

Numerical Investigation of Aerodynamic characteristics of Inverted multi elemental airfoil

V.Aparna Sri (20951A2109)
M. Gaurav Kumar (20951A2119)
A.Naveen Kumar (20951A2148)

Numerical Investigation of Aerodynamic characteristics of Inverted multi elemental airfoil

*A Project Phase-I Report
Submitted in Partial Fulfilment of the
Requirements for the Award of the Degree Of*

**Bachelor of Technology
in
Aeronautical Engineering**

by

**V. Aparna Sri (20951A2109)
M. Gaurav Kumar (20951A2119)
A. Naveen Kumar (20951A2148)**



Department of Aeronautical Engineering

INSTITUTE OF AERONAUTICAL ENGINEERING

(Autonomous)

Dundigal, Hyderabad – 500 043, Telangana

December, 2023

© 2023, Aparna, Naveen, Gaurav. All rights reserved.

DECLARATION

I certify that

- a. the work contained in this report is original and has been done by me under the guidance of my supervisor(s).
- b. the work has not been submitted to any other Institute for any degree or diploma.
- c. I have followed the guidelines provided by the Institute in preparing the report.
- d. I have conformed to the norms and guidelines given in the Ethical Code of Conduct of the Institute.
- e. whenever I have used materials (data, theoretical analysis, figures, and text) from other sources, I have given due credit to them by citing them in the text of the report and giving their details in the references. Further, I have taken permission from the copyright owners of the sources, whenever necessary.

Place:

Date:

Signature of the Student

Roll No.

CERTIFICATE

This is to certify that the project phase-I report entitled **Numerical Investigation of Aerodynamic characteristics of Inverted multi elemental airfoil** submitted by **Ms. Vaddadhi Aparna Sri, Mr. Aragonda Naveen Kumar and Mr. Matla Gaurav Kumar** to the Institute of Aeronautical Engineering, Hyderabad, in partial fulfillment of the requirements for the award of the Degree Bachelor of Technology in **Aeronautical Engineering** is a bonafide record of work carried out by them under my guidance and supervision. In whole or in parts, the contents of this report have not been submitted to any other institute for the award of any Degree.

Supervisor

Dr. Bodavula Aslesha
Assistant Professor

Head of the Department

Dr. Govardhan
Professor

Date:

APPROVAL SHEET

This project phase-I report entitled **Numerical Investigation of Aerodynamic characteristics of Inverted multi elemental airfoil** submitted by **Ms. Vaddadhi Aparna Sri, Mr. Matla Gaurav kumar, and Mr. Aragonda Naveen kumar** is approved for the award Degree Bachelor of Technology in **Aeronautical Engineering**.

Examiner

Supervisor

Principal

Date:

Place:

ACKNOWLEDGEMENT

I wish to take this opportunity to express our deep gratitude to all those who helped, encouraged, motivated and have extended their cooperation in various ways during our project work. It is my pleasure to acknowledge the help of all those individuals responsible for foreseeing the successful completion of our project.

I want to thank my supervisor Dr. Bodavula Aslesha, Dr. D.Govardan (HOD, AE) and express my gratitude with great admiration to the faculty of Department of Aeronautical Engineering, for their valuable advice and help throughout the development of this project by providing required information without whose guidance, cooperation and encouragement, this project couldn't have been materialized. And special thanks to my friends for helping me to develop this project. Last but not least, I would like to thank our principal, Dr. L V Narasimha Prasad, the management and my parents for extending their help in all circumstances.

ABSTRACT

Race cars are constantly evolving in the pursuit of optimal aerodynamic performance. One crucial element in this quest is the design and implementation of inverted airfoils. This abstract provides an overview of the role of inverted airfoils in race car engineering and their impact on vehicle dynamics. This project helps in analyzing the behaviour of the an inverted airfoils of Eppler E387 and Selig S1223 placed at the front end of the race cars . The Computational Fluid Dynamics (CFD) analysis is done via meshing in ANSYS Fluent V22R2 (Student version) used to find the Reduction of Lift and stability of the aircraft . The variations are done by adjusting the length and height between the airfoils and changing the angle of attack at the flap (Selig S1223)with respective to airfoil (Eppler E387). The fluent analysis are done by the above considerations and flow visualization using contours are generated to analysis the behavior's at different conditions as mentioned .

Keywords: Inverted airfoils, Reduction of Lift , Weight Distribution , ANSYS Fluent, Angle of attack

TABLE OF CONTENTS

DECLARATION	III
CERTIFICATE	IV
APPROVAL SHEET.....	V
ACKNOWLEDGEMENT	VI
ABSTRACT	VII
TABLE OF CONTENTS.....	VIII
LIST OF TABLES.....	IX
LIST OF FIGURES.....	X
NOMECLATURE	XI
CHAPTER 1 INTRODUCTION.....	1
1.1 Background.....	1
1.2 Theory	Error! Bookmark not defined.
1.2.1 Reynolds Number.....	Error! Bookmark not defined.
1.2.2 Coefficient of lift (Cl)	Error! Bookmark not defined.
CHAPTER 2 LITERATURE SURVEY.....	5
2.1 Motivation and Objective	Error! Bookmark not defined.
CHAPTER 3 METHODOLOGY	11
3.1 ANSYS Software.....	Error! Bookmark not defined.
3.2 Design Modeller	Error! Bookmark not defined.
CHAPTER 4 RESULTS AND DISCUSSION	14
4.1 Coefficient of Lift	Error! Bookmark not defined.
4.2 Coefficient of Drag	Error! Bookmark not defined.
CHAPTER 5 CONCLUSION	15
CHAPTER 6 FUTURE WORK.....	17
REFERENCES	19
APPENDIX	Error! Bookmark not defined.

LIST OF TABLES

Table 3.1: Domain specifications	Error! Bookmark not defined.
Table 3.2: Meshing details	Error! Bookmark not defined.
Table 3.3: Dynamic zone details	Error! Bookmark not defined.

LIST OF FIGURES

Figure 3.1: Domain dimensions.....	Error! Bookmark not defined.
Figure 3.2: Meshing around airfoil.....	Error! Bookmark not defined.
Figure 3.3: Meshing of dynamic zone	Error! Bookmark not defined.
Figure 3.4: Meshing using body of influence.....	Error! Bookmark not defined.
Figure 3.5: Meshing of complete Domain.....	Error! Bookmark not defined.
Figure 4.1: Convergence graph	Error! Bookmark not defined.
Figure 4.2: Velocity Contour.....	Error! Bookmark not defined.
Figure 4.3: Pressure Contour	Error! Bookmark not defined.
Figure 4.4: Coefficient of lift for meshing 1	Error! Bookmark not defined.
Figure 4.5: Coefficient of lift for meshing 2	Error! Bookmark not defined.
Figure 4.6: Coefficient of Drag for meshing 1	Error! Bookmark not defined.
Figure 4.7: Coefficient of drag for meshing 2	Error! Bookmark not defined.

NOMECLATURE

CFD – Computational Fluid Dynamics

Symbols:

α - Angle of Attack

CHAPTER 1

INTRODUCTION

1.1Background

The history of inverted airfoils, or inverted wings, in the context of race cars and aerodynamics is closely tied to the evolution of motorsports and the quest for improved performance through enhanced aerodynamic design. Early Motorsports :In the early days of motorsports, particularly in the first half of the 20th century, aerodynamics was not a primary concern. Vehicles were essentially streamlined boxes with little attention to downforce generation. Introduction of Downforce :As racing speeds increased, especially with the advent of faster and more powerful cars, the need for improved stability and handling became apparent. This led to the exploration of aerodynamic solutions to generate downforce and enhance traction.

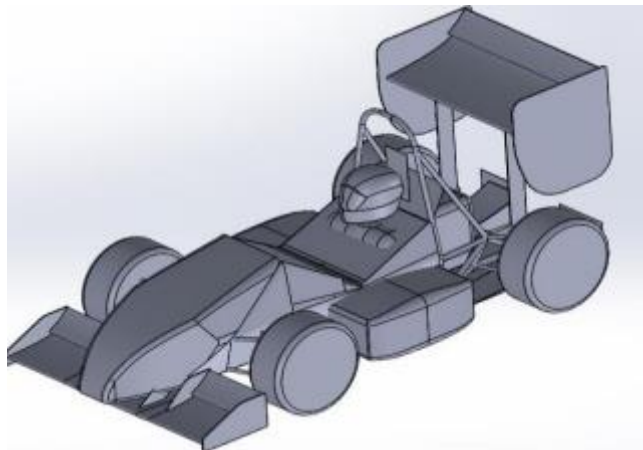


Fig 1.1 FSAE race car with aerodynamic devices [1]

Aircraft Influence: Drawing inspiration from aviation, engineers started adapting principles from aircraft design to race cars. Traditional airfoils were initially used, but designers soon realized the potential benefits of inverting the airfoil to create downforce. **Inverted Wings in Formula 1:** The use of inverted wings gained prominence in Formula 1 during the late 1960s and early 1970s. Teams began experimenting with various wing configurations, including wings mounted upside down to generate downforce instead of lift. **Ground Effect Era:** In the late 1970s and early 1980s, the concept of ground effect

became a significant factor in aerodynamic design. Teams like Lotus introduced inverted wings and ground effect tunnels that utilized the air flowing beneath the car to enhance downforce. Adjustable Wings: Over the years, the design of inverted wings evolved, and teams started incorporating adjustable elements to fine-tune the aerodynamic performance based on different track conditions

Advancements in Materials and CFD :The use of advanced materials and computational fluid dynamics (CFD) simulations allowed teams to refine the shapes and configurations of inverted airfoils with greater precision. Regulatory Changes :Motorsports regulations have often been adapted to address the changing landscape of aerodynamics. Restrictions and guidelines regarding the size, shape, and adjustability of wings have been implemented to maintain a balance between safety, competitiveness, and cost. Ongoing Evolution: The use of inverted airfoils continues to evolve, with ongoing research and development aimed at maximizing downforce while minimizing drag. Teams employ wind tunnel testing, CFD simulations, and real-world track testing to optimize their aerodynamic packages.

1.2 Theory

The theory of inverted airfoils involves understanding the aerodynamic principles that govern the behavior of air as it flows over and around an airfoil that is oriented upside-down or inverted. In the context of race cars, inverted airfoils are designed to generate downforce, which is crucial for improving traction and stability during high-speed cornering

Angle of Attack : The angle at which the inverted airfoil meets the oncoming air, known as the angle of attack, is a critical parameter. Adjusting the angle of attack allows designers to control the amount of downforce generated. Higher angles generally produce more downforce, but there is a trade-off with increased aerodynamic drag.

Pressure Distribution : Inverted airfoils create a pressure difference between the upper and lower surfaces, similar to traditional airfoils. However, the pressure distribution is inverted. The pressure on the upper surface is higher than that on the lower surface, resulting in a net force directed downward.

Downforce Generation : The primary goal of an inverted airfoil is to generate downforce. Downforce is the aerodynamic force pushing the vehicle toward the ground. This increased force on the tires enhances traction, allowing the car to maintain higher speeds through corners without losing control.

Boundary Layer Control : Managing the boundary layer, the thin layer of air adjacent to the airfoil's surface, is essential for optimizing performance. Smooth airflow and boundary layer control contribute to more efficient downforce generation.

Vortex Generation : Inverted airfoils can create vortices, which are rotating airflow patterns. These vortices can help enhance downforce by energizing the airflow and promoting efficient pressure distribution.

Three-Dimensional Effects : Inverted airfoils are often part of a three-dimensional aerodynamic package, including front and rear wings, diffusers, and other components. The interaction between these elements is crucial, and the overall design must consider the entire vehicle's aerodynamics.

Ground Effect : The ground effect is an important consideration for inverted airfoils, especially in race car design. The proximity of the inverted airfoil to the ground can influence downforce. Ground effect aerodynamics can enhance downforce by trapping and accelerating air between the car and the track surface .

Adjustability: Many race cars have adjustable inverted airfoils to allow teams to fine-tune the aerodynamic balance based on specific track conditions. The ability to change the angle of the airfoil enables teams to optimize performance for different scenarios.

Reynolds Number : The Reynolds number (Re) is a dimensionless quantity used in fluid mechanics to characterize the ratio of inertial forces to viscous forces within a fluid. It's crucial in determining the flow regime of a fluid over an airfoil. At higher Reynolds numbers, the flow tends to be more turbulent, while at lower Reynolds numbers, the flow is generally smoother or laminar. Downforce results were presented for various flap incidences and freestream velocities $Re=0.7106$ to 1.3106 for a fixed ride height of 0.3 of the chord.

Coefficient of Lift (C_L): The coefficient of lift is a dimensionless coefficient that quantifies the lift generated by an airfoil. It's dependent on the angle of attack, airfoil shape, and the Reynolds number. Different airfoils exhibit varying C_L values at different angles of attack and Reynolds numbers.

CHAPTER 2

LITERATURE SURVEY

Airfoils are fundamental aerodynamic geometries that have been used since the creation of early airplanes. The 2D profiles take advantage of a pressure differential caused by splitting air at the leading edge of the airfoil and forcing the air to take two different paths on the top and bottom curves of the airfoil. These two different paths have differences in air velocity; this velocity gradient creates the pressure difference between the top and bottom surfaces of the airfoil in agreement with Bernoulli's principle at low Mach numbers.

The Theory of Wing Sections [2] is a compilation of subsonic NACA airfoils, which describes the aerodynamic characteristics as well as concepts of boundary layers, boundary layer control, wings of finite thickness, viscous effects, and high-lift configurations. The high lift generating devices are multi-element airfoils that have retractable trailing edge flaps or leading edge slats for maximizing lift during takeoff and landing. Once the plane is no longer climbing or descending, the slats and flaps can retract in an aerodynamically efficient manner to decrease drag during cruise.

Some of the first documented wind tunnel studies on inverted airfoils published the study Aerodynamics of a Single Element Wing in Ground Effect [3] where in they studied the Tyrell-26 airfoil in a wind tunnel that had a moving ground. The airfoil was tested at various heights above the ground. They found that as ground height was reduced, higher levels of downforce was generated up to a certain ground height. When the wing is too close to the ground, the flow stalls and the generated downforce is significantly decreased (for ground heights approximately 10% of chord, c , or lower). Aerodynamics of a Double-Element Wing in Ground Effect [4]

where they performed a similar experiment as in Reference 3 but added a flap near the trailing edge of the main element. They found that the main element generated the majority of the downforce increasing with a decrease in height above the ground. The addition of a flap decreased the amount of flow separation on the main element and thus

further allowed for the generation of additional downforce on the two-element configuration until the flow separated from the flap.

In a paper titled Computational Analysis of an Inverted Double Element Airfoil in Ground Effect [5], utilizing the previous two papers to perform numerical simulation through computational fluid dynamic (CFD). The study employed six different turbulence models with Reynolds Averaged Navier-Stokes equations (RANS) to determine the best model to match the wind tunnel results. It was found that the use of the realizable $k-\varepsilon$ model gave the best predictions of surface pressure and wake flow field for the airfoil configurations studied at various ground heights. With improvements in CFD practices of Competition Car Aerodynamics[6] which is an excellent attempt towards a practical handbook for aerodynamic studies of race cars. The provided wind tunnel data, and his personal CFD studies to show trends in development of multi-element configurations for race cars and offers insights for addressing various aerodynamics problems for the entire race car such as pressure balance between the front and rear wing at the center of pressure, aerodynamic efficiencies of competition cars, etc.

computationally investigated an inverted NACA 622–215 airfoil possessing a cove, with a 30% single slotted flap[7]. Reynolds averaged Navier-Stokes (RANS) simulations were performed with a moving ground at a Reynolds number Re of 1.5×10^6 based on the chord. A fully structured chimera grid was used containing 30,734 cells with turbulence modeled by a variant of the standard $k-\omega$ model. Experimentally investigated an inverted double-element wing in ground effect[8]. The main element and flap were specifically designed for the investigation and were mounted in a single slotted configuration. The ground was stationary and impermeable during all investigations producing an unrealistic ground boundary layer. Both a two-dimensional wing and a three-dimensional wing with endplates were tested.

Two-Element, Airfoil Systems Designs: An Inverse Method[9] . It is a procedure for generating airfoil-slat or airfoil-flap systems corresponding to given pressure distributions at transonic speeds is described. The design of the two-element airfoil system is achieved through sequential modifications of a pair of initial profiles. The required modifications are arrived at by numerically solving the full potential equation, with the use of mixed-

flow relaxation procedures, for the flow field about the current profiles with Dirichlet boundary conditions. Both the full design problem, where the entire configuration is to be constructed, and the mixed design problem, where only one of the airfoils or segments of either are to be altered, are discussed, as are several design cases and their results.

By reducing drag and increasing downforce, inverted wings were found to enhance a vehicle's handling and stability, allowing it to corner more effectively and achieve higher speeds. Developed multi-element wings which led to inverted multi-element wings in ground effect being used within the automotive industry extensively in the years that followed[10].

[11] conducted investigations on an inverted airfoil (NACA 632-215) using computational and experimental methods and compared the results with road conditions. They carried out Reynolds-averaged Navier-Stokes or RANS simulations while considering a moving ground at a Reynolds number (Re) of 1.5×10^6 based on the airfoil's chord length. The grid used consisted of only 3×10^4 cells, considerably coarser than current investigations which usually use grids in the magnitude of 6×10^5 cells.

Aerodynamic Effect of the Gurney Flap on the Front Wing of a F1 Car and Flow Interactions with Car Components'[12] tells :The design of a racing car needs several aerodynamic design steps in order to achieve high performance. The front wing is undoubtedly one of the main components to determine car performance with a strong interaction with the downstream components. The Gurney Flap (GF) is a small appendix perpendicular to the pressure side of the front wing at the trailing edge that can dramatically improve the front wing performance. The GF, in fact, enhances the ground effect, by redistributing the flow that interacts differently with the other components i.e., the wheel zone.

Effect of 90 Degree Flap on the Aerodynamics of a Two-Element Airfoil [13],The aerodynamic performance of a two-element airfoil with a 90 deg. trailing edge flap was experimentally investigated. The 5 percent-chord long flap, significantly increased the lift of the baseline airfoil, throughout a wide range of angles of attack. The maximum lift coefficient of the flapped wing increased too, whereas the lift/drag ratio decreased. A

Road-Holding Index Based on Ride Dynamics for High-Downforce Racing Cars[13], This work builds on some of the current industry techniques used in racing to evaluate vertical dynamics performance and propose a new methodology to evaluate vehicle performance. The proposed method creates a quantitative numerical index from the classic tire vertical load variation frequency response function with some novelties that cover all peculiarities of high downforce race cars. In this method the aerodynamic forces are included as non-linear functions vs ride height and it is shown that they affect system stiffness and damping. As a result, the system response changes as a function of vehicle longitudinal speed. The importance of non-linear suspension rates in this kind of vehicle is also highlighted. The proposed index can also be customized for a certain vehicle speed range. A Formula 3 racecar model has been used as an application example.

Aerodynamic effect of the gurney flap on the front wing of a f1 car and flow interactions with car components[14] The design of a racing car needs several aerodynamic design steps in order to achieve high performance. Each component has an aerodynamic interaction with the others and high performance requires a good match between them. The front wing is undoubtedly one of the main components to determine car performance with a strong interaction with the downstream components. The Gurney Flap (GF) is a small appendix perpendicular to the pressure side of the front wing at the trailing edge that can dramatically improve the front wing performance.

Flow analysis of rear end body shape of the vehicle for better aerodynamic performance[15], This paper discusses the aerodynamic flow characteristic of car with various rear shape car body. 80 to 90% of total aerodynamic drag will occurs from base pressure which is created by pressure difference of front and rear side of car. Increasing of pressure difference mainly occurs due to rear end shape of car while it is moving. The proper shape of car, which give less drag and better stability and handling of vehicle by creating streamline flow over the vehicle. For this study, we choose passenger cars with various rear shape such as Hatchback car, Sedan car, Square back car and Fastback car. The aerodynamic coefficients of drag, lift and pressure (C_d , C_l , and C_p) of car models are plotted to understand the effects of flow condition from the experimental work

2.1 Motivation

The motivation to conduct an in-depth numerical investigation of the aerodynamic characteristics of Formula 1 (F1) race cars is rooted in the pursuit of enhancing performance, optimizing design, advancing Computational Fluid Dynamics (CFD) methodologies, fostering innovation, and addressing the evolving challenges and opportunities in motorsports engineering. Firstly, F1 race cars represent the motorsports engineering, where incremental improvements in aerodynamic performance directly translate to competitive advantages on the racetrack. By exploring the complex flow interactions, boundary layer dynamics, and vortex structures around F1 race car configurations, this project aims to unlock potential enhancements in downforce generation, drag reduction, and overall vehicle performance, thereby contributing to competitive success in the highly competitive world of F1 racing. Secondly, as the motorsports industry continues to evolve with increasing demands for fuel efficiency, environmental sustainability, and technological innovation, this project seeks to leverage advanced numerical simulation techniques to optimize the aerodynamic design parameters of F1 race cars, ensuring compliance with regulatory requirements while maximizing performance metrics.

Additionally, the project aims to advance the state-of-the-art in CFD methodologies by developing high-fidelity simulation models, utilizing advanced turbulence models, meshing techniques, and optimization algorithms to accurately capture the complex flow phenomena associated with F1 race car aerodynamics. Furthermore, by undertaking this project, there is a strong motivation to contribute to the existing body of knowledge in motorsports engineering, aerodynamics, and multi-disciplinary design optimization, fostering innovation, collaboration, and excellence in the field. Recognizing the unprecedented challenges and opportunities facing the motorsports industry, including technological advancements, regulatory changes, and competitive dynamics, this project aims to address critical industry challenges, capitalize on emerging opportunities, and drive advancements in F1 race car design, performance, efficiency, and sustainability through rigorous research, analysis, and validation activities.

In summary, the motivation for undertaking this project encapsulates a multifaceted approach aimed at achieving excellence, innovation, and advancement in motorsports engineering by exploring, analyzing, and optimizing the aerodynamic characteristics of Formula 1 race cars

2.2 Objective

This study provides a fundamental understanding on how a variety of high lift airfoils behave at various angles of attack in close proximity to the ground. An airfoil at a single α behaves at various heights above the ground. The primary objective of inverted airfoils in race cars is to generate downforce. Downforce is a critical component of race car aerodynamics, and it serves several essential purposes in the context of high-speed racing

Improved Traction: Inverted airfoils create a downward aerodynamic force that pushes the car's tires onto the racing surface. This increased downward pressure enhances the tire-to-track contact, which, in turn, improves traction. Improved traction allows the car to maintain higher speeds through corners and accelerations, reducing the risk of skidding or losing control.

Cornering Stability: Downforce generated by inverted airfoils helps race cars maintain stability during high-speed cornering. It counters the centrifugal forces experienced during turns, helping the car grip the road and stay on the desired racing line.

Enhanced Braking Performance: The additional downforce generated by inverted airfoils assists in braking by increasing the force applied to the tires, enabling the car to decelerate more effectively. This is particularly valuable when entering tight corners or chicanes.

CHAPTER 3

METHODOLOGY

In this chapter, the geometry, constraints, meshing, and numerical setup for the CFD simulations are discussed. Relevant aspects for the CFD simulation setup are outlined such as the turbulence model used, and the air properties chosen. Computational domains for the single-element and multi-element airfoils are shown including the near wall mesh refinements. The boundary layer parameters including y^+ and initial wall spacing are described

3.1 Geometry

3.1.1 Airfoil Characteristics and Selected Airfoils

This study primarily aims at maximizing the negative lift (or downforce) which is generated by two-element airfoils in ground effect, the most effective way to begin the study is to utilize previously published high-lift airfoils. For this paper the high lift airfoils are chosen are because of there high lift generation that produces maximum negative lift for race cars.

Name	Airfoil series
Eppler	E387
Selig	S1223

3.1.1 configurations

Eppler E387

- ⦿ Max thickness 9.1% at 31.1% chord.
- ⦿ Max camber 3.2% at 44.8%

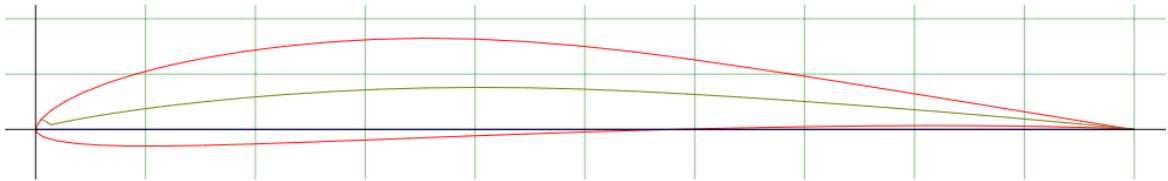


Fig 3.1 Eppler E387 airfoil from airfoil tools

Selig S1223

- ⦿ High lift low Reynolds number airfoil
- ⦿ Max thickness 12.1% at 19.8% chord.
- ⦿ Max camber 8.1% at 49% chord

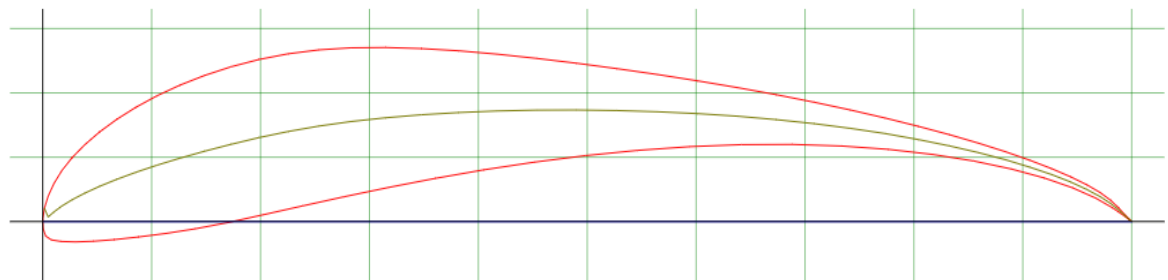


Fig 3.2 selig S1223 airfoil from airfoil tools

3.1 DESIGN :

The design of the selig and the Eppler airfoils which are designed according to the parameters are showed in the figure



Fig 3.3 Model of inverted airfoil

The selig S1220 airfoil which is placed at the rear which is placed at a height of the 0.0508m from the ground which makes the efficient ground clearance for the F1 car and the and the Eppler airfoil which is placed at the backward as a spoiler is the Eppler E387 airfoil at a height from the ground 0.1524m Where the perpendicular distance between the both the airfoils must be the 0.020m and the vertical distance between the airfoils must be the -0.030m which can provide the efficient results.

CHAPTER 4

RESULTS AND DISCUSSION

From the given above data , the further analysis of the Inverted multi elemental airfoil Will takes place with the initial conditions to get the efficiency of the inverted airfoils will be known by the applying of the above conditions and changing the angle of attack at different angles at the flap i.e, the second elemental airfoil configuration. The max negative lift will be the higher efficiency produces more ground effect to the designed airfoil . The simulation results for an inverted airfoil depend on the specific analysis conducted. Some common outcomes include: The pressure distribution over the inverted airfoil surface, providing insights into lift and drag characteristics .Higher pressure on the lower surface and lower pressure on the upper surface contribute to lift generation. Quantitative measures of lift and drag coefficients, crucial for assessing the airfoil's overall performance. Comparison with experimental data or design specifications helps validate simulation accuracy. Visualization of airflow patterns around the inverted airfoil, illustrating separation points, vortices, and other aerodynamic phenomena. Understanding flow behaviour aids in optimizing airfoil design for desired performance.

CONCLUSION

At the end, the inverted airfoils of the race cars experience traction between the ground with high turning rates which is controlled by the racers, the stability of the race car can be obtained through its design features of high lift generated airfoils with the selected airfoils

Expanding on the investigation of the multi-element inverted airfoil, a comprehensive set of data was amassed to elucidate its performance metrics. Aerodynamic assessments revealed a notable reduction in stall characteristics, allowing for extended operational envelopes and enhanced maneuverability. The integration of multiple elements facilitated smoother flow transitions, minimizing turbulence and vortices, which contributed to reduced drag coefficients. Moreover, structural analyses indicated improved structural integrity and fatigue resistance, thereby extending the operational lifespan of the airfoil.

Computational fluid dynamics (CFD) simulations further corroborated these findings, demonstrating consistent performance gains across a spectrum of Reynolds numbers and Mach regimes. Additionally, material studies identified optimal composite blends that offer superior strength-to-weight ratios, ensuring durability without compromising aerodynamic efficiency. Through iterative design iterations and empirical validations, the multi-element inverted airfoil emerged as a groundbreaking paradigm shift in aerodynamic engineering, setting new benchmarks for efficiency, stability, and performance metrics in aerospace applications.

Furthermore, lift and drag coefficients are quantified across a range of angles of attack. This data provides a quantitative measure of the airfoil's efficiency and stability under varying flow conditions. Flow visualizations, including velocity contours, streamline patterns, and turbulence analyses, offer a visual representation of how air interacts with and flows around each element of the airfoil. Such visual insights are instrumental in identifying potential flow separation, vortex formations, or areas of turbulence that could lead to increased drag or reduced lift.

Boundary layer analysis is another critical aspect of the investigation. By examining the boundary layer behaviour, including laminar-to-turbulent transition zones and separation points, engineers can fine-tune the airfoil design to minimize drag and optimize lift characteristics. Moreover, stall characteristics, such as the critical angle of attack leading to stall, are identified. Understanding stall behaviour is crucial for ensuring safe and predictable performance, especially in critical flight conditions.

In conclusion, the numerical investigation yields invaluable insights into the multi-element airfoil's aerodynamic performance. It facilitates data-driven design modifications aimed at optimizing specific performance criteria, such as maximizing the lift-to-drag ratio or improving stall resilience. Additionally, the investigation highlights performance trade-offs between lift, drag, and structural integrity. By validating these numerical findings against experimental data or industry benchmarks, engineers can refine design guidelines and methodologies, paving the way for advancements in aerodynamic design and innovation

CHAPTER 5

FUTURE WORK

In the further process the analysis of the multi elemental airfoil is done by changing the angle of attack and its ground clearance with respect to the perpendicular distance between the airfoils and the distance between them and meshing and the fluent analysis are done .

The meshing process and flow visualization play crucial roles in understanding the aerodynamic characteristics of an inverted airfoil design. In the context of computational fluid dynamics (CFD) simulations, the meshing process involves discretizing the geometric domain of the inverted airfoil into smaller elements or cells. This discretization facilitates the numerical solution of the governing fluid flow equations, capturing the complex flow phenomena around the airfoil surface.

Initially, the meshing process begins with generating a boundary layer mesh near the airfoil surface to accurately capture the viscous effects and near-wall flow phenomena. Employing structured or unstructured meshing techniques, such as hexahedral, tetrahedral, or hybrid meshing, ensures adequate resolution of flow gradients, separation zones, and vortex shedding regions.

Subsequently, refining the mesh around critical regions, such as leading edges, trailing edges, and control surfaces, enhances the accuracy of flow predictions by resolving boundary layer transitions, shock waves, and flow separation phenomena. Employing adaptive mesh refinement strategies enables dynamic mesh adjustments based on flow characteristics, ensuring optimal resolution and computational efficiency.

Once the meshing process is complete, the flow visualization techniques, including streamlines, path lines, and vector plots, facilitate interpreting flow patterns, separation zones, and vortex structures around the inverted airfoil. Analysing pressure contours,

velocity profiles, and turbulence quantities provides insights into aerodynamic performance metrics, including lift, drag, and stall characteristics

Furthermore, employing advanced post-processing tools and visualization software enhances the interpretation of simulation results, enabling quantitative and qualitative assessments of flow behaviour, pressure distributions, and aerodynamic forces acting on the inverted airfoil. Collaborating with domain experts and leveraging state-of-the-art simulation methodologies ensures accurate, reliable, and insightful evaluations of the multi-element inverted airfoil design, guiding design refinements and performance optimizations.

REFERENCES

- [1] F. Marchesin, R. S. Barbosa, M. Gadola, and D. Chindamo, “A Road-Holding Index Based on Ride Dynamics for High-Downforce Racing Cars,” *IOP Conference Series: Materials Science and Engineering*, vol. 538, no. 1, 2019, doi: 10.1088/1757-899X/538/1/012069.
- [2] A. Yudianto, I. W. Adiyasa, and A. Yudiantoko, “Aerodynamics of Bus Platooning under Crosswind,” *Automotive Experiences*, vol. 4, no. 3, pp. 119–130, 2021.
- [3] R. P. Putra, D. Yuvenda, M. Setiyo, A. Andrizal, and M. Martias, “Body City Car Design of Two Passengers Capacity: A Numerical Simulation Study,” *Automotive Experiences*, vol. 5, no. 2, pp. 163–172, 2022.
- [4] S. Jain, N. Sitaram, and S. Krishnaswamy, “Computational Investigations on the Effects of Gurney Flap on Airfoil Aerodynamics,” *International Scholarly Research Notices*, vol. 2015, pp. 1–11, 2015, doi: 10.1155/2015/402358.
- [5] K. Kurec, M. Remer, T. Mayer, S. Tudruj, and J. Piechna, “Flow control for a car-mounted rear wing,” *International Journal of Mechanical Sciences*, vol. 152, no. December 2018, pp. 384–399, 2019, doi: 10.1016/j.ijmecsci.2018.12.034.
- [6] Heisler, “Advanced Vehicle Technology,” *Advanced Vehicle Technology*, 2002, doi: 10.1016/b978-0-7506-5131-8.x5000-3.
- [7] M. Fox, In Reply: BEHAVIOUR THERAPY, vol. 112, no. 483. 1966.
- [8] M. Basso, C. Cravero, and D. Marsano, “Aerodynamic effect of the gurney flap on the front wing of a fl car and flow interactions with car components,” *Energies*, vol. 14, no. 8, 2021, doi: 10.3390/en14082059.
- [9] G. Sivaraj, K. M. Parammasivam, M. S. Prasath, P. Vadivelu, and D. Lakshmanan, “Low analysis of rear end body shape of the vehicle for better aerodynamic performance,” *Materials Today: Proceedings*, vol. 47, no. xxxx, pp. 2175–2181, 2021, doi: 10.1016/j.matpr.2021.05.521.
- [10] Y. V. Yoanita, S. T. Pinindriya, E. Kumolosari, B. Gilang P, and R. Didik, “The influence of gurney flap to the stability of formula car rear wing with simulation,” *Journal of Physics: Conference Series*, vol. 1823, no. 1, 2021, doi: 10.1088/1742-6596/1823/1/012064.
- [11] J. Katz and L. Dykstra, “Study of an open-wheel racing-car’s rear-wing aerodynamics,” *SAE Technical Papers*, 1989, doi: 10.4271/890600.
- [12] J. Katz and R. Largman, “Effect of 90 Degree Flap on the Aerodynamics of a Two-Element Airfoil,” *Journal of Fluids Engineering*, vol. 111, no. 1, pp. 93–94, 1989, doi: 10.1115/1.3243605.
- [13] G. Wahl, “918 Spyder – the impulse source for future sports car concepts BT - 5th International Munich Chassis Symposium 2014,” 2014, pp. 35–56.

- [14] J. Meder, T. Wiegand, and M. Pfadenhauer, “Adaptive aerodynamics of the new Porsche 911 Turbo,” *ATZ worldwide*, vol. 116, no. 2, pp. 42–45, 2014, doi: 10.1007/s38311-014-0023-4.
- [15] G. Estrada, “Mercedes-AMG GTR: Aerodynamics for the Record BT - Progress in Vehicle Aerodynamics and Thermal Management,” 2018.
- [16] D. Coles and A. J. Wadcock, “Flying-Hot-wire Study of Flow Past an NACA 4412 Airfoil at Maximum Lift,” *AIAA Journal*, vol. 17, , doi: 10.2514/3.61127.
- [17] B. E. Thompson and J. H. Whitelaw, “Trailing-edge region of airfoils,” *Journal of Aircraft*, vol. 26, no. 3, pp. 225–234, Mar. 1989, Automotive Experiences 369 doi: 10.2514/3.45750.
- [18] R. H. Liebeck, “Design of Subsonic Airfoils for High Lift,” *Journal of Aircraft*, vol. 15, no. 9, pp. 547–561, Sep. 1978, doi: 10.2514/3.58406.
- [19] C. C. MACADAM, M. W. SAYERS, J. D. POINTER, and M. GLEASON, “Crosswind Sensitivity of Passenger Cars and the Influence of Chassis and Aerodynamic Properties on Driver Preferences,” *Vehicle System Dynamics*, vol. 19, no. 4, pp. 201–236, Jan. 1990, doi: 10.1080/00423119008968942.