Introduction to Turbulence Modeling or Why Turbulence Modeling is Black Magic

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Reynolds Averaged Navier Stokes

Let $\phi = \bar{\phi} + \phi'$, where

$$ar{\phi} = \lim_{T o \infty} \int_{t_0}^{t_0 + T} \phi(t) dt$$

Interested in solving for \bar{u}_i . Nondimensionalizing such that $\rho = 1$,

Continuity Equation

$$\frac{\partial u_i}{\partial x_i} = 0$$

$$\Downarrow$$

$$\frac{\partial \bar{u}_i}{\partial x_i} = 0$$

Momentum Equation

$$\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{1}{Re} \frac{\partial^2 u_i}{\partial x_j \partial x_i}$$

$$\Downarrow$$

$$\frac{\partial \bar{u}}{\partial t} + \bar{u}_j \frac{\partial \bar{u}_i}{\partial x_j} = -\frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} \left[-\overline{u_i' u_j'} + \frac{1}{Re} \frac{\partial \bar{u}_i}{\partial x_j} \right]$$

But the Reynolds stress, $\overline{u'_i u'_i}$ is an unclosed quantity.

Modeling the Reynolds Stress

$$\frac{\partial \bar{u}}{\partial t} + \bar{u}_j \frac{\partial \bar{u}_i}{\partial x_j} = -\frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} \left[-\overline{u_i' u_j'} + \frac{1}{\textit{Re}} \left(\frac{\partial \bar{u}_i}{\partial x_j} + \frac{\partial \bar{u}_j}{\partial x_i} \right) \right]$$

Eddy Viscosity Hypothesis

$$\overline{u_i'u_j'} = -\nu_T \left(\frac{\partial \overline{u}_i}{\partial x_i} + \frac{\partial \overline{u}_j}{\partial x_i} \right) + \frac{2k}{3} \delta_{ij}$$

where $k = \frac{1}{2} \overline{u_i' u_i'}$.

This assumption is fundamentally flawed, but useful.

We still need a model for ν_T and k.

Spalart-Allmaras Model

- Model the eddy viscosity
- Assume *k* is negligible
- Surprisingly decent for the types of flows it was designed for
 - External aerodynamics with minimal flow separation
- · Highly nonlinear equation

$$\begin{split} \frac{\partial \tilde{\nu}}{\partial t} + \bar{u}_{j} \frac{\partial \tilde{\nu}}{\partial x_{j}} &= C_{b1} [1 - f_{t2}] \tilde{S} \tilde{\nu} + \frac{1}{\sigma} \left\{ \nabla \cdot [(\nu + \tilde{\nu}) \nabla \tilde{\nu}] + C_{b2} |\nabla \tilde{\nu}|^{2} \right\} \\ &- \left[C_{w1} f_{w} - \frac{C_{b1}}{\kappa^{2}} f_{t2} \right] \left(\frac{\tilde{\nu}}{d} \right)^{2} + f_{t1} \Delta U^{2} \\ \nu_{T} &= \tilde{\nu} f_{v1}, \quad f_{v1} = \frac{\chi^{3}}{\chi^{3} + v_{v1}^{3}}, \quad \chi = \frac{\tilde{\nu}}{\nu} \quad \cdots \end{split}$$

$k - \epsilon$ Exact Equations

Turbulence Kinetic Energy Equation

$$\frac{\partial k}{\partial t} + \bar{u}_i \frac{\partial k}{\partial x_i} = -\overline{u_i' u_j'} \frac{\partial \bar{u}_i}{\partial x_j} - \frac{\partial}{\partial x_i} \left(\frac{1}{2} \overline{u_j' u_j' u_i'} + \overline{p' u_i'} \right) + \nu \frac{\partial^2 k}{\partial x_i \partial x_i} - \epsilon$$

Dissipation Equation

$$\begin{split} \frac{\partial \epsilon}{\partial t} + \bar{u}_{i} \frac{\partial \epsilon}{\partial x_{i}} &= \nu \frac{\partial^{2} \epsilon}{\partial x_{i} \partial x_{i}} + P_{\epsilon} + D_{\epsilon} - \Phi_{\epsilon} \\ P_{\epsilon} &= -2\nu \left[\overline{\frac{\partial u_{i}'}{\partial x_{j}} \frac{\partial u_{k}'}{\partial x_{i}} \frac{\partial \bar{u}_{i}}{\partial x_{k}} + \overline{\frac{\partial u_{j}'}{\partial x_{k}} \frac{\partial u_{j}'}{\partial x_{k}} \frac{\partial \bar{u}_{i}}{\partial x_{k}} + \overline{u_{k}' \frac{\partial u_{i}'}{\partial x_{j}} \frac{\partial^{2} \bar{u}_{i}}{\partial x_{k} \partial x_{j}} + \overline{\frac{\partial u_{i}'}{\partial x_{k}} \frac{\partial u_{k}'}{\partial x_{m}} \frac{\partial u_{k}'}{\partial x_{m}} \right] \\ D_{\epsilon} &= -\frac{\partial}{\partial x_{k}} \left(\overline{u_{k}' \epsilon'} + 2\nu \overline{\frac{\partial p'}{\partial x_{m}} \frac{\partial u_{k}'}{\partial x_{m}} \right) \\ \Phi_{\epsilon} &= 2\nu^{2} \overline{\frac{\partial^{2} u_{i}'}{\partial x_{k} \partial x_{m}} \frac{\partial^{2} u_{i}'}{\partial x_{k} \partial x_{m}} \frac{\partial^{2} u_{i}'}{\partial x_{k} \partial x_{m}} \end{split}$$

$k - \epsilon$ Approximations

Turbulence Kinetic Energy Equation

$$\frac{1}{2}\overline{u'_{j}u'_{j}u'_{i}} + \overline{p'u'_{i}} \approx -\frac{\nu_{T}}{\sigma_{k}}\frac{\partial k}{\partial x_{i}}$$

Dissipation Equation

From dimensional analysis

$$u_T pprox C_\mu rac{k^2}{\epsilon}$$
 $P_\epsilon pprox - C_{\epsilon 1} rac{\epsilon}{k} rac{u_i' u_j'}{\partial x_j} rac{\partial ar{u}_i}{\partial x_j}$
 $\Phi_\epsilon pprox C_{\epsilon 2} rac{\epsilon^2}{k}$

From gradient transport model

$$D_{\epsilon} \approx \frac{\partial}{\partial x_{i}} \left(\frac{\nu_{T}}{\sigma_{\epsilon}} \frac{\partial \epsilon}{\partial x_{i}} \right)$$

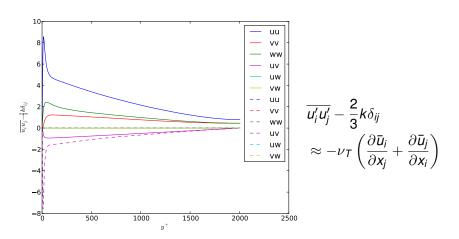


Figure: Reynolds Stress Components

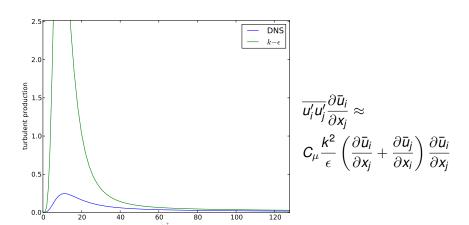


Figure: Production of Turbulent Kinetic Energy

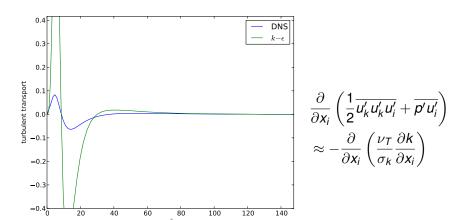


Figure: Transport of Turbulent Kinetic Energy

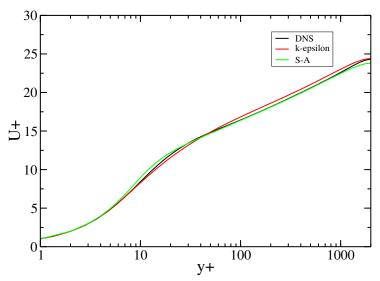


Figure: Time averaged velocity profile Re=2000

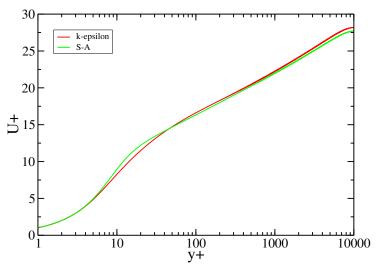


Figure: Time averaged velocity profile *Re* = 10000

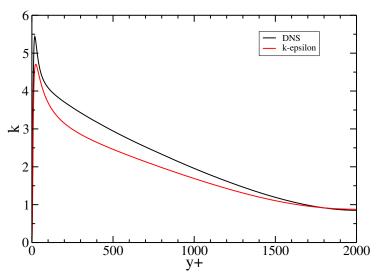


Figure: Turbulence kinetic energy profile Re=2000

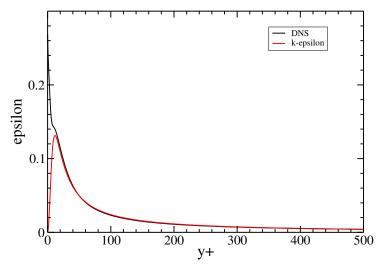


Figure: Dissipation profile *Re* = 2000

Notes on $k - \epsilon$

Fixes near the wall

- Wall functions
- Two-layer models
- SST $k \omega$
- $\overline{v^2} f$ four equation model (f is an elliptic relaxation function)

Boundary conditions

- BC on ϵ is not obvious
- Enforce both k = 0 and $\frac{\partial k}{\partial n} = 0$ at walls.

Popularity

- One of the earliest models
- Physically motivated derivation
- Insensitive to freestream conditions on k and ϵ

Typical Solution Procedure

From a presentation by a Fluent engineer

- Start calculations by performing 100 iterations or so with standard $k-\epsilon$ model and first order upwind differencing. For very simple flows (no swirl or separation) converge with second order upwind and $k-\epsilon$ model.
- If the flow involves jets, separation, or moderate swirl, converge solution with the realizable $k-\epsilon$ model and second order differencing.
- If the flow is dominated by swirl converge solution deeply using RSM and a second order differencing scheme. If the solution will not converge, use first order differencing instead.

Wilcox (1993) $k - \omega$ Model

Turbulence Kinetic Energy Equation

$$\frac{\partial k}{\partial t} + \bar{u}_{j} \frac{\partial k}{\partial x_{j}} = 2\nu_{T} |S|^{2} - C_{\mu} k\omega + \frac{\partial}{\partial x_{j}} \left(\left(\nu + \frac{\nu_{T}}{\sigma_{k}} \right) \frac{\partial k}{\partial x_{j}} \right)$$

Specific Dissipation Equation

$$\frac{\partial \omega}{\partial t} + \bar{u}_j \frac{\partial \omega}{\partial x_j} = 2C_{\omega 1} |S|^2 - C_{\omega 2} \omega^2 + \frac{\partial}{\partial x_j} \left(\left(\nu + \frac{\nu_T}{\sigma_\omega} \right) \frac{\partial \omega}{\partial x_j} \right)$$

$$\omega \equiv \frac{\epsilon}{C_\mu k}$$

$$\nu_T = \frac{k}{\omega}$$

Notes on $k-\omega$

Wall Treatment

- ullet Formally, ω is singular at a perfectly smooth wall
- · Numerically, this is fixed by assuming a finite roughness

•
$$\omega_{\textit{wall}} = \frac{40000\nu_{\textit{wall}}}{k_{s}^{2}}$$
, need $\frac{u_{\tau}k_{s}}{\nu} < 5$

- Produces decent results at the wall
- Can perform the same trick with two BCs on k

Other Notes

- ullet Early models were very sensitive to freestream conditions on ω
- · Menter proposed SST model to overcome this shortcoming
- Possible to write a $k \epsilon$ model with $k \omega$ "physics"

Other Ideas - Unsteady RANS

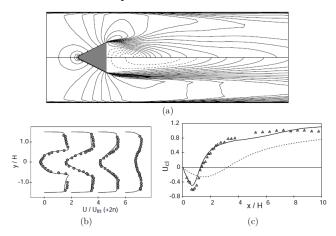
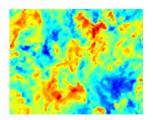


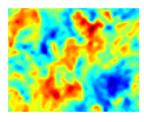
Figure 7.72 Vortex shedding from a triangular cylinder. (a) composite showing time-average contours of U in the upper half, versus a steady solution in the lower half. The dashed lines indicate negative velocity. (b) time-averaged velocity profiles in the wake. (c) velocity along centerline: time-average, ——; steady computation, ---- (Durbin, 1995).

Other Ideas

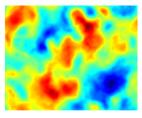
- Large Eddy Simulation
- Reynolds Stress Models
 - Elliptic relaxation model solves 18 coupled, highly nonlinear PDEs
- Variational Multiscale
- Direct Numerical Simulation



DNS Velocity Field



Filtered Velocity Field $\Delta = L/32$



Filtered Velocity Field $\Delta = L/16$