

# **FUNDAMENTALS OF COMBUSTION**

## **Final Report**

### **Hydrogen - air pre-mixed combustion in a rough micro combustor**

## **Group Members:**

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## **OBJECTIVES:**

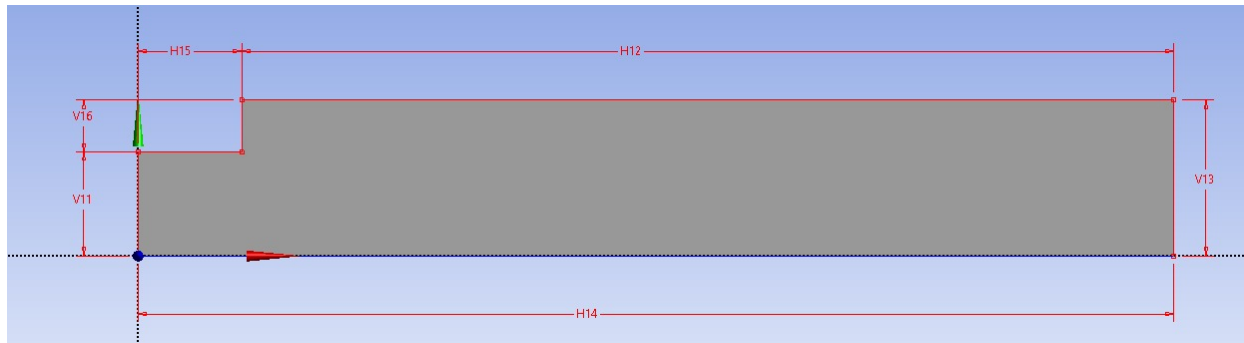
**The objective of this submission is to present final verification work performed in Ansys. We intend to perform the final tests based on a smooth micro combustor.**

## **PROBLEM FORMULATION:**

**In order to obtain proper dimensions of the micro combustor, species and their reactions and boundary conditions for combustion of a premixed H<sub>2</sub>/Air mixture, we have referred to the paper Comparative analysis of hydrogen/air combustion CFD-modelling for 3D and 2D computational domain of micro-cylindrical combustor. This paper is available in the INTERNATIONAL JOURNAL OF HYDROGEN ENERGY. This paper has been authored by Dmitry Pashchenko. We have used a 2-dimensional axisymmetric model in this paper.**

# Micro- Combustor:

- Diagrams and Dimensions:



Dimensions: 6	
<input type="checkbox"/> H12	18 mm
<input type="checkbox"/> H14	20 mm
<input type="checkbox"/> H15	2 mm
<input type="checkbox"/> V11	2 mm
<input type="checkbox"/> V13	3 mm
<input type="checkbox"/> V16	1 mm

## MESH:

**We had initially generated a coarse mesh which has been refined by reducing the size of each mesh element. Details are attached below:**

<input type="checkbox"/> Defaults	
Physics Preference	CFD
Solver Preference	Fluent
Element Order	Linear
<input type="checkbox"/> Element Size	4.e-002 mm
Export Format	Standard
Export Preview Surface Mesh	No

Statistics	
<input type="checkbox"/> Nodes	36826
<input type="checkbox"/> Elements	36250

## MESH DETAILS

## BOUNDARY CONDITIONS:

**Table 2 – Boundary conditions.**

No.	Parameter	Inlet	Outlet
1	Excess air ratio	1.0	—
2	Mass flow rate H <sub>2</sub> +air	$1.8551 \times 10^{-5}$ kg/s	—
3	H <sub>2</sub> mass flow rate	$5.25 \times 10^{-7}$ kg/s	—
4	Mass fraction: H <sub>2</sub> /O <sub>2</sub> /N <sub>2</sub>	0.028301/0.22641/0.745289	—
5	Turbulent intensity	5%	5%
6	Hydraulic diameter	$2 \times 10^{-3}$ m	$3 \times 10^{-3}$ m
7	Gauge pressure	0.0 Pa	0.0 Pa

- The inlet temperature is 1000K

## Chemistry Of Combustion:

**We have used a 9 species, 19 reaction mechanism imported from Chemkin database.**

**Table 2 H<sub>2</sub>/air chemical reaction mechanism**

Reaction	A (m <sup>3</sup> /kmol s)	$\beta$	E (J/kmol)
H + O <sub>2</sub> = O + OH	$5.1 \times 10^{13}$	-0.82	$6.91 \times 10^7$
H <sub>2</sub> + O = H + OH	$1.8 \times 10^7$	1.0	$3.7 \times 10^7$
H <sub>2</sub> + OH = H <sub>2</sub> O + H	$1.2 \times 10^7$	1.3	$1.52 \times 10^7$
OH + OH = H <sub>2</sub> O + O	$6.0 \times 10^6$	1.3	0.0
H + OH + M = H <sub>2</sub> O + M <sup>a</sup>	$7.5 \times 10^{17}$	-2.6	0.0
O <sub>2</sub> + M = O + O + M	$1.9 \times 10^8$	0.5	$4.001 \times 10^8$
H <sub>2</sub> + M = H + H + M <sup>b</sup>	$2.2 \times 10^9$	0.5	$3.877 \times 10^8$
H <sub>2</sub> + O <sub>2</sub> = OH + OH	$1.7 \times 10^8$	0.0	$2.0 \times 10^8$
H + O <sub>2</sub> + M = HO <sub>2</sub> + M <sup>c</sup>	$2.1 \times 10^{12}$	-1.0	0.0
H + O <sub>2</sub> + O <sub>2</sub> = HO <sub>2</sub> + O <sub>2</sub>	$6.7 \times 10^{13}$	-1.42	0.0
H + O <sub>2</sub> + N <sub>2</sub> = HO <sub>2</sub> + N <sub>2</sub>	$6.7 \times 10^{13}$	-1.42	0.0
HO <sub>2</sub> + H = H <sub>2</sub> + O <sub>2</sub>	$2.5 \times 10^{10}$	0.0	$2.9 \times 10^6$
HO <sub>2</sub> + H = OH + OH	$2.5 \times 10^{11}$	0.0	$7.9 \times 10^6$
HO <sub>2</sub> + O = OH + O <sub>2</sub>	$4.8 \times 10^{10}$	0.0	$4.2 \times 10^6$
HO <sub>2</sub> + OH = H <sub>2</sub> O + O <sub>2</sub>	$5.0 \times 10^{10}$	0.0	$4.2 \times 10^6$
HO <sub>2</sub> + HO <sub>2</sub> = H <sub>2</sub> O <sub>2</sub> + O <sub>2</sub>	$2.0 \times 10^{10}$	0.0	0.0
H <sub>2</sub> O <sub>2</sub> + M = OH + OH + M	$1.3 \times 10^{14}$	0.0	$1.905 \times 10^8$
H <sub>2</sub> O <sub>2</sub> + H = HO <sub>2</sub> + H <sub>2</sub>	$1.7 \times 10^9$	0.0	$1.57 \times 10^7$
H <sub>2</sub> O <sub>2</sub> + OH = H <sub>2</sub> O + HO <sub>2</sub>	$1.0 \times 10^{10}$	0.0	$7.5 \times 10^6$
<sup>a</sup> Enhancement factors: H <sub>2</sub> O/20/.			
<sup>b</sup> Enhancement factors: H <sub>2</sub> O/6/, H/2/, H <sub>2</sub> /3/.			
<sup>c</sup> Enhancement factors: H <sub>2</sub> O/21/, H <sub>2</sub> /3.3/, O <sub>2</sub> /0/, N <sub>2</sub> /0/.			

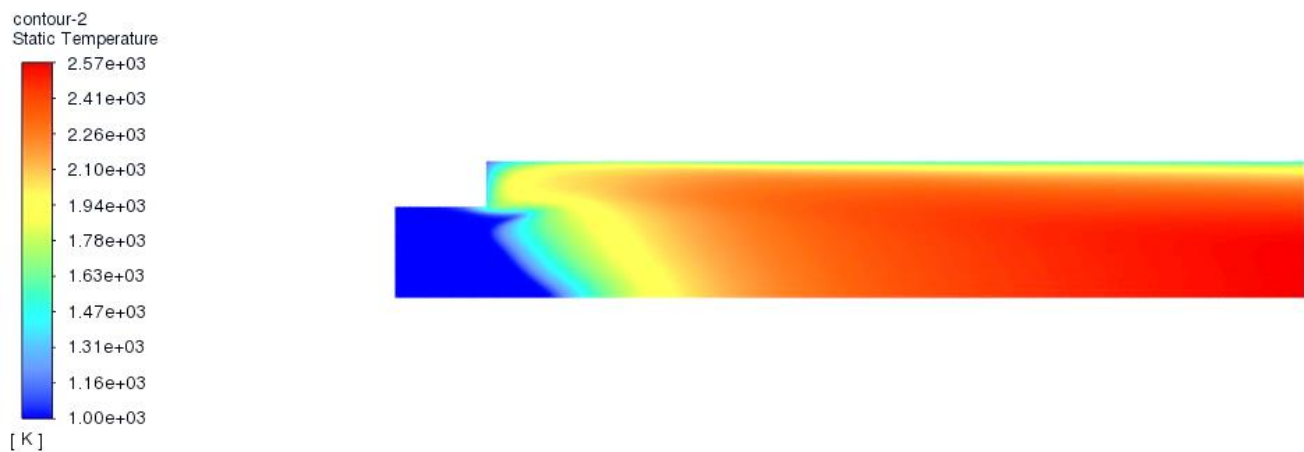
## **VISCOUS MODEL:**

Model	Model Constants
<input type="radio"/> Inviscid	Cmu
<input type="radio"/> Laminar	<input type="text" value="0.0845"/>
<input type="radio"/> Spalart-Allmaras (1 eqn)	C1-Epsilon
<input checked="" type="radio"/> k-epsilon (2 eqn)	<input type="text" value="1.42"/>
<input type="radio"/> k-omega (2 eqn)	C2-Epsilon
<input type="radio"/> Transition k-kl-omega (3 eqn)	<input type="text" value="1.68"/>
<input type="radio"/> Transition SST (4 eqn)	Wall Prandtl Number
<input type="radio"/> Reynolds Stress (5 eqn)	<input type="text" value="0.85"/>
<input type="radio"/> Scale-Adaptive Simulation (SAS)	Production Limiter Clip Factor
<input type="radio"/> Detached Eddy Simulation (DES)	<input type="text" value="10"/>
<b>k-epsilon Model</b>	
<input type="radio"/> Standard	
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<input type="radio"/> Realizable	
<b>RNG Options</b>	
<input type="checkbox"/> Differential Viscosity Model	
<b>Near-Wall Treatment</b>	
<input checked="" type="radio"/> Standard Wall Functions	
<input type="radio"/> Scalable Wall Functions	
<input type="radio"/> Non-Equilibrium Wall Functions	
<input type="radio"/> Enhanced Wall Treatment	
<input type="radio"/> Menter-Lechner	
<input type="radio"/> User-Defined Wall Functions	
<b>Options</b>	
<input type="checkbox"/> Viscous Heating	
<input type="checkbox"/> Production Kato-Launder	
<input checked="" type="checkbox"/> Production Limiter	
<b>User-Defined Functions</b>	
Turbulent Viscosity	
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<b>Prandtl Numbers</b>	
Wall Prandtl Number	
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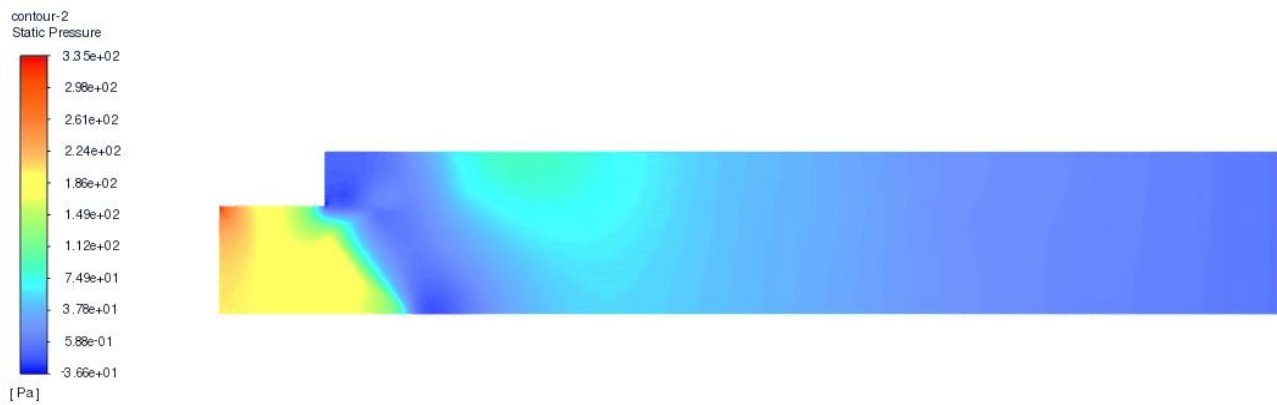
## **Radiation Model:**

**In this study, P-1 radiation model is used for developed numerical model. The chosen P-1 radiation model is the simplest case of the more general P-N model, which considers the influence of geometry on the radiative heat transfer.**

# Result Obtained from Our Simulation:

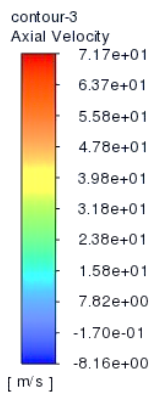


## STATIC TEMPERATURE DISRTIBUTION

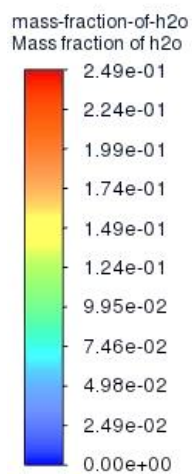


## STATIC PRESSURE DISTRIBUTION

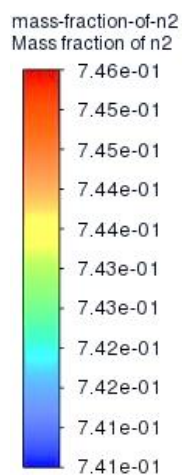




## AXIAL VELOCITY DISTRIBUTION



## MASS FRACTION OF H<sub>2</sub>O



## MASS FRACTION OF N<sub>2</sub>