245 1254 1263 1264 1265 1268 1269 1275 1277 1286 1287 1289 1294 1297 1298 1301 1310 1311 1313 1314 1315 1317 1319 1326 1328 1331 1340 1342 1348 1354 1358 1360 1369 1378 1382 1389 1391 1393 1403 1412 1425 1436 1442 1447 1472
VIR	0764 0978 1130 1219 1233
VLE	0085 0189 0190 0343 0420 0436 1125 1132 1152 1162 1179 1198 1244 1259 1308 1322 1451
WGT	1212

(c) Effectivity by MSN and Kit/Configuration

The kits and configurations applicable to these MSNs are given at the end of this list: 0002-0008 0010-0014 0016-0078 0080-0104 0106-0373 0375-0432 0434-0480 0482-0535 0537-0707 0709-1277 1279-1355 1357-1366 1368-1370 1372 1374 1376-1394 1396-1400 1402-1404 1406

1410-1413 1415-1416 1419-1423 1425-1428 1434 1436-1438

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No.: 01 - Mar 06/02 Page: 6

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SERVICE BULLETIN

1440-1443 1446-1451 1453 1456 1458 1460 1468 1471-1472 1474 1477

KIT No. QTY PER A/C CONFIGURATION

211129A01R02 1 None

(3) Spares

None

B. <u>CONCURRENT REQUIREMENTS</u>

None

C. REASON

(1) History

The skin air inlet valve closure command in the Avionics Equipment Ventilation Computer (AEVC) can be aborted by a voltage spike related to the valve internal relay.

Some electrical characteristics of wiring and the normal spectral reponse of the valve lead to an unstable oscillation in the command circuit and cross over the threshold of the current limitation protection of the AEVC.

(2) Objective/Action

In order to reduce the occurence of aborted closure command, this Service Bulletin proposes addition of a resistor on the closure command wiring.

(3) Advantages

By an amplitude modulation rejection and an uncoupling between the valve, the wiring and the computer, the resistor shifts the oscillating point to an other which has shown through experience to be out of AEVC range.

(4) Operational/Maintenance Consequences

None

D. <u>DESCRIPTION</u>

To accomplish this Service Bulletin it is necessary to :

- (1) Modify the rear plate of the shelf 86VU.
- (2) Install a resitor on the rear plate of the shelf 86VU.

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No.: 01 - Mar 06/02 Page: 7

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SERVICE BULLETIN

(3) Modify the wiring in the shelf 86VU.

E. COMPLIANCE

(1) Classification

DESIRABLE

(2) Accomplishment Timescale

In accordance with operators' maintenance schedule.

F. APPROVAL

The technical content of this Service Bulletin has been approved under the authority of the DGAC Design Organisation Approval No. F.JA.02.

If an aircraft listed in the effectivity has a modification or repair embodied that is not of AIRBUS origin, and which affects the content of this Service Bulletin, the operator is responsible for obtaining approval by its airworthiness authority for any adaptation necessary before incorporation of the Service Bulletin.

G. MANPOWER

The manpower estimates given in this Service Bulletin are based on the direct labor cost to do the work. These estimates assume that the work will be done by experienced personnel, and may need to be revised upwards to suit operator's circumstances. The estimates do not include the time to prepare, plan or inspect the work. Manufacture and procurement of parts and tools, drying times for paints, sealants, etc, and general administration work are also not included.

Kit 211129A01R02

Get access	1.0
Modification of rear plate in 86VU	0.5
Installation of resitor in 86VU	0.5
Modification of wiring in 86VU	0.5
Tests	0.5
Close-up	1.0
TOTAL MANHOURS	4.0
ELAPSED TIME (HOURS)	4.0

H. WEIGHT AND BALANCE

Not changed

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

SERVICE BULLETIN

I. <u>ELECTRICAL LOAD DATA</u>

Not changed

J. REFERENCES

: 06-41-53 11-00-00 12-34-24 21-26-34 25-54-11 52-10-00 Aircraft Maintenance Manual (AMM)

52-30-00 52-41-00

Consumable Material List (CML)

Elec. Std. Practices Manual (ESPM) : 20-30-00 Structural Repair Manual (SRM) : 51-44-11

K. PUBLICATIONS AFFECTED

Illustrated Parts Catalog (IPC) : 21-26-08

Aircraft Wiring List (AWL)

L. <u>INTERCHANGEABILITY/MIXABILITY</u>

None

SERVICE BULLETIN No.: A320-21-1129 DATE : Apr 03/01



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DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 10

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SERVICE BULLETIN

2. MATERIAL INFORMATION

A. MATERIAL - PRICE AND AVAILABILITY

(1) Material

Customers with aircraft shown in the effectivity of this Service Bulletin should send a purchase order to AIRBUS. Quote the number of this Service Bulletin. The address is:

AIRBUS MATERIEL SUPPORT CENTER P.O. Box 630262 22312 HAMBURG GFRMANY

(2) Price and Availability

Kit 211129A01R02

Cost : 260 US Dollars

Availability: 120 calendar days from receipt of order

The Kit availability given above is the standard lead time from the date of your purchase order. If you require the Kit(s) before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.

The sales terms (costs and availability) are estimated in relation to economic conditions at the issue date of the Service Bulletin.

B. **INDUSTRY SUPPORT INFORMATION**

None

C. <u>LIST OF COMPONENTS</u>

Kit 211129A01R02

ITEM	NEW PART No.	QTY UM	KEYWORD	ITEM	OLD PART No	o. INT INST DISP
1	D9251488200000	1	Resistor			
2	NAS1100-04-5	2	Screw			
3	ASNA2397-4L	4	Washer			
4	MS21042-04	2	Nut			
	D9000095214895	1	Bundle			
	D9100095209695	1	Set			
	NSA935401-03	1	Tie			
	NSA935401-05	1	Tie			
	NSA935401-08	1	Tie			

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 11

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SERVICE BULLETIN

NOTE: Refer to the AIRBUS Standards Manual (SM) if you find part numbers of hardware components in the related kit(s) which you cannot identify in the LIST OF COMPONENTS of this Service Bulletin. The SM will give you the correct part number to part number relationship.

D. <u>LIST OF MATERIALS - OPERATOR SUPPLIED</u>

(1) Consumable Materials

DESCRIPTION	REFERENCE TO CML MAT. No.	QTY PER A/C INST DISP				
Anti-Corrosion Primer (Polyurethane)	16-001	As required				
Polyurethane Topcoat Grey	16-002	As required				

(2) Components

None

E. PARTS TO BE RE-IDENTIFIED BY THE OPERATOR

None

F. TOOLING - PRICE AND AVAILABILITY

None

G. SPECIAL TOOLS

None

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

SERVICE BULLETIN

3. ACCOMPLISHMENT INSTRUCTIONS

A. GENERAL

<u>WARNING</u>: MAKE SURE THAT YOU OBEY ALL THE WARNINGS AND THE CAUTIONS INCLUDED IN THE REFERENCED PROCEDURES.

(1) Preparation

- (a) Electrically ground the aircraft (Refer to AMM 12-34-24 Page block 201).
- (b) Put the access platforms in position at the access doors.
- (c) Open the passenger/crew door 841 (Refer to AMM 52-10-00 Page block 201).
- (d) Open the FWD cargo-compartment door 825 (Refer to AMM 52-30-00 Page block 201).
- (e) Open the avionics-compartment door 824 (Refer to AMM 52-41-00 Page block 201).
- (f) In the FWD cargo-compartment at FR24A, remove the partition panels 132EW, 131CW, 131AW and 132BW (Refer to AMM 06-41-53 and AMM 25-54-11 Page block 401).
- (g) Do the preparation procedure as specified in the removal/installation of the AEVC (10HQ) (Refer to AMM 21-26-34 Page block 401).

(2) Standard Practices

- (a) For the specification of the material number (Mat No.), refer to CML.
- (b) Drill and deburr the holes (Refer to SRM 51-44-11).
- (c) After drilling, apply Anti-Corrosion Primer (Polyurethane) (Mat No. 16-001) and Polyurethane Topcoat Grey (Mat No. 16-002).
- (d) Bond the placards (Refer to AMM 11-00-00 Page block 201).
- (e) To install the new wires, cut them to the necessary length, crimp the terminals and connect them.
- (f) Do a continuity test of the modified and new wires.
- (g) Route the new wires with the wires that are in the aircraft.
- (h) Obey the instructions of the wiring installation (Refer to ESPM 20-30-00).

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

SERVICE BULLETIN

B. MODIFICATION

- (1) Modify the rear plate of the shelf 86VU. Refer to figure 1
 - (a) Drill two holes to 3.5 mm (0.138in.) dia., at the dimension given in the figure.
- (2) Install a resitor on the rear plate of the shelf 86VU. Refer to figure 2
 - (a) Near the 1862VT and 1880VT, Install:

FIN 33HQ

1 Resistor assy D9251488200000 Item 1

with:

2 Screw NAS1100-04-5 Item 2 4 Washer ASNA2397-4L Item 3 2 Nut MS21042-04 Item 4

(b) Near the resistor, bond the placard 33HQ, supplied in :

Placard set D9100095209695

(3) Modify the wiring in the shelf 86VU. Refer to figure 3

Refer to figure 4

- (a) Remove the AEVC (10HQ) (Refer to AMM 21-26-34 Page block 401).
- (b) Modify the connection of wire shown on the lines 1 and 2 (Refer to figure 4).
- (c) Install the wire shown on the line 6 (Refer to figure 4) supplied in :

Bundle D9000095214895

(d) Attach the wires with:

Cable-ties NSA935401-03 Cable-ties NSA935401-05 Cable-ties NSA935401-08

(e) Install the AEVC (10HQ) (Refer to AMM 21-26-34 Page block 401).

C. TEST

(1) Do the test after the removal/installation of the AEVC (10HQ) (Refer to AMM 21-26-34 Page block 401).

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 14

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SERVICE BULLETIN

D. CLOSE UP

- (1) Make sure that the work areas are clean and clear of tools and other items of equipment.
- (2) Do the close-up procedure as specified in the removal/installation of the AEVC 10HQ (Refer to AMM 21-26-34 Page block 401).
- (3) In the FWD cargo-compartment at FR24A, install the partition panels 132EW, 131CW, 131AW and 132BW (Refer to AMM 06-41-53 and AMM 25-54-11 Page block 401).
- (4) Close the avionics-compartment door 824 (Refer to AMM 52-41-00 Page block 201).
- (5) Close the FWD cargo-compartment door 825 (Refer to AMM 52-30-00 Page block 201).
- (6) Close the passenger/crew door 841 (Refer to AMM 52-10-00 Page block 201).
- (7) Remove the access platforms from the access doors.
- (8) Disconnect the aircraft electrical ground-connection (Refer to AMM 12-34-24 Page block 201).
- (9) Restore the systems and the aircraft to normal operating condition.

E. **DOCUMENTATION**

Write in the applicable aircraft records that you have done all the work given in the Service Bulletin.

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

SERVICE BULLETIN

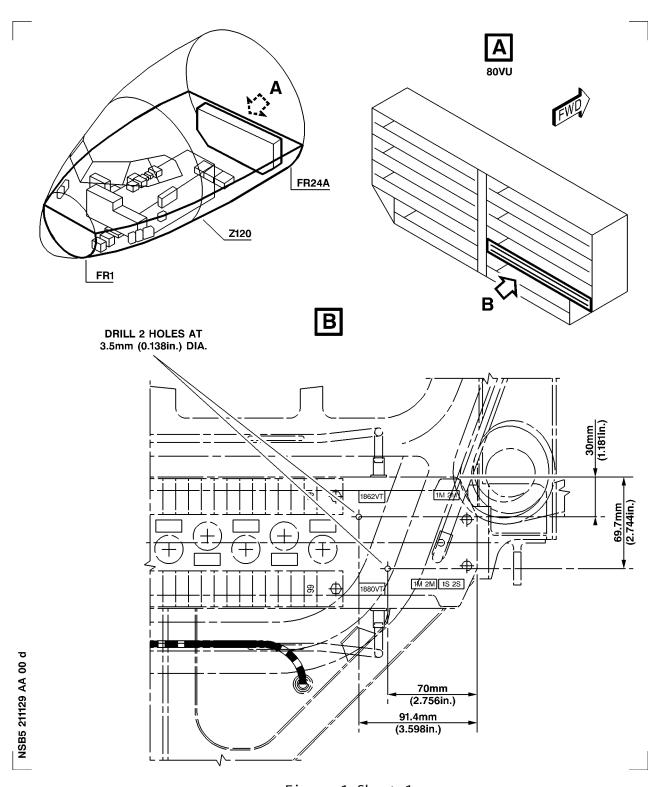


Figure 1 Sheet 1 Modification of Rear Plate of the Shelf 86VU

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 16

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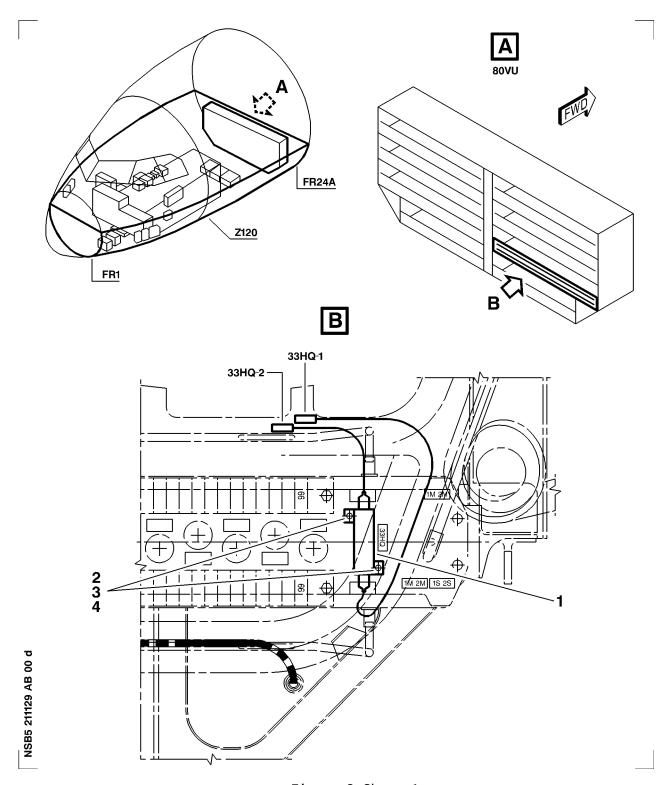


Figure 2 Sheet 1 Installation of the Resistor Assy in the Shelf 86VU

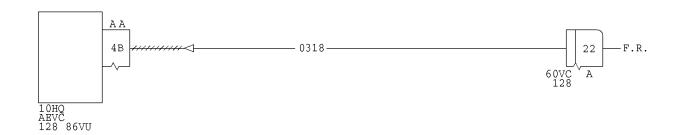
DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 17

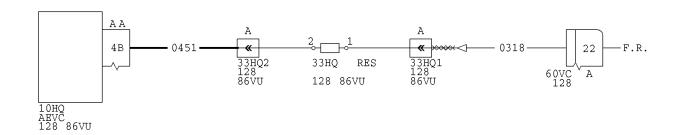
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BEFORE



AFTER



D//// OLD HOOK-UP

>>>> NEW HOOK-UP

---- NEW WIRE

F.R. FOR REFERENCE

NOTE : UNLESS OTHERWISE SPECIFIED PREFIX ALL WIRE IDENTIFICATION WITH ATA2126 UNLESS OTHERWISE SPECIFIED ALL WIRES ARE CF24 GAUGE UNLESS OTHERWISE SPECIFIED ALL ROUTES ARE 1M

Figure 3 Sheet 1 Modification of the Wiring in the Shelf 86VU

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 18

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SERVICE BULLETIN

Figure 4 Sheet 1 Hook-up Chart (Goto A3 section)

DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 19

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SERVICE BULLETIN REPORTING SHEET

TITLE: AIR CONDITIONING - AVIONICS EQUIPMENT VENTILATION - ADD RESISTOR ON AEVC

INLET SKIN AIR VALVE CONTROL LINE.

MODIFICATION No.: 30934P6961

Please	complete	the	appropriate	item	(A	٥r	R`)
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A - SB <u>WILL BE</u> embodied YES/NO (if NO please comment of YES, aircraft concerned (as per SB effectivity by default) and planned date	t) es
(month/year) of embodiment:	
	•
B - SB <u>HAS BEEN</u> embodied on aircraft:	
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REVISION No. : 01 - Mar 06/02 Page : 1 of 1

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5 DATE : Apr 03/01 SERVICE BULLETIN No. : A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 1 of 1

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SERVICE BULLETIN

L	END 1				L E A D				END2							
n e	Zone or Panel	Elec.Ident.	Term	Terminal P/N	Wire Ident.	Со	Rte	Gauge	Lenç mm	gth Inch	Zone or Panel	Elec.Ident.	Term	Terminal P/N	Inst	ructions
1 2 3 4 5 6 7 8 9 10	86VU 86VU 86VU	10HQ-AA 33HQ1 10HQ-AA		EN3155-016M2018	2126-0318 2126-0318 2126-0318		1M 1M	CF24 CF24 CF24	2500	100	128 128 86VU	60VC 60VC 33HQ2	22 22 A	EN3155-016M2018	M10 M1N	FIG3 (1) FIG3 FIG3
11 12 13 14 15 16 17 18 19 20 21 22 23 24 25																

(a) = BUNDLE D9000095214895

M1N = NEW HOOK-UP ON END 1

A = ADDED WIRE

(1) = TERMINAL SUPPLIED IN BUNDLE

M10 = OLD HOOK-UP ON END 1

D9000095214895

Figure 4 Sheet 1 Hook-up Chart

DATE: Apr 03/01 SERVICE BULLETIN No.: A320-21-1129

REVISION No. : 01 - Mar 06/02 Page : 19/20

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