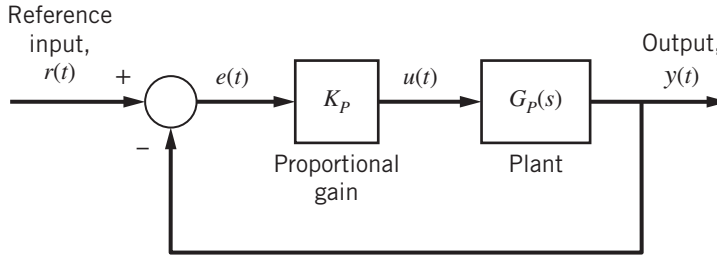


**10.3** Figure P10.3 shows a general closed-loop control system. The plant transfer function is

$$G_P(s) = \frac{1}{s^2 + 6s + 8}$$



**Figure P10.3**

- Determine whether the closed-loop system is stable for control gain  $K_P = 2$ .
- Compute the controller gain  $K_P$  so that step response shows 25% overshoot.
- Estimate the settling time for a step reference input if the control gain is  $K_P = 0.5$ .

**10.3** The closed-loop transfer function (CLTF) is

$$T(s) = \frac{K_P G_P(s)}{1 + K_P G_P(s) H(s)} = \frac{K_P \frac{1}{s^2 + 6s + 8}}{1 + K_P \frac{1}{s^2 + 6s + 8}} = \frac{K_P}{s^2 + 6s + 8 + K_P}$$

- a) When gain  $K_P = 2$ , the denominator of the closed-loop transfer function is

$$\text{Characteristic equation: } s^2 + 6s + 10 = 0 \quad \text{Closed-loop roots: } s = -3 \pm j$$

Since the closed-loop roots have negative real parts the closed-loop system is **stable** for  $K_P = 2$ .

- b) The CLTF is second order. Maximum overshoot is solely a function of damping ratio  $\zeta$

$$M_{OS} = e^{-\zeta\pi / \sqrt{1-\zeta^2}} = 0.25$$

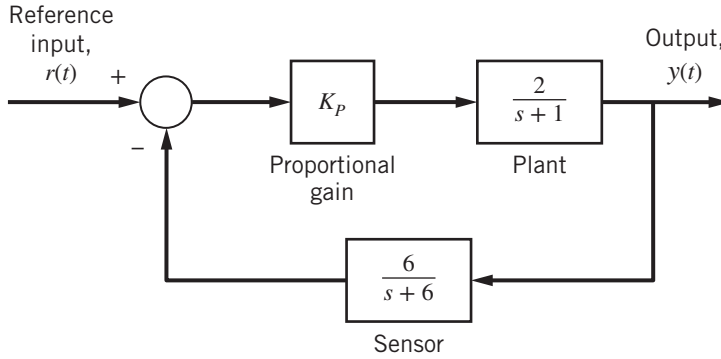
Hence, for  $M_{OS} = 0.25$  we require  $\zeta = 0.4037$ . The first-order denominator term of the CLTF is

$$6 = 2\zeta\omega_n = 2(0.4037)\omega_n$$

where the undamped natural frequency is  $\omega_n = \sqrt{8 + K_P} = 7.4313$  rad/s. Therefore  **$K_P = 47.224$** .

- c) When gain  $K_P = 0.5$ , the denominator of the closed-loop transfer function is  $s^2 + 6s + 8.5$ . The closed-loop roots are  $s_1 = -3.7071$  and  $s_2 = -2.2929$ . The “slowest” closed-loop root is  $s_2$  and its associated time constant is  $\tau_2 = 1/2.2929 = 0.4361$  s. The settling time is  $t_S = 4\tau_2 = \mathbf{1.7445}$  s.

**10.4** Figure P10.4 shows a closed-loop control system.



**Figure P10.4**

- Compute the controller gain  $K_p$  so that the undamped natural frequency of the closed-loop system is  $\omega_n = 4 \text{ rad/s}$ .
- Compute the controller gain  $K_p$  so that the damping ratio of the closed-loop system is  $\zeta = 0.7$ .
- Compute the steady-state output for a step reference input  $r(t) = 4U(t)$  and controller gain  $K_p = 2$ .

**10.4** The closed-loop transfer function (CLTF) is

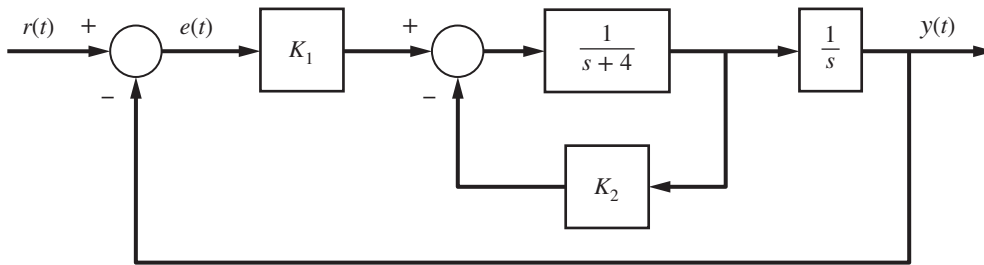
$$T(s) = \frac{K_p G_p(s)}{1 + K_p G_p(s) H(s)} = \frac{K_p \frac{2}{s+1}}{1 + K_p \frac{2}{s+1} \frac{6}{s+6}} = \frac{2K_p(s+6)}{s^2 + 7s + 6 + 12K_p}$$

- The (closed-loop) undamped natural frequency is obtained from the zeroth-order term, or  $\omega_n = \sqrt{6 + 12K_p} = 4 \text{ rad/s}$ . Hence the gain setting must be  **$K_p = 0.8333$** .
- The (closed-loop) damping ratio is obtained from the first-order term  $7 = 2\zeta\omega_n$ , where the undamped natural frequency is  $\omega_n = \sqrt{6 + 12K_p}$ . Setting  $\zeta = 0.7$  and solving for the gain yields the solution  **$K_p = 1.5833$** .
- With gain setting  $K_p = 2$ , the CLTF is

$$T(s) = \frac{4(s+6)}{s^2 + 7s + 30}$$

Because the input is a constant (magnitude of 4) we can use the CLTF DC gain:  $T(s=0) = 24/30 = 0.8$  and hence the steady-state output is  $y_{ss} = (4)(0.8) = 3.2$ .

**10.5** A closed-loop control system is shown in Fig. P10.5.



**Figure P10.5**

- Compute the closed-loop transfer function for the overall system,  $T(s) = Y(s)/R(s)$ .
- Two gain pairs are considered: Option 1 ( $K_1 = 15$ ,  $K_2 = 2$ ) and Option 2 ( $K_1 = 30$ ,  $K_2 = 3$ ). Which gain pair provides the greatest closed-loop damping ratio? Justify your answer.

**10.5** a) The complete closed-loop transfer function is

$$T(s) = \frac{Y(s)}{R(s)} = \frac{K_1 G(s) \frac{1}{s}}{1 + K_1 G(s) \frac{1}{s}}$$

where  $G(s)$  is the closed-loop transfer function of the “inner” closed-loop feedback system:

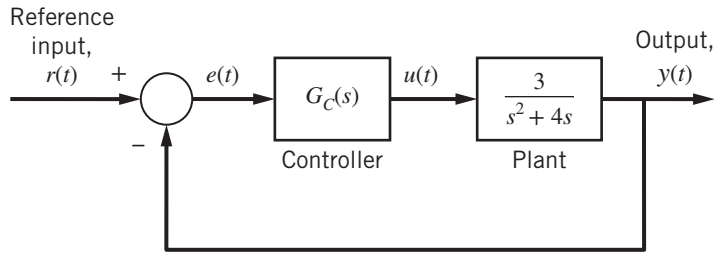
$$G(s) = \frac{\frac{1}{s+4}}{1 + \frac{K_2}{s+4}} = \frac{1}{s+4+K_2}$$

Substituting this result for  $G(s)$  yields the overall CLTF:

$$T(s) = \frac{Y(s)}{R(s)} = \frac{K_1 \frac{1}{s+4+K_2} \frac{1}{s}}{1 + K_1 \frac{1}{s+4+K_2} \frac{1}{s}} = \boxed{\frac{K_1}{s^2 + (4+K_2)s + K_1}}$$

b) The (closed-loop) damping ratio is obtained from the first-order term  $4 + K_2 = 2\zeta\omega_n$ , where the undamped natural frequency is  $\omega_n = \sqrt{K_1}$ . Substituting the Option 1 gains ( $K_1 = 15$ ,  $K_2 = 2$ ) yields  $\omega_n = 3.873$  rad/s and damping ratio  $\zeta = 0.7746$ . Substituting the Option 2 gains ( $K_1 = 30$ ,  $K_2 = 3$ ) yields  $\omega_n = 5.4772$  rad/s and damping ratio  $\zeta = 0.6390$ . Therefore **Option 1** gains provide the largest closed-loop damping ratio.

**10.6** Figure P10.6 shows a unity-feedback closed-loop system. The reference input is a ramp,  $r(t) = 0.2t$ .



**Figure P10.6**

- a. Compute the steady-state tracking error if the controller  $G_C(s)$  is a simple proportional gain  $K_P = 2$ .
- b. Compute the steady-state tracking error if we use a PI controller with gains  $K_P = 3$  and  $K_I = 1.5$ .

**10.6** a) We may use Table 10-3 to compute the steady-state tracking error. Note that the plant is a type 1 system (it has one “pure integrator”). For a ramp input (with slope 0.2) the steady-state error for a type-1 system is

$$\text{Type-1 system: } e_{ss} = \frac{0.2}{K_{sv}}$$

where the “static velocity error constant” is  $K_{sv} = \lim_{s \rightarrow 0} sG(s)$ . The forward transfer function is

$$G(s) = \frac{3K_P}{s(s+4)}$$

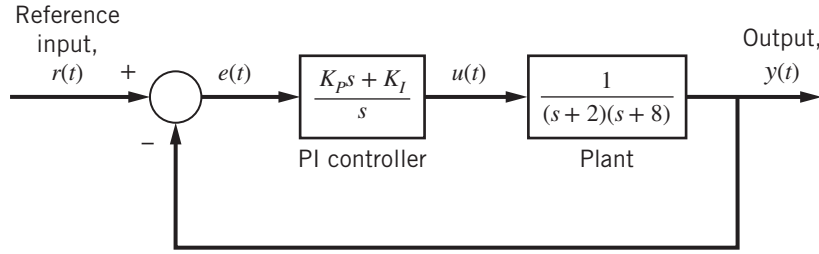
When the control gain is  $K_P = 2$ , we get  $K_{sv} = 6/4 = 1.5$ , and the steady-state error  $e_{ss} = \mathbf{0.1333}$ .

b) If we use a PI controller the forward transfer function becomes

$$G(s) = G_C(s)G_P(s) = \frac{K_P s + K_I}{s} \cdot \frac{3}{s(s+4)} = \frac{3(K_P s + K_I)}{s^2(s+4)}$$

Because the forward transfer function is type 2 the steady-state error for a ramp input is **zero**.

**10.7** A simple closed-loop PI control system is shown in Fig. P10.7.



**Figure P10.7**

- Show that the closed-loop system is stable for PI gains  $K_p = 5$  and  $K_I = 25$ .
- Determine “by hand” the closed-loop output  $y(t)$  at time  $t = 8$  s if the reference input is a ramp function  $r(t) = 1.4t$ . Use the PI gains from part (a).

**10.7** The closed-loop transfer function (CLTF) is

$$T(s) = \frac{Y(s)}{R(s)} = \frac{\frac{K_p s + K_I}{s} \cdot \frac{1}{(s+2)(s+8)}}{1 + \frac{K_p s + K_I}{s} \cdot \frac{1}{(s+2)(s+8)}} = \frac{K_p s + K_I}{s^3 + 10s^2 + (16 + K_p)s + K_I}$$

- Using gains  $K_p = 5$  and  $K_I = 25$  the CLTF becomes

$$T(s) = \frac{5s + 25}{s^3 + 10s^2 + 21s + 25}$$

The three roots of the closed-loop transfer function are  $s_1 = -7.6926$  and  $s_{2,3} = -1.1537 \pm j1.3852$ . Since all closed-loop roots have negative real parts the closed-loop system is **stable** for this gain setting.

- We want the output at time  $t = 8$  s for a ramp input  $r(t) = 1.4t$ . The complex closed-loop root at  $s_{2,3} = -1.1537 \pm j1.3852$  is the “slowest” root and its settling time is approximately 3.5 s. Hence, the closed-loop transient response has died out by  $t = 8$  s. Using Table 10-1 we see that the for a type 1 system the steady-state error for a ramp input (with slope 1.4) is

$$\text{Type-1 system: } e_{ss} = \frac{1.4}{K_{sv}}$$

The system is clearly type 1 due to the PI controller in the forward path. The “static velocity error constant” is  $K_{sv} = \lim_{s \rightarrow 0} sG(s)$ . The forward transfer function is

$$G(s) = G_c(s)G_p(s) = \frac{K_p s + K_I}{s} \cdot \frac{1}{(s+2)(s+8)} = \frac{5s + 25}{s(s+2)(s+8)} \quad (\text{using the PI gains})$$

Hence  $K_{sv} = 25/16 = 1.5625$ , and the steady-state error for a ramp input is  $e_{ss} = 0.8960$ . The output at time  $t = 8$  s can be computed using the tracking error since  $e = r - y$ . Therefore

$$y(8) = r(8) - e_{ss} = (1.4)(8) - 0.8960 = \mathbf{10.3040}$$